

1 SUPPLEMENTARY MATERIALS

2 Strengthening of Wood-like Materials via
3 Densification and Nanoparticle Intercalation

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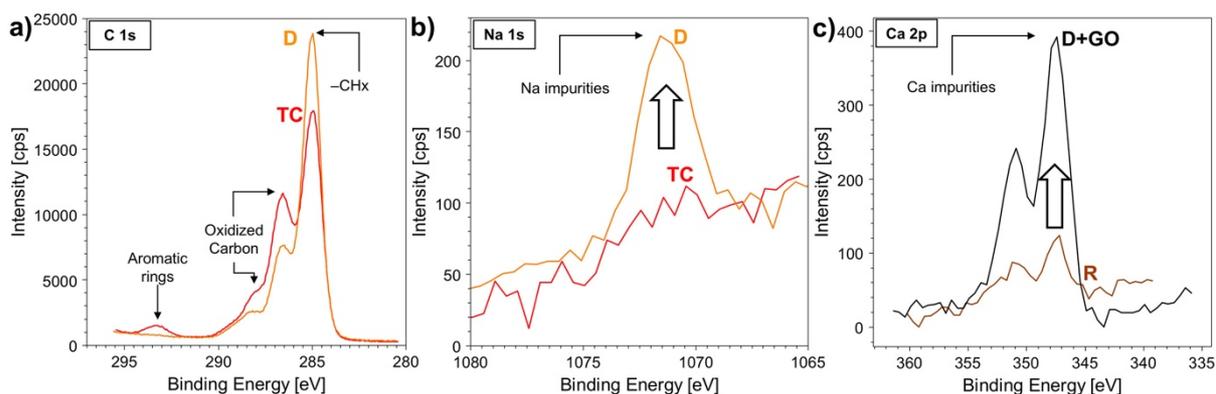
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16 Note 1 - XPS:



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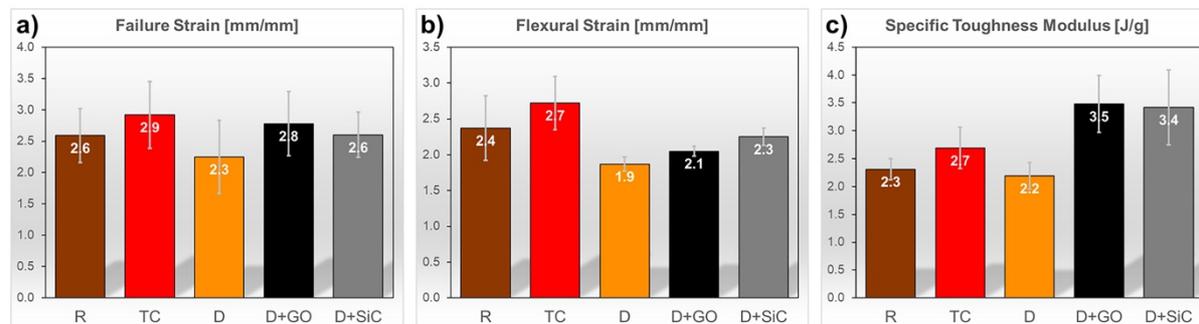
18 **Figure S1.** a) The difference in XPS spectra between TC and D shows the effects of delignification. TC
19 and D samples differs only for the delignification process. b) presence of Na impurities. They are
20 increased after the chemical etching c) presence of Ca impurities. They are increased from <0.1% to
21 ~0.5% after the chemical etching. Ca impurities have an important effect on the D+GO as they interact
22 with GO during burning.

23 The binding energy that is associated with the structure at ~294.3 eV in the C1s spectrum of “TC”
24 in Fig. S1 has energies that are too high to justify the bonding of carbon with any element. It could
25 instead be explained with a shake-up structure that is formed due to the presence of aromatic rings.
26 This structure appears only in R and TC, and it disappears in all the other specimens, which were
27 subject to delignification process. Thus, it could arise from the aromatic rings in the molecular
28 structure of lignin [1]. As Fig. S1 shows, delignification took place going from TC to D and this
29 corresponds with the decrease of the structure at ~294.3 eV. However, the decrease of this feature is
30 not enough to explain the increase of the component associated with -CH_x at 285.5 eV [2] to the
31 detriment of the oxidized carbon that occurs as a result of the chemical delignification process.

32 A possible concurrent phenomenon is the degradation of hemicellulose and cellulose [3]. Indeed,
33 alkaline treatments on wood-like materials remove part of the lignin and partially depolymerize
34 hemicellulose and cellulose [4]. The subsequent degradation is turned into an extraction of the

35 different sugars forming hemicellulose or the glucose from cellulose. In particular, in solutions with
 36 increased alkalinity, the extraction of glucose from the depolymerization of cellulose in natural fibers
 37 is increased [5]. This extraction can reduce the overall amount of oxidized carbons.

38 Note 2 - Mechanical properties:



39

40 **Figure S2.** Histograms summarizing the mechanical properties of R, TC, D, D+GO and D+SiC
 41 samples: a) failure strain, b) flexural strain and c) specific toughness modulus, which was derived
 42 from the toughness modulus as described in the article. Error bars represent the standard deviation.

43 Note 3 - Interfacial fracture energy:

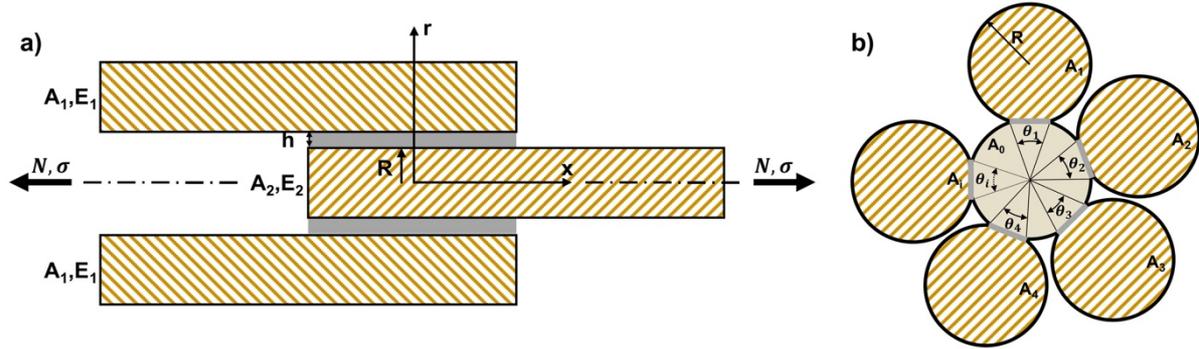
44 As for wood-like structures, the mechanical performance of densified and composite reed is
 45 dominated by the shear load transfer between cellulose microfibrils [6].

46 Thus, failure mechanics is prevalently governed by the shear failure at interfibrillar interface.
 47 Griffith linear elastic fracture mechanics have proven to be suitable describing the failure of adhesive
 48 joints subjected to complex loading conditions [7, 8]. It was suggested [9] that there is a correlation
 49 between the load necessary for the crack to propagate and the fracture energy of the interface joint.
 50 This insight could result in a better understanding of the adhesion in our nanocomposite materials.
 51 The microscale interactions between nanoparticles and cellulose microfibrils can be treated within a
 52 precise geometrical configuration that is modelled to estimate the increase in toughness reported after
 53 the introduction of nanoparticles. We can imagine the interface as an adhesive joint that unbinds
 54 when crack propagates. Fig. S2 shows a hypothetical loading setup for a group of fibrils in mutual
 55 contact. An isotropic linear elastic behavior is assumed for the materials involved, i.e. CMFs and
 56 interface materials. If the adhesive layer thickness h tends to zero, then the elastic strain energy
 57 absorbed by the layer is negligible. A full analytical derivation was proposed [9] for a similar
 58 geometry of concentric tubes joined by an adhesive at their interface and subjected to axial load. The
 59 link between critical load for crack propagation, adhesion and geometry holds to be:

60

$$N_c = \sqrt{4\pi R G_a \frac{E_2 A_2}{E_1 A_1} (E_1 A_1 + E_2 A_2)} \quad (1)$$

61 where A_i are cross-section areas, E_i are Young's moduli and their product represent the axial rigidity
 62 of tubular elements in tension and the internal element have radius R (Fig. S3a). N_c is the critical load
 63 for which the critical strain energy release rate G_a is reached. Eq.1 is applied to bodies #1, #2 in Fig.
 64 S3a where the body #2 is cylindrical and it could be extended to a modified configuration shown in
 65 Fig. S3b that resembles to the CMFs structure of D, D+GO and D+SiC.



66

67 **Figure S3.** a) Depiction of tubular adhesive joint subjected to axial load, simplified for a tube-cylinder
 68 circular and co-axial contact mediated by an adhesive film, in grey. b) Schematic representation of the
 69 cross-section of previous joint for the case of several identical aligned microfibrils with partial
 70 adhesive contact described by θ_i . A_i are the cross-sections of the CMFs surrounding the central one,
 71 they have an Elastic Modulus of E_i . This configuration is depicting D, D+GO and D+SiC sample.

72 D, D+GO and D+SiC have equivalent cellulose microstructures (Fig. 2a), thus the only structural
 73 differences can be found in the interface layer, where nanoparticles are located. The differences
 74 between the concentric tubular configuration presented in [9] and our systems lies in the area of
 75 contact between different CMFs that is partial but have multiple areas of contact and does not extend
 76 to the entire external surface as for the tubes (Fig. S3a,b).

77 We propose to estimate if the nanoparticles at the CMF interfaces generate an increase in critical
 78 strain energy release at the microscale, which could cause the macroscopic enhancements in the stress
 79 of failure and toughness modulus in the D+GO and D+SiC samples (Fig. 3). Passing from a discrete
 80 analytical to a local model requires to substitute the axial force N with a constant distributed pressure
 81 σ on the x -axis that is generated as a consequence of the uniaxial tensile load conditions. The two
 82 models are linked if a catastrophic failure of the samples is considered. In particular, the bulk material
 83 exhibits a stable crack propagation when the applied load σ reaches a critical load σ_c , which is
 84 computed as the tensile strength of the mechanical tests.

85 Taking into account these considerations, Eq.1 can be rewritten to better suit the configuration
 86 of CMF in cellulose nanocomposites. The partial contact between CMFs is depicted in Fig. S3b where
 87 the central fibril is in contact with multiple fibrils, each for an arc of $\theta_i R$. As the nanoparticles are
 88 filling vacancies inside the structure, the key in comparing different samples is that the contact area
 89 between microfibrils remains unchanged in D, D+GO, D+SiC, hence nanoparticles play a role only in
 90 the determination of fracture energy G_a . Comparing the results with the analytical case in [9], we can
 91 consider crack nucleation to be a statistical phenomenon since the problem is symmetrical. This
 92 scrutiny implies that two terms in the ratio between axial rigidities in Eq.1 can indifferently be
 93 inverted and thus the ratio can be considered equal to 1 for cellulose nanocomposites. Finally, the
 94 following correlation is obtained:

95
$$\sigma_c \propto \sqrt{\sum_i (\theta_i R) \cdot G_a \cdot \left[E_0 A_0 + \sum_i (E_i A_i) \right]} \quad (2)$$

96 Eq. 2 links the failure strength to the critical value of strain energy release rate (G_a), i.e. fracture
 97 energy of the interface, for which the crack propagates [8, 9]. E_i and E_0 are to be considered identical
 98 since the load bearing structure of the composite is the same, so they can be reduced to E and
 99 extracted from the summation. To compare different sample groups (D, D+GO, D+Si), a second index
 100 j is inserted to label the quantities dependent on the three groups.

101
$$\sigma_{c,j} \propto \sqrt{\sum_i (\theta_{i,j} R_j) \cdot \sum_i A_{i,j} \cdot \sqrt{E_j G_{a,j}}} \quad (3)$$

The first two terms in the product in Eq.3 are identical under the hypothesis of identical configuration and microstructure, therefore the relation can be rewritten as follows:

$$\sigma_{c,j} \propto \sqrt{E_j G_{a,j}} \quad (4)$$

where the tensile strength σ_c is a function of E_j , $G_{a,j}$, which are sample-related values. E_j was computed as the elastic modulus measured upon tensile testing for each sample. Then, the average improvements in interfacial fracture energy for the two set of nanoparticles can be computed:

$$\frac{\bar{G}_{a,D+GO}}{\bar{G}_{a,D}} = \frac{\bar{\sigma}_{c,D+GO}^2}{\bar{\sigma}_{c,D}^2} \cdot \frac{\bar{E}_D}{\bar{E}_{D+GO}} \quad (5a)$$

$$\frac{\bar{G}_{a,D+SiC}}{\bar{G}_{a,D}} = \frac{\bar{\sigma}_{c,D+SiC}^2}{\bar{\sigma}_{c,D}^2} \cdot \frac{\bar{E}_D}{\bar{E}_{D+SiC}} \quad (5b)$$

where the two ratios $\bar{G}_{a,D+GO}/\bar{G}_{a,D}$ and $\bar{G}_{a,D+SiC}/\bar{G}_{a,D}$ represent the average interfacial fracture energy improvements for the 2 sets of nanocomposite reed. They are linked with the average failure stress $\sigma_{c,j}$ and elastic modulus E_j of each set of samples. The results show that the intercalation of GO the interfacial fracture energy is improved by 28%, and by 47% after SiC intercalation.

| Interfacial fracture energy improvements | |
|---|-----------|
| $\frac{\bar{G}_{a,D+GO}}{\bar{G}_{a,D}}$ | 1.28±0.33 |
| $\frac{\bar{G}_{a,D+SiC}}{\bar{G}_{a,D}}$ | 1.47±0.36 |

Table S1. Interfacial fracture energy improvements for D+GO and D+SiC samples.

In summary, the estimation of the interfacial fracture energy improvements in Table S1 are in line with the results of the mechanical tests that show a higher tensile and flexural performance after the intercalation of SiC nanoparticles rather than GO. Finally, a linear correlation is obtained between elastic strain energy release rate and toughness modulus reported in Fig. 3f.

Note 4 - Video of the burning tests

Attached file: Video S1 - Burning tests.

List of Abbreviations

GO graphene oxide; SiC silicon carbide; R giant reed; TC thermo-compressed; D densified; D+GO densified + graphene oxide; D+SiC densified + silicon carbide; CMF cellulose microfibril; XPS X-ray photoelectron spectroscopy; DSC differential scanning calorimetry; TGA thermogravimetric analysis; EDX energy dispersive X-ray; SEM scanning electron microscope

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