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Tensor Product Alternatives for Nonlinear Field-Oriented Control of Induction Machines

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Abstract: The paper presents a nonlinear field-oriented control technique based on the tensor product representation of the nonlinear induction machine model and the solvability of linear matrix inequalities. The nonlinear model has 32 quasi linear parameter-varying equivalent variants, and it is shown that only half of the models result in feasible controller. Two control goals are realized: torque control and speed control. The controller is a nonlinear state feedback controller completed by integral action. A new block diagram is investigated for speed control. The controller gains are designed by the solution of linear matrix inequalities to solve the Lyapunov inequality to obtain a stable and fast response and constraints on the control signal. The presented methods are verified and compared by simulations.

Keywords: induction machine model; field-oriented control; nonlinear control; tensor product model; state feedback; linear matrix inequality



Citation: Kuczmán, M.; Horváth, K. Tensor Product Alternatives for Nonlinear Field-Oriented Control of Induction Machines. *Electronics* **2024**, *13*, 1405. <https://doi.org/10.3390/electronics13071405>

Academic Editor: Pedro J. Villegas

Received: 6 March 2024

Revised: 5 April 2024

Accepted: 6 April 2024

Published: 8 April 2024



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1. Introduction

Induction machines with a squirrel-cage rotor type have a simple and robust structure due to the lack of a commutator and brushes. In addition, these machines do not contain permanent magnets, resulting in a relatively low price and increased reliability since the demagnetization problem does not occur. These features make induction machines the most common in industrial drives [1], and also an attractive solution for automotive electric propulsion systems [2–4]. However, the performance of induction machine drives depends significantly on the control method used.

The earliest control approaches for induction machines are the so-called scalar control methods. By using them, the frequency and magnitude of the sinusoidal excitation voltages or currents can be adjusted. This allows the steady-state speed or torque to be controlled while keeping flux at the desired level. Since scalar control methods are based on steady-state equations of the machine, the control algorithms are simple and require low computational effort. In addition, an adequate steady-state characteristic can be achieved by using them. However, the scalar control has no effect on transients, so only weak or moderate dynamics are provided. Nevertheless, scalar control is widely used in low-cost and low-performance industrial applications such as pumps, compressors and fans [5,6].

Unlike scalar control, vector control techniques affect not only the frequency and magnitude of stator voltages or currents but also the direction of the space vectors since vector control methods are based on the transient description of the machine. As a result, much better dynamics can be achieved than with scalar control. The first vector control method is the well-known field-oriented control (FOC) introduced in [7,8]. The FOC principle is the indirect control of flux and torque through current controls, similar to the control of a separately excited DC machine. To separate the flux- and torque-producing components of the stator current vector, the coordinate system of the magnetic field is used for FOC. Originally, the rotor flux vector was used for field orientation, and it is still the most widespread. As alternatives to rotor flux orientation, stator flux [9] or air gap

flux [10] reference frames can also be applied. However, these are rarely used in practical applications. Note that these traditional FOCs use linear control design, although induction machine models are nonlinear in nature. To compensate for the effects of nonlinearity, an additional decoupling algorithm is required, and the control performance depends on how accurately this algorithm works [11,12]. In contrast to scalar control, FOC provides better dynamics and very good properties at both low and high speeds [13]. It is also particularly important that the torque ripple is low over the entire speed range. The main drawbacks of FOC are its relatively high computational burden due to the coordinate transformations and its sensitivity to parameter mismatches as discussed in [1].

The second group of vector control includes the direct torque control (DTC) methods, which were introduced parallel to each other in [14,15]. The basic principles used in [14,15] are very similar, but the stator flux vector follows a circular trajectory in [14]; it is the so-called classical DTC, while direct self control (DSC) in [15] operates with a hexagonal stator flux vector trajectory. Both DTC methods are based on the effect of the possible switching states of the inverter on the behavior of the machine. Instead of using inner current control loops, DTCs directly control the stator flux and torque. To determine flux and torque feedback signals without direct measurement, estimators are applied in general. The main advantage of the conventional DTC and DSC methods is the extremely fast transient operation. In addition, these techniques do not require the information of the rotor angular velocity for torque control. In other words, the traditional DTCs are inherently rotary sensorless methods, unlike FOC. Nevertheless, both FOC and DTC require angular velocity feedback for speed control. To replace the rotary sensors, fundamental frequency model-based estimators [16–19] and signal injection-based estimators [20–22] are also widely used. Due to the absence of coordinate transformation, pulse width modulation and inner current control loops, the conventional DTCs have a very simple structure and lower computational time than the FOC as presented in [23]. Furthermore, DTCs are less sensitive to parameter variations than FOC techniques. However, the original DTC methods are mostly not applicable in practice due to the extremely high current and torque ripples. To reduce current and torque ripples, several improvements have been developed for DTC, which are reviewed in [24,25]. But solving this problem increases the complexity of the control algorithm as discussed in [12].

In commercial drives, scalar control, FOC and improved DTC methods are the most popular due to their simplicity and easy tuning. However, more sophisticated control methods can be found in the literature. Since the difficulty of control design for induction machines is caused by the nonlinear description, nonlinear control theory offers attractive solutions. Among these, feedback linearization control (FLC) is used for induction machine drives in [26–30]. By using FLC, nonlinear transformation can be defined to represent the nonlinear induction machine model by independent linear subsystems. For these linear subsystems, advanced state feedback controllers, such as optimal or robust controllers, can be easily synthesized. It is important to highlight a theoretical similarity between FOC and FLC, namely that both control strategies use transformations to achieve a favorable mathematical description for control design. However, induction machine models in different coordinate systems can also be used for FLC. For example, Refs. [26–28] use stationary reference frame models, while rotor flux-oriented models are applied in [29,30]. The most significant disadvantages of FLC are the complex design procedure and the relative high computational burden.

A further promising approach in control design for nonlinear systems is the tensor product (TP) transformation [31–34]. TP model transformation is a modern framework to numerically approximate the linear time invariant (LTI), linear parameter varying (LPV) and quasi LPV (qLPV) representations of state-space models. The framework can replace the complicated derivations of the closed formulae of the state-space models with straightforward and numerically appealing solutions. Furthermore, it should be irrelevant how the state-space model is given (e.g., a set of closed formulae and a black box model). All in all, the nonlinear model is represented by the parameter-varying combination of LTI models;

moreover, the TP model transformation was introduced as the higher-order singular value decomposition (HOSVD) of LPV models.

The TP model representation belongs to the class of polytopic models. The polytopic model-based state-space control design has three key steps:

1. Defining the state-space model;
2. Finding the optimal polytopic model (TP model in this paper);
3. Deriving the controller.

The crucial point is that the model structure of steps 1 and 2 has a key role in determining the effectiveness of the controller design in step 3. The system matrix elements influence the controller design process in a strong sense. The convex hull defined by the polytopic structure in step 2 also directly influences the design. The system matrix elements can be obtained in a systematic way to check all the possible model representations. Controller design is usually carried out by the appropriate linear matrix inequalities (LMIs) based on the Lyapunov stability condition and constraint on the control input and on the control value [35].

Papers [36–38] deal with the fuzzy-state feedback controller design of induction machine with optimal performance. The Takagi–Sugeno fuzzy model is employed to approximate the nonlinear machine model in the FOC representation. A fuzzy controller is designed to stabilize the machine, and minimum disturbance attenuation is guaranteed via the $\mathcal{H}_\infty$ -controller design. The LMI method is used to find controller gains.

Paper [39] presents a FOC-based technique; however, a nonlinear controller is derived using TP model transformation methodology. The classical FOC method and TP modeling are merged to control permanent magnet synchronous motors. Speed control is designed by the TP model approach, but the d-component of the current is controlled by a PI-controller. Simulation results show better performance in terms of overshoot, speed and disturbance rejection when compared to the results of conventional cascade PI + PI control solution.

The paper [40] proposes the design of an induction machine controller based on the TP method, and the sum of square (SOS) method is used to set up performance criterion constraints in polynomial form instead of the LMIs. Reference signal tracking is realized efficiently.

The induction machine nonlinear TP-based controller is complemented by an integral action and observer in [41,42]. The designed controller ensures stable and accurate operation over the full operating range of the machine, taking into account the wide range of temperature values, and possible variations in inductances, which are outside the range of the parameters under consideration.

In summary, the main challenge in the control of induction machines is caused by the nonlinear mathematical description. To solve the problem of nonlinear models, different solutions are used in FOC, DTC, FLC and the above discussed TP-based control methods. Among them, the FOC is the most widespread, which uses simple linear controllers and compensates for nonlinear effects by decoupling. Therefore, the performance depends on the decoupling algorithm. The conventional DTCs use nonlinear hysteresis controllers, which result in high current and torque ripples. Although FLC avoids decoupling, as well as high current and torque ripples, it requires complex formulae to be defined. In contrast, the TP model transformation-based control is numerical and easy to algorithmize. Therefore, high performance and easy to implement control methods can be achieved by combining FOC and TP transformation, similarly to [40–42]. However, the idea of [43] shows that a TP model has a huge number of alternatives. These variants can be readily derived by the TP model transformation that can vary the number of fuzzy rules, the number of antecedent and consequent sets, and, further, the shape of the antecedent fuzzy sets. The goal of this paper is to analyze the behavior of model alternatives from the controller design viewpoint. The FOC-based state-space model is used in the frame of TP modeling and LMI-based control synthesis for torque control and speed control.

Although this paper focuses on the control of induction machines, the proposed methodology can be applied in general to other problems of which the TP model representation can be used for control design.

2. Materials and Methods

2.1. Tensor Product Model Transformation

The most important results of the TP model transformation will be summarized shortly based on the papers [31–34].

Consider the linear parameter-varying state-space model

$$\begin{aligned}\dot{\mathbf{x}}(t) &= \mathbf{A}(\mathbf{p}(t))\mathbf{x}(t) + \mathbf{B}(\mathbf{p}(t))\mathbf{u}(t), \\ \mathbf{y}(t) &= \mathbf{C}(\mathbf{p}(t))\mathbf{x}(t) + \mathbf{D}(\mathbf{p}(t))\mathbf{u}(t),\end{aligned}\quad (1)$$

where $\mathbf{x} = \mathbf{x}(t) \in \mathbb{R}^m$, $\mathbf{u} = \mathbf{u}(t) \in \mathbb{R}^k$ and $\mathbf{y} = \mathbf{y}(t) \in \mathbb{R}^l$ are the state vector, the input vector, and the output vector with the dimension of m , k , and l , respectively. Matrices $\mathbf{A}$, $\mathbf{B}$, $\mathbf{C}$, and $\mathbf{D}$ depend on the parameter vector $\mathbf{p} = \mathbf{p}(t) \in \Omega$. The parameter vector consists of N parameters, depending on the model description, $\mathbf{p} = [p_1 \ p_2 \ \dots \ p_N]^T$. The set Ω is a hyper space, i.e., $\Omega = \omega_1 \times \omega_2 \times \dots \times \omega_N$, where ω_i is an interval of the according parameter p_i , i.e., $\omega_i = [\omega_i^{\min}, \omega_i^{\max}]$. The applied hyper space Ω as well as the selected parameter vector $\mathbf{p}$ are discussed later in Sections 2.2 and 2.4.

The system matrix is defined by

$$\mathbf{S} = \mathbf{S}(\mathbf{p}) = \begin{bmatrix} \mathbf{A}(\mathbf{p}(t)) & \mathbf{B}(\mathbf{p}(t)) \\ \mathbf{C}(\mathbf{p}(t)) & \mathbf{D}(\mathbf{p}(t)) \end{bmatrix} \in \mathbb{R}^{(m+l) \times (m+k)}. \quad (2)$$

The main idea of TP model transformation is to discretize the given model state-space representation over a hyper rectangular grid M in Ω , then the tensor product structure of the model is obtained via executing higher-order singular value decomposition. By ignoring singular values, the TP model of reduced complexity and accuracy can be obtained.

The tensor product structure can be written as follows:

$$\mathbf{S} = \mathcal{S} \boxtimes_{n=1}^N \mathbf{w}_n(p_n) = \sum_{i_1=1}^{I_1} \sum_{i_2=1}^{I_2} \dots \sum_{i_N=1}^{I_N} \prod_{n=1}^N w_{n,i_n}(p_n) S_{i_1, i_2, \dots, i_N}, \quad (3)$$

where $\mathcal{S} \in \mathbb{R}^{I_1 \times I_2 \times \dots \times I_N \times (m+l) \times (m+k)}$ is the core tensor, the vector $\mathbf{w}_n$ contains the weighting functions $w_{n,i_n}(p_n)$, I_n denotes the number of LTI systems in the n th dimension of the domain Ω , and $S_{i_1, i_2, \dots, i_N}$ is a LTI system. The symbol $\boxtimes$ denotes multiple n -mode products of a tensor by a matrix.

The transformation steps and the corresponding controller design scheme are detailed below.

2.2. Induction Machine Model

The nonlinear ordinary differential equations of a set of nonlinear systems have the following general formula [44]:

$$\dot{\mathbf{x}} = \mathbf{f}(\mathbf{x}) + \mathbf{g}(\mathbf{x})\mathbf{u}, \quad (4)$$

where $\mathbf{x}$ is the state variable vector, the nonlinear mapping $\mathbf{f}(\mathbf{x})$ has the special form of $\mathbf{f}(\mathbf{x}) = \mathbf{A}(\mathbf{x})\mathbf{x}$, and $\mathbf{g}(\mathbf{x})$ is constant, denoted by the matrix $\mathbf{B}$, i.e., systems with the differential equations $\dot{\mathbf{x}} = \mathbf{A}(\mathbf{x})\mathbf{x} + \mathbf{B}\mathbf{u}$ are studied in this paper, meaning that the parameter vector $\mathbf{p}$ can be equal to the state vector $\mathbf{p} = \mathbf{x}$ or a part of the state vector—furthermore, a function of state vector elements. The system input is denoted by $\mathbf{u}$.

The nonlinear mapping and the constant matrix of an induction machine can be written in the rotor flux-oriented model as [45–48]

$$\mathbf{f}(\mathbf{x}) = \begin{bmatrix} -\frac{R_s L_r^2 + R_r L_m^2}{\sigma L_s L_r^2} i_{sd} + \frac{R_r L_m}{\sigma L_s L_r^2} \psi_{rd} + p \omega_m i_{sq} + \frac{R_r L_m}{L_r} \frac{i_{sq}^2}{\psi_{rd}} \\ -\frac{R_s L_r^2 + R_r L_m^2}{\sigma L_s L_r^2} i_{sq} - \frac{p L_m}{\sigma L_s L_r} \omega_m \psi_{rd} - p \omega_m i_{sd} - \frac{R_r L_m}{L_r} \frac{i_{sd} i_{sq}}{\psi_{rd}} \\ \frac{R_r L_m}{L_r} i_{sd} - \frac{R_r}{L_r} \psi_{rd} \\ \frac{3}{2} \frac{p}{J} \frac{L_m}{L_r} i_{sq} \psi_{rd} - \frac{D_f}{J} \omega_m - \frac{1}{J} T_L \end{bmatrix}, \quad (5)$$

and

$$\mathbf{g}(\mathbf{x}) = \mathbf{B} = \begin{bmatrix} \frac{1}{\sigma L_s} & 0 \\ 0 & \frac{1}{\sigma L_s} \\ 0 & 0 \\ 0 & 0 \end{bmatrix}. \quad (6)$$

The resistance and self-induction coefficient of the three windings of the stator and of the rotor are denoted by R_s , L_s , R_r and L_r ; furthermore, L_m is the mutual inductance between the stator and rotor coils, and σ is a parameter depending on the the inductances $\sigma = 1 - \frac{L_m^2}{L_s L_r}$. The pole pair number is denoted by p , and J and D_f are the moment of inertia and the viscous friction coefficient. The state variables are the d and q components of the stator current, denoted by i_{sd} and i_{sq} , the d component is of the rotor flux ψ_{rd} , and ω_m is the mechanical speed of the rotor, i.e.,

$$\mathbf{x} = [i_{sd} \quad i_{sq} \quad \psi_{rd} \quad \omega_m]^T. \quad (7)$$

The load torque T_L is a disturbance model input. Finally, the motor torque is obtained by

$$T = \frac{3}{2} p \frac{L_m}{L_r} i_{sq} \psi_{rd}. \quad (8)$$

Dividing by ψ_{rd} is replaced by variable p_5 , i.e., $p_5 = 1/\psi_{rd}$ is a new model parameter (element of the parameter vector $\mathbf{p}$), which, by dividing by zero, can be handled.

Varying the state vector elements, the matrix $\mathbf{A} = \mathbf{A}(\mathbf{p})$ of the model (5) can be rewritten as the following general form:

$$\mathbf{A} = \begin{bmatrix} -\frac{R_s L_r^2 + R_r L_m^2}{\sigma L_s L_r^2} & \frac{R_r L_m}{L_r} i_{sq} p_5 + p \omega_m A & \cdots \\ -p \omega_m B - \frac{R_r L_m}{L_r} i_{sq} p_5 C & -\frac{R_s L_r^2 + R_r L_m^2}{\sigma L_s L_r^2} - \frac{R_r L_m}{L_r} i_{sd} p_5 (1 - C) & \cdots \\ \frac{R_r L_m}{L_r} & 0 & \cdots \\ 0 & \frac{3}{2} \frac{p}{J} \frac{L_m}{L_r} \psi_{rd} E & \cdots \\ \cdots & \frac{R_r L_m}{\sigma L_s L_r^2} p i_{sq} (1 - A) & \cdots \\ \cdots & -\frac{p L_m}{\sigma L_s L_r} \omega_m D & -p i_{sd} (1 - B) - \frac{p L_m}{\sigma L_s L_r} \psi_{rd} (1 - D) \\ \cdots & -\frac{R_r}{L_r} & 0 \\ \cdots & \frac{3}{2} \frac{p}{J} \frac{L_m}{L_r} i_{sq} (1 - E) & -\frac{D_f}{J} \end{bmatrix}. \quad (9)$$

With the five new parameters A , B , C , D and E , all the possible 32 qLPV models can be analyzed from the viewpoint of modeling and controller design. The new parameters can be selected from the discrete set of $\{0, 1\}$. Table 1 depicts all the 32 qLPV models, where the dependency of the matrix $\mathbf{A}$ on the parameters can also be seen (for example, $\mathbf{p} = [i_{sd} \quad i_{sq} \quad \psi_{rd} \quad p_5]^T$ for model #1, or $\mathbf{p} = [i_{sq} \quad \omega_m \quad p_5]^T$ for model #15, i.e., the parameter vector contains state variables, functions of state variables, or other relevant parameters).

The following output matrices $\mathbf{C}_i$ ($i = 0, 1, 2, 3$) are chosen in the analysis to define the output vector $\mathbf{y} = \mathbf{C}_i \mathbf{x} = [y_1 \ y_2]^T$. The index i represents the different output vectors as follows.

The classical field-oriented control scheme can be given by

$$\mathbf{C}_0 = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \end{bmatrix}, \quad (10)$$

i.e., $y_1 = i_{sd}$ and $y_2 = i_{sq}$. The reference currents along the d and q axes can be obtained from the reference flux and torque as it will be presented later.

Table 1. The 32 qLPV models of the studied induction machine.

#	E	D	C	B	A	i_{sd}	i_{sq}	ψ_{rd}	ω_m	p_5	R
0	0	0	0	0	0	x ^a	x	x	-	x	16
1	0	0	0	0	1	x	x	x	x	x	32
2	0	0	0	1	0	x	x	x	x	x	32
3	0	0	0	1	1	x	x	x	x	x	32
4	0	0	1	0	0	x	x	x	-	x	16
5	0	0	1	0	1	x	x	x	x	x	32
6	0	0	1	1	0	-	x	x	x	x	16
7	0	0	1	1	1	-	x	x	x	x	16
8	0	1	0	0	0	x	x	-	x	x	16
9	0	1	0	0	1	x	x	-	x	x	16
10	0	1	0	1	0	x	x	-	x	x	16
11	0	1	0	1	1	x	x	-	x	x	16
12	0	1	1	0	0	x	x	-	x	x	16
13	0	1	1	0	1	x	x	-	x	x	16
14	0	1	1	1	0	-	x	-	x	x	8
15	0	1	1	1	1	-	x	-	x	x	8
16	1	0	0	0	0	x	x	x	-	x	16
17	1	0	0	0	1	x	-	x	x	x	16
18	1	0	0	1	0	x	x	x	x	x	32
19	1	0	0	1	1	x	-	x	x	x	16
20	1	0	1	0	0	x	x	x	-	x	16
21	1	0	1	0	1	x	x	x	x	x	32
22	1	0	1	1	0	-	x	x	x	x	16
23	1	0	1	1	1	-	x	x	x	x	16
24	1	1	0	0	0	x	x	x	x	x	32
25	1	1	0	0	1	x	-	x	x	x	16
26	1	1	0	1	0	x	x	x	x	x	32
27	1	1	0	1	1	x	-	x	x	x	16
28	1	1	1	0	0	x	x	x	x	x	32
29	1	1	1	0	1	x	x	x	x	x	32
30	1	1	1	1	0	-	x	x	x	x	16
31	1	1	1	1	1	-	x	x	x	x	16

^a Notation x means that the variable appears in the system model.

The reference flux and torque can be directly controlled by the output matrices

$$\mathbf{C}_1 = \begin{bmatrix} 0 & 0 & 1 & 0 \\ 0 & \frac{3}{2}p\frac{L_m}{L_r}\psi_{rd} & 0 & 0 \end{bmatrix}, \quad (11)$$

and

$$\mathbf{C}_2 = \begin{bmatrix} 0 & 0 & 1 & 0 \\ 0 & 0 & \frac{3}{2}p\frac{L_m}{L_r}i_{sq} & 0 \end{bmatrix}, \quad (12)$$

i.e., $y_1 = \psi_{rd}$ and $y_2 = T = \frac{3}{2}p\frac{L_m}{L_r}i_{sq}\psi_{rd}$.

The reference flux and speed can be directly controlled by the output matrix of

$$\mathbf{C}_3 = \begin{bmatrix} 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix}, \quad (13)$$

i.e., $y_1 = \psi_{rd}$ and $y_2 = \omega_m$.

The matrix $\mathbf{D}$ is zero.

All in all, $32 \times 4 = 128$ models are set up and analyzed.

2.3. Model Parameters

The asynchronous motor model nominal parameters are chosen as follows, where their physical meaning is also given [49]:

- Number of pole pairs, $p = 2$;
- Stator resistance, $R_s = 4.7 \, \Omega$;
- Rotor resistance, $R_r = 5.2 \, \Omega$;
- Stator self inductance, $L_s = 0.1788 \, \text{H}$;
- Rotor self inductance, $L_r = 0.1790 \, \text{H}$;
- Mutual inductance, $L_m = 0.1690 \, \text{H}$;
- Moment of inertia, $J = 0.00108 \, \text{kg m}^2$;
- Viscous friction coefficient, $D_f = 0.00475 \, \text{Nms}$.

and $\sigma = 1 - \frac{L_m^2}{L_s L_r}$ is the leakage coefficient.

2.4. Model Discretization Parameters

The following parameters are set when the model matrices are discretized in the grid M . Here, the problem region Ω as well as the region limits are listed:

- The problem region for models 0, 4, 16, and 20 is

$$\Omega = [i_{sd}^{\min}, i_{sd}^{\max}] \times [i_{sq}^{\min}, i_{sq}^{\max}] \times [\psi_{rd}^{\min}, \psi_{rd}^{\max}] \times [p_5^{\min}, p_5^{\max}];$$

- The problem region for models 8, 9, 10, 11, 12, and 13 is

$$\Omega = [i_{sd}^{\min}, i_{sd}^{\max}] \times [i_{sq}^{\min}, i_{sq}^{\max}] \times [\omega^{\min}, \omega^{\max}] \times [p_5^{\min}, p_5^{\max}];$$

- The problem region for models 1, 2, 3, 5, 17, 18, 19, 21, 24, 25, 26, 27, 28, and 29 is

$$\Omega = [i_{sd}^{\min}, i_{sd}^{\max}] \times [i_{sq}^{\min}, i_{sq}^{\max}] \times [\psi_{rd}^{\min}, \psi_{rd}^{\max}] \times [\omega^{\min}, \omega^{\max}] \times [p_5^{\min}, p_5^{\max}];$$

- The problem region for models 6, 7, 22, 23, 30, and 31 is

$$\Omega = [i_{sq}^{\min}, i_{sq}^{\max}] \times [\psi_{rd}^{\min}, \psi_{rd}^{\max}] \times [\omega^{\min}, \omega^{\max}] \times [p_5^{\min}, p_5^{\max}];$$

- The problem region for models 14 and 15 is

$$\Omega = [i_{sq}^{\min}, i_{sq}^{\max}] \times [\omega^{\min}, \omega^{\max}] \times [p_5^{\min}, p_5^{\max}].$$

The limit values are set to

- $i_{sd}^{\min} = -10 \, \text{A}$, $i_{sd}^{\max} = 10 \, \text{A}$;
- $i_{sq}^{\min} = -10 \, \text{A}$, $i_{sq}^{\max} = 10 \, \text{A}$;
- $\psi_{rd}^{\min} = 0.1 \, \text{mVs}$, $\psi_{rd}^{\max} = 2 \, \text{Vs}$;
- $\omega^{\min} = -200 \, \text{rad/s}$, $\omega^{\max} = 200 \, \text{rad/s}$;
- $p_5^{\min} = 0.1 \, \text{Vs}$, $p_5^{\max} = 10 \, \text{kVs}$.

The limit values are relevant for the studied motor from the paper [49]. All the intervals are divided into 20 segments.

2.5. Tensor Product-Based Controller Design

The mentioned control system is a nonlinear state feedback controller completed by the integral action to eliminate the steady-state error of the step response and to attenuate disturbance and noise effects [41,42,50].

2.5.1. Reference and Disturbance Data

The flux and torque reference values are set as follows:

$$\psi_{rd}^{ref} = 0.2 \text{ Vs}, \quad T^{ref} = 0.4 \text{ Nm}.$$

From these reference values, the reference currents i_{sd} and i_{sq} can be calculated as

$$i_{sd}^{ref} = \frac{\psi_{rd}^{ref}}{L_m} = 1.1834 \text{ A}, \quad i_{sq}^{ref} = \frac{2}{3} \frac{L_r}{p L_m} \frac{T^{ref}}{\psi_{rd}^{ref}} = 0.7061 \text{ A},$$

i.e., the reference current components contains the nominal value of the inductances L_m and L_r .

In the case of speed control,

$$\omega_{ref} = \frac{T_{ref}}{D_f} = 84.2105 \frac{\text{rad}}{\text{s}}$$

is adopted.

The load torque is defined as

$$T_L = \begin{cases} 0, & \text{if } t < T_1; \\ 0.4, & \text{if } T_1 \leq t < T_2; \\ -0.4, & \text{if } t \geq T_2. \end{cases}$$

In the case of torque control, $T_1 = 1.5 \text{ s}$, and $T_2 = 3 \text{ s}$; furthermore, in the case of speed control, $T_1 = 6 \text{ s}$, $T_2 = 9 \text{ s}$.

2.5.2. Controller Block Diagrams

Two control goals are realized in the frame of tensor product modeling: torque control and speed control.

The torque control scheme

The torque control scheme is depicted in Figure 1. The plant is the induction machine model with two inputs and two outputs: u_{sd} and u_{sq} are the d and q components of the stator voltage in the dq reference frame; y_1 and y_2 can be defined by the output matrix of C_0 , C_1 or C_2 , as well as the reference components y_1^{ref} and y_2^{ref} . The state x is applied in the state feedback part of the controller, denoted by $K = K(x)$. The error components between the reference signals and the outputs are plugged into the feedforward block of the controller, denoted by the matrix $K_I = K_I(x)$. The feedforward block contains an integrator for the two channels independently. The motor is affected by the load torque T_L .

The plant input in Figure 1 can be written as follows:

$$\mathbf{u} = \begin{bmatrix} u_{sd} \\ u_{sq} \end{bmatrix} = -\mathbf{K}\mathbf{x} + \mathbf{K}_I \int_{-\infty}^t \left(\begin{bmatrix} y_1^{ref} - y_1 \\ y_2^{ref} - y_2 \end{bmatrix} \right) d\tau, \quad (14)$$

where the dimensions of the matrices K and K_I are 2×4 and 2×2 , respectively.

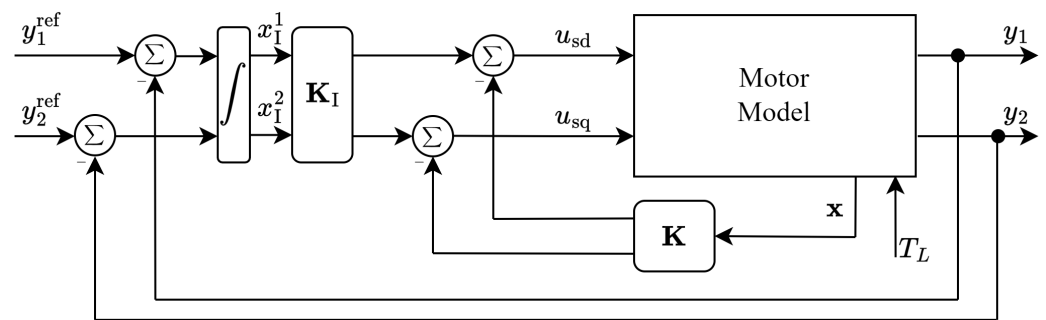


Figure 1. Block diagram for torque control.

The errors are defined as

$$\dot{x}_I^1 = y_1^{\text{ref}} - y_1, \quad \text{and} \quad \dot{x}_I^2 = y_2^{\text{ref}} - y_2,$$

with which the following augmented system can be defined for controller design using the notation $\mathbf{x}_I = [x_I^1 \ x_I^2]^T$:

$$\begin{bmatrix} \dot{\mathbf{x}} \\ \dot{\mathbf{x}}_I \end{bmatrix} = \begin{bmatrix} \mathbf{A}^{4 \times 4} & \mathbf{0}^{4 \times 2} \\ \mathbf{C}_i^{2 \times 4} & \mathbf{0}^{2 \times 2} \end{bmatrix} \begin{bmatrix} \mathbf{x} \\ \mathbf{x}_I \end{bmatrix} + \begin{bmatrix} \mathbf{B}^{4 \times 2} \\ \mathbf{0}^{2 \times 2} \end{bmatrix} \mathbf{u}, \quad (15)$$

$$\mathbf{u} = -[\mathbf{K} \ \mathbf{K}_I] \begin{bmatrix} \mathbf{x} \\ \mathbf{x}_I \end{bmatrix}, \quad (16)$$

and $i = 0, 1, 2$. The matrices can be designed by the solution of an LMI problem.

The speed control scheme

The control algorithm mentioned in Figure 1 is extended to the control rotational speed. The block diagram of the speed control is shown in Figure 2, where a new state variable x_I^3 and a new feedforward block $\mathbf{K}_\omega$ are introduced (this is a column vector with two rows); moreover, $y_1 = \psi_{rd}$, and $y_2 = \omega_m$ are chosen, i.e., the flux and rotor speed are the reference signal components.

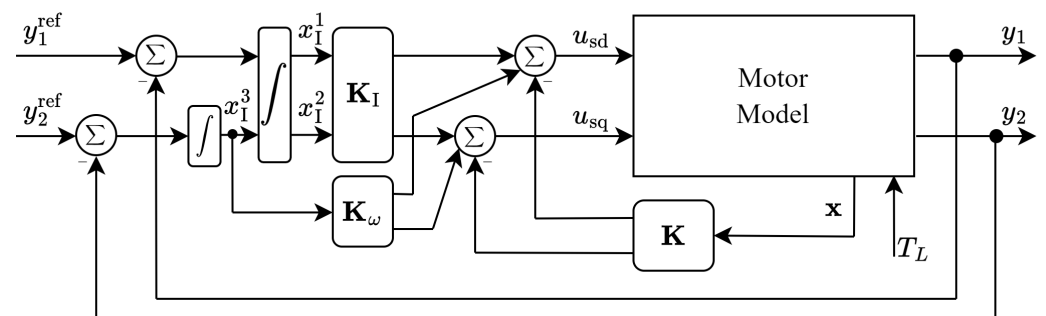


Figure 2. Block diagram for speed control.

The error $y_2^{\text{ref}} - y_2$ is processed by a PI controller block in the original field-oriented control scheme to generate a reference torque ($y_2^{\text{ref}} = \omega_m^{\text{ref}}$ and $y_2 = \omega_m$) [7,8,45–48]. This cascade PI controller must be reformulated to have the state feedback and integrator feedforward formula.

The details of the PI controller reformulation based on Figure 3 are as follows.

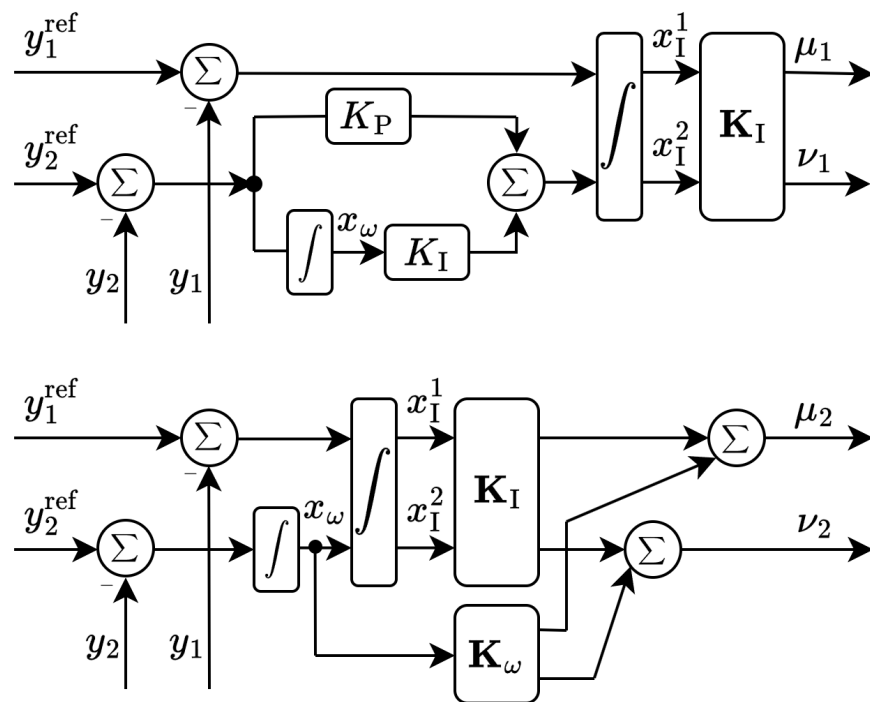


Figure 3. Block diagram conversion.

First, the original field-oriented control scheme extended for speed control contains a PI controller whose input is the error signal $y_2^{ref} - y_2$. This PI block is copied to the second channel of Figure 3, resulting in the upper scheme. Let us denote the error signal by

$$\dot{x}_\omega = y_2^{ref} - y_2. \quad (17)$$

The PI output signal is the input of the next integrator, i.e.,

$$\dot{x}_I^2 = K_P \dot{x}_\omega + K_I x_\omega, \quad (18)$$

in other words

$$x_I^2 = K_P x_\omega + K_I \int_{-\infty}^t x_\omega d\tau. \quad (19)$$

With these, the integrator block output followed by the matrix K_I is

$$\begin{aligned} \mu_1 &= K_{11} x_I^1 + K_{12} \left(K_P x_\omega + K_I \int_{-\infty}^t x_\omega d\tau \right), \\ \nu_1 &= K_{21} x_I^1 + K_{22} \left(K_P x_\omega + K_I \int_{-\infty}^t x_\omega d\tau \right). \end{aligned} \quad (20)$$

Second, the integrator output signal on the second channel in the bottom of Figure 3 is

$$\dot{x}_I^2 = x_\omega, \quad (21)$$

i.e.,

$$x_I^2 = \int_{-\infty}^t x_\omega d\tau. \quad (22)$$

The P and I part constants of the original field-oriented control scheme can be moved to the vector $\mathbf{K}_\omega$. With these, the integrator block output followed by the matrix $\mathbf{K}_I$ and the vector $\mathbf{K}_\omega$ is

$$\begin{aligned}\mu_2 &= K_{11}x_1^1 + K_{12}x_1^2 + K_{\omega,1}x_\omega = K_{11}x_1^1 + K_{12}\int_{-\infty}^t x_\omega d\tau + K_{\omega,1}x_\omega, \\ \nu_2 &= K_{21}x_1^1 + K_{22}x_1^2 + K_{\omega,2}x_\omega = K_{21}x_1^1 + K_{22}\int_{-\infty}^t x_\omega d\tau + K_{\omega,2}x_\omega.\end{aligned}\quad (23)$$

By the appropriate choice of the integrator parameters, $\mu_2 = \mu_1$ and $\nu_2 = \nu_1$ can be realized, i.e., the mentioned block diagram can be applied to control motor speed via state feedback control.

The control signals of Figure 2 are given as:

$$\begin{bmatrix} u_{sd} \\ u_{sq} \end{bmatrix} = -\mathbf{K}\mathbf{x} + \mathbf{K}_I \int_{-\infty}^t \left(\begin{bmatrix} y_1^{\text{ref}} - y_1 \\ y_2^{\text{ref}} - y_2 \end{bmatrix} d\tau \right) + \mathbf{K}_\omega \int_{-\infty}^t (y_2^{\text{ref}} - y_2) d\tau. \quad (24)$$

In this case, the augmented system can be written as

$$\begin{bmatrix} \dot{\mathbf{x}} \\ \dot{\mathbf{x}}_I \\ \dot{x}_\omega \end{bmatrix} = \begin{bmatrix} \mathbf{A}^{4 \times 4} & \mathbf{0}^{4 \times 3} \\ \begin{bmatrix} 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix} & \begin{bmatrix} 0 & 0 & 0 \\ 0 & 0 & 1 \\ 0 & 0 & 0 \end{bmatrix} \end{bmatrix} \begin{bmatrix} \mathbf{x} \\ \mathbf{x}_I \\ x_\omega \end{bmatrix} + \begin{bmatrix} \mathbf{B}^{4 \times 2} \\ \mathbf{0}^{3 \times 2} \end{bmatrix} \mathbf{u}, \quad (25)$$

$$\mathbf{u} = -\begin{bmatrix} \mathbf{K} & \mathbf{K}_I & \mathbf{K}_\omega \end{bmatrix} \begin{bmatrix} \mathbf{x} \\ \mathbf{x}_I \\ x_\omega \end{bmatrix}, \quad (26)$$

whose representation can be used to design a nonlinear controller which contains the PI-term for torque generation in a nonlinear form which is applicable in the TP-based framework.

2.5.3. Controller Design

Controller synthesis is performed via the following steps:

- Define the motor parameters, references, and other data such as the load torque signal (see Sections 2.3 and 2.5.1).
- Define parameter vector $\mathbf{p}$ according to the analyzed TP model alternative (i.e., select parameters A, B, C, D , and E). It is based on Table 1.
- Define the parameter space Ω , and its discretization (see Section 2.4).
- Set up system matrix $\mathbf{A}$, input and output matrices $\mathbf{B}$ and $\mathbf{C}_i$ ($i = 0, 1, 2, 3$) based on Table 1 and on the equations in Section 2.2.
- Set up the matrices according to the block diagram in Figure 1 for torque control or in Figure 2 for rotor speed control, i.e., according to (15) or (25). For example, set up the matrix

$$\begin{bmatrix} \mathbf{A}^{4 \times 4} & \mathbf{0}^{4 \times 2} & \mathbf{B}^{4 \times 2} \\ \mathbf{C}_i^{2 \times 4} & \mathbf{0}^{2 \times 2} & \mathbf{0}^{2 \times 2} \end{bmatrix},$$

or

$$\begin{bmatrix} \mathbf{A}^{4 \times 4} & \mathbf{0}^{4 \times 3} & \mathbf{B}^{4 \times 2} \\ \begin{bmatrix} 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix} & \begin{bmatrix} 0 & 0 & 0 \\ 0 & 0 & 1 \\ 0 & 0 & 0 \end{bmatrix} & \mathbf{0}^{3 \times 2} \end{bmatrix};$$

- Run higher-order singular value decomposition using the above mentioned matrix;

- Set up weighting functions and vertex systems according to the approximation by

$$\dot{\mathbf{x}} = \sum_{n=1}^R w_n(\mathbf{p})(\mathbf{A}_n \mathbf{x} + \mathbf{B}_n \mathbf{u}), \quad (27)$$

where R is the number vertex systems (see Table 1 for the applied linear systems of the different TP model alternatives), $\mathbf{A}_n$ and $\mathbf{B}_n$ are the linear system matrices in the vertices, and $w_n(\mathbf{p})$ is the weighting function. In this study, only CNO (Close-To-Normality) and IRNO (Inverse-Relaxed-Normality) functions are studied; however, other weighting functions can be selected, such as SNNN (Sum-Normalized and Non Negativeness) [31–34];

- The control goal is to speed up the system response. In this case, the so-called decay rate control can be applied, i.e., to solve the following generalized eigenvalue minimization problem with linear matrix inequalities [35]:

$$\begin{aligned} & \text{maximize } \alpha, \text{ subject to} \\ & \mathbf{X} \succ \mathbf{0}, \\ & -\mathbf{X}\mathbf{A}_n^T - \mathbf{A}_n\mathbf{X} + \mathbf{M}_n^T\mathbf{B}_n^T + \mathbf{B}_n\mathbf{M}_n - 2\alpha\mathbf{X} \succ \mathbf{0}, \\ & -\mathbf{X}\mathbf{A}_n^T - \mathbf{A}_n\mathbf{X} - \mathbf{X}\mathbf{A}_m^T - \mathbf{A}_m\mathbf{X} + \mathbf{M}_m^T\mathbf{B}_n^T + \mathbf{B}_n\mathbf{M}_m + \mathbf{M}_n^T\mathbf{B}_m^T + \mathbf{B}_m\mathbf{M}_n - 4\alpha\mathbf{X} \succeq \mathbf{0}, \end{aligned} \quad (28)$$

where $n = 1, \dots, R$, and $m = n + 1, \dots, R$.

The control goal is augmented by the control value constraint. Assuming that the state initial value is bounded, i.e., $\|\mathbf{x}(0)\| \leq \phi$ (ϕ is predefined), the constraint $\|\mathbf{u}(t)\|_2 \leq u_{\max}$ can be enforced at all times $t \geq 0$ if the following linear matrix inequalities hold:

$$\begin{aligned} & \phi^2 \mathbf{I} \preceq \mathbf{X}, \\ & \begin{bmatrix} \mathbf{X} & \mathbf{M}_n^T \\ \mathbf{M}_n & u_{\max}^2 \mathbf{I} \end{bmatrix} \succeq \mathbf{0}, \quad n = 1, \dots, R. \end{aligned} \quad (29)$$

The maximum value of the parameter α can be found by the simple bisection method as follows:

- Set the interval $[a, \dots, b]$ where α is assumed;
- While $|b - a| > \varepsilon$ do the following iteration (ε is a small positive limit, $\varepsilon = 10^{-5}$ is used):
 - * $x = \frac{a+b}{2}$;
 - * Solve the above mentioned LMI with $\alpha = x$;
 - * If LMI is feasible, then $a = x$; otherwise, $b = x$.
- $\alpha = x$.
- Vertex gain matrices are obtained by

$$\mathbf{K}_n = \mathbf{M}_n \mathbf{X}^{-1}, \quad n = 1, \dots, R, \quad (30)$$

from which the feedback and feedforward gains in (16) or in (26) can be calculated:

$$\mathbf{u} = - \sum_{n=1}^R w_n(\mathbf{p}) \mathbf{K}_n \begin{bmatrix} \mathbf{x} \\ \mathbf{x}_I \end{bmatrix}, \quad \text{or} \quad \mathbf{u} = - \sum_{n=1}^R w_n(\mathbf{p}) \mathbf{K}_n \begin{bmatrix} \mathbf{x} \\ \mathbf{x}_I \\ x_\omega \end{bmatrix}, \quad (31)$$

if torque control or speed control is running.

- Check the closed-loop control system. Feasibility and applicability checks are performed, and the results are shown in the next section.

3. Results

3.1. The Weighting Functions

Figure 4 shows the CNO-type weighting functions of the model #1 when the parameter p_5 is not used. The effect of ψ_{rd} in the denominator of the A matrix elements can be seen in $w_{3,j}$ ($j = 1, 2, 3$), especially when the flux is small, close to zero. This behavior can be eliminated by introducing the parameter denoted by p_5 , i.e., $p_5 = 1/\psi_{rd}$. The other weighting functions $w_{i,j}$ ($i = 1, 2, 4, j = 1, 2$) are simple linear ones. By introducing p_5 , all the weighting functions become such simple linear ones.

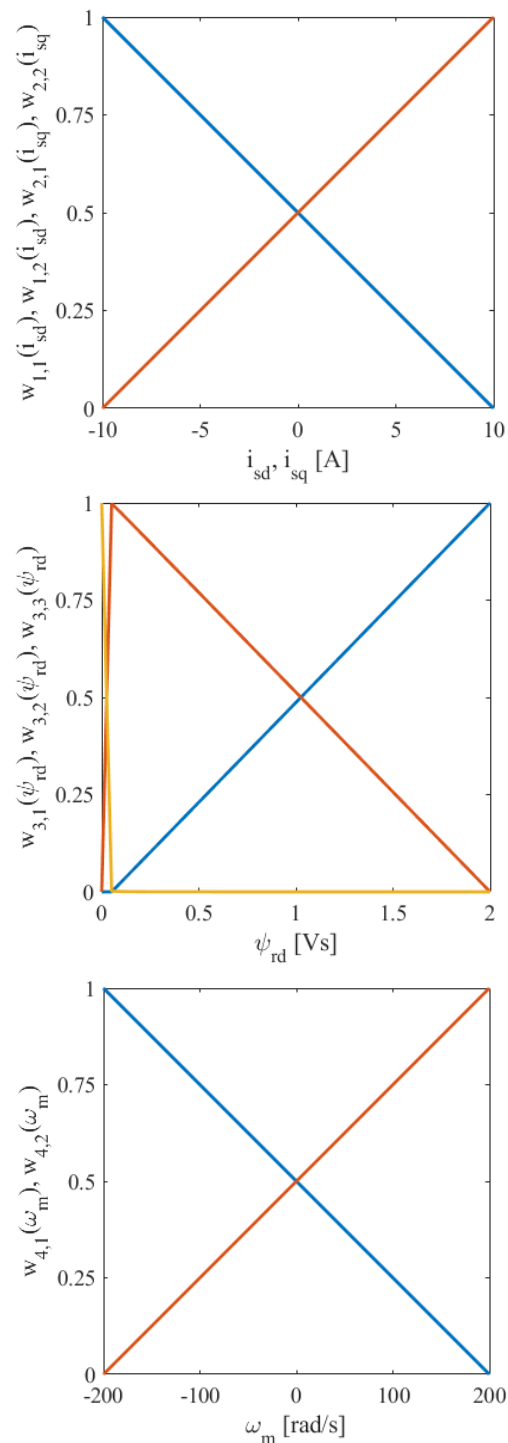


Figure 4. Typical CNO basis functions $w_{i,j}$ of the model without the parameter p_5 —model #1, (blue— $j = 1$, red— $j = 2$, orange— $j = 3$).

Figure 5 shows the IRNO-type weighting functions of the model #0 when the parameter p_5 is used.

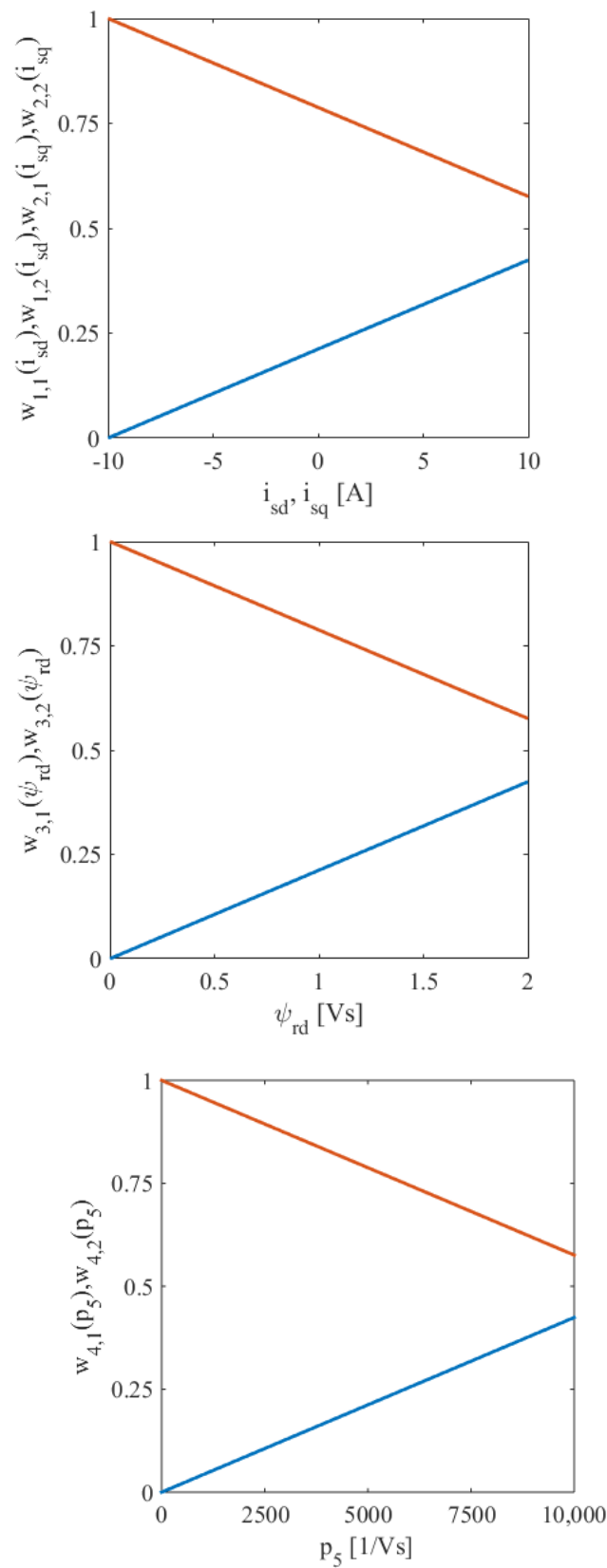


Figure 5. Typical IRNO basis functions $w_{i,j}$ of the model with the parameter p_5 —model #0—model #1, (blue— $j = 1$, red— $j = 2$).

These functions are used in (3), and consequently in (27) and in (31) to approximate the nonlinear system as well as the nonlinear controller. Applying weighting functions gives a combination of the vertex systems A_n , B_n in (27), and K_n in (31); finally, the shape of the weighting functions is dependent on the type (CNO, IRNO, SNNN, etc.) and on the qLPV system representation. In this case, simple linear functions appear.

3.2. Results with the Output Matrix C_0

The output matrix C_0 gives the original field-oriented control scheme idea, but the TP-based controller is nonlinear. The current components i_{sd} and i_{sq} are controlled directly by using the reference value of i_{sd}^{ref} and i_{sq}^{ref} .

The parameter u_{max} is chosen from the interval $[100, \dots, 400]$ V and the parameter $|x(0)|$ from $[0.01, \dots, 0.1]$. The maximum value of the decay rate is selected by the bisection method as it was presented before. A typical function of $\alpha = \alpha(u_{max}, |x(0)|)$ can be seen in Figure 6 obtained by using model #4. It is noted that similar functions can be given by the other model variants, too. The maximum and the minimum values of the decay rate parameter are depicted in the figure: $\alpha_{max} = 4.282$ at $u_{max} = 400$ V and $|x(0)| = 0.01$; $\alpha_{min} = 0.3336$ at $u_{max} = 100$ V and $|x(0)| = 0.1$. The effect of selecting the decay rate is illustrated in the following.

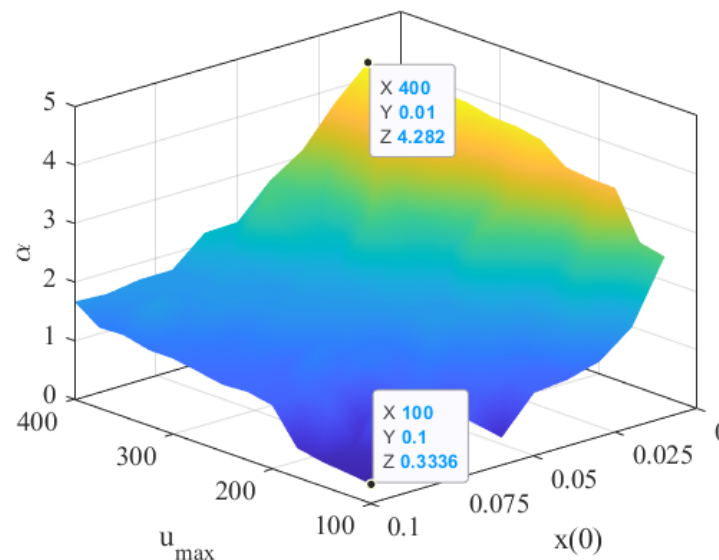


Figure 6. Parameter α as the function of u_{max} and $|x(0)|$ —model #4.

Figures 7–9 show the time function of the state variable i_{sd} , i_{sq} , ψ_{rd} , and ω_m , respectively. The slow and fast transients according to the minimum and maximum decay rate parameters are highlighted. It is noted that the current i_{sq} is practically not changed if the parameter α is changed. The other current i_{sd} can be sped up efficiently; however, there is a peak at the very beginning when model #4 is applied. The current overshoot is typically around 10–20% for model #4 when α is changed. There is no overshoot in the time function of the flux and of the rotational speed.

The torque is shown in Figure 10; it follows the flux time function. The d and q components of the control input can be studied in Figure 11.

It must be underlined that the load torque change has no practical effect on the controller scheme because the state variables as well as the torque are not changing, except the motor speed. The speed is decreased to zero and increased to twice the nominal value according to the load torque change.

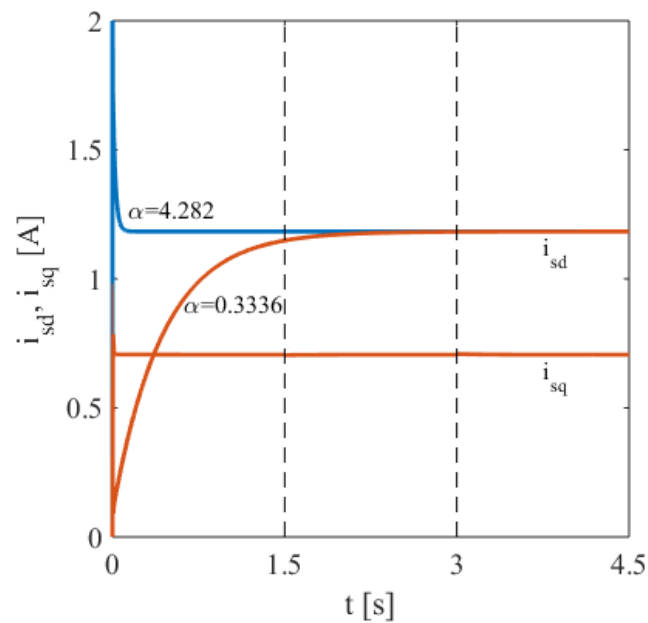


Figure 7. Slow and fast transients of i_{sd} and i_{sq} —model #4.

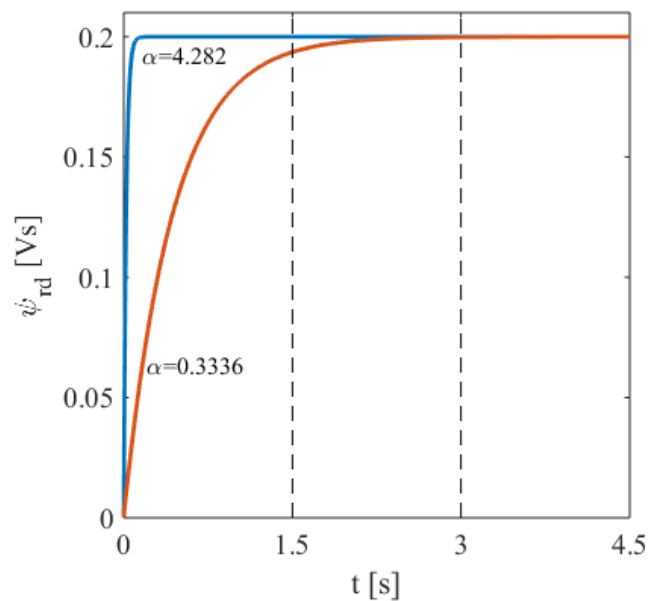


Figure 8. Slow and fast transients of ψ_{rd} —model #4.

It is found that the behavior of models #4–#7 is practically the same. It is noted that the above mentioned overshoot in the d-current component can be eliminated by the variants #4–#7 and an appropriate choice of α . On the other hand, a faster transient can be reached as well, e.g., with model variants #12–#15 and #28–#31, when the currents can be step-wise. Models #20–#23 are not the proper way to choose because overshoot appears in the q-current component as well as in the torque function at the load torque change.

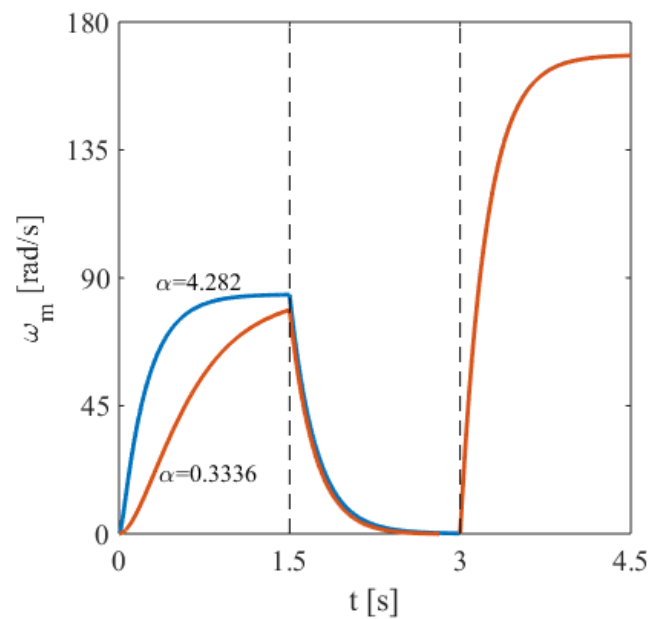


Figure 9. Slow and fast transients of ω_m —model #4.

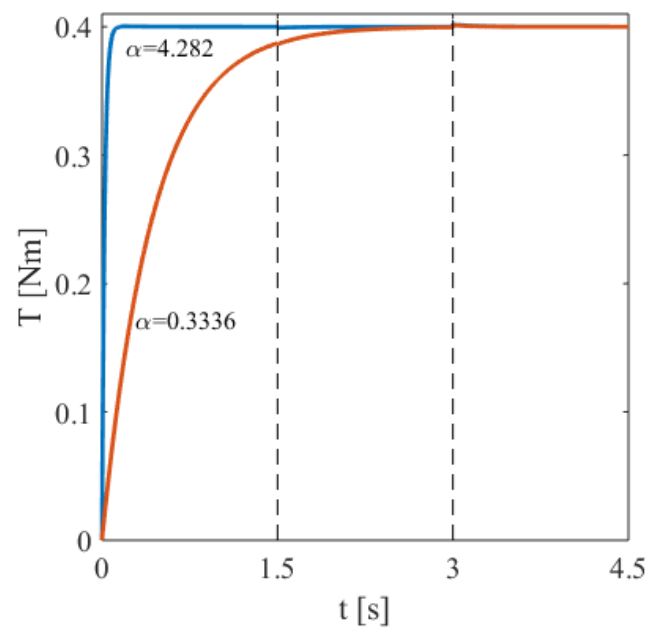


Figure 10. Slow and fast transients of T —model #4.

Next, the robustness analysis is shown. The stator and rotor resistances depend on the temperature inside the motor. The temperature change on the resistance has the following physical rule:

$$\begin{aligned} R_s(T) &= R_{s,20^\circ\text{C}}(1 + \alpha_T \Delta T), \\ R_r(T) &= R_{r,20^\circ\text{C}}(1 + \alpha_T \Delta T), \end{aligned} \quad (32)$$

with the notation 20°C in the index, meaning the resistance at room temperature, i.e., the nominal value of the resistance. The parameter α_T is the temperature coefficient for resistivity, and finally, ΔT is the temperature change. The value of α_T is $3.93 \cdot 10^{-3} \frac{1}{^\circ\text{C}}$ for copper.

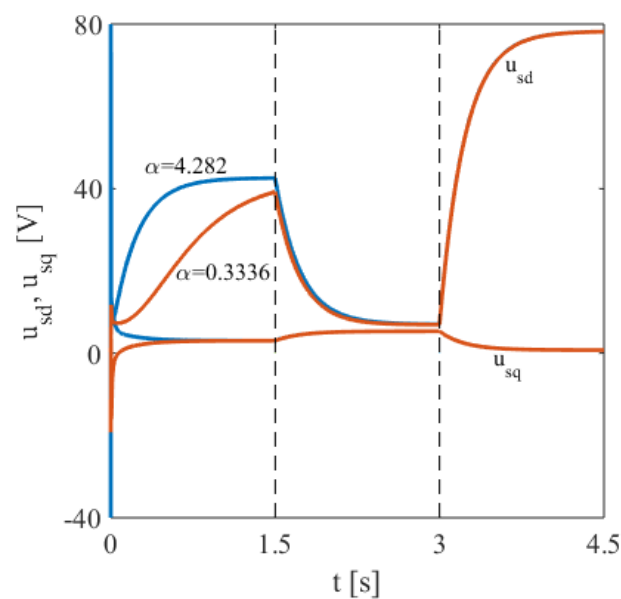


Figure 11. Slow and fast transients of u_{sd} and u_{sq} —model #4.

Taking the resistivity change into account in the simulations is a kind of robustness analysis. The controller is designed when room temperature and nominal resistances are valid. In the present example, the following parameters are obtained by controller design: $u_{\max} = 200$ V, $|\mathbf{x}(0)| = 0.05$, and $\alpha = 1.666$. Here, the temperature is changed between -60 °C and 200 °C, resulting in a very small change in the state variables, as well as in the output functions. As an illustration of the robustness, the resulting torque function is depicted in Figure 12. It is easy to see that the torque change is negligible; moreover, the load torque change has practically no effect on the output torque.

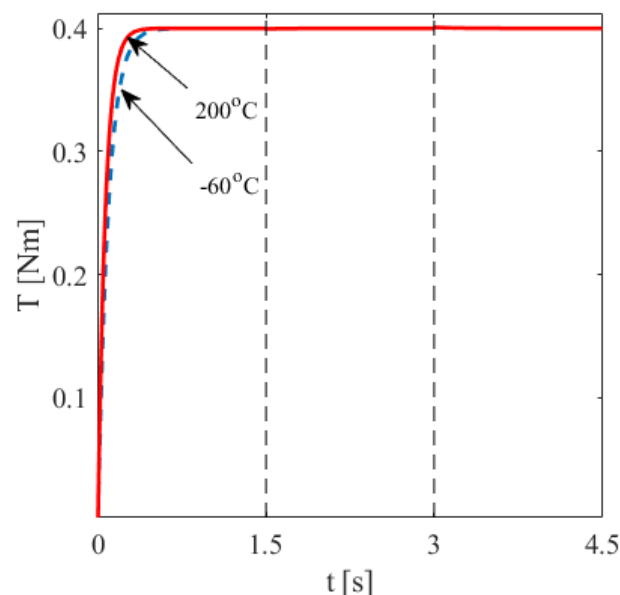


Figure 12. Torque change vs. the temperature—model #4.

3.3. Results with the Output Matrix $\mathbf{C}_1$

The flux component ψ_{rd} and the torque are the model outputs when matrix $\mathbf{C}_1$ is applied. In this case, only the models #4–#7, #20–#23 and #28–#31 result in an applicable solution. The models #12–#15 give also a feasible controller, but the designed controller works infeasibly because the current i_{sq} is zero. From engineering point of view, this solution is not

allowed. The best results can be obtained when one of the models #28–#31 is used for the controller design.

It is noted that the different qLPV models are different representations of the same nonlinear system but with different weighting functions and different vertex systems in (27). The controller gain vertex systems in (31) are designed by a numerical LMI solver. An infeasible solution means a noncontrollable system because the appropriate LMIs cannot be solved. If the controller is feasible but at the same time the solution is inadequate, then the model representation is not a good candidate.

Model #28 is selected to present the results (the following parameters are obtained by controller design: $u_{\max} = 400$ V, $|\mathbf{x}(0)| = 0.04$, and $\alpha = 1.3106$). It is noted that similar findings can be obtained by models #28–#31.

Figure 13 presents the current component time functions. A magnified plot of the starting transient is also shown. First, the input is turned on at $t = 0$, resulting in huge overshoot in the q-component of the currents (see red curves), but there is no peak in the other component. Second (blue curves), a time shift $\Delta t = 0.1$ s is applied in the torque reference, which decreases the overshoot of the q-component current efficiently. On the other hand, overshoot appears in the d-component. The overshoot in the q-component affects the torque transient as it is highlighted in Figure 14, but the time shift in the torque reference efficiently decreases the peak in the torque function. The time shift has neither a practical effect on the rotor speed (Figure 15).

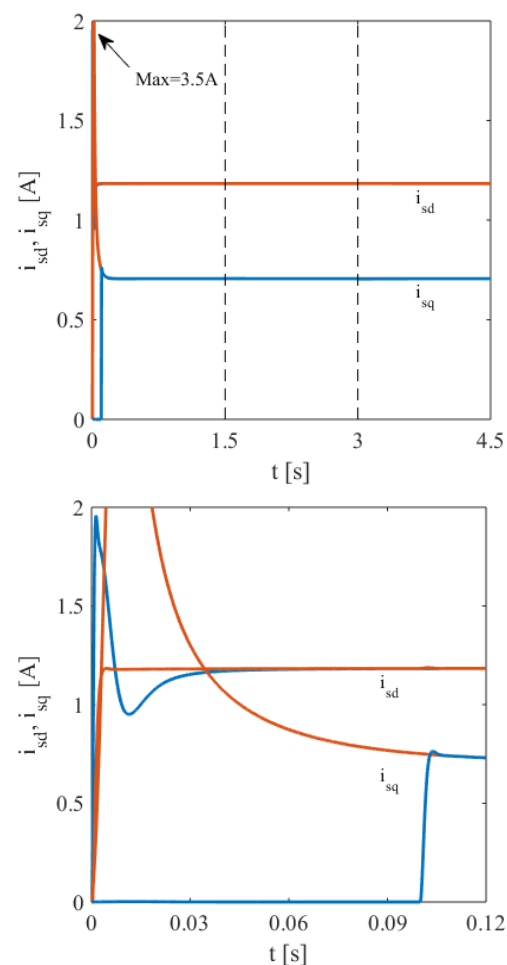


Figure 13. Current transient with (blue) and without (red) time delay in torque reference—model #28.

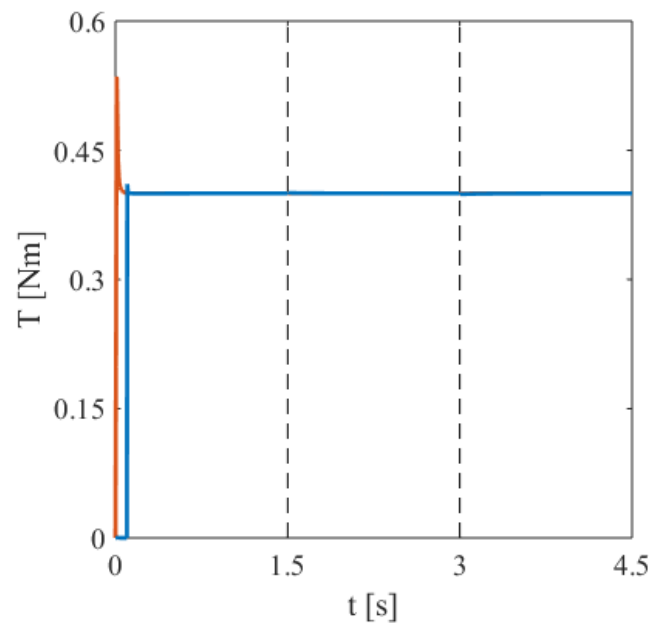


Figure 14. Typical time function of torque with (blue) and without (red) time delay in torque reference—model #28.

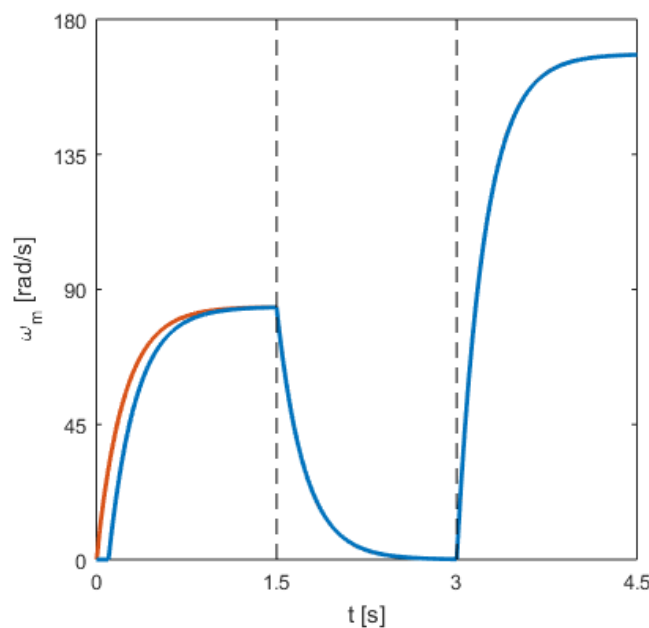


Figure 15. Time function of rotor speed with (blue) and without (red) time delay in torque reference—model #28.

It must be underlined again that the load torque change has no practical effect on the controller scheme behavior because the state variables as well as the torque are not changing, except the motor speed. Of course, the speed is decreased to zero and increased to twice the nominal value according to the load torque change.

The resistance change caused by temperature change has exactly the same effect as it was shown in the case output matrix C_0 : negligible change in the time functions can be observed.

Here, another robustness analysis is presented. The inductance drop in L_m has an effect only in the time function of the i_{sd} component, it is increased by 25%, i.e., the stationary value is around 1.48 A when the inductance drop is 80%. Anyway, the reference flux and torque can be tracked without stationary error according to the integral action in the control block diagram.

Models #20–#23 are not useful variants again because the load torque change affects the load characteristics.

3.4. Results with the Output Matrix C_2

On the other hand, applying the output matrix C_2 does not give any applicable solution. The flux component ψ_{rd} and the torque are the model outputs; the two nonzero elements are both in the third column.

Only models #4–#7 are feasible mathematically, but there is a huge overshoot in the time function of the current i_{sq} , resulting in a very slow controlling algorithm, i.e., the results are not useful from an engineering point of view.

3.5. Results with the Output Matrix C_3

The flux component ψ_{rd} and the rotor speed ω_m are the model outputs when matrix C_3 is applied. In this case, only the model variants #20–#23 and #28–#31 result in an applicable solution. The results below are presented via model #31 with the following parameters: $u_{\max} = 100$ V, $|x(0)| = 0.01$, and $\alpha = 0.434$.

First, the input is turned on at $t = 0$, resulting in huge overshoot in the q component of the currents (see red curves in Figure 16), but there is no peak in the d component. Second (blue curves), a time shift is applied in the speed reference by linearly increasing the reference in the range of $t \in [0, \dots, 3]$ s. It decreases the overshoot of the q-component current efficiently. The d component is practically not changed. The overshoot in the q component affects the torque transient as it is highlighted in Figure 17, but the time shift in the speed reference efficiently decreases the peak in the torque function. Unfortunately, the rotor speed has a huge overshoot as it is shown in Figure 18, without the linearly increasing speed reference. It can only be eliminated by the mentioned time function of the reference signal. The time shift has no practical effect on the flux. The control signal components are shown in Figure 19.

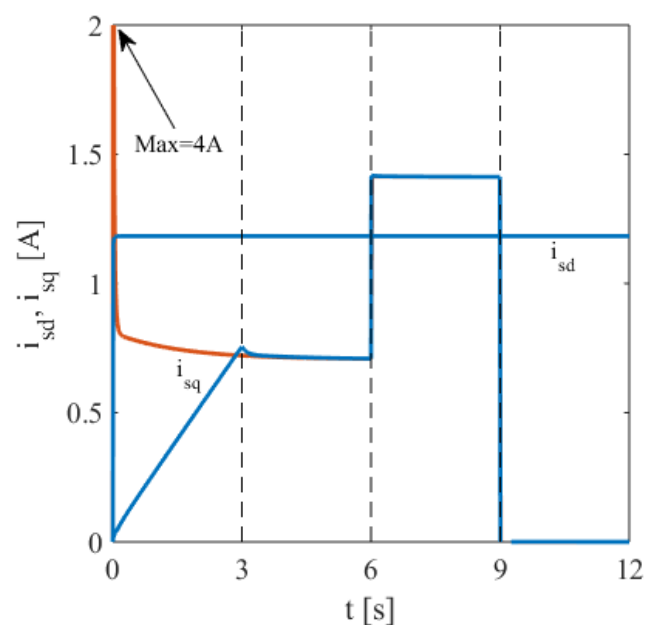


Figure 16. Current transient with (blue) and without (red) time delay in speed reference—model #31.

The load torque change has no practical effect on the controller scheme because the state variables as well as the speed are not changing, except the q-component current and the torque. The torque is decreased to zero and is increased to twice the nominal value according to the load torque change.

An interesting check can be studied in Figure 20. Here, the reference speed function has linearly increasing and decreasing parts as well as stationary intervals. The system output follows this signal with a small tracking error.

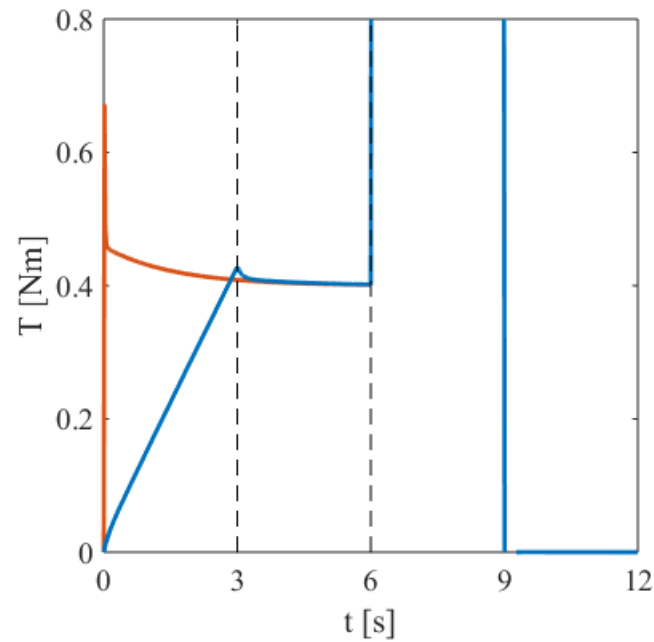


Figure 17. Torque transient with (blue) and without (red) time delay in speed reference—model #31.

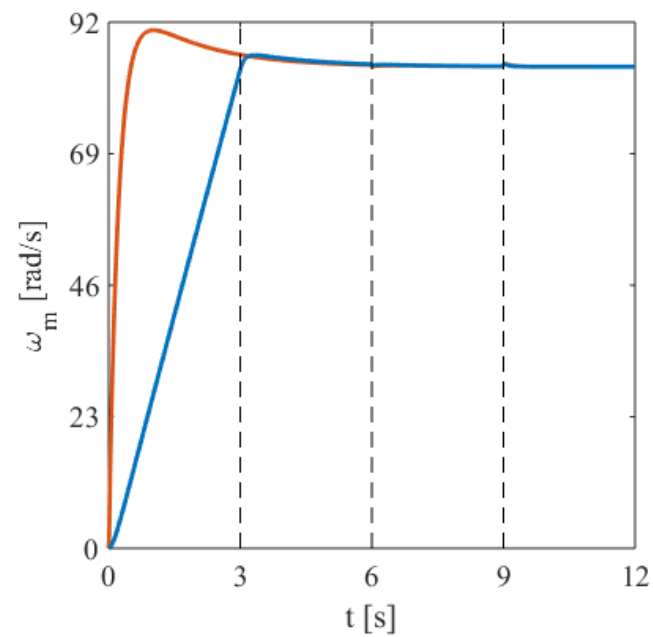


Figure 18. Speed transient with (blue) and without (red) time delay in speed reference—model #31.

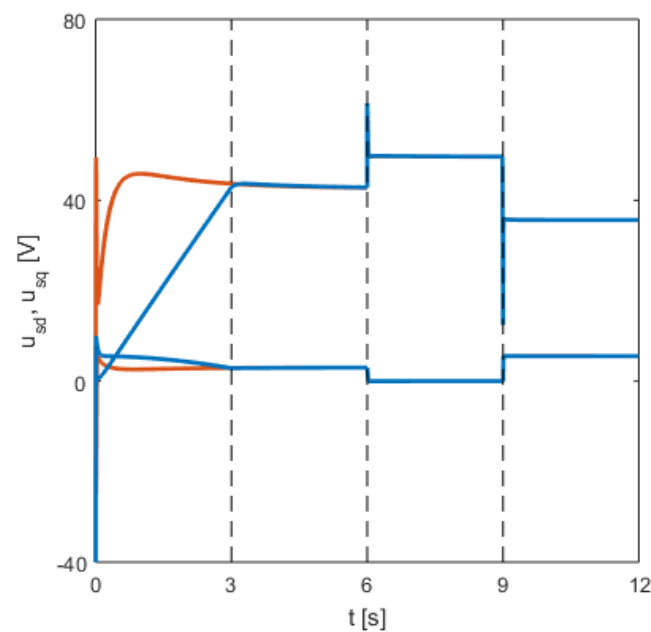


Figure 19. Control signal transient with (blue) and without (red) time delay in speed reference—model #31.

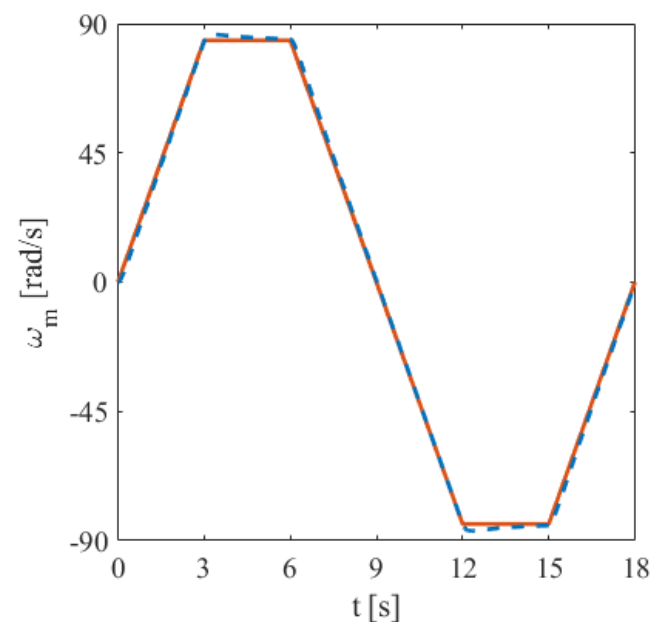


Figure 20. Linearly varying reference speed (—) and the output (---)—model #31.

3.6. Changing of the Weighting Functions

At the end, it is noted that there is practically no difference between the results by CNO-type weighting functions and IRNO-type weighting functions. This can be explained by the fact that the original nonlinear model is the same, giving the same results, while it can be approximated by different weighting functions and different vertex systems as well as vertex gain matrices in (27) and in (31), respectively.

3.7. The Control Matrix

The original field-oriented control scheme contains two independent linear PI controllers. The two channels in the TP-based controller are not independent because of the matrix $\mathbf{K}_I$. Figure 21 shows the time function of the $\mathbf{K}_I$ matrix elements. The diagonal elements K_I^{11} and K_I^{22} are constants after a short transient at the initial state; the values

are 1792 and 2437, respectively. It is noted that the values are not changing when the load torque is changed. The off-diagonal elements K_I^{12} and K_I^{21} are also constants after a short transient, and the values are around 0.9 and 1.2, respectively, and these are slightly changing when the load torque is changed. It can be stated that the off-diagonal elements are negligible compared to the diagonal ones, i.e., the nonlinear feedforward integral action is approximately independent.

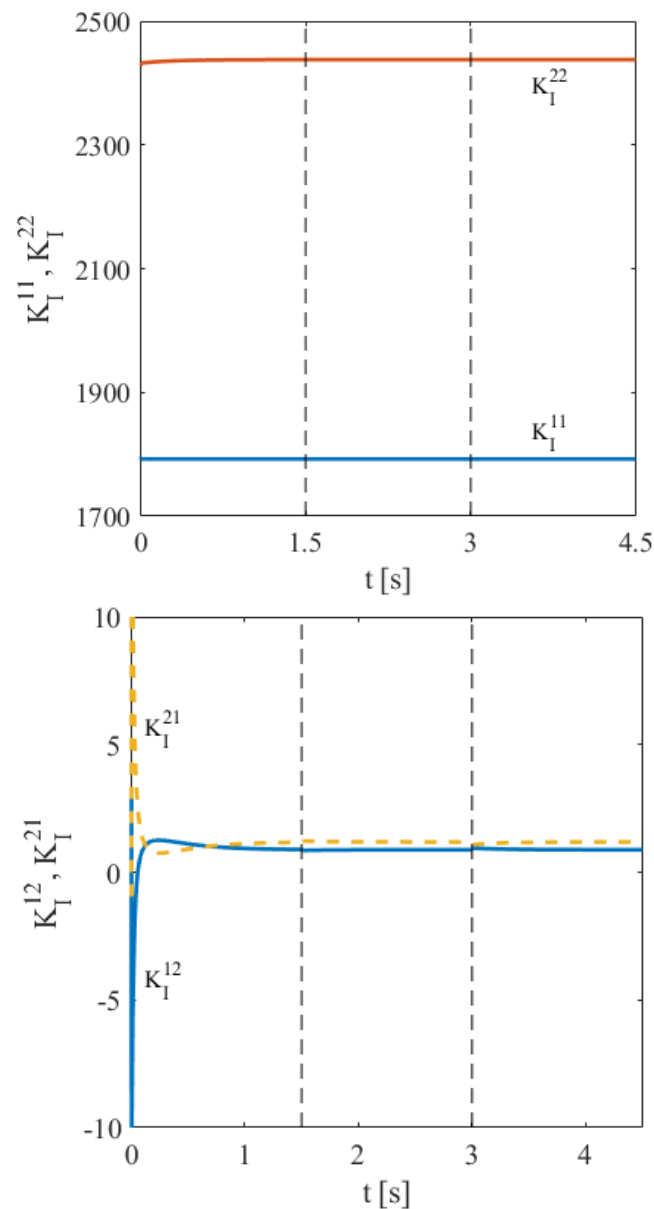


Figure 21. Time variation of $\mathbf{K}_I$ matrix elements—model #4.

The matrix elements in the stationary state are slightly constant, resulting in a diagonal $\mathbf{K}_I$ matrix. The matrix elements are not changing when the load torque appears; however the state variables are changing, resulting in a change in the control signal (see, for example, Figure 11 or Figure 19). It must be highlighted that the closed-loop system is always stable because the LMIs are based on Lyapunov's stability theorem.

4. Conclusions

The main contribution of this paper is a new methodology for TP-based nonlinear controller design. Following this methodology, it is possible to select all qLPV models,

which are suitable for controller design. Although the proposed methodology is generally applicable for nonlinear systems, it was used to create a family of controllers for induction machines, which combines the principles of FOC and TP transformation.

The first part of the paper discusses the TP modeling of induction machines and its description in the LMI framework. The mathematical formalism of the FOC-based nonlinear state-space representation is investigated, and all the possible 128 qLPV models for torque control and speed control are studied extensively. A new block diagram is investigated for speed control, which is applicable in the TP model-based framework.

With the TP model transformation, it is possible to systematically investigate all possible qLPV models that can be written. This, of course, requires a rather lengthy and resource-intensive run. However, it is possible to find the optimal model–controller pair. It must be underlined that not all of the models give a feasible and applicable controller, which is why all the possible combinations must be analyzed.

The designed nonlinear controller ensures stable, accurate, and robust operation over the full operating range. It is confirmed that the wide range of temperature value affecting the resistance of stator and rotor coils can be taken into account. Thus, the presented TP-based FOC provides high performance even under extreme parameter variations, unlike the traditional FOC using linear controllers and decoupling. In addition, the proposed controller keeps the current and torque ripples low, outperforming the conventional DTCs in this respect. Moreover, TP-based control design is numerical, which is easy to algorithmize, so unlike FLC, it does not require complex mathematical formulas.

A possible future research task is the observer synthesis in the above mentioned comprehensive investigation. Other effects can also be modeled, such as eddy current effects or other nonmodeled nonlinearities. The method can also be applied to other type of machines as well as other nonlinear systems.

Author Contributions: Conceptualization, M.K. and K.H.; methodology, M.K.; software, M.K.; validation, M.K. and K.H.; formal analysis, M.K.; investigation, M.K.; resources, K.H.; data curation, M.K. and K.H.; writing—original draft preparation, M.K. and K.H.; writing—review and editing, M.K. and K.H.; visualization, M.K.; supervision, M.K.; project administration, K.H. All authors have read and agreed to the published version of the manuscript.

Funding: This research received no external funding.

Data Availability Statement: Data is contained within the article.

Acknowledgments: Special thanks to Péter Baranyi for supervising the TP modeling and control design.

Conflicts of Interest: The authors declare no conflicts of interest.

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