



Article Analysis of Trajectory Tracking Characteristics of a Magnetically Driven Oil-Free Scroll Compressor

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Abstract: The conventional scroll compressor cannot run oil-free because of wear and tear and lubrication problems during operation due to some parts, such as anti-rotation devices. The magnetic drive oil-free scroll compressor (MDOFSC) uses a contactless drive method to avoid this drawback. In order to solve the swing problem of the orbiting scroll during the operation of the MDOFSC, decentralized control and centralized control are used to study the trajectory tracking characteristics. Firstly, the structure and working principle of the MDOFSC are introduced, and the system's magnetic circuit and differential control principle are analyzed. Then, the dynamic model of the MDOFSC under the condition of non-compressed gas is established, and the coordinate matrix decoupling method is used to analyze the relationship between the degree of freedom of the system and the measurement distance of the displacement sensor. Finally, the system is simulated and experimentally studied under centralized PID control, and the experimental comparison study between decentralized control and centralized control is conducted. The results show that centralized control dramatically improves the trajectory control ability of the system.

Keywords: oil-free scroll compressor; magnetic drive; centralized control; matrix decoupling; PID control; experimental study

1. Introduction

Scroll compressors have many unique advantages, such as a simple structure, low noise, high efficiency, and good reliability. As the application of scroll compressors extends to medicine, food, and other fields requiring clean compressed gas, developing an oil-free scroll compressor has essential practical significance [1,2].

Li proposed a sealing mechanism model to achieve sealing of scroll tooth axial clearance by installing self-lubricating material sealing strips and springs in the sealing groove opened on the tooth end face [3]. In existing oil-free scroll compressors, the first type uses self-lubricating bearings instead of oil-lubricated bearings; the second type coats a layer of self-lubricating material on the surface of the anti-rotation device and the contact surface of the moving and stationary scroll. Although the two methods achieve the goal of being oil-free, there are some problems with friction and heating. Sun proposed an oil-free scroll compressor with a solenoid instead of an anti-rotation device and no wearing parts. Only modeling and simulation were performed in the article, and no experimental study was conducted [4,5]. Magnetic levitation technology, with its fast response, high accuracy, and clean and pollution-free features, provides new solutions for several industries and has been widely used under the intensive research of many scholars, with typical applications such as magnetic levitation bearings [6,7], magnetic levitation trains [8], and magnetic levitation planar motors [9]. Zhao proposed a new magnetic levitation system with two rotating magnets that can achieve a zero-power horizontal levitation state under eccentric loading [10]. Soni investigated the dynamical behavior and stability of energy-efficient electromagnetic levitation with time-lag rotors.



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The control formulas were transformed to the frequency domain, the time lag values of the feedback signals were modeled using the Pade approximation, and stability analysis was performed by analyzing the system poles [11]. Zhang X provided a six-degreeof-freedom levitation platform. This study manipulated the levitation carrier completely without a tether in a large area and calculated the wrench matrix quickly and accurately using the magnetic node method and the Lorentz force law [12]. Graphene levitation and magnetic field orientation control were studied in the literature [13]. Torques induced by antimagnetic forces in all three spatial directions were used to predict stability conditions for millimeter-scale graphite plates of different shapes. Article [14] introduces the design principle, and the initial model of radial-axial magnetic bearing proposes a multi-objective optimization method of bearing and provides the analytical expression of the equivalent magnetic circuit, verified by finite element analysis and experiments. For the problem that the magnetic saturation of the core reduces the maximum output force of the magnetic bearing, an improved core design criterion for the cross-polar magnetic bearings is proposed in the paper [15]. The three-dimensional finite element method and experiments also successfully verified the optimization method. The method can provide a reference for designing and selecting active magnetic bearings. In order to predict the performance and guide the design of permanent magnet-biased three-degree-of-freedom magnetic levitation bearings, Yun proposed an accurate analytical model including the eddy current effect and leakage effect. The stiffness was investigated by static and transient finite element methods (fem). Finally, a prototype was designed and fabricated, and the validity of the analysis was verified using the finite element results and experimental results [16]. In the articles [17], the dynamics of a six-degree-of-freedom bearingless linear motor system were modeled using a set-sum modeling approach. The validity of the established model was verified by comparing the time-domain simulation results with the experimental results.

The model can be applied to time-domain simulation, real-time control system development, and various system analyses. The article [18] proposes the integration of magnetic bearings into a tubular linear actuator (TLA), resulting in a new self-bearing (bearingless) TLA. The system is derived from a standard TLA by changing its stator geometry. The working principle is illustrated using the finite element method, and critical design aspects are investigated. Sun proposes a three-degree-of-freedom permanent magnetic levitation platform for cleanroom conveying systems, which is supported by four magnetic poles and uses a variable flux path mechanism. Based on the analysis of the dynamics, the magnetic levitation platform's mathematical model and state space are established. A double closed-loop controller for stable levitation and motion control is designed, and the control system is decoupled using a coordinate transformation matrix. Finally, simulations and experiments were conducted to achieve stable levitation and three-degree-of-freedom motion control with remarkable positioning accuracy [19–21].

Magnetic levitation has been widely used in the biomedical industry due to its advantages of cleanliness and non-contact. Ke proposed a planar Litz coil sandwiched between two ferrite substrates optimized design method that wirelessly powers a novel mechanical artificial anal sphincter system for treating severe fecal incontinence [22]. Zhou designed a novel implantable puborectal-like artificial anal sphincter (PAAS) that replicates rectal perception with a low risk of ischemic necrosis. A pressure sensor embedded in the PAAS was used to determine the relationship between the stool mass and pressure and to develop a stool mass estimation model based on in vitro studies [23]. Articles [24–26] applied magnetic levitation technology to studies on artificial hearts to optimize the improvement of magnetic pumps. Forrai investigated the system identification and control of a nonlinear electromagnetic actuator that could be used in an artificial heart actuator [27]. Haisler described a 3D culture method, the magnetic levitation method (MLM), in which cells are combined with magnetic nanoparticle assemblies overnight to provide them with magnetic properties. When resuspended in the medium, an external magnetic field causes the cells to levitate and concentrate at the gas-liquid interface, where they aggregate to form larger 3D cultures [28]. Parfenov proposed the rapid creation of 3D scaffolds using

the magnetic levitation of calcium phosphate particles. Label-free suspension assembly was achieved in the presence of gadolinium salts by using a custom-designed magnetic system that allowed the suspension of calcium phosphate particles. The chemical conversion of tricalcium phosphate to octacalcium phosphate under magnetic levitation conditions in a non-uniform magnetic field was also demonstrated [29]. Yaman developed a magnetization-rate-based protein detection scheme using a low-cost, miniaturized magnetic levitation device consisting of two opposing magnets to generate a magnetic field gradient, a glass capillary channel to retain the sample, and two lateral mirrors to monitor the interior of the channel. The method involves using polymeric microspheres as a mobile analysis surface and magnetic nanoparticles as markers. The assay is achieved by trapping the target protein in the polymer microspheres [30].

The MDOFSC proposed in this paper has the advantages of no oil and stable operation at a low-pressure ratio. It can be used in the medical environment to provide oxygen for patients and improve cardiac extracorporeal membrane oxygenation. Based on the electromagnetic drive principle, this paper proposes the MDOFSC, the relationship between the nonlinear magnetic model and the working air-gap length is analyzed, and the decoupling control model of MDOFSC is established by applying the coordinate transformation strategy. With the goal of trajectory control, a centralized controller was designed and experimentally verified.

2. Description of the System

2.1. Structure and Principle

The structure of the MDOFSC is shown in Figure 1, which is mainly composed of orbiting scroll, static scroll, electromagnet, armature, magnetic ring, sensor, etc. The working principle of the MDOFSC is that three groups of electromagnets attract the armature and orbital scrolls to achieve orbital motion and compress the gas. In order to make the MDOFSC work with a specific stiffness, each group of electromagnets is controlled differentially. The MDOFSC is equipped with six electromagnets inside. The open-loop electromagnetic drive system is unstable, and difficult to control the running trajectory. In order to realize the closed-loop control of the running track of MDOFSC, each group of electromagnetic units is executed by a servo system composed of a controller, a sensor, and a power amplifier. The sensor in the Y direction measures the position of the armature and the two sensors in the X direction measure the position and rotation angle of the armature. Through real-time detection and control, the distance between the two X directions is kept straight and equal to ensure the translational movement of the orbiting scroll, which replaces the anti-rotation mechanism, reducing friction and realizing oil-free operation. As shown in Figure 1b, four sets of permanent magnet rings are mounted in four bracket recesses, forming symmetrically distributed mutually exclusive upper and lower sections to support the suspension of the orbiting scroll.

2.2. Magnetic Circuit Analysis

Figure 2 shows the magnetic flux density distribution of the electromagnet and armature. The electromagnetic drive system is composed of three groups of electromagnets and armatures. The air-gap distance between the armature and the electromagnet shown in the figure is 1.5 mm, the number of coil turns is N_x = 235, two coils in the Y direction are connected in series, N_y = 260, and the current is 2A. The magnetic flux that leaked into the air cannot be ignored, and the leakage coefficient varying with the air-gap distance needs to be introduced. As seen in Figure 2, the magnetic flux density of the armature is almost zero. It can be seen from Figure 3 that the flux path starts from the north pole and returns to the south pole through the core. In order to form a specific stiffness when controlling the operation of the orbiting scroll, the system adopts the method of differential control. When the orbiting scroll is stationary, the two electromagnets in the up and down direction of Y are input with bias current and pull the orbiting scroll mutually. When the positive



signal in the direction is given, the upper electromagnet current increases, and the lower electromagnet current decreases.

Figure 1. Structure of MDOFSC (a) Overall structure; (b) Suspension mechanism.



Figure 2. Structure of MDOFSC Magnetic flux density distribution on the electromagnet. Current = 2 A, Air gap x,y = 1.5 mm, and N_x = 235, N_y = 260.



Figure 3. Magnetic flux path through the U-core electromagnet.

2.3. System Model

The distance measured by the sensor needs to be converted into the air gap between the electromagnet and the armature, and the relationship is shown in Formula (1). As shown in Figure 4, when the displacement of the MDOFSC in the X direction is x, the

displacement in the *Y* direction is *y*, and the rotation angle of the orbiting scroll around the *Z*-axis is θ . The transformation relationship between the change of air gap between the three groups of magnetic poles and armature and the three degrees of freedom of the platform is shown in Formula (2). After linear approximation, the coordinate transformation relationship in matrix form can be obtained, as shown in Formula (3). As shown in Formula (4), solve the pseudo-inverse matrix of the coordinate transformation matrix, and the coordinate inverse transformation relationship of the system can be obtained, as shown in Formula (5).



Figure 4. Relationship between the working air gap and sensor measurement distance.

As shown in Figure 4 and Formula (1): D_x , and D_y are the distances between the sensor and the corresponding end face of the electromagnet, which are fixed values; d_1 , d_2 , and d_3 are the distances between the sensor and the armature; x_1 , x_2 , and x_3 are the air-gap distances between the end face of the electromagnet and the armature. x, y, and θ are the three degrees of freedom of the suspended solids. $2L_1$ is the distance between two sensors in the X direction.

$$\begin{bmatrix} x_1 & x_2 & y_1 \end{bmatrix}^T = \begin{bmatrix} D_x & D_x & D_y \end{bmatrix}^T - \begin{bmatrix} d_1 & d_2 & d_3 \end{bmatrix}^T$$
, (1)

$$\begin{cases} x_1 = x + y \cdot \tan\theta - H \cdot \sin\theta \\ x_2 = x + y \cdot \tan\theta + H \cdot \sin\theta \\ y_1 = -x \cdot \tan\theta + y \end{cases}$$
(2)

$$\begin{bmatrix} x_1 & x_2 & y_1 \end{bmatrix}^T = N_1 \begin{bmatrix} x & y & \theta \end{bmatrix}^T,$$
(3)

 $N_1 N_2 = E_{3 \times 3},$ (4)

$$\begin{bmatrix} x & y & \theta \end{bmatrix}^T = N_2 \begin{bmatrix} x_1 & x_2 & y_1 \end{bmatrix}^T,$$
(5)

where N_1 is the coordinate transformation matrix, and N_2 is the inverse coordinate transformation matrix.

$$N_1 = \begin{bmatrix} 1 & 1 & -H \\ 1 & 1 & H \\ -1 & 1 & 0 \end{bmatrix}, \quad N_2 = \begin{bmatrix} 1/4 & 1/4 & -1/2 \\ 1/4 & 1/4 & 1/2 \\ -1/(2H) & 1/(2H) & 0 \end{bmatrix},$$

As shown in Figure 5, 2*H* is the center distance between Electromagnet 2 and Electromagnet 4; 2*L* is the distance from the center of the magnetic spring composed of an annular permanent magnet and a cylindrical permanent magnet to the center of the actuating platform.



Figure 5. Force and motion of the platform.

In order to analyze the feasibility of the MDOFSC, the control system needs to be simulated. Therefore, the mathematical model describing the controlled object should be established according to the dynamic characteristics of the controlled object. In order to simplify mathematical modeling, this paper assumes that the following complex elements can be ignored. (1) The mass of the suspension platform of the MDOFSC is evenly distributed, and the center of gravity overlaps the geometric center of the platform. (2) The suspended platform is an ideal rigid body; the deformation is negligible. (3) The dimensional accuracy is sufficient, and the magnetic characteristics of each magnetic pole are the same. (4) It is difficult to derive the friction formula between the orbiting scroll and the static scroll, supplemented by increasing the system damping. The force and motion of the platform are shown in Figure 4. Then, the differential formula is established through the force and motion analysis of each degree of freedom, as shown in Formula (6).

$$\begin{cases} m \ddot{x} = F_{qx1} + F_{qx2} + 4F_m - c \dot{x} \\ m \ddot{y} = F_{qy} + 4F_m - c \dot{y} \\ J \ddot{\theta} = (2(F_{qx2} - F_{qx1})H + 8F_mL)\theta - c \dot{\theta} \end{cases},$$
(6)

$$F_v = k_w \frac{i_2}{(d_w + \lambda_w)^2}; \ v = x_1, \ x_2, \ x_3, \ x_4, \ y_1, \ y_2; \ w = x, \ y$$
(7)

$$F_w = k_w \left(\left(\frac{i_0 + i_w}{d_0 - d_w - \lambda_w} \right)^2 - \left(\frac{i_0 - i_w}{d_0 + d_w + \lambda_w} \right)^2 \right),\tag{8}$$

where *m* is the mass of suspended solids, F_{qx1} , F_{qx1} , and F_{qy} are three groups of differential electromagnetic forces in X_1 , X_2 , and *Y* directions, respectively, and F_m is the magnetic force of a single group of permanent magnetic rings. The derivation process of the magnetic force of permanent magnetic rings is shown in the paper [5]. *J* is the rotational inertia of the suspended solids, and F_v is the magnetic force of a single electromagnet. i_2 is the input current of the coil, *d* is the air-gap length between the electromagnet and the magnetized target, F_w is the differential electromagnetic force of a single group of electromagnets, λ_w is the compensation constant for the air gap, k_w is the constant of magnetic force, i_0 is the bias current, i_w is the control current, d_0 is the air-gap distance at the balance position, and d_w is the displacement of the platform. The structure of the electromagnetic system determines the above two parameters.

$$F_m = k_m \delta, \tag{9}$$

$$\Delta F = k_{iw}(i_0 + i_w) + k_{dw}(d_0 + d_w); \ w = x, y \tag{10}$$

where K_m is the stiffness coefficient of the permanent magnetic force, δ . It is the horizontal displacement change between a pair of permanent magnetic rings and the linearization result of a single group of differential electromagnetic forces. K_{iw} is the current stiffness after linearizing a single group of differential electromagnetic forces, and k_{dw} is the displacement stiffness after linearizing a single group of differential electromagnetic forces. k_{iw} and k_{dw} are current and displacement stiffness coefficients, respectively.

According to the system dynamics formula, the system state space formula is established as follows:

$$\begin{cases} \dot{X} = AX + BU \\ Y = CX \end{cases}$$
(11)

where,

$$X = \begin{bmatrix} x & \dot{x} & \theta & \dot{\theta} & y & \dot{y} \end{bmatrix}^{T},$$
 (12)

$$Y = \begin{bmatrix} y_1 & y_2 & y_3 \end{bmatrix}^l, \tag{13}$$

$$U = \begin{bmatrix} i_1 & i_2 & i_3 \end{bmatrix}^T, \tag{14}$$

$$A = \begin{bmatrix} 0 & 1 & 0 & 0 & 0 & 0 \\ \frac{2k_x + 4k_m}{m} & \frac{-c}{m} & 0 & 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 1 \\ 0 & 0 & 0 & 0 & 0 & \frac{k_y + 4k_m}{m} & \frac{-c}{m} \end{bmatrix}$$
$$B = \begin{bmatrix} 0 & 0 & 0 \\ \frac{k_{ix}}{m} & \frac{k_{ix}}{m} & 0 \\ 0 & 0 & 0 \\ \frac{-k_{ix}}{J} & \frac{J}{J} & 0 \\ 0 & 0 & 0 \\ 0 & 0 & \frac{k_{iy}}{m} \end{bmatrix}$$
$$C = \begin{bmatrix} 1 & 0 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & 1 & 0 \end{bmatrix}$$

3. Controller Design and System Simulation

3.1. Motion Controller Design

To solve the deviation problem caused by the different magnetic characteristics of each magnetic pole of the MDOFSC under a decentralized control, a three-degree freedom centralized control strategy is designed to realize the direct control of the freedom of the MDOFSC. There is a coupling between the input and output of the system. The system can be decoupled into three single input and single output systems by Formulas (1) and (5), and PID controller. The integration link can correct the control current of each magnetic pole in real time, compensate for the influence of the difference in magnetic characteristics, and correct the inclination of the orbiting scroll. Figure 6 shows the principle of the controller, and the expression is written as follows:

$$e_{du} = \operatorname{Ref} u - u(t); \ u = x, y, \theta,$$

$$i_u(t) = k_{pu} \cdot e_{du}(t) + k_{iu} \int e_{du}(t) dt + k_{du} \cdot \frac{de_{du}(t)}{dt},$$
(15)

where e_{du} is the input error signal, Ref *u* is the reference input of three degrees of freedom, u(t) is the actual degree of freedom of feedback, $i_u(t)$ is the input current. The k_{pu} , k_{iu} , and k_{du} are the proportional gain, integral gain, and differential gain of the PID controller, respectively.



Figure 6. Block diagram of the control system.

3.2. Control System Simulation

The feasibility of the double closed-loop controller is verified by numerical simulation using Simulink software. Table 1 shows the parameters. In this program, the simulation lasts for 0.5 s, and at 0 s, each electromagnetic unit inputs a signal, respectively. Figure 7 shows the simulation results (track tracking states and corresponding control currents in the X and Y directions). The control parameters in the X direction are P = 100, I = 2, and D = 10. During the movement in the X direction, a negative sinusoidal signal is input in the X direction at 0 s, and the controller quickly inputs a negative current to the electromagnetic unit. This system simulation does not consider the friction between the orbiting scroll and the static scroll, so the system stability is poor, and a larger D is required to enhance the system damping and control effect. However, the differential effect amplifies noise disturbances, causing the control current to oscillate rapidly. The threshold of the control current is ± 10 A, and the starting current of track tracking reaches the peak. The peak time of track tracking is 0.053 s, and the maximum overshoot is 3%. The control parameters in the Y direction are P = 30, I = 2, and D = 20. In the trajectory motion in the Y direction, a cosine signal is an input in the Y direction at 0 s, at which time the controller quickly outputs the maximum negative current. The larger D value makes the system have the effect of starting overshoot and strengthening control, which can effectively reduce the dynamic deviation of the control process. There is no overshooting in the trajectory tracking, and the tracking effect is good.

Table 1. System parameters.

Description	Parameter	Quantity
Moment of inertia	J	0.0015097 kg⋅m ²
Mass of suspended solids	т	1.7 kg
Current stiffness coefficient in the X direction	k_{ix}	150
Displacement stiffness coefficient in the X direction	k_{dx}	-75
Current stiffness coefficient in the Y direction	k_{iy}	180
Displacement stiffness coefficient in the Y direction	k_{dy}	-57
Magnetic stiffness coefficient of a magnetic ring group	k_m	4.544 N/mm
Damping of the system	С	0.015 N/(mm/s)
Center distance between two electromagnets in the X direction	Н	19.5 mm
Center distance of adjacent magnetic rings	L	21 mm



Figure 7. Simulation results of trajectory tracking. (a) Trajectory in *X* direction; (b) Control current in *X* direction; (c) Trajectory in *Y* direction; (d) Control current in *Y* direction.

4. Experiment and Analysis

4.1. Experimental System

Figure 8 shows the experimental system of the MDOFSC, mainly including the prototype, hardware equipment, and control system. The control system is based on the MicroLabBox produced by the dSPACE company. MATLAB and dSPACE software kits are installed on the upper computer. The power amplifier adopts the current control mode. Pu-05 eddy current displacement sensor of the AEC company is used for air-gap detection, with a range of 0–2 mm and a resolution of 0.5 μ m. The analog output voltage range is -5 V to 5 V.

4.2. Experimental Result

Trajectory tracking experimental parameters: P = 30, I = 2, D = 0.01. The parameters of PID come from many experiments. The appropriate parameters are determined by the step experiment's positioning effect and the trajectory tracking stability. Figure 8 shows the trajectory tracking experiment of the MDOFSC in the X direction. The period is 0.2 s, and the motion track is a circle with a radius of 0.5 mm. It can be seen from Figure 9a that at 0 s, a negative sinusoidal trajectory in the X-axis direction is input. Under PID decentralized control, the track difference between x_1 and x_2 increases gradually from the track center to the track peak, the maximum time difference reaches 0.148 mm, and the maximum relative error rate is 29.6%. As shown in Figure 9b, the trajectory tracking effect is better. In the whole tracking process, the maximum error of x_1 and x_2 is 0.024 mm, and the maximum relative error rate is 4.8%. The results show that the centralized control system has a good tracking effect.



Figure 8. Control system hardware.



Figure 9. Trajectory tracking experiment. (a) *X*-direction trajectory tracking under decentralized control; (b) *X*-direction trajectory tracking under centralized control.

Figure 10 shows the tracking track in the Y direction. The cosine displacement signal is input at 0 s. The track tracking is not complete at the second and third peaks, which is quite different from the cosine signal. The reasons are as follows: 1. After the trajectory signal is given, the static scroll changes from static to motion and vibrates due to inertia and magnetic ring interference. It is not easy to control the system. From the fourth peak, the tracking effect is better; 2. PID control is more suitable for linear systems. At the trajectory's peak, the magnetic unit's nonlinearity is strong, and the control effect is poor.

Figure 11 shows the current control diagram during track tracking. Figure 8 shows that coils 1, 2, 3, and 4 are in the *X* direction. When the time is 0, the trajectory signal begins to input, and the orbiting scroll makes a negative sinusoidal motion along the *X* direction. At this time, the currents of coil 2 and coil 4 are 4.38 A and 6.77 A, which provides the system with high response-ability and provides acceleration. At this time, the current of coil 3 is 0 because the magnetic ring will provide suction to fight against the current of coil 2,4, and the permanent magnet is a passive magnetic force, which is challenging to maintain the stable movement in the *X* direction, so the current of coil 1 constantly changes. At 0 s, the current of coil 1 starts to rise from 0, which can not only cooperate with the

permanent magnetic ring to control the torsion of the orbiting scroll but also pull with coil 2,4 to form a specific stiffness.



Figure 10. Trajectory tracking in Y-direction under centralized control.



Figure 11. Control current.

Figure 12 shows the trajectory tracking of MDOFSC under decentralized control and centralized control, respectively. Under decentralized control, the circles fitted by the X_1 and X_2 direction trajectories and the Y direction trajectories have poor coincidence. While under centralized control, x_1 and x_2 , respectively, coincide with the circular trajectory formed by the trajectory in the Y direction. Since the trajectories of x_1 and x_2 are controlled separately under a decentralized control, the value of x_1 and x_2 is not always equal due to the hysteresis of position control. By adopting the centralized control method, the three degrees of freedom of the system are directly controlled, and the control rotation angle is always zero. The effect of track tracking is improved, and the coincidence degree of tracks is higher.

An experimental prototype of MDOFSC was built to verify the trajectory tracking experiment. The experimental results show that under a decentralized control, the maximum difference between the synchronization trajectories of x_1 and x_2 is 0.148 mm, and the maximum relative error rate is 29.6%. Through centralized control after decoupling, the maximum difference between the synchronization trajectories of x_1 and x_2 is 0.024 mm, and the maximum relative error rate is 4.8%. The results show that through the decoupled centralized control, the system has a self-tuning characteristic, which can ensure that the trajectories of x_1 and x_2 directions remain constant at all times, reduce the left and right wobble of the orbiting scroll, and reduce the wobble and air leakage during the actual operation of the MDOFSC, which is of great significance for further research on the MDOFSC.





5. Conclusions

This paper introduces the structure and working principle of MDOFSC. Then, the magnetic circuit of the electromagnetic unit is analyzed through finite element simulation, and the mathematical model of MDOFSC no-load operation is established. Finally, the relationship between the air-gap distance and the degree of freedom is decoupled, and the decoupling controller is designed. The trajectory tracking characteristics of MDOFSC are studied through simulation and experiment. Experimental results: Under decentralized control, the maximum difference between x_1 and x_2 is 0.148 mm, and the maximum relative error rate is 29.6%. Through the centralized control after decoupling, the maximum difference between x_1 and the maximum relative error rate is 4.8%. The experimental results show that the system has self-tuning characteristics through decoupling centralized control, which can ensure that the trajectories of x_1 and x_2 directions remain unchanged, reduce the left and right swing of the orbiting scroll, and improve the tracking effect of the system. MDOFSC has good trajectory tracking characteristics.

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