



# Communication Ultra-Sensitive Intensity Modulated Strain Sensor by Tapered Thin-Core Fiber Based Modal Interferometer

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**Abstract:** In this paper, to enhance practicality, a novel tapered thin-core fiber (*t*-TCF) based modal interferometer is proposed and demonstrated experimentally. The light field distribution of *t*-TCF structure is investigated by a beam propagation method, and the quantitative relationship is gained between light intensity loss and waist diameter. Under ~30 µm waist diameter, multiple *t*-TCF based sensor heads are fabricated by arc-discharged splicing and taper techniques, and comprehensive tests are performed with respects to axial strain and temperature. The experimental results show that, with near-zero wavelength shift, obvious intensity strain response is exhibited and negative-proportional to the reduced length of TCF. Thus, the maximum sensitivity reaches 0.119 dB/µ $\epsilon$  when the TCF length is equal to 15 mm, and a sub-micro-strain detection resolution (about 0.084 µ $\epsilon$ ) is obtained. Besides, owing to the flat red-shifted temperature response, the calculated cross-sensitivity of our sensor is compressed within 0.32 µ $\epsilon$ /°C, which is promising for high precision strain related engineering applications.

Keywords: fiber-optic sensing; modal interference; thin-core fiber; strain; intensity; taper

## 1. Introduction

Fiber-optic strain sensors have the advantages of a compact structure, light weight and anti-electromagnetic interference, which has been widely used in high precision measurement, health-monitoring of building structures and aerospace engineering [1-5]. Strain sensors based on fiber Bragg grating (FBG) [6,7], long-period fiber grating (LPFG) [8,9], polymer optical fiber (POF) [10–12] and photonic crystal fiber (PCF) [13,14] are easy to fabricate, but the sensitivity is usually only  $\sim 1 \text{ pm}/\mu\epsilon$ . To improve sensitivity, the modal interference based fiber optic strain sensors, derived from the excitation of higher-order cladding modes, has received much attention [15–17]. PCF-based modal interferometers are proposed, and a 2–3 pm/ $\mu\epsilon$  strain sensitivity is reported with ultralow temperature cross-talk [18–20]. Du et al. enhanced the sensitivity of fiber grating sensors based on a four-wave mixing (FWX) mechanism, and the corresponding strain sensitivity reached 13.3 pm/ $\mu\epsilon$  [21]. Furthermore, Han made high birefringence PCF via a filling high refractive index (RI) liquid, and the maximum strain sensitivity of the interferometer was  $25 \text{ pm}/\mu\epsilon$  [22]. Liu et al. prepared a hybrid silica-polymer fiber sensor to gain the sensitivity of 28 pm/ $\mu\epsilon$  under an ultrahigh pressure condition [23]. In addition, Ruan and Yin respectively fabricated the bubble based micro-cavity interferometers through precise arc-discharge control, and the strain sensitivities were further increased to more than 30 pm/ $\mu\epsilon$  [24,25]. Moreover, the highest sensitivity so far reached 1.15 nm/ $\mu\epsilon$  in the range of 0~230 µɛ, through a cascaded micro-cavity structure [26]. Compared with wavelength sensitive structures, the intensity demodulation schemes can greatly improve the practicability and portability of sensors without an expensive and heavy high-precision



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**Copyright:** © 2021 by the authors. Licensee MDPI, Basel, Switzerland. This article is an open access article distributed under the terms and conditions of the Creative Commons Attribution (CC BY) license (https:// creativecommons.org/licenses/by/ 4.0/). spectrometer [27]. Zhang et al. proposed an intensity modulation scheme by using a piece of micro-structure hollow-core fiber (MS-HCF), but the presented sensitivity was less than 0.004 dB/ $\mu\epsilon$  [28]. Wang et al. prepared a Mach–Zehnder interferometer (MZI) based on the series connection of LPG and up-taper, with a strain sensitivity of 0.026 dB/ $\mu\epsilon$  in the range of 0–590  $\mu\epsilon$  [29]. The intensity sensitivity was further enhanced to 0.051 dB/ $\mu\epsilon$  by a multimode and microfiber assisted open-cavity structure (MMA-OC), but with a complex fabrication process [30]. Comparatively, thin-core fiber based modal interferometers have been widely investigated and used in the RI sensing and gas concentration measurement [31,32], and the intensity sensitivity of 442.59 dB/RIU was obtained by a down-taper structure [33]. Similarly, our group developed the study on the thin-core fiber based axial strain sensing, and the sensitivity of ~0.02 dB/ $\mu\epsilon$  was presented with a high temperature consistency [34,35].

In this paper, through arc-discharge splicing and taper techniques, a novel tapered thin core fiber (*t*-TCF) based in-fiber MZI is proposed and completed. With the varied waist diameter, the evanescent wave field distributions of *t*-TCF are analyzed, and the quantitative light intensity loss is obtained by beam propagation method. Under the suitable waist diameter, multiple *t*-TCF structures with different lengths are fabricated, and their axial strain characteristics are comprehensively tested. The experimental results show that our sensors have an obvious intensity change with the increased axial strain, and the strain sensitivity is negatively proportional to the length of TCF. The maximum sensitivity reaches 0.119 dB/ $\mu\epsilon$ , and less than a 0.1- $\mu\epsilon$  detection resolution is gained. In addition, owing to the flat wavelength shift, the cross-sensitivity caused by the ambient temperature is effectively constrained in 0.318  $\mu\epsilon$ /°C.

#### 2. Principles and Simulations

As shown in Figure 1, the *t*-TCF structure is composed of lead-in and lead-out singlemode fibers (SMFs) and a section of tapered TCF, which is connected by a core-offset splicing technique (the core-offset value is denoted by  $\alpha$ ). When the incident light reaches the first fusion point (Offset Joint 1, OJ1) through the lead-in SMF, one beam of light continues to transmit along the core of TCF and the other beam enters the cladding and excites the corresponding high-order cladding modes. Due to the different RI of the core and cladding of TCF, when two beams reach the second fusion joint (Offset Joint 2, OJ2), they will have a significant optical path difference and form a Mach–Zehnder interference.



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**Figure 1.** The schematic diagram of the *t*-TCF structure. (**a**) Top-view, (**b**) side-view and (**c**) cross-sectional view.

According to the theory of dual-beam interference, the intensity of the transmission spectrum of an in-fiber MZI can be written as:

$$I = I_{co} + I_{cl} + 2\sqrt{I_{co}I_{cl}\cos\Delta\varphi}$$
(1)

where  $I_{co}$  and  $I_{cl}$  are the light intensities in the core and cladding modes, respectively.  $\Delta \varphi$  is the phase difference and equal to  $2\pi(n_{co} - n_{cl}) \cdot L_{TCF} / \lambda_i$ , where  $\lambda_i$  is the wavelength of incident light,  $L_{TCF}$  is the length of thin-core fiber,  $n_{co}$  and  $n_{cl}$  are the effective RIs of core and cladding modes. Thus, we get  $\Delta n_{eff} = n_{co} - n_{cl}$ . Furthermore, when  $\Delta \varphi = (2m + 1)\pi$  (m = 1, 2, 3...), the resonance wavelength (denoted by  $\lambda_m$ ) will be:

$$\lambda_m = \frac{2\Delta n_{eff} L_{TCF}}{2m+1} \tag{2}$$

Additionally, the free spectral range (FSR) of the fringes can be expressed as:

$$FSR = \lambda_m - \lambda_{m-1} \approx \frac{\lambda_m^2}{\Delta n_{eff} L_{TCF}}$$
(3)

Further, we define the normalized extinction ratio (ER) as:

$$ER = \frac{2\sqrt{I_{co}I_{cl}}}{I_{co} + I_{cl}} \tag{4}$$

Equation (4) shows that the *ER* value can reach its maximum when  $I_{co} = I_{cl}$ , which is very important for an intensity modulation based fiber-optic sensor.

For an axial strain test, we define the distance between two fixed points as  $L_S = L_{SMF} + L_{TCF}$ , where  $L_{SMF}$  is the length of SMF. Assume that  $\Delta L_S$  is the variation of  $L_S$ , the applied axial strain is then expressed as  $S = \Delta L_S / L_S$ . Different from the conventional modal interferometers reported in [34], the axial strain of *t*-TCF structure can be expressed as:

$$S_{TCF} = \frac{L_S S}{L_{TCF} + L_{SMF} \frac{d^2}{D^2}}$$
(5)

where *d* is the diameter of taper waist, *D* is the cladding diameter of SMF. Obviously, Equation (5) means that there is a negatively proportional relationship between  $S_{TCF}$  and *d* for a given D. Furthermore, the differential operation of Equation (5) is performed and we get:

$$\Delta S_{TCF} = -k \frac{2L_s d}{D} \Delta d \tag{6}$$

where  $k = L_S S / (L_{TCF} + L_{SMF} d^2/D^2)^2$ . Equation (6) means that when an axial strain is applied, the increased  $\Delta S_{TCF}$  will lead a tiny decrease in *d*. According to the principle of evanescent wave field, this reduced diameter of taper must bring a continuous loss of light energy. Therefore, in addition to the wavelength drift caused by photo-elastic effect [36], the intensity of fringes of *t*-TCF structure will be also changed significantly, which provides the possibility of intensity demodulation for axial strain sensing. Moreover, when  $\Delta L_S$  is a small value, *k* can be regarded as a constant and the intensity variation may be linearly decreased with the added axial strain.

In order to get a transmission spectrum with a high *ER*, the parameters of preparation and structure should be optimized. Then, the light field distribution of *t*-TCF structure is simulated by a beam propagation method. Typically, the central wavelength of incident light (i.e.,  $\lambda_i$ ) is 1550 nm, and the background RI is equal to 1.0. Table 1 shows other key parameters, and the simulated results are presented in Figure 2. The light field energy distributions of *t*-TCF structure when  $d = 10-50 \mu m$  are shown in Figure 2a. With the decrease in *d*, the energy loss of evanescent wave field gradually increases. Furthermore, from Figure 2b, the normalized energy change of light field is highly linear with the decreased *d*. By calculation, the loss coefficient is about 1% per micrometer. This means that with the increase (decrease) of axial strain (waist diameter), our tapered structure may exhibit an obvious and linear intensity variation. It is worth noting that due to the fact that a smaller waist diameter (e.g., less than 20 µm) will reduce the fusion efficiency, the target diameter of taper is set as ~30 µm in the following fabrication process.



Table 1. The main parameters of the structure.

**Figure 2.** (a) Light field distributions of *t*-TCF structure and (b) normalized power with a varied diameter of taper waist.

### 3. Fabrication

As shown in Figure 3a, the fabrication of *t*-TCF structures includes two parts: symmetrical core-offset splicing and taper. The core-offset structure with  $L_{TCF} = 50$  mm is completed in manual-mode by a commercial fusion splicer (KL-300T, with an adjustment resolution of ~0.5 µm). The offset value is set at  $\alpha = 12$  µm and the fusion splicing loss is about 0.04 dB. The key parameters for taper structure are set as follows: pre-discharge intensity and time are 40 bit and 180 ms, the main discharge intensity and time are 70 bit and 2200 ms, the waiting time is 1200 ms and the splicing speed is 0.17 µm/ms. In addition, for comparison, three *t*-TCF structures are fabricated with similar waist diameters (~31 µm) and different  $L_{TCF}$  (from 15 mm to 50 mm). Table 2 shows the key structural parameters of the samples. The average waist diameter is 31. 25 µm with an error of less than ±1.1 µm. In addition, the maximum difference of offset values is well constrained within 0.4 µm. This means that a high reproducibility (over 95%) of our *t*-TCF structures can be achieved by simply an arc-discharge technique. Moreover, the fabricated taper

structure with  $L_{TCF} = 50$  mm is shown in Figure 3b. The diameter of taper is 30.12 µm and the length of transition area is 774.4 µm. Then, three *t*-TCF structures are connected with a broadband light source (BBS, CONNET VENUS, with the range of 1525–1610 nm) and an optical spectrum analyzer (OSA, Agilent 86142B, with the resolution of 0.06 nm/0.01 dB). Their outputted transmission spectra are presented in Figure 3c. It is clear that when  $L_{TCF}$  decreases from 50 mm to 15 mm, the corresponding FSR of transmission spectrum inversely increases from 5.92 nm to 15.73 nm. Moreover, the *ER* values are also gradually increased and the maximum reaches 20.82 dB when  $L_{TCF} = 15$  mm.



**Figure 3.** (a) Fabrication flow chart of *t*-TCF structure, (b) micro image of taper region, (c) transmission spectra with different  $L_{TCF}$ .

Table 2. Structural	parameters	of the	samp	les
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Samples	L <sub>TCF</sub> (mm)	Transition Areas (µm)	Diameters (µm)	Offset Values (µm)
S1	50	774.4	30.12	11.78/12.15
S2	30	765.6	32.25	12.37/11.93
S3	15	770.2	31.37	12.28/12.05

#### 4. Experiments and Results

As shown in Figure 4, the experimental setup for axial strain sensing is mainly completed by a micro-displacement platform (Newport, CA, USA, Model ESP-300), with the minimum accuracy of 0.1  $\mu$ m. For protection, the above three *t*-TCF structures are packaged into a thin steel tube with the diameter of ~500  $\mu$ m (see the sub-figure). Then, the sensor heads are placed horizontally on both sides of the platform and fast fixed with UV glue. Specially, *L*<sub>S</sub> is equal to 10 cm and the ambient temperature is 25 ± 0.2 °C. The axial strain tests are then performed, and their transmission spectra are demonstrated with the varied strain in Figure 5.

From Figure 5a, when  $L_{TCF} = 50$  mm, its spectrum mainly shows an intensity variation as the strain increases. The intensity of dip is linearly increased at about 6.27 dB. The corresponding sensitivity is 0.011 dB/µ $\epsilon$ , and the linearity is 0.993 in the range of 0–600 µ $\epsilon$ . When  $L_{TCF}$  is reduced to 30 mm (see Figure 5b), the intensity sensitivity is lightly increased to 0.019 dB/µ $\epsilon$ , but the corresponding linear range is reduced to 0–250 µ $\epsilon$ . Comparatively, in the case of  $L_{TCF} = 15$  mm, the intensity sensitivity is greatly and linearly enhanced about 11-fold, and reaches 0.121 dB/µ $\epsilon$  in the range of 0–120 µ $\epsilon$ . The corresponding detection resolution is 0.08 µ $\epsilon$  theoretically. Furthermore, the quantitative relationships are compared between intensity sensitivity/linear range and  $L_{TCF}$ . As shown in Figure 6, it is obvious that by simply reducing  $L_{TCF}$  from 50 mm to 15 mm, there is a gain of about 11-fold in the enhancement of intensity sensitivity, but the penalty is 5 times a deduction in the linear range (from 600  $\mu\epsilon$  to 120  $\mu\epsilon$ ).



Figure 4. The experimental setup for axial strain sensing.



**Figure 5.** The transmission spectra of axial strain responses with different length of TCF. (a)  $L_{TCF} = 50 \text{ mm}$ , (b)  $L_{TCF} = 30 \text{ mm}$  and (c)  $L_{TCF} = 15 \text{ mm}$ .



**Figure 6.** Comparison of sensitivities and linearity ranges with different *L*<sub>TCF</sub>.

In addition, for these three samples, their wavelength responses are all blue-shifted, and the maximum sensitivity is merely  $-1.89 \text{ pm}/\mu\epsilon$  occurring in the sample of  $L_{TCF} = 15 \text{ mm}$ . Moreover, the sample with  $L_{TCF} = 15 \text{ mm}$  is selected and multiple-time axial strain measurements are further conducted. As shown in Figure 7, with the increased and decreased axial strain, our structure exhibits better performance in term of repeatability. By calculation, the average intensity sensitivity is about 0.119 dB/ $\mu\epsilon$ , and the maximum deviation is less than  $\pm 0.0021 \text{ dB}/\mu\epsilon$ , which means that the error of repeatability is superior to  $\pm 2\%$ .



Figure 7. The relationships between intensity variation and increased/decreased axial strain.

Further, the sensor head with  $L_{TCF} = 15$  mm is placed into a temperature chamber (LICHEN, 202-00T, Shanghai, China, with a resolution of  $\pm 0.1$  °C) to characterize the temperature response. As shown in Figure 8a, the transmission spectrum is red-shifted with the rise of temperature and the intensity of dip is increased by about 1.88 dB. From Figure 8b, the wavelength sensitivity is 0.0736 nm/°C with the linearity of ~0.99. The corresponding intensity sensitivity is merely 0.0381 dB/°C in the range from 25 °C to 75 °C. Therefore, the calculated cross-sensitivity caused by ambient temperature is ~0.32  $\mu\epsilon$ /°C.



**Figure 8.** (a) Transmission spectrum of *t*-TCF structure ( $L_{TCF} = 15$  mm) with varied temperature and (b) the relationship between intensity/wavelength variation and increased temperature.

Moreover, according to [27], the measured axial strain and temperature changes can be effectively discriminated by using the inverse matrix method shown as Equation (7).

$$\begin{bmatrix} \Delta \varepsilon \\ \Delta T \end{bmatrix} = \frac{1}{W} \begin{bmatrix} k_{TI} - k_{T\lambda} \\ -k_{\varepsilon I} & k_{\varepsilon\lambda} \end{bmatrix} \begin{bmatrix} \Delta \lambda \\ \Delta I \end{bmatrix}$$
(7)

where  $W = k_{\epsilon\lambda}k_{TI} - k_{\epsilon I}k_{T\lambda}$ ,  $\Delta\epsilon$  and  $\Delta T$  represent the changes in axial strain and temperature, respectively.  $\Delta\lambda$  and  $\Delta I$  represent the change of wavelength and intensity of fringe, respectively.  $k_{\epsilon\lambda} = -0.00189$  and  $k_{\epsilon I} = 0.119$  are the wavelength and intensity response of strain,  $k_{T\lambda} = 0.0736$  and  $k_{TI} = 0.0381$  are the wavelength and intensity response of temperature. Therefore, we get the corrected sensitivity as ~0.121 dB/µ $\epsilon$ , which means the measurement error of our sensor is less than 1.7%. Table 3 compares the performance of fiber-optic strain sensors in terms of sensitivity, detection resolution and cross-sensitivity of temperature (note that the same resolution of 0.06 nm/0.01 dB is adopted). It is clear that as far as intensity response is concerned, the *t*-TCF structure updates our previous record obtained by a microfiber based open cavity, and presents a sub-high and less than 0.1-µ $\epsilon$  detection resolution, but with the advantage of ease of fabrication. Meanwhile, our sensor has the potential to be applied in a highly discriminative dual-parameter measurement because of the high temperature response (73.6 pm/°C) and low cross-talk (~0.32 µ $\epsilon$ /°C).

Sturctures	Sensitivity	<b>Detection Resolution</b>	Cross Senstivity	Linear Range	Refs
Few-mode FBG	2 pm/με	30 με	17.15 με/°C	0–450 με	[15]
SMF-PCF-SMF	2.1 pm/με	28.6 με	6.3 με/°C	0–3000 με	[18]
FBG with FWX	13.3 pm/με	4.51 με	10.6 με/°C	0–137 με	[21]
Filled high birefring-ent PCF	25 pm/με	2.4 με	_	0–61 με	[22]
Panda-type PMF	32 pm/με	1.875 με	_	0–900 με	[24]
Bubble based micro-cavity	30.66 pm/με	1.95 με	0.04 με/°C	0–600 με	[25]
Dual-micro cavity	1.15 nm/με	0.052 με	0.06 με/°C	0–230 με	[26]
MS-HCF	0.0036 dB/με	2.78 με	-	0–1000 με	[28]
Up-tapered LPFG	0.026 dB/με	0.38 με	-	0–590 με	[29]
Core-offset TCF	0.024 dB/με	0.42 με	-	0–700 με	[35]
MMA-OC	0.051 dB/με	0.196 με	0.106 με/°C	0–500 με	[30]
tapered-TCF	0.119 dB/με	0.084 με	0.32 με/°C	0–120 με	Our work

Table 3. Performance comparisons of the reported fiber-optic strain sensors.

## 5. Discussions

According to Equation (5), the applied axial strain in fact can be magnified due to the reduced diameter of the taper waist. In theory, similar to the schemes reported in [24–26], the corresponding wavelength shift should be larger than that in an un-tapered structure. In our structure, under a suitable waist diameter, we experimentally prove that the energy loss of an evanescent wave field is very sensitive to the variation of the waist diameter caused by the increased or decreased axial strain. Thus, with an almost near-zero wavelength shift, the obvious intensity response of axial strain is gained by the proposed structure, which may provide a new method for intensity modulation based fiber-optic sensing and measurements.

In addition, the intensity sensitivity of our structure is negatively proportional to the length of the sensing unit. For a modal interferometer, the minimum length of TCF will be ~10 mm, which means that the sensitivity of our structure can be further improved but limited. Developing a new tapered TCF structure may be a possible solution, by combining the techniques of arc-discharge tapering and flame brush. More importantly, it is clear that there is a conflict between the sensitivity and linear range. In fact, the response range of the axial strain of the *t*-TCF structure can reach over 1000  $\mu\epsilon$  although the waist taper is reduced to ~30  $\mu$ m. Therefore, our future work will focus on gaining a tradeoff of strain sensitivity and linear range, through improving the package technique and introducing a lasing-sensor based scheme.

#### 6. Conclusions

In this paper, a novel in-fiber Mach–Zehnder interferometer is proposed and fabricated based on the tapered TCF structure. The quantitative relationship between the energy loss of evanescent wave field and diameter of taper waist is gained, and the comprehensive tests are conducted in terms of axial strain and temperature under a suitable taper diameter (~31 µm). The experimental results prove that the intensity response of axial strain can be enhanced by simply reducing the length of the TCF. The ~0.12 dB/µ $\epsilon$  strain sensitivity is obtained when  $L_{TCF} = 15$  mm with high repeatability, and the detection resolution can reach 0.084 µ $\epsilon$ . Moreover, the temperature cross-sensitivity is constrained within 0.32 µ $\epsilon$ /°C because of <0.04 dB/°C intensity fluctuation. With the merits of ease of fabrication and practicality, such an ultra-sensitive scheme is very potential in axial strain related engineering sensing.

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