



Article Numerical and Experimental Determination of the Wind Speed Value Causing Catastrophe of the Scissor Lift

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Abstract: The current work is devoted to the numerical determination of the wind speed value, which can cause the overturning of the mobile elevating work platform of the scissor lift type. In the first step of the analysis, the scaled model of the real vehicle is prepared. In the second step, the model is used in the aerodynamic tunnel to determine the aerodynamic force values and moment, which act on the vehicle. The three different configurations of the work platform are considered, namely: (a) The work platform raised to the maximum height with an additional bridge extended, (b) the work platform raised to the maximum height, and (c) the work platform half raised. In each position, the direction of the wind is changed from the range from 0° to 180° with an increment equal to 15°. In the next step of the analysis, the CFD simulations are carried out. The ANSYS Fluent R22 software is used. As a model of turbulent airflow, the standard k- ε with standard wall function is adopted. The obtained experimental results are used to verify the numerical model. A very good agreement between the results of the experiment and the results of numerical simulations is obtained. As the main result of the numerical study, the values of the tipping moment and corresponding wind speed that cause the overturning of the analyzed real scissor lift are determined. It occurred that the lowest value of the wind speed is obtained for the first variant of the vehicle configuration V_{crt}^1 = 22.315 m/s for the angle of the wind speed direction β = 30° and the highest one for the third variant V³_{crt} = 34.534 m/s and β = 15°, without any persons on the work platform. The presence of human beings on the work platform is also considered.

Keywords: scissor lift; wind load; CFD simulation; aerodynamic forces; tip-over

1. Introduction

Among the different kinds of mobile elevating work platforms (MEWP), the scissor lift seems to be very useful and effective in practical usage. However, as in the case of other devices of this kind, the workers are exposed to various hazards. As is reported in the available literature [1], according to the Census of Fatal Occupational Injuries, a Bureau of Labor Statistics database, in 1992–1999 the number of deaths of construction workers because of different accidents was equal to 339 of which 19% were different accidents with the use of scissor lifts. According to another source [2], namely, Census of Fatal Occupational Injuries, 306 fatalities between 1992 and 2003 were caused by the usage of the different kinds of aerial lifts, and 78 of these fatalities specifically involved scissor lifts. The causes of these accidents can be quite different [3], but it seems that the most dangerous is the tip-over of the whole mobile platform [4]. The overturning can be caused by the wind load [5,6]. According to media information, the issue of scissor lifts accidents and safety is described in [7–11].

It should be stressed here that according to the different standards (European Standard, Eurocode 1–4 [12], British Standard, BS 2573-1 [13], American Society of Civil Engineers, ASCE 7-16 [14], Japanese standard [15] or Chinese standard, GB/T 3811-2008 [16]), the load caused by the wind is treated as static load where the impact of the adjacent structures is



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Copyright: © 2023 by the authors. Licensee MDPI, Basel, Switzerland. This article is an open access article distributed under the terms and conditions of the Creative Commons Attribution (CC BY) license (https:// creativecommons.org/licenses/by/ 4.0/). omitted. In the case of the mobile elevating work platforms (MEWP), including scissor lifts, the appropriate design rules are defined in the standard EN 280-1:2022 [17].

However, such a simple approach can be inadequate in comparison with the wind load phenomenon, especially in the case of large-scale engineering structures like gantry cranes, tower cranes, or MEWP of a different kind. Therefore, the important aspect of the analysis of the wind load is the measurement of the wind load in the case of real structures [18–20]. Based on these results, the appropriate wind conditions can be imitated in the aerodynamic tunnels, where the scaled models of the real large-scale structures can be tested in order to determine the wind load (forces and moments coefficients) [21–23]. Together with the rapid development of computers and software, it is also possible to carry out numerical simulations of the airflow around crane-like structures [24–26]. Moreover, having the distribution of the static pressure (caused by wind) on the surface of the studied structure, it is possible to carry out further mechanical analysis based on the fluid-solid interaction [27].

Here, it should be noted that the papers which are devoted to the impact of the wind load on the scissor lifts are rather rare, thus the below brief survey of the literature also contains works, which consider the actions of the wind on other types of crane-like devices. However, the standard approach is to prepare the scaled model of the real large-scale structure for experimental tests in the aerodynamic tunnel. The obtained results are used to validate the numerical model [28].

It is worth noting that the wind load, especially the gust of wind, which is very often two times greater in comparison with the mean wind speed and a frequency of about 1 Hz [19], can also cause the vibration of structures. Chen and Li [29] studied the displacement of a lattice tower. The results of the nonlinear dynamic analysis show that the displacement obtained from the time-history analysis is higher by about 5–28% in comparison with the static one. Next, Takahashi et al. [30] studied the runaway of quayside container cranes subjected to transient gusty winds. They also proposed the sliding state while approaching the runaway and after a runaway caused by a wind gust. Chen et al. [31] studied the interference effect of the different segments of a tower crane subjected to wind load. The wind coefficients of a full-scale model of a tower crane were calculated by CFD, and then the time history of wind loads, simulated through the autoregressive method, was applied to the finite element model of a tower crane. Although the maximum wind load direction of the tower crane was perpendicular to its jib, the obtained results reveal that the maximum along-wind load direction was deflected 30° - 60° , and the mean ratio of the absolute value of the across-wind coefficient to the along-wind coefficient of the tower crane was about 8.56%. Su et al. [32] studied the effect of stochastic dynamic transient gusty winds on the sliding and overturning of quayside container cranes. The main conclusion is that dynamic transient gusty wind-induced peak response follows type III (Weibull) extreme value distribution. Azzi et al. [33] performed wind tunnel tests on an aeroelastic lattice tower model. The obtained results reveal the resonance contribution could reach a maximum of 18% of the peak response of the tower.

In 2022, several interesting works were published, which concern the problem of the impact of the wind load on crane-like structures. Lu et al. [34] took into consideration the problem of the outer-attached tower cranes installed on super high-rise buildings and exposed to wind-induced vibration. The numerical CFD simulations and structural FEM calculations were performed. He et al. [35] presented a study, which concerns a similar problem. To avoid sophisticated finite element analysis of the main building, a modified generalized flexural-shear model (FSM-MS) was proposed to estimate the along-wind and across-wind displacement response of the main building and the response at the connection support of the tower crane. Yeon et al. [36] studied the lift effect on wind load estimation for a semi-submersible rig. Wind loads on the analyzed rig were calculated under the maritime atmospheric boundary layer. The obtained results match well with those from the wind tunnel within a $\pm 20\%$ error.

The influence of the lifted load on the stability of crane-like devices has also been investigated. Here, two exemplary works can be quoted, namely Monteiro et al. [37] and Cekus et al. [38].

At the end of this brief survey of the literature, it is also worth mentioning the windinduced interference effect (IE). The analysis of this problem provides insights into how the neighborhood's other structures influence the studied structure by changing the surface pressure caused by the wind. However, the cases of crane-like devices are rarely found. Here, the work by Wu et al. [39] can be mentioned. The author of this work investigated the wind load and wind-induced dynamic response of three quayside container cranes. Other similar works concerning lattice structures refer to the mutual influence of a system of antennas, for example, Holmes et al. [40], Carril et al. [41], and Martín et al. [42]. The interference factor of microwave antenna dishes is found to be greater than one for some wind directions. The interferon effect has also been investigated for tall buildings [43], low-rise buildings [44–47], cooling towers [48,49], and scaffoldings [50].

This work should be treated as a continuation of the previous one [51]. The current work is devoted to the problem of determining the wind speed, which causes the tip-over of the scissor lift in the case when there are no workers on the platform and in the case when there are one, two, or more persons on the platform. The main aim of this effort is to increase the safety of the workers when the device is in service. Basically, the value of the moment induced by the wind, which causes a tip-over of the studied scissor lift, is determined based on the results obtained from the CFD simulation. However, the used numerical model is verified by the experimental tests performed in the aerodynamic tunnel, namely, the values of aerodynamic forces obtained from simulations and experiments are compared. Taking into consideration the geometrical dimensions of the real device, the scaled model of the scissor lift is made. Having verified the numerical model, the critical wind speed value can be determined for a real-scale scissor lift. Moreover, knowing the value of the mean aerodynamic force, which acts on the human being, the impact of the presence of the workers on the platform can also be estimated on the critical wind speed.

2. Materials and Methods

2.1. Object of Investigation

As it was mentioned above, the object of the current analysis, shown in Figure 1, is the scissor lift MEC 4191RT [52]. Table 1 presents the technical data and geometric dimensions of the actual structure.



Figure 1. The scissor lift MEC 4191RT: (a) isometric projection [52]; (b) real object of investigation [53].

Technical Data	Value
Platform Height [mm]	12,500
Stowed Height (Top Guardrail) [mm]	3000
Platform Entry Height [mm]	1880
Overall length [mm]	3660
Overall length with outriggers [mm]	4570
Width [mm]	2310
Platform length extended [mm]	4570
Platform length retracted [mm]	3350
Lift capacity [kg]	454
Roll-out deck capacity [kg]	227
Self-weight Q [kg]	4400
Maximal number of persons on the platform	4

Table 1. Selected technical data and geometrical dimensions of the scissor lift MEC 4191RT [52].

The MEC 4191RT mobile platform is adapted to work in difficult terrain. All four wheels are driven and steerable, equipped with off-road tires. The source of the traction drive is a liquid-cooled diesel engine with a power (Liquid-cooled, Kubota DF752 Dual Fuel or Kubota D1105 Diesel engine) of 8.7 kW. The scissor mechanism is built of two parallel, interconnected scissor kinematic systems, each of which consists of five members. The basic element of the scissor system consists of two double-armed levers (rectangular steel profiles) with arms of identical length connected in the middle with a pin. The members being kinematic pairs relate to each other by the ends of the two-arm levers and by means of pins.

The unfolding and folding of the scissor mechanism are carried out by means of two hydraulic cylinders, one of which is attached to the first pair and the other to the last pair. At the same time, both hydraulic cylinders are coupled to the middle pair. The entire scissor mechanism is connected to the chassis frame at the bottom and to the platform frame at the top. One arm of the lever is connected to the slider moving along the guides, and the other is connected to the frame through a pivot enabling rotation.

The working platform is made of metal. The floor of the working deck is made of special steel mesh, the balustrade is made of rectangular steel profiles. The platform has a manually extended side platform, which increases its surface by about 1/3.

2.2. Experimental Setup

For experimental research in the aerodynamic tunnel, the model of the investigated scissor lift with an approximate scale of 1:14 is prepared. Due to the specificity of the research, it is important to maintain the geometric dimensions of the model in relation to the real object, while some elements were simplified or omitted. Simplifications used in the model: (a) Arms of the scissor mechanism made of full profiles (originally an open rectangular profile), (b) connections of the scissor mechanism members made with the use of screws and bushings (originally pins), (c) stabilizing supports permanently integrated with the chassis (in the original it is possible to disassemble), (d) rigidly mounted road wheels, (e) elements of the platform and covers of the engine and hydraulic system made of 3 mm thick PVC.

Making the scissor mechanism from full profiles and the use of PVC with a thickness inconsistent with the scale of the model is reasonable because the overall dimensions have been preserved, and thus the reference surface, which is needed to determine the force and moment coefficients. Moreover, the whole described simplifications of the scaled model do not affect the surface in the orthogonal projection. The scaled model with the geometrical dimensions is shown in Figure 2.



Figure 2. The model of the scissor lift (approximate scale 1:14): (**a**) The general geometrical dimension of the model, (**b**) the model in isometric projection.

The experimental tests are carried out for three configurations of the scissor lift, namely:

- configuration 1, where the work platform of the model is positioned 891 mm from the floor level (height measured to the top guardrail, as is shown in Figure 2 with extended roll-out deck),
- configuration 2, the configuration of the scissor lifts the same as in the case of variant 1, but the roll-out deck is retracted, and finally,
- configuration 3, where the work platform is positioned 450 mm from the tunnel floor level to the top guardrail with retracted roll-out deck.

In all mentioned cases, it is assumed that the wind direction varies from 0° to 180° with an increment equal to 15° . At this stage of analysis, the presence of workers on the platform is not considered. The impact of the presence of people will be studied further.

The assumed coordinate system and the way the angle of wind direction is defined are shown in Figure 3. For the tested model of the scissor lift, the aerodynamic moment M_z and aerodynamic forces F_x and F_y are measured using the three-component aerodynamic balance based on the electric resistance strain gauges [54]. The orientation of the fixed coordinate system x, y, z is as follows: x—along wind direction, y—across wind direction, z—vertical direction.

The sensitivity of the aerodynamic balance in the range of 0 to 2 N in increments of 0.1 N is 89.4%, in the range of 2 to 10 N in increments of 2 N is 92.9%, and in the range of 10 to 50 N in increments of 10 N is 99.7% [54]. A similar topic, including research in a wind tunnel, was described in [55–57].

2.3. Aerodynamic Tunnel

The experimental tests were carried out in the boundary layer wind tunnel at the Wind Engineering Laboratory at the Cracow University of Technology [58]. The wind tunnel, with a length of 10 m and a measurement area of 2.2×1.4 m, allows for conducting tests in both closed and open circuits. The wind flow was generated by an axial fan with a diameter of 2.7 m and an efficiency of 0.8–0.9, and a blade tip speed of about 100 m/s. The fan was driven by a 200-kW alternating current motor with a nominal speed of 750 rpm, controlled

by an inverter. The maximum wind speed was approximately 40 m/s (144 km/h). Figures 4 and 5 show the wind tunnel and CFD simulation with the study models in the 1st and 3rd configurations for the selected angle of wind attack. Figure 4a shows the mast, where the two wind speed sensors are installed: One of them 0.5 m and the second 0.92 m above the tunnel floor. The intensity of the turbulence was controlled by a system of spires about 0.8 m high and 0.3 m wide (Figure 4).



Figure 3. The orientation of the model in the coordinate system *x*, *y*, *z*; and the aerodynamic test conditions: The assumed wind direction *W*; the angle of diverting the wind β ; the aerodynamic moment M_z and aerodynamic forces F_x and F_y .



Figure 4. Model of the scissor lift at maximum working height with extended roll-out deck (conf. 1): (a) During the aerodynamic tests, (b) the CFD simulation.



Figure 5. Model of the scissor lift at medium working height (conf. 3): (**a**) During the aerodynamic tests, (**b**) the CFD simulation.

2.4. Measurement of the Force Components Acting on the Model and Corresponding Wind Speed

The schema of the applied measuring system is presented in Figure 6. The model of the scissor lifts 1 was installed on the aerodynamic balance 2. A, B—special segments with strain gauges. It is the device that enables the determination of components F_x and F_y of the aerodynamic forces as well as the moment M_z with respect to the coordinate system shown in Figure 3. The aerodynamic balance relates to the electronic system 3 controlling the stepper motor of the turntable and with the strain gauge bridge 4 measuring the aerodynamic forces. The wind speed was evaluated via two thermo-anemometer probes 5 and 6, which were connected to the thermo-anemometer ANT 2000 7. Aerodynamic forces and wind speed measurements were recorded via a PCI 1710 card 8 on a PC 9.



Figure 6. The schema of the measuring system (1—the scissor lift, 2—the aerodynamic balance, 3—the electronic system, 4—the stepper motor, 5 and 6—the thermo-anemometer probes, 7—the thermo-anemometer, 8—PCI 1710 card, 9-PC, and A, B—the segments with strain gauges).

The sampling time of the wind speed is equal to 5000 ms. The exemplary graph of the wind speed variations measured via sensors 5 and 6 is presented in Figure 7. Here, it should be noted that the average value of the wind speed measured by sensor 6 is about 15% lower in comparison with the value indicated by sensor 5. Moreover, based on the graph in Figure 7, the average intensity of turbulence I_v can also be evaluated according to the following formula:

$$I_v = \frac{\sigma_v}{V_{ref}},\tag{1}$$

where σ_v is the standard deviation of the measuring wind speed and V_{ref} is the reference speed, which is equal:

$$p_{ref} = \sqrt{\frac{2 \cdot p_{ref}}{\rho}},\tag{2}$$

where p_{ref} is the reference pressure. The standard deviation can be computed based on the dynamic component $v_{dyn}(t)$ of the instantaneous velocity V(t), where $V(t) = V_{ref} + v_{dyn}(t)$.



Figure 7. Wind speed measured with two anemometers placed at different heights.

The wind speed, which was measured at the point localized 500 mm over the floor of the aerodynamic tunnel, was generally lower in comparison with the wind speed measured at 920 mm over the floor. It was caused by the set of "spires", which were installed at the beginning of the tunnel in order to induce the initial turbulence. The spires were much wider at the basis in comparison with their summits. Therefore, the wind speed was much more disturbed at the level of 500 mm and, in consequence, the average wind speed was lower.

The formulas determining the coefficient of aerodynamic drag C_x , the coefficient of the lateral aerodynamic force C_y , the aerodynamic coefficient of torque C_{Mz} , and the aerodynamic coefficient of the moment, which caused the tip-over M_C are as follows:

$$C_{x} = \frac{2F_{x}}{\rho V^{2} \cdot A_{ref}}, \ C_{y} = \frac{2F_{y}}{\rho V^{2} \cdot A_{ref}}, \ C_{Mz} = \frac{2M_{z}}{\rho V^{2} \cdot A_{ref} \cdot r_{ref}}, \ C_{Mc} = \frac{2M_{C}}{\rho V^{2} \cdot A_{ref} \cdot h_{ref}}$$
(3)

where: F_x —aerodynamic drag N, F_y —lateral force respectively N, V—the average wind speed m/s, ρ —the mass air density, A_{ref} —effective area of one of the supporting structures of the model, i.e., the area of the shadow normal projected by its members on a plane parallel to the wall equal 0.051 m², r_{ref} —the reference dimension adopted by convention, the width of the structure of the model equal 0.116 m and h_{ref} is half of the total height of the model equal to 0.446 m.

2.5. CFD Simulations

The airflow simulations were carried out with the use of the Fluent software in the ANSYS Workbench R22 environment (Figure 8). The main aim of these simulations was to determine the aerodynamic forces and moments acting on the studied scissor lift, more specifically, force, and moment coefficients. These quantities were defined by Equation (3).



Figure 8. The CAD model of the studied scissor lift created with the use ANSYS Design Modeler R22.

In the performed simulations, we assumed standard air properties, therefore, $\rho = 1.225 \text{ kg/m}^3$, T = 15 °C, $p_0 = 101,325.25 \text{ Pa}$. Moreover, the model of the scissor lift was immersed in a rectangular space filled with air with external dimensions $2.2 \times 1.4 \text{ m}$ and a length of 2 m, which corresponded to the dimensions of the wind tunnel. The tunnel walls were modeled as stationary boundaries.

One of the most important problems was the appropriate choice of the turbulent model. In the current work, we considered three most frequently applied in practice models, namely: Reynolds stress model, model k- ω , and model k- ε . The first mentioned model, used by Wu et al. [39], caused the computations to converge unacceptably slowly. The second model, used by He et al. [35], Yeon et al. [36], and Monteiro et al. [37], requires preparing the specific mesh at the boundary conditions. For estimation made for the studied structure, the first layer of the finite volume should be about 7×10^{-5} m in height. Unfortunately, it leads to an enormous number of nodes and finite element volumes.

Finally, the standard k- ε model with a standard wall function has been used. It is worth noting that this model is still in use, for example, the works of Zan et al. [26], Chen et al. [31], Lu et al. [34]. The application of this model caused the computations are converging relatively quickly. Obtaining the solution demands less than 100 iterations.

3. Results

To validate the numerical model, the obtained computation values of C_x and C_y are compared with those, which are determined as the results of experimental tests (Tables 2–4).

Angle β [°]	F_x [N]	<i>F</i> _y [N]	<i>M</i> _z [Nm]	V [m/s]	<i>I</i> _v [%]
0	3.682	0.099	-0.099	8.362	6.79
15	4.062	-0.580	-0.060	8.325	4.79
30	4.321	-1.360	-0.032	8.325	4,79
45	4.058	-1.761	-0.002	8.399	4.92
60	3.389	-1.743	-0.027	8.186	7.88
75	2.492	-1.258	-0.023	8.371	5.58
90	1.561	0.232	-0.001	8.482	6.52
105	2.049	1.841	-0.029	8.381	5.50
120	2.951	2.257	-0.002	8.270	8.59
135	3.523	2.191	0.010	8.399	5.14
150	3.939	1.765	0.030	8.094	5.82
165	3.927	0.950	0.074	8.223	5.61
180	3.794	0.298	0.104	8.371	4.93

Table 2. Experimental results obtained for configuration 1.

Table 3. Experimental results obtained for configuration 2.

Angle β [°]	F_x [N]	<i>F</i> _y [N]	M_z [Nm]	<i>V</i> [m/s]	<i>I</i> _v [%]
0	3.564	0.012	-0.067	8.605	5.06
15	3.931	-0.620	-0.036	8.448	4.56
30	4.220	-1.398	-0.002	8.396	5.27
45	3.934	-1.769	0.004	8.417	4.10
60	3.315	-1.733	-0.033	8.185	5.65
75	2.348	-1.175	-0.018	8.280	6.72
90	1.557	0.252	0.000	8.285	5.82
105	2.054	1.657	-0.003	8.378	5.38
120	2.928	2.230	-0.026	8.465	4.32
135	3.461	2.131	-0.007	8.476	4.93
150	3.857	1.779	0.022	8.093	5.82
165	3.780	1.004	0.021	8.120	7.06
180	3.732	0.353	0.008	8.461	5.04

Table 4. Experimental results obtained for configuration 3.

Angle β [°]	F_x [N]	<i>F</i> _y [N]	M_z [Nm]	<i>V</i> [m/s]	<i>I</i> _v [%]
0	3.279	-0.057	-0.018	8.094	8.55
15	3.357	-0.558	0.034	7.752	8.37
30	3.391	-1.240	0.063	8.057	8.39
45	3.117	-1.594	0.087	8.168	8.21
60	2.339	-1.482	0.067	8.205	9.14
75	1.646	-0.981	0.019	7.086	8.43
90	1.172	0.061	0.000	7.992	7.66
105	1.390	0.806	-0.033	8.371	6.48
120	2.024	1.423	-0.049	7.779	7.56
135	2.741	1.478	-0.044	8.159	7.28
150	3.130	1.052	-0.041	8.029	7.06
165	3.207	0.532	-0.016	7.844	7.07
180	3.179	-0.004	0.026	7.752	8.04

3.1. Results of the Experimental Tests

Tables 2–4 show the collected results of the experimental tests, which are obtained in the case of configurations 1st, 2nd, and 3rd, respectively. In the 5th column, the average wind speed is computed based on indications obtained from sensors 5 and 6. As can be observed, the highest values of the F_x component of the aerodynamic force are obtained for angle β equal to 30° and 150° for all configurations. The highest values of the F_y component are for the angle β from the range of 45° to 60° and, respectively, 120° to 135°. The average

turbulence computed for all cases is equal to I_v = 6.32%. The average wind speed is equal to 8.355, 8.322, 7.945 m/s, for configurations 1st, 2nd, and 3rd, respectively.

3.2. Results of the Numerical Simulations

The values of the forces and moments induced by the wind, which act on the investigated scissor lift, are estimated based on the distribution of the static pressure. Mentioned distribution is obtained from CFD simulation. The exemplary distribution of the static pressure is depicted in Figure 9.



Figure 9. The exemplary distribution of the static pressure ($\beta = 30^{\circ}$).

The impact on the values of the forces and moment coefficients are analyzed for different sizes of the volume cells. Table 5 shows the values of the force and moment coefficients, which are obtained for different approximate cell sizes. The finite volume mesh creates the cells of the tetrahedron shape of variable edge length. It is assumed that on the surface of the studied structure, the length of the cell edge is about 1.5, 2.0, ..., 3 mm while the length of the edge cells on the channel walls, inlet, and outlet boundaries is equal to about 70 mm. The computations are performed for variant 1, where it is assumed turbulence intensity $I_V = 6.32\%$, the hydraulic diameter $D_H = 1.712$ m, wind speed V = 7.872 [m/s], and angle $\beta = 30^\circ$.

Table 5. The force and moment coefficients for different finite volume mesh.

Cell Size [mm]	Nodes	Cells	C_x	Cy	C_{Mc}	
1.5	2,186,819	1,330,171	2.202	-0.725	-2.299	
2.0	1,172,976	717,266	2.206	-0.718	-2.271	
2.5	1,068,763	663,681	2.208	-0.740	-2.296	
3.0	775,510	483,151	2.185	-0.708	-2.262	
$a = 1.225 \text{ kg}/\text{m}^3 \text{ s}_1 = 7.872 \text{ m/s}_2$						

 $\rho = 1.225 \text{ kg/m}^3$, v = 7.872 m/s.

As can be observed, the obtained numerical solution shows a good global convergence, thus, for further computations, we assumed the following mesh sizes: 1.5 mm on the surface of the scissor lift and 70 mm on the rest of the external boundaries.

The influence of the turbulence intensity on the force and moment values and moment coefficients is also investigated. As is reported in Tables 2–4, the estimated value of the turbulence intensity varies in the range from 4% to 10%. The computations are performed for the initial parameters the same as in the previous convergence analysis and for the finally assumed mesh sizes. The obtained results are shown in Table 6. As can be observed, together with the increase in the turbulent intensity, the values of the analyzed coefficients also increase slightly. Therefore, it seems that the choice of the turbulent intensity equal to $I_V = 6.32\%$ seems to be reasonable.

Turbulent Intensity I _v [%]	C_x	C_y	C_M
4	2.137	-0.711	-2.226
7	2.220	-0.731	-2.319
10	2.301	-0.750	-2.409

Table 6. The force and moment coefficients for different turbulent intensity.

 $\rho = 1.225 \text{ kg/m}^3$, v = 7.872 m/s.

3.3. Comparison of the Experimental and Numerical Results

As can be observed in Figures 10–12, the comparison of the results obtained from experimental tests and numerical simulations reveals a very good agreement. In the case of the C_x force coefficients, the average error is defined as follows:

$$\varepsilon_{AVG} = \left| \frac{C_X^{exp} - C_X^{num}}{C_X^{num}} \right|,\tag{4}$$

where C_X^{exp} , C_X^{num} are the force coefficients obtained from the experiment and numerical computations, respectively. For variant 1 the average error is equal to $\varepsilon_{AVG}^1 = 8.177\%$, and for variants 2 and 3 is equal to $\varepsilon_{AVG}^2 = 8.982\%$ and $\varepsilon_{AVG}^3 = 8.055\%$. It seems that the results of the numerical simulation slightly overestimate the value of the aerodynamic force F_x , which plays an important role when the tip-over of the scissor lift could happen. Moreover, the best match is obtained for configuration 3 of the investigated device. Finally, it is worth noting that in the case of the force coefficient C_y , the results from experimental tests and numerical simulations seem to be an even better match than in the case of C_x force coefficients.



Figure 10. Comparison of the force coefficient C_x and C_y for the experimental and numerical results as function of angle of wind attack β in the case of configuration 1.



Figure 11. Comparison of the force coefficient C_x and C_y for the experimental and numerical results as function of angle of wind attack β in the case of configuration 2.



Figure 12. Comparison of the force coefficient C_x and C_y for the experimental and numerical results as function of angle of wind attack β at the case of configuration 3.

4. Discussion

4.1. Calculation of Overturning and Stabilizing Moments

The maximum overturning M_O and corresponding stabilizing moments M_S shall be calculated for the most unfavorable tipping lines. In this case, tipping line is between supports of the scissor lift. For solid and foam-filled tires, according to ISO 4305, the tipping lines determined may be taken at 1/4 of the tire ground contact width from the outside of the ground contact width.

The calculations are made with the scissor lift in the most unfavorable extended (configuration 1) with the maximum allowable inclination of the chassis equal to 0.5° , as shown in Figure 11. We assume that the scissor lift used out-of-doors is being affected by wind at a pressure of 100 N/m², equivalent to a wind speed of 12.5 m/s (Beaufort Scale 6) according to EN 280-1:2022. All loads and forces which can act simultaneously are considered in their most unfavorable combinations according to the graphical method for on-slope stationary example presented in Figure 13.

The platform is stable when the condition is met:

$$M_S > M_O \tag{5}$$

where: M_S is the sum stabilizing moments, M_O is the sum overturning moments. The stabilizing moments, according to the standard EN 280-1:2022 (and level ground stationary), include:

$$M_S = M_{S1} + M_{S2}.$$
 (6)

 M_{S1} —is the moment resulting from the product of the structural load S = 43,164 N acting on the arm r_1 equal to 1.1 m from the tipping line. *S*—vertical direction. The self-weight and all technical data of a real object are shown in Table 3. The masses of the components of the scissor lift are taken to be static structural loads (not moving), Table 7—factor equal to 1.

 $\frac{M \times 1.1}{r_2}$ $\frac{S \times 1}{r_1}$ $\frac{S \times 1}{r_2}$ $\frac{S \times 1}{r_1}$ $\frac{S \times 1}{r_2}$ $\frac{S \times 1}{r_1}$ $\frac{S \times 1}{r_2}$ $\frac{S \times 1}{r_1}$

Figure 13. The schema of the load acting on real object scissor lift for level ground stationary example according to [17].

Table 7. The load and force directions for stabilit	y calculations for scissor lift [17]
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Working	80 kg Up to	Rated Load	Structural	Loads (S)	Manual l	Force (M)	Wind (V	Load V)
Condition	\times 1.0	\times 0.1	\times 1.0	$\times 0.1$	\times 1.0	$\times 0.1$	\times 1.0	$\times 0.1$
level ground stationary	V	-	V	-	А	А	Н	Н

Key: V = vertical; H = horizontal, A = angular, S = at slope angle.

 M_{S2} = 181.29 Nm—is the moment resulting from the product of the person load equal to 784.8 N acting on the arm r_3 equal to 0.231 m.

To the overturning moments include:

$$M_{\rm O} = M_{\rm O1} + M_{\rm O2} \tag{7}$$

 $M_{O1} = 7497.73$ Nm—is the moment resulting from the product of the wind loads W = 956.65 N acting on the arm h_1 equal to 7.125 m, W—horizontal direction. If the scissor lift is being affected by wind at a pressure of 100 N/m², equivalent to a wind speed of 12.5 m/s that the wind load acting on the scissor lift at the full area of the real object is A = 9.996 m² is equal to W = 956.65 N. Wind forces are assumed to act horizontally at the center of the area of the parts of the scissor lift and persons and equipment on the work platform h_1 . Wind load W is taken to be dynamic forces with factor 1.1 (Table 7).

 $M_{O2} = 5593.28$ N—is the moment resulting from the product of the manual force dynamic M = 400 N acting on the arm r_2 equal to 12.71 m. The minimum value for the manual force M is taken as 400 N for the scissor lift designed to carry more than one person (Table 1), applied at the height of 1.1 m above the work platform floor. The manual force—dynamic M is taken to be dynamic forces with factor 1.1 (Table 7).

Taking the above data into account, condition (5) is as follows:

$$M_{\rm S} = 47,679.82 \,\,{\rm Nm} > M_{\rm O} = 13,091.01 \,\,{\rm Nm},$$
 (8)

and conclude that the condition for the stability of the scissor lift is satisfied.

4.2. Calculation of Overturning and Stabilizing Moments with Used to Aerodynamic Coefficients Obtained from Experiment and CFD

The obtained values of aerodynamic coefficients from the experiment and from the CFD simulation are used to determine forces F_X , and F_Y acting relative to the tipping line of the scissor lift of the real object of investigation.

$$F_X = F_x \cos(\beta), \quad F_Y = F_y \sin(\beta), \tag{9}$$

Next, we determined the reduced value of force F_{rd} and moment M_{rd} acting relative to the tipping line of the scissor lift as a function of the angle of wind attack β .

$$F_{rd} = \sqrt{(F_X)^2 + (F_Y)^2}, \quad M_{rd} = \sqrt{(M_{F_{rd}})^2 + (M_z)^2}, \quad M_{F_{rd}} = F_{rd} \cdot h_1.$$
 (10)

The moment M_{rd} is taken to be dynamic moment with factor 1.1.

Finally, the sum overturning moments from experiment M_{Oexp} and CFD simulation $M_{O CFD}$ due to wind direction was determined:

$$M_{Oexp} = M_{rd\ exp} + M_{O2}, \qquad M_{O\ CFD} = M_{rd\ CFD} + M_{O2},$$
 (11)

and the comparison of the results for three different wind speeds acting on the scissor lift is shown in Figures 14–16.



Figure 14. Comparison of the overturning and stabilizing moments received from the experimental and numerical tests with the calculation based on [17] for the wind speed V = 12.5 m/s (config. 1).

Taking the above discussed results into account, the characteristics of the overturning moment as a function of wind speed for the angle of 15° can be determined (Figure 17).

According to Figure 17, the scissor lift will be overturned at approximately 21.95 m/s (76.825 km/h). Note, however, that according to the standard and manufacturer's data, the maximum wind speed is 12.5 m/s (45 km/h).



Figure 15. Comparison of the overturning and stabilizing moments received from the experimental and numerical tests with the calculation based on [17] for the wind speed V = 20 m/s (config. 1).



Figure 16. Comparison of the overturning and stabilizing moments received from the experimental and numerical tests with the calculation based on [17] for the wind speed V = 25 m/s (config. 1).



Figure 17. Comparison of the overturning and stabilizing moments received from the experimental and numerical tests for 15° angle of wind attack with the calculation based on standard EN280 for the wind speed V = 12.5-25 m/s.

4.3. Calculation of Overturning and Stabilizing Moments with Used to Aerodynamic Coefficients Obtained CFD

Figure 18a shows the values of the moment coefficients C_{Mc} , which could cause the overturning of the investigated scissor lift as a function of the wind direction described by angle β together with the assumed schema of the acting loads.



Figure 18. Calculation based on the CFD (**a**) the moment coefficient C_{Mc} as a function of the wind direction described by angle β , (**b**) the schema of the loads acting on the investigated scissor lift.

As can be observed, the most dangerous situation is when the wind blows along the direction, which creates the angle $\beta = 30^{\circ}$ with the X-axis of the coordinate system shown in Figure 3 for configurations 1st and 2nd. In the case of configuration 3rd, the most dangerous wind direction is equal to $\beta = 15^{\circ}$. It is worth noting that the extended movable deck has no significant impact on the moment M induced by the wind. The maximal values of the coefficients are as follows $C_{Mc}^1 = 2.299$, $C_{Mc}^2 = 2.212$, and $C_{Mc}^3 = 1.920$ for configurations 1st, 2nd, and 3rd, respectively.

To estimate the wind speed, which could cause the tip over the investigated scissor lift, it is assumed that $M_S < M_C$, where M_S is the static moment generated by the self-weight of the structure, Figure 18b. This moment can be computed as follows:

$$M_S = S \cdot r_1, \tag{12}$$

where *S* = 44 kN is the self-weight of the scissor lift (Table 1) and r_1 = 2.31 m, according to the technical properties of the scissor lift presented in Table 1. Next, assuming the density of the air is equal to $\rho = 1.225 \text{ kg/m}^3$, the critical wind speed can be determined as:

$$V_{crt} = \sqrt{\frac{2 \cdot S \cdot r_1}{C_{Mc} \cdot \rho \cdot A_{ref} \cdot h_{ref}}},$$
(13)

where $A_{ref} = 9.996 \text{ m}^2$ and $h_{ref} = 7.25 \text{ m}$ for the configuration 1st, 2nd and $h_{ref} = 3.625 \text{ m}$ for the configuration 3rd for the real structure. Finally, the critical values of the wind speed for studied variants of the scissor lift are as follows: $V_{crt}^1 = 22.315 \text{ m/s}$ (80.337 km/h), $V_{crt}^2 = 22.750 \text{ m/s}$ (81.902 km/h), and $V_{crt}^3 = 34.534 \text{ m/s}$ (124,322 km/h). The obtained values seem to be relatively high, but the main impact on the critical values of the wind speed possess the presence of the workers on the platform.

The estimation of the impact of the presence of human beings on the work platform on the critical wind speed should be considered quite problematic. First, the people who stand on the platform of the scissor lift are not "permanently installed" on the floor. Therefore, they can transfer the load induced by wind only when they lean on the guardrails. Moreover, the highest load induced by the silhouette of a man is when the people are positioned facing the wind direction. Besides, people differ from each other in height, silhouette, and weight. Thus, there are an almost infinite number of possibilities. Therefore, it seems that the most reasonable approach to estimating the impact of the presence of people on the work platform of the scissor lifts on the critical wind speed is to involve the superposition principles. In the available literature, one can find papers where the authors try to determine the drag force acting on the human body induced by wind, for example, Gjeta et al. [59], Koo et al. [60], Thomas [61], or Hunt et al. [62].

The authors of the first-mentioned work [59] performed several CFD simulations to estimate the drag force induced by the human body. Calculations are made for a person of medium build, 172 cm tall. The results of their simulations are reprinted in Table 8.

The Position of the Human Body against the Wind	Wind Speed [m/s]	Drag Force F _w [N]	Coefficient C
	20	125.956	0.900
Frontal position	30	272.009	0.857
	40	479.672	0.850
	20	69.4320	0.840
Lateral position	30	155.612	0.837
	40	277.650	0.840

Table 8. The values of the drag force acting on the human body according to [59].

To estimate the impact of the presence of workers on the platform on the value of the critical wind speed, in the first step, the moment M_C^{1} induced by the wind at the speed of $V_0 = 20 \text{ m/s}$ acting on the scissor lift is computed. The assumed density of the air is equal to $\rho = 1.225 \text{ kg/m}^{3}$. Next, the moment M_W generated by a single worker of the approximate weight $G_W = 800 \text{ N}$ is evaluated, namely $M_W = F_W h_2$, where F_W is the drag force, whose value can be found in Table 8 for the wind speed V₀ (frontal position), and the $h_1 = 12.5 \text{ m}$ is the distance between the ground and the top guardrail level for configuration 1st, 2nd and $h_1 = 7.25 \text{ m}$ for configuration 3rd. Finally, the total moment M_C induced by the wind blowing with a speed equal to V₀ can be computed with the use of the following formula:

$$M_C = \frac{1}{2} \cdot C_{Mc} \cdot \rho \cdot V_0^2 \cdot A_{ref} \cdot H_{ref} + n_W \cdot M_w, \tag{14}$$

where n_w is the number of workers on the platform. According to [51], the reference area of the human body is equal to about 0.555 m². Therefore, it can be neglected. Now, the coefficient C_T of the total moment M_T induced by the wind can be computed again according to Equation (3). The presence of the people on the platform also increases the total weight of the structure:

$$S_T = S + n_w \cdot G_W. \tag{15}$$

Finally, the values of the critical wind speed causing the tip-over of the investigated scissor lift with the various number of workers on the platform can be determined with the use of Equation (6). Table 9 presents the appropriate values of the critical wind speed for studied variants of scissor lift configurations as a function of the number of workers.

Table 9. The critical value of the wind speed V_{CRT} [m/s].

Number of Workers n_w	Configuration 1st	Configuration 2nd	Configuration 3rd
1	22.096	22.510	33.961
2	21.889	22.284	33.412
3	21.696	22.073	32.917
4	21.513	21.874	32.459

In the case of configurations 1st and 2nd, the presence of a single worker does not reduce the critical wind speed dramatically. However, the maximal number of persons on the platform causes the value of the critical wind speed to be lower in comparison with the value for the empty platform, about 7 m/s (2 km/h). In the case of configuration 3rd, the reduction of the value of the critical wind speed is the greatest, about 18 m/s (6 km/h).

5. Conclusions

The current work concerns the problem of the critical wind speed determination, which can cause the tip-over of the scissor lift. The analysis is performed for the device MEC 4191RT. The estimation of the critical value of the wind speed is based on results obtained from the CFD analysis of the scaled model (assumed approximate scale of the model 1:14) of the scissor lift. The computations are performed with the use of ANSYS Fluent R22. Three different configurations of the scissor list at analyzed, namely: (a) The work platform raised to the maximum height with an additional bridge extended, (b) the work platform raised to the maximum height, and (c) the work platform half raised. The numerical model is verified by the experimental test in the wind tunnel. To verify the CFD results, the values of the F_x and F_y components of the aerodynamic forces are compared, which are obtained from numerical simulations and experimental tests. A relatively good agreement is observed. In the case of the force coefficient C_x , the value of the average error does not exceed 9% for all investigated configurations. It is established that the most dangerous wind direction is the one that makes an angle of 30° with the X-axis of the global coordinate system, Figure 3, for all studied configurations of the scissor lift. The lowest critical wind speed is obtained for configuration 1st and is equal to $V_{CRT}^1 = 22.315$ m/s. For other configurations, the critical wind speed is higher and equal to $V_{CRT}^2 = 22.75 \text{ m/s}$, and V_{CRT}^{3} = 33.534 m/s, respectively. The impact of the workers on the platform of the scissor lift on the critical wind speed is estimated based on the values of the drag force of the human body, which are available in the literature. The presence of the workers decreases the critical wind speed and, in the case, when the four persons are on the platform, the critical wind speed is equal to $V_{CRT}^1 = 21.513 \text{ m/s}, V_{CRT}^2 = 21.874 \text{ m/s}, \text{ and } V_{CRT}^3 = 32.459 \text{ m/s}$ for the configurations 1st, 2nd, and 3rd, respectively. The obtained results are in good agreement with these, which are obtained based on code EN 280-1:2022.

The maximum wind speed that the standard and the manufacturer specify of 12.5 m/s gives a large safety margin, as shown in Figure 14. In this case, the capsizing moment values obtained from experimental and CFD studies do not differ significantly from the values calculated from the standard (the graph is flattened). The standard gives moment values regardless of wind direction. However, it should be borne in mind that the wind during a sudden change in weather very quickly increases in strength and the terrain and buildings cause turbulence or gust conditions. Currently, every summer in our climatic conditions, there are weather anomalies, where the determination of the wind speed is impossible, much less the direction from which it will hit this supporting structure, which is the scissor lift. It is worth referring to charts in Figures 14–16, which describe for which ranges of wind angle of attack the highest values of capsizing moment occur. The proposed system is unable to determine the timing of the speed increase from 12.5 m/s to higher speeds, such as 21.95 m/s. This is the speed that, for an angle of 15° , can add a corresponding moment to the sum of overturning moments, causing a loss of stability in the structure. The time given by the manufacturer that the operator needs to lower the platform from the maximum height is 50 s. The open question is if a structure of this type (operating outdoors in the open air) should not be equipped with a system for continuous measurement of wind speed and wind direction.

In the future, it is planned to study the dynamic response of this structure to strong gusts of wind. Moreover, it will take into consideration the so-called interference effect, which is connected with the existence of other objects (buildings, other crane devices, trees, etc.) which are localized close to the investigated device.

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