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Robust Interval Type-2 Fuzzy Sliding Mode Control Design for Robot Manipulators

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Abstract: This paper develops a new robust tracking control design for n-link robot manipulators with dynamic uncertainties, and unknown disturbances. The procedure is conducted by designing two adaptive interval type-2 fuzzy logic systems (AIT2-FLSs) to better approximate the parametric uncertainties on the system nominal. Then, in order to achieve the best tracking control performance and to enhance the system robustness against approximation errors and unknown disturbances, a new control algorithm, which uses a new synthesized AIT2 fuzzy sliding mode control (AIT2-FSMC) law, has been proposed. To deal with the chattering phenomenon without deteriorating the system robustness, the AIT2-FSMC has been designed so as to generate three adaptive control laws that provide the optimal gains value of the global control law. The adaptation laws have been designed in the sense of the Lyapunov stability theorem. Mathematical proof shows that the closed loop control system is globally asymptotically stable. Finally, a 2-link robot manipulator is used as case study to illustrate the effectiveness of the proposed control approach.

Keywords: nonlinear systems; robot manipulators; sliding mode control; type-2 fuzzy systems; uncertain systems

1. Introduction

The tracking control problem of robot manipulators is a very complicated issue due to undesirable characteristics, such as high nonlinearities, strong dynamic coupling, parameter perturbations, un-modeled dynamics, and unknown disturbances. Therefore, to achieve the good tracking control performance for such complex process, several researchers have developed robust control approaches, most of which use the fuzzy logic control (FLC), sliding mode control (SMC), feedback linearization technique, Neural Network (NN), adaptive control, and H_∞ technique [1–11].

Over the past years, intelligent algorithms using fuzzy logic systems (FLSs) are increasingly used and successfully applied in control problem of robot manipulators in the presence of dynamic uncertainties and unknown disturbances [12–14]. However, conventional type-1 fuzzy logic system (T1-FLS) cannot directly handle rule and measurement uncertainties because it uses T1-fuzzy sets (T1-FSs) that are certain. Therefore, these last years, an advanced form of FLS, called type-2 FLS (T2-FLS), has attracted considerable attention and becomes more and more imposed in designing robust controllers for uncertain complex processes, including robot systems [15–18]. One reason is that a T2-FS is characterized by a membership function (MF) that includes a footprint of uncertainty (FOU), which makes it possible to handle linguistic uncertainties more effectively than T1-FS [19–21].

On the other hand, SMC is known as an efficient tool well suitable for controlling complex uncertain processes due to its higher robustness against dynamic uncertainties and unknown disturbances [22–24]. However, the SMC has a major drawback, which consists in using a

discontinuous control law with large control gains that generate the chattering [25]. This phenomenon can cause severe damage to system actuators. In order to overcome or reduce the chattering, boundary layer method (BL), and higher order SMC approach (HO-SMC) are commonly employed by many researchers [26–31]. However, these approaches have a drawback that limits their performance, which consists in the fact that they still require the knowledge of the upper bounds of the uncertainties to ensure the desired control performance. The overestimation of the control gains to cover a wide range of uncertainties can cause the chattering and a dynamic response with overshoot, and the small gains can deteriorate the control accuracy performance and affect the system robustness. Moreover, BL method constrains the system trajectories not to the desired dynamics, but to their vicinities, thus both the control accuracy and robustness are affected. The HO-SMC approach requires in general higher order derivative of the sliding variable. The second order super-twisting SMC (SOST-SMC) is among the most effective HO-SMC algorithms that is widely used in the literature for controlling complex uncertain processes [4,32–35], it is developed by Levant [36] to avoid the chattering and to ensure the finite time convergence of the system state trajectories. However, the choice of its optimal control gains values remains a challenging matter for this kind of controllers.

In order to increase accurate tracking control performance and to guarantee the robustness of robot manipulators against dynamic uncertainties and unknown disturbances, several approaches have been developed. Among them, those that combine the benefits of FLS and robust control techniques, such as SMC, H_∞ Technique, NN, and adaptive control have recently been the focus of many researchers [37–39]. In [40], a FLS and a fuzzy sliding mode controller are employed to achieve the best tracking performance for the robot manipulators in presence of uncertainties. Most recently, in [41], the authors used an adaptive fuzzy sliding mode controller in order to improve the precision trajectory tracking of a designed winding hybrid-driven cable parallel manipulator subject to un-modeled dynamics and random disturbances. In [42], in order to regulate the vertical displacement of a bioinspired robotic dolphin, a sliding mode fuzzy control method is successfully applied. In [43], a hierarchically improved fuzzy dynamical sliding-model control is proposed for the autonomous ground vehicle to ensure the best path tracking performance in the presence of different payloads.

When compared to the existing works in the literature, the main contributions of the present study are listed, as follows:

- (1) a new robust algorithm is proposed for n-link robot manipulator systems to deal with the tracking control problems, with the following considerations are taken into account:
 - The dynamics of the robot manipulator systems are only partially known and present parametric variations.
 - The studied systems are subject to unknown disturbances.
 - No prior knowledge of the upper bound of the parametric uncertainties, unknown dynamics, un-modeled dynamics, and unknown disturbances that affect the studied system dynamics is required.
- (2) Based on T2-FLS, two adaptive interval T2-FLSs (AIT2-FLSs) are designed in order to efficiently estimate the parametric uncertainties of the system dynamics. FSs are chosen to be interval T2 (IT2), firstly, because they do not require a lot of computation, and, secondly, for their efficiency to capture severe uncertainties.
- (3) In order to handle errors approximation of parametric uncertainties and effectively reject the effects of unknown dynamics, un-modeled dynamics, and unknown disturbances on the control system without generating the undesired chattering, a new enhanced robust AIT2-FSMC law is designed so as to generate three adaptive control laws in order to provide the optimal estimation of the control law gains that effectively reject all of the undesired effects that perturb the control system while yielding a smooth global control law. Thus, the best tracking control performance is guaranteed. The adaptation laws of the synthesized controller parameters have been designed

in the sense of the Lyapunov stability approach. Finally, a 2-link robot manipulator is used as a study case to validate the effectiveness of the proposed control approach.

The rest of the paper is organized as follows. In Section 2, the problem formulation is presented. In Section 3, we propose the controller design method for n-link uncertain robot manipulator systems. Section 4 presents the simulation results for a robot manipulator system to illustrate the superiority of the proposed control approach in achieving the desired performance.

2. Problem Formulation

The main objective of this study is to design an enhanced tracking control for n-link robot manipulators in the presence of un-modelled dynamics, unknown payload dynamics, unknown friction force, parametric variations, and other unknown perturbations. The Euler-Lagrange dynamic equation for n-link robot manipulator systems can be written as:

$$J(q)\ddot{q} + C(q, \dot{q})\dot{q} + F(\dot{q}) + G(q) = u + d \quad (1)$$

where $q, \dot{q}, \ddot{q} \in \mathbb{R}^n$ are the vectors of joint angular position, velocity, and acceleration, respectively; $J(q) = (J_0(q) + \Delta J(q)) \in \mathbb{R}^{n \times n}$ is the bounded symmetric positive definite inertia matrix; $C(q, \dot{q}) = (C_0(q, \dot{q}) + \Delta C(q, \dot{q})) \in \mathbb{R}^n$ is the vector of centripetal Coriolis matrix; $G(q) = (G_0(q) + \Delta G(q)) \in \mathbb{R}^n$ denotes the gravity vector, with $J_0(q)$, $C_0(q, \dot{q})$, and $G_0(q)$ denote the nominal matrices, and $\Delta J(q)$, $\Delta C(q, \dot{q})$ and $\Delta G(q)$ represent the parametric variations on the nominal system; $F(\dot{q}) \in \mathbb{R}^n$ is the unknown friction vector; $u \in \mathbb{R}^n$ represents the vector of input torques; $d \in \mathbb{R}^n$ is the disturbance vector including un-modelled dynamics, unknown payload dynamics, and other unknown perturbations.

The Equation (1) can be reformulated as:

$$\begin{aligned} \ddot{q} &= A(Q) + B(q)u + D \\ &= A_0(Q) + B_0(q)u + (\Delta A(Q) + \Delta B(q)u) + D \end{aligned} \quad (2)$$

where $A(q, \dot{q}) = A_0(q, \dot{q}) + \Delta A(q, \dot{q}) = -J^{-1}(q)(C(q, \dot{q})\dot{q} + G(q))$, $B(q) = B_0(q) + \Delta B(q) = J^{-1}(q)$, and $D = J^{-1}(q)(d - F(\dot{q}))$, with $\Delta A(q, \dot{q})$ and $\Delta B(q)$ representing the parametric uncertainties on the system nominal, $Q = \begin{bmatrix} q^T & \dot{q}^T \end{bmatrix}^T \in \mathbb{R}^{2n}$ denotes the state vector of the system (1) assumed to be available to measurement, and $q = \begin{bmatrix} q_1 & q_2 & \dots & q_n \end{bmatrix}^T$ is the first element of the state vector.

For convenience, it is assumed that: $\dot{J}(Q) - 2C(Q)$ is skew symmetric, $J^{-1}(Q)$ exists, and D is bounded.

3. Proposed Control Approach

3.1. Introduction to Type-2 Fuzzy Logic Systems

A T2-FLS is characterized by MFs that are themselves fuzzy. Output sets of inference engine are T2-FSs. Therefore, a reducer is required to convert them into T1-FS. The obtained type reducer set is then defuzzified to obtain a crisp output.

An example of a T2 fuzzy MF is the Gaussian MF represented in Figure 1, with the associated FOU being shown as a bounded blue area.

Upper MF and Lower MF are two T1 MFs. μ_1 is the intersection of the crisp input x with the lower MF, and μ_2 is the intersection of x with the upper MF.

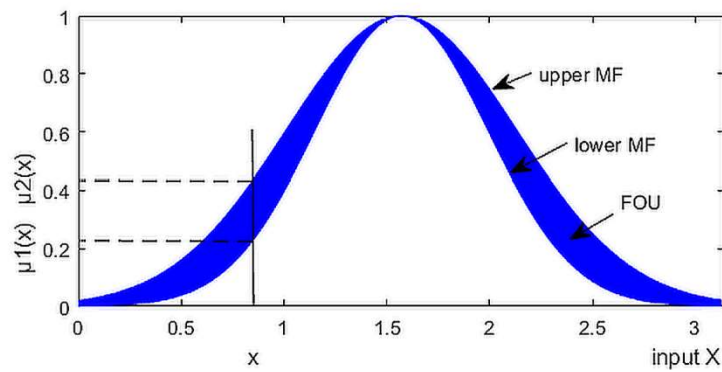


Figure 1. A type-2 fuzzy set.

3.1.1. Interval Type-2 Fuzzy System

For an IT2-FLS with a rule base of M rules, each having m antecedents, the j th rule can be expressed as [44]:

$$R^j: \text{ if } x_1 \text{ is } \tilde{F}_1^j \text{ and } x_2 \text{ is } \tilde{F}_2^j \dots \text{ and } x_m \text{ is } \tilde{F}_m^j \text{ then } y \text{ is } \tilde{\theta}^j \quad (3)$$

where $\tilde{F}_i^j$ and $\tilde{\theta}^j$ are IT2-FSs that are characterized by the fuzzy MFs $\mu_{\tilde{F}_i^j}(x_i)$ and $\mu_{\tilde{\theta}^j}(y)$, respectively, ($j = 1, 2, \dots, M, i = 1, 2, \dots, m$); $X = [x_1 \ x_2 \ \dots \ x_m]^T \in \mathbb{R}^m$ and $y \in \mathbb{R}$ are the input vector and the output of the IT2-FLS, respectively.

For the IT2-FLS described in (3), the meet operation is implemented by the product t-norm. Thus, the firing interval of the j -th fuzzy rule is the following IT1-FS:

$$Z^j(X) = [z_l^j(X), z_r^j(X)] \quad (4)$$

where $z_l^j(X) = \prod_{i=1}^m \mu_{\tilde{F}_i^j}^{low}(x_i)$ and $z_r^j(X) = \prod_{i=1}^m \mu_{\tilde{F}_i^j}^{upp}(x_i)$, with $\mu_{\tilde{F}_i^j}^{low}(x_i)$ and $\mu_{\tilde{F}_i^j}^{upp}(x_i)$ are the lower and upper MFs of $\mu_{\tilde{F}_i^j}(x_i)$, respectively.

3.1.2. Type Reduction for Interval Type-2 Fuzzy Sets

The output of the inference engine must be reduced to a T1-FS before defuzzification. The type reduction using the center of sets (COS) method is adopted in this study for the IT2-FSs and it is given by [45]:

$$Y_{\cos}(\theta^1, \theta^2, \dots, \theta^M, Z^1, Z^2, \dots, Z^M) = \int_{y^1} \int_{y^2} \dots \int_{y^M} \int_{z^1} \int_{z^2} \dots \int_{z^M} \frac{\sum_{j=1}^M z^j}{\sum_{j=1}^M y^j z^j} \quad (5)$$

where $Y_{\cos}$ is an IT1-FS defined by two end points $y_l(X)$ and $y_r(X)$; $y^j \in \theta^j = [\theta_l^j, \theta_r^j]$ with θ^j is the centroid of the associated IT2 fuzzy consequent set $\tilde{\theta}^j$; and, $z^j \in Z^j(X) = [z_l^j(X), z_r^j(X)]$.

The defuzzified crisp out by using the center of gravity is then obtained, as follows:

$$y = \frac{y_l + y_r}{2} \quad (6)$$

where y_l and y_r can be expressed as:

$$\left\{ \begin{array}{l} y_l = \min_{Z^j} \frac{\sum_{j=1}^M \theta_l^j z^j}{\sum_{j=1}^M z^j} = \theta_l^T \xi_l \\ y_r = \max_{Z^j} \frac{\sum_{j=1}^M \theta_r^j z^j}{\sum_{j=1}^M z^j} = \theta_r^T \xi_r \end{array} \right. \quad (7)$$

where $\xi_l = [\xi_l^1 \ \xi_l^2 \ \dots \ \xi_l^M]^T$ and $\xi_r = [\xi_r^1 \ \xi_r^2 \ \dots \ \xi_r^M]^T$ are two vectors of fuzzy basis functions, such that: $\xi_l^j = \frac{z^j}{\sum_{j=1}^M z^j}$ and $\xi_r^j = \frac{\bar{z}^j}{\sum_{j=1}^M \bar{z}^j}$, with $(z^j, \bar{z}^j) \in Z^j(X)$; $\theta_l = [\theta_l^1 \ \theta_l^2 \ \dots \ \theta_l^M]^T$ and $\theta_r = [\theta_r^1 \ \theta_r^2 \ \dots \ \theta_r^M]^T$ are the adjustable parameter vectors.

In this study, $\underline{z}^j$ and $\bar{z}^j$ are determined while using the iterative algorithm that was developed by Mendel and Karnik [46]. Therefore, y_l and y_r can be easily computed.

3.2. Control Law Design

In order to ensure that the state q of the system (1) effectively tracks a desired reference q_r in the presence of dynamic uncertainties and unknown disturbances without generating the chattering, a new robust AIT2-FSMC law is proposed.

3.2.1. Sliding Mode Control Law

The main objective of SMC is to force the system dynamics to reach and then remain on the sliding surface $s(Q, t) = \underline{0}$, with $\underline{0} \in \mathbb{R}^n$ denotes the null vector.

Define the tracking error $e = q_r - q$. Then, in order to ensure that the tracking error converges asymptotically to zero when the sliding surface $s(Q, t) = \underline{0}$ is established, we adopted in this study the following sliding surface defined by Slotine for a j th order system as [47]:

$$\begin{aligned} s(Q, t) &= \left(\frac{\partial}{\partial t} + \lambda \right)^{(p-1)} e \\ &= \sum_{j=0}^{p-1} \frac{(p-1)!}{j!(p-j-1)!} \left(\frac{\partial}{\partial t} \right)^{(p-j-1)} \lambda^j e \\ &= [s_1 s_2 \dots s_n]^T \in \mathbb{R}^n \end{aligned} \quad (8)$$

where $\lambda = \text{diag}(\lambda_i)_{1 \leq i \leq n} \in \mathbb{R}^{(n \times n)}$ is a diagonal matrix, with λ_i is the positive slope of the sliding surface s_i ; p denotes the system order. In this study, for the system (1), we have $p = 2$. Therefore:

$$s(Q, t) = \dot{e} + \lambda e \quad (9)$$

The time derivative of the above equation can be given, for the system (2), as:

$$\dot{s}(Q, t) = \ddot{q}_r - (A_0(Q) + B_0(q)u) - (\Delta A(Q) + \Delta B(q)u) - D + \lambda \dot{e} \quad (10)$$

In this paper, the IT2-FLS (6) is used to approximate the uncertainties $\Delta A(Q)$ and $\Delta B(q)$. Therefore, $\Delta A(Q)$ and $\Delta B(q)$ are substituted by their AIT2-FLSs, respectively:

$$\begin{aligned} \Delta \hat{A}(i) &= \xi_a^T(i) \theta_a(i) & i &= 1, \dots, n \\ \Delta \hat{B}(i, j) &= \xi_b^T(i, j) \theta_b(i, j) & j &= 1, \dots, n \end{aligned} \quad (11)$$

where $\xi_a^T(i) = \frac{1}{2}(\xi_a(i)_l^T + \xi_a(i)_r^T)$ and $\xi_b^T(i, j) = \frac{1}{2}(\xi_b(i, j)_l^T + \xi_b(i, j)_r^T)$, with $\xi_a(i)_l = [\xi_a(i)_l^1 \ \xi_a(i)_l^2 \ \dots \ \xi_a(i)_l^M]^T$, $\xi_a(i)_r = [\xi_a(i)_r^1 \ \xi_a(i)_r^2 \ \dots \ \xi_a(i)_r^M]^T$, $\xi_b(i, j)_l = [\xi_b(i, j)_l^1 \ \xi_b(i, j)_l^2 \ \dots \ \xi_b(i, j)_l^M]^T$, and $\xi_b(i, j)_r = [\xi_b(i, j)_r^1 \ \xi_b(i, j)_r^2 \ \dots \ \xi_b(i, j)_r^M]^T$ are the vectors of fuzzy basis functions as they were described in (7); $\theta_a(i) = [\theta_a^1(i) \ \theta_a^2(i) \ \dots \ \theta_a^M(i)]^T$ and $\theta_b(i, j) = [\theta_b^1(i, j) \ \theta_b^2(i, j) \ \dots \ \theta_b^M(i, j)]^T$ are the parameter vectors free to be designed by adaptive law; and, M is the number of rules.

The system (11) can be rewritten as:

$$\begin{aligned} \Delta \hat{A} &= [\Delta \hat{A}(1) \ \Delta \hat{A}(2) \ \dots \ \Delta \hat{A}(n)]^T = \xi_a^T \theta_a \\ \Delta \hat{B} &= \begin{bmatrix} \Delta \hat{B}(1,1) & \Delta \hat{B}(1,2) & \dots & \Delta \hat{B}(1,n) \\ \Delta \hat{B}(2,1) & \Delta \hat{B}(2,2) & \dots & \Delta \hat{B}(2,n) \\ \vdots & \vdots & \ddots & \vdots \\ \Delta \hat{B}(n,1) & \Delta \hat{B}(n,2) & \dots & \Delta \hat{B}(n,n) \end{bmatrix} = \xi_b^T \theta_b \end{aligned} \quad (12)$$

$$\begin{aligned} \text{where } \xi_a^T &= \begin{bmatrix} \xi_a^T(1) & 0 & 0 & \dots & 0 \\ 0 & \xi_a^T(2) & 0 & \dots & 0 \\ 0 & 0 & . & \dots & \vdots \\ \vdots & \vdots & \vdots & \ddots & 0 \\ 0 & 0 & \dots & 0 & \xi_a^T(n) \end{bmatrix} \quad \text{and} \quad \theta_a = \begin{bmatrix} \theta_a(1) \\ \theta_a(2) \\ \vdots \\ \theta_a(n) \end{bmatrix}; \\ \xi_b^T &= \begin{bmatrix} \xi_b^T(1) & 0 & 0 & \dots & 0 \\ 0 & \xi_b^T(2) & 0 & \dots & 0 \\ 0 & 0 & . & 0 & 0 \\ \vdots & \vdots & \vdots & \ddots & \vdots \\ 0 & 0 & \dots & 0 & \xi_b^T(n) \end{bmatrix} \quad \text{and} \quad \theta_b = \begin{bmatrix} \theta_b(1) \\ \theta_b(2) \\ \vdots \\ \theta_b(n) \end{bmatrix}, \quad \text{with} \\ \xi_b(i) &= [\xi_b^T(i,1) \ \xi_b^T(i,2) \ \dots \ \xi_b^T(i,n)]^T \quad \text{and} \quad \theta_b(i) = \begin{bmatrix} \theta_b(i,1) & 0 & 0 & \dots & 0 \\ 0 & \theta_b(i,2) & 0 & \dots & 0 \\ 0 & 0 & . & \dots & \vdots \\ \vdots & \vdots & \vdots & \ddots & 0 \\ 0 & 0 & \dots & 0 & \theta_b(i,n) \end{bmatrix}. \end{aligned}$$

Define the optimal parameters of $\Delta \hat{A}(Q)$ and $\Delta \hat{B}(q)$:

$$\begin{aligned} \theta_a^* &= \underset{\theta_a}{\operatorname{argmin}} \left(\sup_Q \|\Delta \hat{A} - \Delta A\| \right) \\ \theta_b^* &= \underset{\theta_b}{\operatorname{argmin}} \left(\sup_q \|\Delta \hat{B} - \Delta B\| \right) \end{aligned} \quad (13)$$

The minimum approximation error of $\Delta A(Q)$ and $\Delta B(q)$ is then given by:

$$\varepsilon = \Delta A^*(Q) - \Delta A(Q) + (\Delta B^*(q) - \Delta B(q))u \quad (14)$$

where $\Delta A^*(Q) = \xi_a^T \theta_a^*$ and $\Delta B^*(q) = \xi_b^T \theta_b^*$ are the optimal approximation of $\Delta A(Q)$ and $\Delta B(q)$, respectively.

In order to ensure the desired control performance, a new control law is designed, as follows:

$$u = \begin{bmatrix} u_1 & u_2 & \dots & u_n \end{bmatrix}^T = (B_0(q) + \Delta \hat{B}(q))^{-1} (\ddot{q}_r - (A_0(Q) + \Delta \hat{A}(Q)) + \lambda \dot{e} - u_{sl}) \quad (15)$$

$$= u_{eq} - (B_0(q) + \Delta \hat{B}(q))^{-1} u_{sl}$$

where $u_{eq} = (B_0(q) + \Delta \hat{B}(q))^{-1} (\ddot{q}_r - (A_0(Q) + \Delta \hat{A}(Q)) + \lambda \dot{e})$.

The fuzzy equivalent control u_{eq} describes the sliding mode of the system dynamics, it drives the system trajectories to the desired dynamics and it is obtained when $\dot{s} = 0$. However, dynamic uncertainties and unknown disturbances may cause a deterioration of the sliding mode. To overcome this problem, u_{sl} is introduced, and it describes the reaching phase of the system dynamics towards the sliding surface $s = 0$. Thus, the new reaching control law is designed, as follows:

$$u_{sl} = -\alpha s(Q, t) - k \int_0^{t^r} \text{sign}(s(Q, t)) dt - \mu \omega(s(Q, t)) \quad (16)$$

where $\omega(s(Q, t)) = \begin{bmatrix} \omega_1(s_1) & \omega_2(s_2) & \dots & \omega_n(s_n) \end{bmatrix}^T \in \mathbb{R}^n$ such that $\omega_i(s_i) = \begin{cases} s_i + \varepsilon_i \text{sign}(s_i) & s_i \in \Omega \\ \frac{\text{sign}(s_i)}{\log^2(|s_i|)} & s_i \notin \Omega \end{cases}$, with $\Omega = \{s_i | |s_i| \geq \frac{N_i}{2}, 0 < N_i \leq 1\}$, and $\varepsilon_i = \frac{1}{\log^2(\frac{N_i}{2})} - \frac{N_i}{2}$ in

order to ensure a continuous signal in $|s_i| = \frac{N_i}{2}$; $\alpha = \text{diag}(\alpha_i)_{1 \leq i \leq n}$, $\mu = \text{diag}(\mu_i)_{1 \leq i \leq n}$ and $k = \text{diag}(k_i)_{1 \leq i \leq n}$ are diagonal matrices of the positive reaching control gains α_i , μ_i and k_i , respectively,

$$i = 1, \dots, n; \int_0^{t^r} \text{sign}(s(Q, t)) dt = \begin{bmatrix} \int_0^{t_1^r} \text{sign}(s_1) dt & \int_0^{t_2^r} \text{sign}(s_2) dt & \dots & \int_0^{t_n^r} \text{sign}(s_n) dt \end{bmatrix}^T,$$

with $t^r = \begin{bmatrix} t_1^r & t_2^r & \dots & t_n^r \end{bmatrix}^T$ such that $t_i^r = \begin{cases} t & |s_i| > \vartheta_i \\ t_{\vartheta_i} & |s_i| \leq \vartheta_i \end{cases}$ denotes the reaching time to a neighborhood ϑ_i of the sliding surface $s_i = 0$.

The adaptive laws for the synthesized AIT2-FLSs are designed, as follows:

$$\begin{aligned} \dot{\theta}_a &= -\gamma_a \xi_a s \\ \dot{\Phi}_b &= -\gamma_b u_d \xi_b s \end{aligned} \quad (17)$$

where $\Phi_b = \theta_b I_n$, with $I_n = \begin{bmatrix} 1 & 1 & \dots & 1 \end{bmatrix}^T \in \mathbb{R}^n$; $u_d = \begin{bmatrix} I_d^1 & 0 & 0 & \dots & 0 \\ 0 & I_d^2 & 0 & \dots & 0 \\ 0 & 0 & . & \dots & \vdots \\ \vdots & \vdots & \vdots & \ddots & 0 \\ 0 & 0 & \dots & 0 & I_d^n \end{bmatrix}$, with $I_d^1 = I_d^2 =$

$\dots = I_d^n = I_d$ such as $I_d = \begin{bmatrix} u_1 I & 0 & 0 & \dots & 0 \\ 0 & u_2 I & 0 & \dots & 0 \\ 0 & 0 & . & \dots & \vdots \\ \vdots & \vdots & \vdots & \ddots & 0 \\ 0 & 0 & \dots & 0 & u_n I \end{bmatrix}$, $I \in \mathbb{R}^{M \times M}$ denotes the identity matrix; γ_a

and γ_b are positive constants.

Theorem 1. For the corollled system (1) with the AIT2-FLSs (11) and adaptive laws (17), the control law defined in (15) is globally asymptotically stable in closed loop system with the tracking error converges asymptotically to zero despite dynamic uncertainties and unknown disturbances.

Proof. In order to ensure the desired dynamics and guarantee the stability of the closed loop control system, the following Lyapunov function is adopted:

$$v = \frac{1}{2}s^T s + \frac{1}{2\gamma_a}\tilde{\theta}_a^T \tilde{\theta}_a + \frac{1}{2\gamma_b}\tilde{\Phi}_b^T \tilde{\Phi}_b \quad (18)$$

where $\tilde{\theta}_a = \theta_a - \theta_a^*$ and $\tilde{\Phi}_b = \tilde{\theta}_b I_n$, with $\tilde{\theta}_b = \theta_b - \theta_b^*$.

The time derivative of (18) is:

$$\dot{v} = s^T \dot{s} + \frac{1}{\gamma_a}\dot{\tilde{\theta}}_a^T \tilde{\theta}_a + \frac{1}{\gamma_b}\dot{\tilde{\Phi}}_b^T \tilde{\Phi}_b \quad (19)$$

Substitute $\dot{s}$ defined in (10) into (19), gives:

$$\dot{v} = s^T [\ddot{q}_r - (A_0(Q) + B_0(q)u)] - s^T [(\Delta A(Q) + \Delta B(q)u) + D - \lambda \dot{e}] + \frac{1}{\gamma_a}\dot{\tilde{\theta}}_a^T \tilde{\theta}_a + \frac{1}{\gamma_b}\dot{\tilde{\Phi}}_b^T \tilde{\Phi}_b \quad (20)$$

From (15), we get:

$$\ddot{q}_r = (A_0(Q) + \Delta \hat{A}(Q)) + (B_0(q) + \Delta \hat{B}(q))u + u_{sl} - \lambda \dot{e} \quad (21)$$

Substituting (21) into (20) gives:

$$\begin{aligned} \dot{v} &= s^T [(\Delta \hat{A}(Q) - \Delta A(Q)) + (\Delta \hat{B}(q) - \Delta B(q))u + (u_{sl} - D)] + \frac{1}{\gamma_a}\dot{\tilde{\theta}}_a^T \tilde{\theta}_a + \frac{1}{\gamma_b}\dot{\tilde{\Phi}}_b^T \tilde{\Phi}_b \\ &= s^T [(\Delta \hat{A}(Q) - \Delta A^*(Q)) + (\Delta A^*(Q) - \Delta A(Q)) + (\Delta \hat{B}(q) - \Delta B^*(q))u + (\Delta B^*(q) - \Delta B(q))u + (u_{sl} - D)] + \\ &\quad \frac{1}{\gamma_a}\dot{\tilde{\theta}}_a^T \tilde{\theta}_a + \frac{1}{\gamma_b}\dot{\tilde{\Phi}}_b^T \tilde{\Phi}_b \\ &= s^T [\xi_a^T \tilde{\theta}_a + \xi_b^T \tilde{\theta}_b u + u_{sl} - \varphi] + \frac{1}{\gamma_a}\dot{\tilde{\theta}}_a^T \tilde{\theta}_a + \frac{1}{\gamma_b}\dot{\tilde{\Phi}}_b^T \tilde{\Phi}_b \end{aligned} \quad (22)$$

where $\varphi = D - \varepsilon = [\varphi_1 \quad \varphi_2 \quad \dots \quad \varphi_n]^T$ is assumed to be bounded ($|\varphi_i| \leq \phi_i, \phi_i \geq 0, i = 1, \dots, n$).

We have $\tilde{\theta}_b u = u_d \tilde{\Phi}_b$. Then, substitute $\tilde{\theta}_b u$ by $u_d \tilde{\Phi}_b$ into (22), gives:

$$\begin{aligned} \dot{v} &= s^T \left(\xi_a^T \tilde{\theta}_a + \xi_b^T u_d \tilde{\Phi}_b + u_{sl} - \varphi \right) + \frac{1}{\gamma_a}\dot{\tilde{\theta}}_a^T \tilde{\theta}_a + \frac{1}{\gamma_b}\dot{\tilde{\Phi}}_b^T \tilde{\Phi}_b \\ &= \left(s^T \xi_a^T + \frac{1}{\gamma_a}\dot{\tilde{\theta}}_a^T \right) \tilde{\theta}_a + \left(s^T \xi_b^T u_d + \frac{1}{\gamma_b}\dot{\tilde{\Phi}}_b^T \right) \tilde{\Phi}_b + s^T (u_{sl} - \varphi) \end{aligned} \quad (23)$$

Substitute $\dot{\tilde{\theta}}_a$ and $\dot{\tilde{\Phi}}_b$ defined in (17) into (23), then we have:

$$\dot{v} = s^T (u_{sl} - \varphi) \quad (24)$$

Substituting u_{sl} by its expression gives:

$$\dot{v} = -s^T \left(\alpha s(Q, t) + k \int_0^t \text{sign}(s(Q, t)) \, dt + \mu \omega(s(Q, t)) \right) - s^T \varphi \quad (25)$$

The above equation becomes negative if the following condition is verified:

$$-s^T \left(\alpha s(Q, t) + k \int_0^t \text{sign}(s(Q, t)) \, dt + \mu \omega(s(Q, t)) \right) \leq s^T \varphi \quad (26)$$

The condition (26) is guaranteed if:

$$\alpha_i |s_i| + k_i t_i^r + \mu_i |\omega_i| \geq \phi_i, \quad i = 1, \dots, n \quad (27)$$

An adequate choice of the reaching control gains k_i , α_i , and μ_i makes it possible that the condition (27) can be guaranteed. Hence, the function (19) becomes negative. $\square$

In practice, and because the upper bounds ϕ_i of φ_i are unknown, it becomes very difficult to obtain the optimal reaching control gains k_i , α_i , and μ_i that ensure the rejection of φ_i without deteriorating the system robustness or generating the undesired chattering. Indeed, the large gains can cover a wide range of uncertainties. However, they can cause the chattering and a dynamic response with overshoot. On the other hand, the small gains can deteriorate the system robustness and affect the tracking control accuracy. In this paper, for handling this problem, a new AIT2-FLS is designed to better estimate the gains (k_i , α_i , and μ_i) of the control law u_{sl} that provide the best tracking control performance of (1) by guaranteeing the condition (27) without generating the chattering.

3.2.2. Adaptive Interval Type-2 Fuzzy Sliding Mode Control Law

Based on the IT2-FLS (6), and with the sliding surface $s(Q, t)$ as input vector, the terms $u_\alpha = -\alpha s$, $u_k = -k \int_0^{t^r} \text{sign}(s) dt$ and $u_\mu = -\mu \omega$ of the control law defined in (16) are substituted by their AIT2-FLSs, respectively:

$$\begin{aligned} \hat{u}_\alpha(i) &= \zeta_\alpha^T(i) \theta_\alpha(i) s_i \\ \hat{u}_k(i) &= \zeta_k^T(i) \theta_k(i) t_i^r \\ \hat{u}_\mu(i) &= \begin{cases} \left(\zeta_\mu^T(i) \theta_\mu(i) \right) (|s_i| + \varepsilon_i), & s_i \in \Omega, \quad i = 1, \dots, n \\ \frac{\zeta_\mu^T(i) \theta_\mu(i)}{\log^2 |s_i|}, & s_i \notin \Omega \end{cases} \end{aligned} \quad (28)$$

where $\theta_\alpha(i) = [\theta_\alpha^1(i) \ \theta_\alpha^2(i) \ \dots \ \theta_\alpha^M(i)]^T$, $\theta_k(i) = [\theta_k^1(i) \ \theta_k^2(i) \ \dots \ \theta_k^M(i)]^T$, and $\theta_\mu(i) = [\theta_\mu^1(i) \ \theta_\mu^2(i) \ \dots \ \theta_\mu^M(i)]^T$ are the parameter vectors free to be designed by adaptive law; $\zeta_k(i) = \frac{1}{2}(\zeta_k(i)_l + \zeta_k(i)_r) = [\zeta_k^1(i) \ \zeta_k^2(i) \ \dots \ \zeta_k^M(i)]^T$, and $\zeta_\mu(i) = \frac{1}{2}(\zeta_\mu(i)_l + \zeta_\mu(i)_r) = [\zeta_\mu^1(i) \ \zeta_\mu^2(i) \ \dots \ \zeta_\mu^M(i)]^T$ are the vectors of fuzzy basis functions, as they were described in (7).

The system (28) can be rewritten as:

$$\begin{aligned} \hat{u}_\alpha &= [\hat{u}_\alpha(1) \ \hat{u}_\alpha(2) \ \dots \ \hat{u}_\alpha(n)]^T = \zeta_\alpha^T \theta_\alpha s \\ \hat{u}_k &= [\hat{u}_k(1) \ \hat{u}_k(2) \ \dots \ \hat{u}_k(n)]^T = \zeta_k^T \theta_k t^r \\ \hat{u}_\mu &= [\hat{u}_\mu(1) \ \hat{u}_\mu(2) \ \dots \ \hat{u}_\mu(n)]^T = \zeta_\mu^T \theta_\mu \omega \end{aligned} \quad (29)$$

where

$$\begin{aligned} \zeta_\alpha^T &= \begin{bmatrix} \zeta_\alpha^T(1) & 0 & 0 & \dots & 0 \\ 0 & \zeta_\alpha^T(2) & 0 & \dots & 0 \\ 0 & 0 & . & \dots & \vdots \\ \vdots & \vdots & \vdots & \ddots & 0 \\ 0 & 0 & 0 & \dots & \zeta_\alpha^T(n) \end{bmatrix}, \quad \zeta_k^T = \begin{bmatrix} \zeta_k^T(1) & 0 & 0 & \dots & 0 \\ 0 & \zeta_k^T(2) & 0 & \dots & 0 \\ 0 & 0 & . & \dots & \vdots \\ \vdots & \vdots & \vdots & \ddots & 0 \\ 0 & 0 & 0 & \dots & \zeta_k^T(n) \end{bmatrix} \\ \zeta_\mu^T &= \begin{bmatrix} \zeta_\mu^T(1) & 0 & 0 & \dots & 0 \\ 0 & \zeta_\mu^T(2) & 0 & \dots & 0 \\ 0 & 0 & . & \dots & \vdots \\ \vdots & \vdots & \vdots & \ddots & 0 \\ 0 & 0 & 0 & \dots & \zeta_\mu^T(n) \end{bmatrix}, \quad \theta_\alpha = \begin{bmatrix} \theta_\alpha(1) & 0 & 0 & \dots & 0 \\ 0 & \theta_\alpha(2) & 0 & \dots & 0 \\ 0 & 0 & . & \dots & \vdots \\ \vdots & \vdots & \vdots & \ddots & 0 \\ 0 & 0 & 0 & \dots & \theta_\alpha(n) \end{bmatrix} \end{aligned}$$

$$\theta_k = \begin{bmatrix} \theta_k(1) & 0 & 0 & \dots & 0 \\ 0 & \theta_k(2) & 0 & \dots & 0 \\ 0 & 0 & \cdot & \dots & \vdots \\ \vdots & \vdots & \vdots & \ddots & 0 \\ 0 & 0 & 0 & \dots & \theta_k(n) \end{bmatrix}, \theta_\mu = \begin{bmatrix} \theta_\mu(1) & 0 & 0 & \dots & 0 \\ 0 & \theta_\mu(2) & 0 & \dots & 0 \\ 0 & 0 & \cdot & \dots & \vdots \\ \vdots & \vdots & \vdots & \ddots & 0 \\ 0 & 0 & 0 & \dots & \theta_\mu(n) \end{bmatrix}$$

$$\omega = [\omega_1 \ \omega_2 \ \dots \ \omega_n]^T \text{ such that } \omega_i = \begin{cases} (|s_i| + \varepsilon_i), & s_i \in \Omega \\ \frac{1}{\log^2 |s_i|}, & s_i \notin \Omega \end{cases}, i = 1, \dots, n.$$

Define the optimal parameters of the AIT2-FLSs $\hat{u}_\alpha$, $\hat{u}_k$, and $\hat{u}_\mu$:

$$\begin{aligned} \theta_\alpha^* &= \operatorname{argmin}_{\theta_\alpha} \left(\sup_s \|\hat{u}_\alpha - u_\alpha\| \right) \\ \theta_k^* &= \operatorname{argmin}_{\theta_k} \left(\sup_s \|\hat{u}_k - u_k\| \right) \\ \theta_\mu^* &= \operatorname{argmin}_{\theta_\mu} \left(\sup_s \|\hat{u}_\mu - u_\mu\| \right) \end{aligned} \quad (30)$$

The global AIT2-FSMC law of the proposed control approach is designed as:

$$\begin{aligned} u &= [u_1 \ u_2 \ \dots \ u_n]^T \\ &= (B_0(q) + \Delta \hat{B}(q))^{-1} (\ddot{q}_r - (A_0(Q) + \Delta \hat{A}(Q)) + \lambda \dot{e} - \hat{u}_{sl}) \end{aligned} \quad (31)$$

where $\hat{u}_{sl} = \hat{u}_\alpha + \hat{u}_k + \hat{u}_\mu$. The adaptive laws for the synthesized AIT2-FLSs defined in (29) are designed, as follows:

$$\begin{aligned} \dot{\Phi}_\alpha &= -\gamma_\alpha s_d \zeta_\alpha s \\ \dot{\Phi}_k &= -\gamma_k t_d^r \zeta_k s \\ \dot{\Phi}_\mu &= -\gamma_\mu W_d \zeta_\mu s \end{aligned} \quad (32)$$

$$\begin{aligned} \text{where } s_d &= \begin{bmatrix} s_1 I & 0 & 0 & \dots & 0 \\ 0 & s_2 I & 0 & 0 & 0 \\ 0 & 0 & \cdot & \dots & \dots \\ \vdots & \vdots & \vdots & \ddots & 0 \\ 0 & 0 & 0 & \dots & s_n I \end{bmatrix}, \quad t_d^r = \begin{bmatrix} t_1^r I & 0 & 0 & \dots & 0 \\ 0 & t_2^r I & 0 & 0 & 0 \\ 0 & 0 & \cdot & \dots & \dots \\ \vdots & \vdots & \vdots & \ddots & 0 \\ 0 & 0 & 0 & \dots & t_n^r I \end{bmatrix}, \quad \text{and } W_d = \\ & \begin{bmatrix} \omega_1 I & 0 & 0 & \dots & 0 \\ 0 & \omega_2 I & 0 & 0 & 0 \\ 0 & 0 & \cdot & \dots & \dots \\ \vdots & \vdots & \vdots & \ddots & 0 \\ 0 & 0 & 0 & \dots & \omega_n I \end{bmatrix}; \quad \tilde{\Phi}_\alpha = \tilde{\theta}_\alpha I_n, \tilde{\Phi}_k = \tilde{\theta}_k I_n \text{ and } \tilde{\Phi}_\mu = \tilde{\theta}_\mu I_n \text{ such as } \tilde{\theta}_\alpha = \theta_\alpha - \theta_\alpha^*, \tilde{\theta}_k = \theta_k - \theta_k^* \\ & \text{and } \tilde{\theta}_\mu = \theta_\mu - \theta_\mu^*; \gamma_\alpha, \gamma_k \text{ and } \gamma_\mu \text{ are positive constants.} \end{aligned}$$

Theorem 2. For the n -link robot manipulator system (1), with AIT2-FLSs defined in (11) and (29), and adaptive laws expressed by (17) and (32), the proposed AIT2-FSMC law (31) is smooth and globally asymptotically stable in closed loop system with the tracking error converge asymptotically to zero despite dynamic uncertainties and unknown disturbances.

Proof. Consider the following augmented Lyapunov function:

$$v = \frac{1}{2} s^T s + \frac{1}{2\gamma_a} \tilde{\theta}_a^T \tilde{\theta}_a + \frac{1}{2\gamma_b} \tilde{\Phi}_b^T \tilde{\Phi}_b + \frac{1}{2\gamma_\alpha} \tilde{\Phi}_\alpha^T \tilde{\Phi}_\alpha + \frac{1}{2\gamma_k} \tilde{\Phi}_k^T \tilde{\Phi}_k + \frac{1}{2\gamma_\mu} \tilde{\Phi}_\mu^T \tilde{\Phi}_\mu \quad (33)$$

According to (24) and (31), the time derivative of (33) gives:

$$\dot{v} = s^T(\hat{u}_{sl} - \varphi) + \frac{1}{\gamma_\alpha} \dot{\Phi}_\alpha^T \tilde{\Phi}_\alpha + \frac{1}{\gamma_k} \dot{\Phi}_k^T \tilde{\Phi}_k + \frac{1}{\gamma_\mu} \dot{\Phi}_\mu^T \tilde{\Phi}_\mu \quad (34)$$

Let $\hat{u}_\alpha^* = \xi_\alpha^T \theta_\alpha^* s = -\alpha^* s$, $\hat{u}_k^* = \xi_k^T \theta_k^* t^r = -k^* \int_0^{t^r} \text{sign}(s) dt$ and $\hat{u}_\mu^* = \xi_\mu^T \theta_\mu^* \omega = -\mu^* \omega$ denote, respectively, the optimal control laws of $u_\alpha = -\alpha s$, $u_k = -k \int_0^{t^r} \text{sign}(s) dt$ and $u_\mu = -\mu \omega$ that ensure the best tracking control performance of the robot manipulator (1) by providing the optimal gains $\alpha^* = \text{diag}(\alpha_i^*)$, $\mu^* = \text{diag}(\mu_i^*)$, and $k^* = \text{diag}(k_i^*)$, $i = 1, \dots, n$ of u_{sl} , which allows for effectively rejecting the effect of φ without generating the undesired chattering.

Considering (27), and as the adaptive gains α^* , μ^* and k^* are, respectively, the optimal estimation of α , μ , and k . Thus, the following condition is verified:

$$\alpha_i^* |s_i| + k_i^* t_i^r + \mu_i^* |\omega_i| \geq \phi_i, \quad i = 1, \dots, n \quad (35)$$

By introducing the optimal control law $u_{sl}^* = u_\alpha^* + u_k^* + u_\mu^*$ into (34), it gives:

$$\begin{aligned} \dot{v} &= s^T((\hat{u}_{sl} - u_{sl}^*) + (u_{sl}^* - \varphi)) + \frac{1}{\gamma_\alpha} \dot{\Phi}_\alpha^T \tilde{\Phi}_\alpha + \frac{1}{\gamma_k} \dot{\Phi}_k^T \tilde{\Phi}_k + \frac{1}{\gamma_\mu} \dot{\Phi}_\mu^T \tilde{\Phi}_\mu \\ &= s^T(\xi_\alpha^T \tilde{\theta}_\alpha s + \xi_k^T \tilde{\theta}_k t^r + \xi_\mu^T \tilde{\theta}_\mu \omega) + s^T(u_{sl}^* - \varphi) + \frac{1}{\gamma_\alpha} \dot{\Phi}_\alpha^T \tilde{\Phi}_\alpha + \frac{1}{\gamma_k} \dot{\Phi}_k^T \tilde{\Phi}_k + \frac{1}{\gamma_\mu} \dot{\Phi}_\mu^T \tilde{\Phi}_\mu \end{aligned} \quad (36)$$

Substituting $\hat{u}_{sl}$ and u_{sl}^* by their expression into (36), and taking into account that $\tilde{\theta}_\alpha s = s_d \tilde{\Phi}_\alpha$, $\tilde{\theta}_k t^r = t_d^r \tilde{\Phi}_k$ and $\tilde{\theta}_\mu \omega = W_d \tilde{\Phi}_\mu$, then we have:

$$\begin{aligned} \dot{v} &= s^T(\xi_\alpha^T \tilde{\theta}_\alpha s + \xi_k^T \tilde{\theta}_k t^r + \xi_\mu^T \tilde{\theta}_\mu \omega) + s^T(u_{sl}^* - \varphi) + \frac{1}{\gamma_\alpha} \dot{\Phi}_\alpha^T \tilde{\Phi}_\alpha + \frac{1}{\gamma_k} \dot{\Phi}_k^T \tilde{\Phi}_k + \frac{1}{\gamma_\mu} \dot{\Phi}_\mu^T \tilde{\Phi}_\mu \\ &= s^T(\xi_\alpha^T s_d \tilde{\Phi}_\alpha + \xi_k^T t_d^r \tilde{\Phi}_k + \xi_\mu^T W_d \tilde{\Phi}_\mu) + s^T(u_{sl}^* - \varphi) + \frac{1}{\gamma_\alpha} \dot{\Phi}_\alpha^T \tilde{\Phi}_\alpha + \frac{1}{\gamma_k} \dot{\Phi}_k^T \tilde{\Phi}_k + \frac{1}{\gamma_\mu} \dot{\Phi}_\mu^T \tilde{\Phi}_\mu \\ &= s^T \xi_\alpha^T s_d \tilde{\Phi}_\alpha + \frac{1}{\gamma_\alpha} \dot{\Phi}_\alpha^T \tilde{\Phi}_\alpha + s^T \xi_k^T t_d^r \tilde{\Phi}_k + \frac{1}{\gamma_k} \dot{\Phi}_k^T \tilde{\Phi}_k + s^T \xi_\mu^T W_d \tilde{\Phi}_\mu + \frac{1}{\gamma_\mu} \dot{\Phi}_\mu^T \tilde{\Phi}_\mu + s^T(u_{sl}^* - \varphi) \\ &= \left(s^T \xi_\alpha^T s_d + \frac{1}{\gamma_\alpha} \dot{\Phi}_\alpha^T\right) \tilde{\Phi}_\alpha + \left(s^T \xi_k^T t_d^r + \frac{1}{\gamma_k} \dot{\Phi}_k^T\right) \tilde{\Phi}_k + \left(s^T \xi_\mu^T W_d + \frac{1}{\gamma_\mu} \dot{\Phi}_\mu^T\right) \tilde{\Phi}_\mu + s^T(u_{sl}^* - \varphi) \end{aligned} \quad (37)$$

Substituting $\dot{\Phi}_\alpha$, $\dot{\Phi}_k$ and $\dot{\Phi}_\mu$ by their expressions gives:

$$\dot{v} = s^T(u_{sl}^* - \varphi) \quad (38)$$

Substituting u_{sl}^* by its expression into (38), then we have:

$$\dot{v} = s^T(-\alpha^* s - k^* t^r \text{sign}(s) - \mu^* \omega \text{sign}(s) - \varphi) \quad (39)$$

According to (35), the above equation is negative. Thus, the desired tracking control performance of the proposed approach is guaranteed. $\square$

The proposed control approach is depicted in the Figure 2 below.

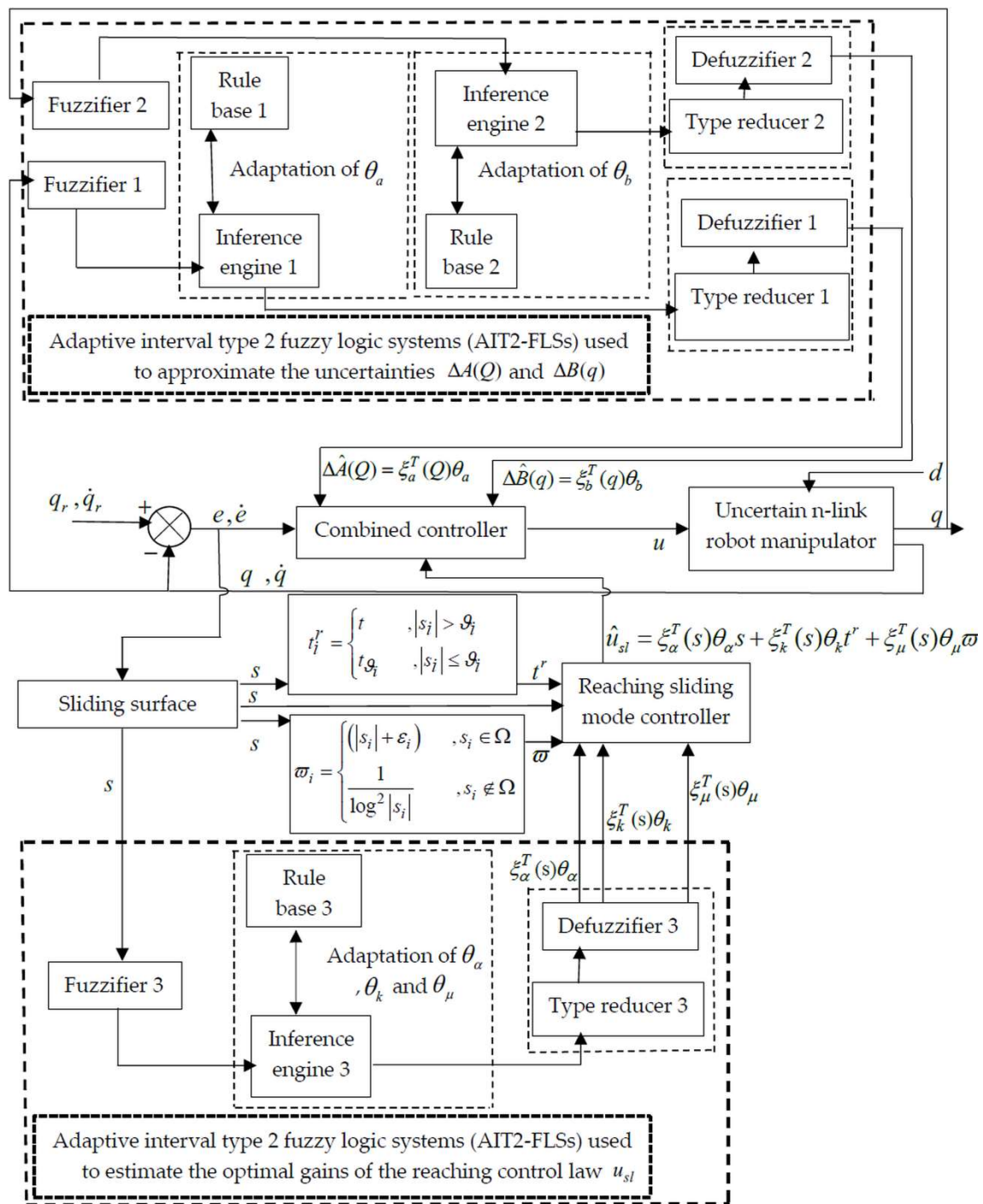


Figure 2. A schematic representation of the proposed control approach.

4. Simulation Results

For simplicity, consider a 2-link robot manipulator, as shown in Figure 3, to validate the developed approach of control.

Let $l_1 = l_2 = 0.5$ m be arm lengths, $m_1 = 2$ kg and $m_2 = 1$ kg the masses at the end of each joint axe, $g = 9.8(\text{m/s}^2)$ the gravity acceleration, and $q = [q_1 \ q_2]^T$ the joint angular position vector.

The robot manipulator is described by the following equation:

$$\ddot{q} = A_0(Q) + B_0(q)u + D \quad (40)$$

where $A_0(Q) = -J^{-1}(q)(C(Q)\dot{q} + G(q))$, $B_0(q) = J^{-1}(q)$ and $D = J^{-1}(q)(d - F(\dot{q}))$ with, the inertia matrix $J(q) = \begin{bmatrix} (m_1 + m_2)l_1^2 & m_2l_1l_2(\sin(q_1)\sin(q_2) + (\cos(q_1)\cos(q_2))) \\ m_2l_1l_2(\sin(q_1)\sin(q_2) + (\cos(q_1)\cos(q_2))) & m_2l_2^2 \end{bmatrix}$, the centripetal Coriolis matrix $C = m_2l_1l_2(\cos(q_1)\sin(q_2) - \sin(q_1)\cos(q_2)) \begin{bmatrix} 0 & -\dot{q}_2 \\ -\dot{q}_1 & 0 \end{bmatrix}$, the gravity vector $G = \begin{bmatrix} -(m_1 + m_2)l_1g\sin(q_1) & -m_2l_2g\sin(q_2) \end{bmatrix}^T$, the joint torque input vector $u = \begin{bmatrix} u_1 & u_2 \end{bmatrix}^T$, the state vector $Q = \begin{bmatrix} q^T & \dot{q}^T \end{bmatrix}^T \in \mathbb{R}^4$, the disturbance vector $d = \begin{bmatrix} d_1 \\ d_2 \end{bmatrix} \in \mathbb{R}^2$, including un-modeled dynamics, unknown payload dynamics, and other unknown perturbations; and, the unknown friction vector $F(\dot{q}) = \begin{bmatrix} F_1(\dot{q}_1) \\ F_2(\dot{q}_2) \end{bmatrix} \in \mathbb{R}^2$.

Assume that the robot manipulator system (40) presents time-varying uncertainties on the mass of joints, as follows: $dm(\text{kg}) = \begin{bmatrix} dm_1 & dm_2 \end{bmatrix}^T = \begin{bmatrix} 0.2\sin(2t) & 0.4\sin(2t) \end{bmatrix}^T$. Therefore, the dynamic Equation (40) can be rewritten as:

$$\ddot{q} = (A_0(Q) + \Delta A(Q)) + (B_0(q) + \Delta B(q))u + \bar{D} \quad (41)$$

where $A_0(Q)$ and $B_0(q)$ represent the nominal dynamics; $\Delta A(Q)$ and $\Delta B(q)$ represent the parametric variations on the nominal system caused by dm ; and, $\bar{D} = (J(q) + \Delta J(q))^{-1}(d - F(\dot{q}))$.

Un-modeled dynamics, unknown payload dynamics, unknown friction force, parametric variations, and other unknown perturbations are represented as:

$$\bar{D} = (J(q) + \Delta J(q))^{-1} \begin{bmatrix} 0.2\sin(\dot{q}_1) + 0.8\sin(2t) + 0.4q_1 + 0.2\text{sign}(\dot{q}_1) \\ 0.2\sin(\dot{q}_2) + 0.8\sin(2t) + 0.4q_2 + 0.2\text{sign}(\dot{q}_2) \end{bmatrix} \quad (42)$$

where $\Delta J(q)$ represents the parametric variations on the inertia matrix $J(q)$.

Set the initial joint angular position vector $q(\text{rad}) = \begin{bmatrix} 1.2 & 0.4 \end{bmatrix}^T$; the control objective is to maintain the system to track the desired trajectory $q_d = \begin{bmatrix} q_{1d} & q_{2d} \end{bmatrix}^T = \begin{bmatrix} \sin(t) & \cos(t) \end{bmatrix}^T$.

Set the sliding surfaces $s_1 = \dot{e}_1 + \lambda_1 e_1$ and $s_2 = \dot{e}_2 + \lambda_2 e_2$, where $e_1 = q_{1d} - q_1$ and $e_2 = q_{2d} - q_2$ are the tracking errors.

Assume that $q_1, q_2, \dot{q}_1$, and $\dot{q}_2$ belong to $\left[-\frac{\pi}{2} \quad \frac{\pi}{2} \right]$.

The proposed AIT2-FSMC law is designed as

$$u = \begin{bmatrix} u_1 \\ u_2 \end{bmatrix} = (B_0(q) + \Delta \hat{B}(q))^{-1}(\ddot{q}_d - (A_0(Q) + \Delta \hat{A}(Q)) + \lambda \dot{e} - \hat{u}_{sl}) \quad (43)$$

where the AIT2-FLS $\Delta \hat{A}(Q)$ has four inputs $q_1, q_2, \dot{q}_1$, and $\dot{q}_2$, and each of them is defined by three MFs, as represented in Figure 4. The AIT2-FLS $\Delta \hat{B}(q)$ has two inputs q_1 and q_2 , and each of them is defined by three MFs, as depicted in Figure 5. Likewise, for the AIT2-FLS $\hat{u}_{sl} = \hat{u}_\alpha + \hat{u}_k + \hat{u}_\mu$, three MFs are designed for each of its inputs s_1 and s_2 , as depicted in Figure 6.

To show the effectiveness of the proposed approach of control, a comparison was made with the adaptive fuzzy SOST-SMC algorithm (AFSOST-SMC) that uses AT1-FLSs $\Delta \bar{A}(Q)$ and $\Delta \bar{B}(q)$ to approximate $\Delta A(Q)$ and $\Delta B(q)$, and it uses a SOST-SMC law to handle the approximation errors and unknown disturbances.

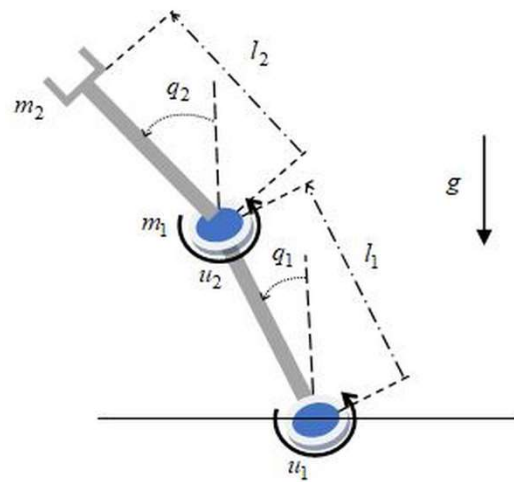


Figure 3. A 2-link robot manipulator.

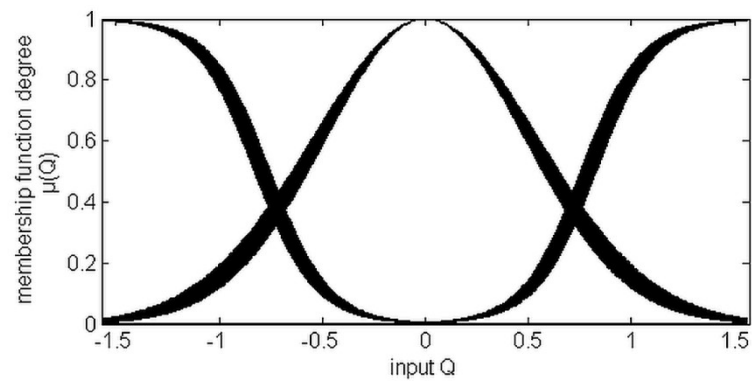


Figure 4. Interval type-2 fuzzy sets used by the AIT2-FLS $\Delta\hat{A}(Q)$.

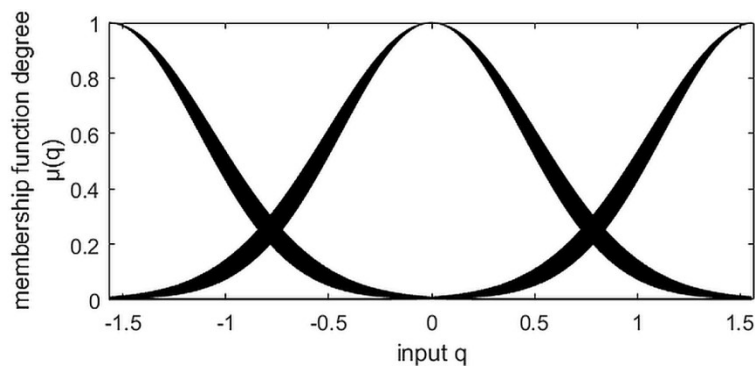


Figure 5. Interval type-2 fuzzy sets used by the AIT2-FLS $\Delta\hat{B}(q)$.

The global control law of the AFSOST-SMC approach is given as:

$$V = \begin{bmatrix} V_1 \\ V_2 \end{bmatrix} = (B_0(q) + \Delta\bar{B}(q))^{-1}(\ddot{q}_d - (A_0(Q) + \Delta\bar{A}(Q)) + \lambda\dot{e} - u_v) \quad (44)$$

where, $u_v = - \begin{bmatrix} \alpha_1 & 0 \\ 0 & \alpha_2 \end{bmatrix} \begin{bmatrix} \int_0^t \text{sign}(s_1) dt \\ \int_0^t \text{sign}(s_2) dt \end{bmatrix} - \begin{bmatrix} \beta_1 & 0 \\ 0 & \beta_2 \end{bmatrix} \begin{bmatrix} |s_1|^{0.5} \\ |s_2|^{0.5} \end{bmatrix}$; $\alpha_1, \alpha_2, \beta_1, \beta_2$ are the gains of the control law u_v .

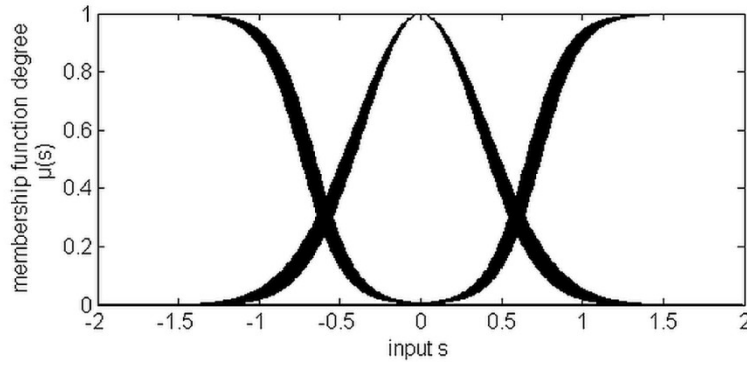


Figure 6. Interval type-2 fuzzy sets used by the AIT2-FLS $\hat{u}_{sl}$.

The T1-FSs used by $\Delta\bar{A}(Q)$ and $\Delta\bar{B}(q)$ to approximate $\Delta A(Q)$ and $\Delta B(q)$, are depicted in Figures 7 and 8, respectively:

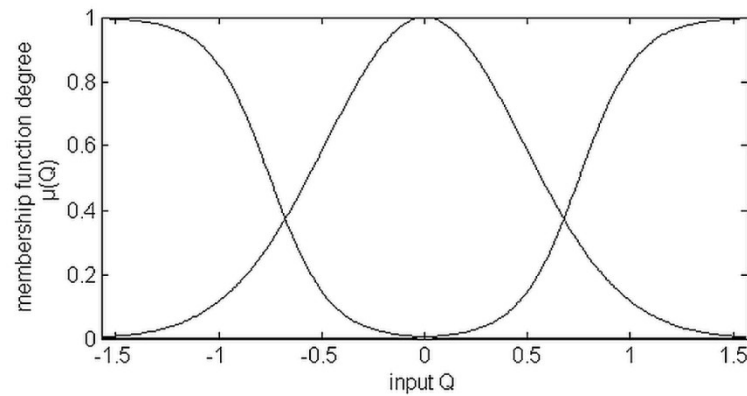


Figure 7. Type-1 fuzzy sets used by the fuzzy system $\Delta\bar{A}(Q)$.

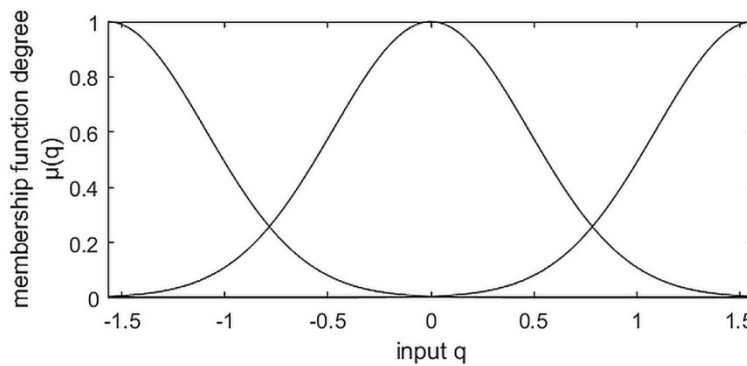


Figure 8. Type-1 fuzzy sets used by the fuzzy system $\Delta\bar{B}(q)$.

For the constant parameters of the two approaches of control, we take the following values, as shown in the Table 1 below:

Table 1. Constant parameters of both control approaches.

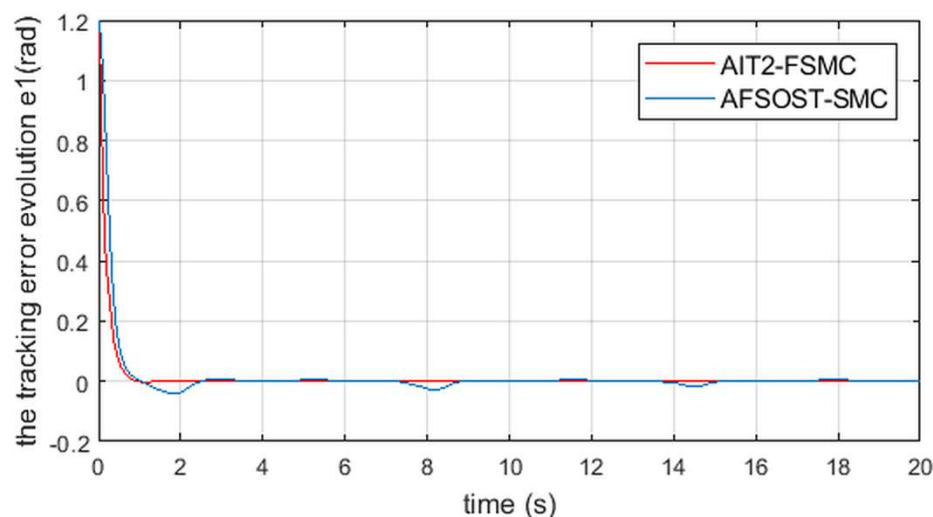
Parameters	AIT2-FSMC	AFSOST-SMC
λ_1	6	6
λ_2	14	14
α_1	-	5
α_2	-	2
β_1	-	10
β_2	-	12
γ_a	800	20
γ_b	0.001	0.001
γ_α	110	-
γ_k	880	-
γ_μ	24	-
N_1	0.1	-
N_2	0.1	-

The simulation results are depicted in Figures 9–18. They illustrate the comparison between the two control methods, namely the proposed AIT2-FSMC and AFSOST-SMC.

Figures 9 and 10, they show the evolution of the tracking errors. Figures 11 and 12, they depict the robot manipulator angular positions q_1 and q_2 trajectories, and their desired references q_{d1} and q_{d2} , respectively. Figures 13 and 14 they represent the control laws of both control approaches.

The comparison between the AIT2-FSMC and AFSOST-SMC methods shows that the AIT2-FSMC provides better tracking control performance with a smooth control law. This is thanks to the fact that the AIT2-FSMC, firstly, it provides better approximations of the uncertainties $\Delta A(Q)$ and $\Delta B(q)$, and secondly, it rejects the effect of un-modeled dynamics, approximation errors, and other unknown disturbances more efficiently than the AFSOST-SMC.

Figures 15 and 16 below show that the tracking accuracy of the AFSOST-SMC approach is improved when we increase the gains α_1 , α_2 , β_1 , and β_2 of the control law u_v . However, this implies control inputs with chattering, as shown in Figures 17 and 18. Even with this improvement in accuracy, which generates the chattering in the AFSOST-SMC method, it is concluded that the AIT2-FSMC approach still shows a better tracking accuracy with smooth control inputs.

**Figure 9.** The tracking error $e_1(rad)$ of both control approaches.

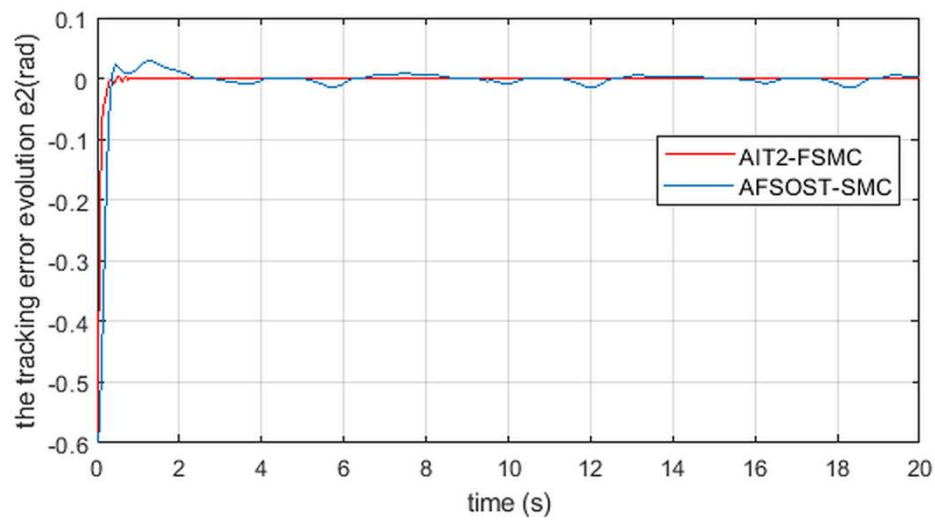


Figure 10. The tracking error $e_2(rad)$ of both control approaches.

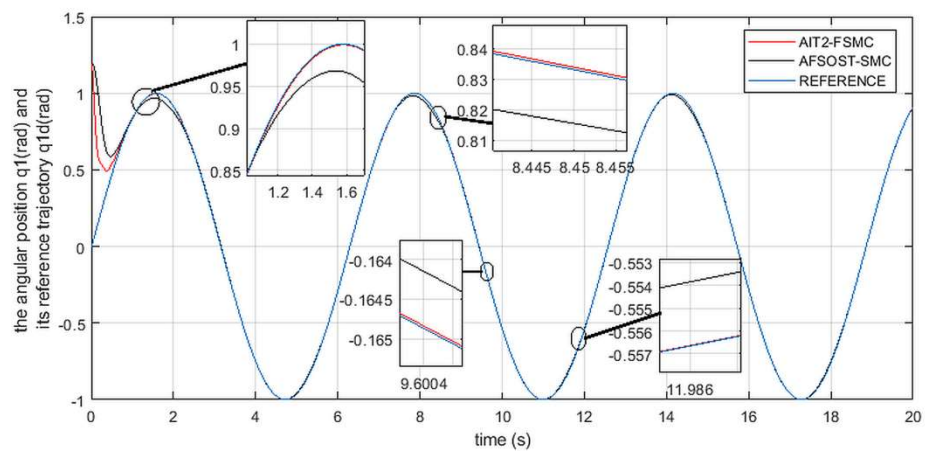


Figure 11. The angular position $q_1(rad)$ of both control approaches, and its reference trajectory $q_{1d}(rad)$.

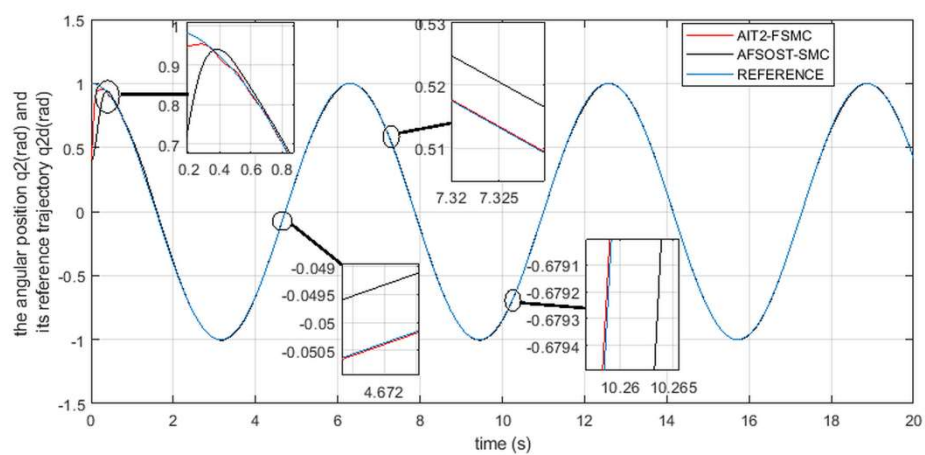


Figure 12. The angular position $q_2(rad)$ of both control approaches, and its reference trajectory $q_{2d}(rad)$.

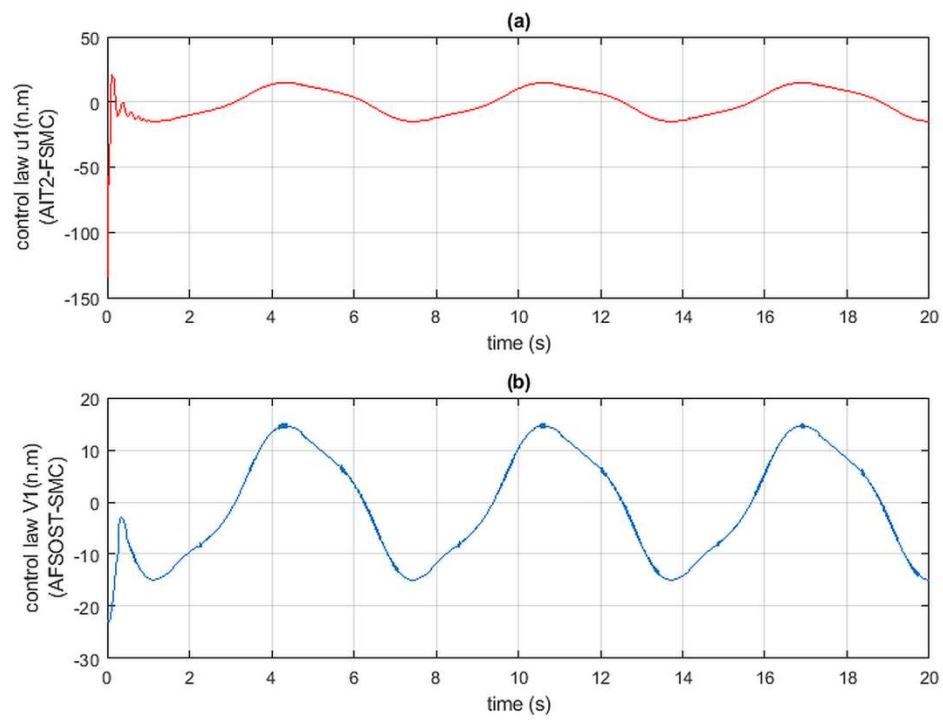


Figure 13. (a) The control law $u_1(n.m)$ of the control approach AIT2-FSMC; (b) The control law $V_1(n.m)$ of the control approach AFSOST-SMC.

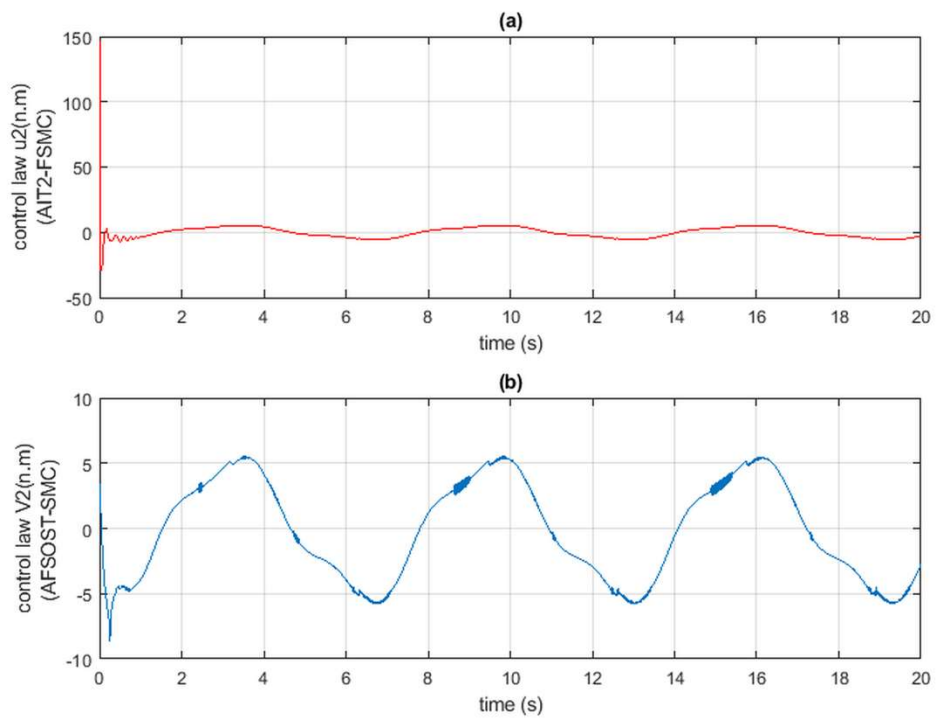


Figure 14. (a) The control law $u_2(n.m)$ of the control approach AIT2-FSMC; (b) The control law $V_2(n.m)$ of the control approach AFSOST-SMC.

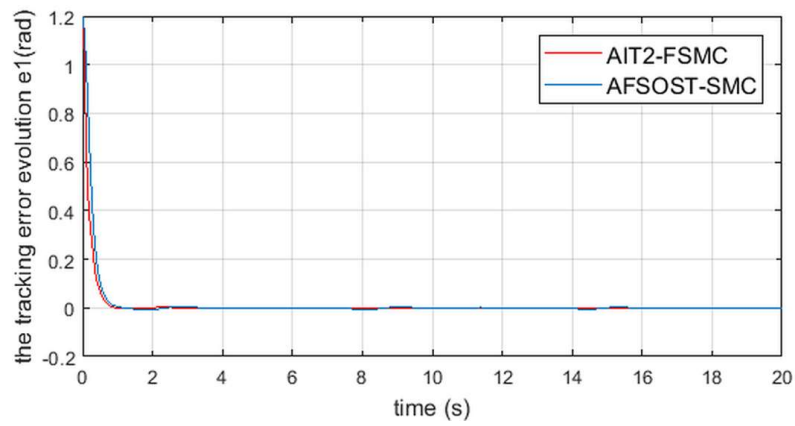


Figure 15. The tracking error $e_1(rad)$ of both control approaches, for $\alpha_1 = 7, \alpha_2 = 6, \beta_1 = 14, \beta_2 = 17$.

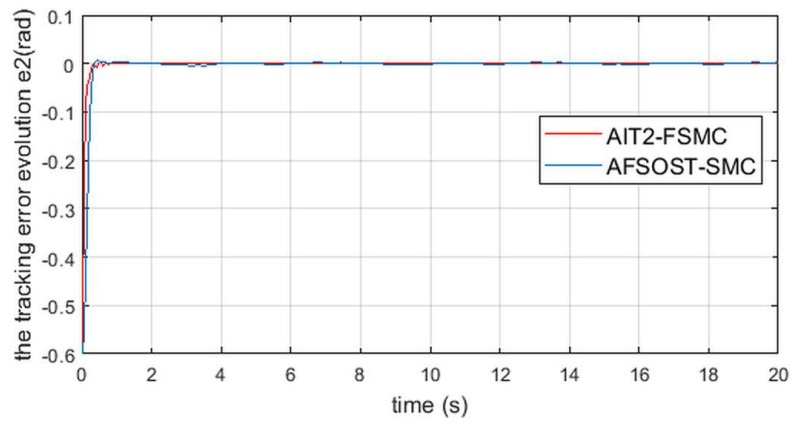


Figure 16. The tracking error $e_2(rad)$ of both control approaches, for $\alpha_1 = 7, \alpha_2 = 6, \beta_1 = 14, \beta_2 = 17$.

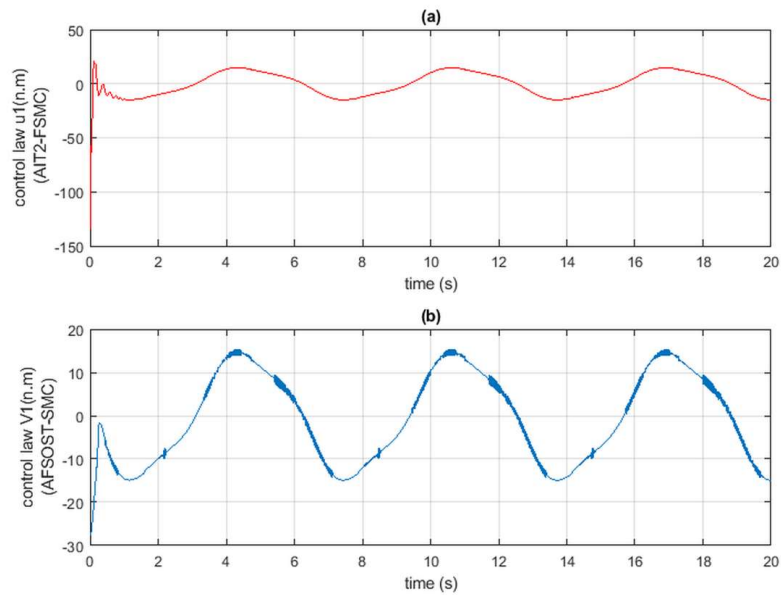


Figure 17. (a) The control law $u_1(n.m)$ of the control approach AIT2-FSMC; (b) The control law $V_1(n.m)$ of the control approach AFSOST-SMC, for $\alpha_1 = 7, \alpha_2 = 6, \beta_1 = 14, \beta_2 = 17$.

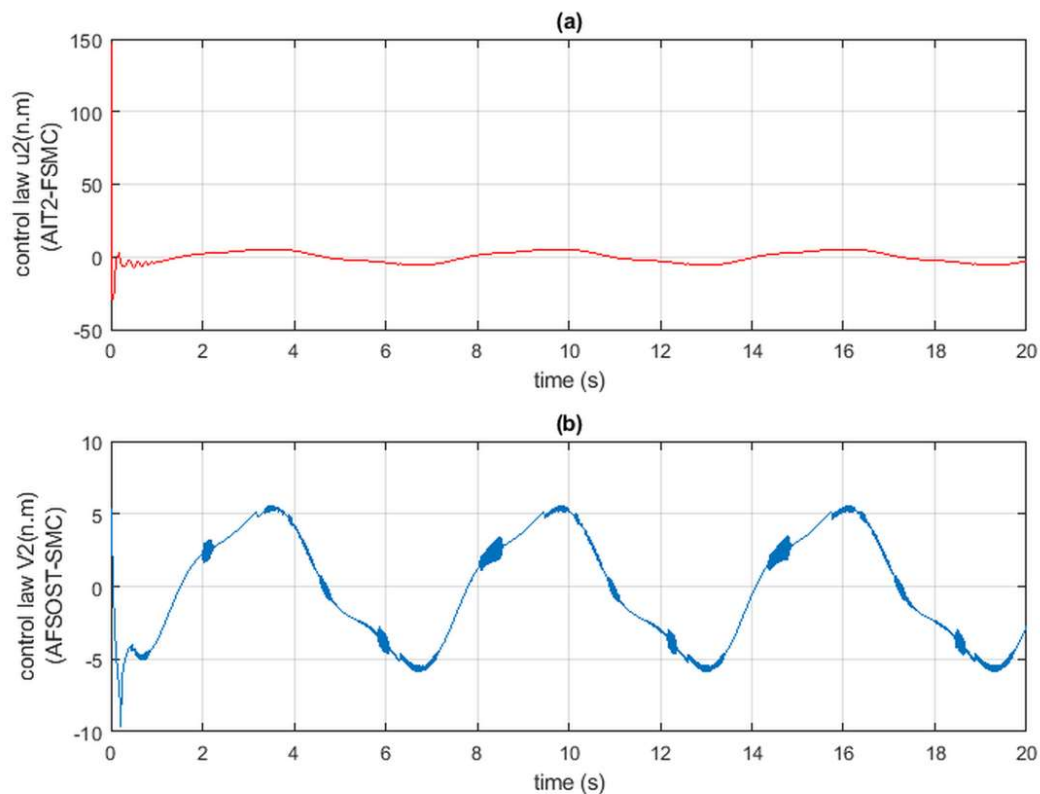


Figure 18. (a) The control law $u_2(n.m)$ of the control approach AIT2-FSMC; (b) The control law $V_2(n.m)$ of the control approach AFSOST-SMC, for $\alpha_1 = 7, \alpha_2 = 6, \beta_1 = 14, \beta_2 = 17$.

5. Conclusions

In this paper, we presented a new enhanced tracking control design for n-link robot manipulators in the presence of un-modelled dynamics, unknown payload dynamics, unknown friction force, parametric variations, and other unknown perturbations. Firstly, two AIT2-FLSs are designed to better approximate the parametric uncertainties, then secondly, a new control algorithm, which uses a new designed AIT2-FSMC law, is introduced in order to handle approximation errors and unknown disturbances that affect the robot manipulator systems. In order to overcome the chattering without deteriorating the system robustness, the AIT2-FSMC generates three adaptive control laws to guarantee the best estimation of the optimal smooth control law that ensures the best tracking control performance, despite the uncertainties and disturbances. The closed loop control system is globally asymptotically stable and mathematically proven. The simulation example confirms the effectiveness of the developed control approach in achieving the desired objectives. In the future, we intend to extend the study to cover a wide range of nonlinear systems, such as underactuated nonlinear systems and non affine nonlinear systems.

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