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Enhancement of Fatigue Endurance by Al-Si Coating in Hot-Stamping Boron Steel Sheet

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Abstract: Most structural components undertake cyclic loads in engineering and failures always cause catastrophic economic losses and casualties. In the present work, the phase evolution of Al-Si coating of high-strength boron steel during hot stamping was investigated. Two types of 1500 MPa grade boron steel sheets, one with Al-Si coating and the other without, were studied to reveal the effect on the high-cycle fatigue behavior. The as-received continuously hot-dip Al-Si coating was composed of α (Al), eutectic Al-Si and τ_5 . After hot stamping at 1193 K, three phases formed in this coating: β_2 , Fe(Al,Si)₂ and α (Fe). The experimental results showed that the endurance limit of the coated steel sheet was 370 MPa under 10⁷ fully reversed tension-compression loading cycles as opposed to 305 MPa in the uncoated sheet. Both the coated and the uncoated specimens showed surface-induced transgranular fatigue fractures. In the uncoated sheet, the fatigue cracks were generated from the decarburization surface, but the Al-Si coating effectively prevented the occurrence of near-surface decarburization during high-temperature hot stamping, and the only cracks in the coated steel sheet were initiated at wire-cutting surfaces.

Keywords: boron steel; Al-Si coating; hot stamping; high-cycle fatigue; decarburization

1. Introduction

Demands for increasing safety, reducing the weight of automobile bodies and fuel consumption in automotive manufacturing have promoted technological innovation [1–4]. Due to high stamping loads, large spring-backs, bad formability and severe wear such as galling and seizure, cold stamping of high strength steels is facing challenges [5,6]. Hot stamping has been rapidly developed since it effectively overcomes/avoids these problems, and presently it can be roughly classified into direct and indirect methods according to technological routes [7]. For the direct hot stamping process, the steel sheet is heated firstly, transferred to the press and subsequently formed and quenched in a closed tool [8]. In contrast, a pre-formed part is cold-stamped firstly in case of the indirect hot stamping process, and only quenching and calibration are performed after austenitizing in the press [9]. On the other hand, boron steel sheets have been widely used for the production of high strength panels of car bodies and vehicle chassis structures due to its excellent formability [10,11]. The ultimate tensile strength of hot-stamping boron steel sheets would reach up to 1500 MPa with a full martensite microstructure [12,13]. Nevertheless, it was inevitable that bare (uncoated) steel sheets were thermally oxidized in earlier hot stamping technology and the oxidized scale had to be removed by shot blasting or shot peening [14]. Subsequently, the attempt to use lubricant oils was abandoned due to complex



cleaning procedures and a high risk for hydrogen-induced embrittlement [15]. Currently, various types of coatings, including aluminized coating [16–20], galvanized and galvannealed coatings [21–23], Zn-Ni alloy coating [24,25], sol-gel hybrid coating [26] and Al-Zn alloy coating [27], were developed industrially. Among which, Type 1 aluminized coating, Al-Si with a near eutectic composition, has been widely applied on high-strength boron steel sheets [28].

Currently, the relationship between mechanical properties (e.g., toughness and hardness etc.), microstructure (e.g., phase constitution and morphology, etc.) and hot-stamping parameters (e.g., autenitization temperature and holding duration, etc.) of boron steel sheets have been extensively reported [29–31]. However, most structural components of automobiles are indeed exposed to cyclic loads or continuous vibrations during long-term service, and these components may fracture due to repeated stress, which are much lower than the conventional static strength [32–35]. For example, Lara et al. [32] found that cutting technology affected the fatigue properties of Al-Si coated 22MnB5 high-strength steel sheets with ~1333 MPa of yield strength (YS) and ~1560 MPa of ultimate tensile strength (UTS). The endurance limits under 2×10^6 tension-tension loading cycles (load ratio R = 0.1) were 391 MPa and 453 MPa for sheared specimens and laser-cut specimens, respectively. With further grinding and polishing on the cutting edges of the sheared specimens, it dramatically increased up to 652 MPa. Considering that fatigue failure always bring catastrophic economic loss and casualties, it has received more interest in recent decades due to scientific as well as engineering requirements. However, investigations on the tension-compression fatigue behavior of hot-stamping high-strength boron steel sheet are limited. The main objective of this work was to characterize the phase evolution of Al-Si coating during hot stamping, and to reveal the effect of the high-cycle fatigue behavior on a commercial boron steel sheet. Our work indicated that the Al-Si coating could significantly improve the fatigue life.

2. Experimental Procedures

A comparison was made between two 1500 MPa-grade high-strength boron steel sheets, each 1.80 mm thick (B1500HS, Chinese grade, largely corresponds to 22MnB5). One sheet was uncoated and the other was coated with Al-Si by continuously using the hot-dipping method. Aside from the coating, both sheets had the same chemical composition (Table 1). Both types of cold-rolled steel sheets were heated to 1193 K at 30 K/s, held for 180 s, and then the austenized sheets were quickly cooled at ~70 K/s to the desired stamping temperature with the help of high pressure air. The stamping was done at a start temperature of ~973 K with a pressure time of 5 s in the present work. The YS and UTS of both sheets were ~1250 MPa and ~1500 MPa, respectively. High-cycle fatigue tests were conducted using a servo hydraulic tabletop test system (Bionix 370.02, MTS, Eden Prairie, MN, USA) with the load axis parallel to the rolling direction. The specimens for the fatigue test were wire-cut from the hot-stamped sheets, and the dimensions have been illustrated in Figure 1. More than five stress levels based on the standard staircase method were undertaken in stress control at room temperature. A sinusoidal waveform was employed at 20 Hz with fully reversed tension-compression loading (stress ratio R = -1). Care was taken to ensure the proper alignment of the specimen in the grips and to eliminate buckling of the specimen during fixing and testing. Three experiments were performed under each stress level, and if two of them reached 10^7 cycles, the endurance limit was considered to be reached.

Table 1. The chemical composition of the studied boron steel sheet (wt.%).

С	Si	Mn	Р	S	Ni	Мо	Al	В	Cr	Ti	Cu
0.23	0.25	1.35	0.015	0.006	0.028	0.04	0.04	0.003	0.19	0.03	0.016



Figure 1. Illustration for dimensions of the fatigue test specimens (unit: mm).

The microstructure and fracture morphology were observed using stereomicroscope (SteREO Discovery V8, Zeiss International, Berlin, Germany) and an optical microscope (Zeiss A2m, Zeiss International, Berlin, Germany). Quantitative elemental analysis was conducted using a field emission electron probe microanalyzer (EPMA-8050G, Shimadzu, Tokyo, Japan). High-temperature differential scanning calorimetric measurements (DSC 404 F3, NETZSCH-Gerätebau GmbH, Selb, Germany) were carried out at a heating rate of 10 K/min. The martensitic transformation start temperature was determined on a dilatometer system (TA DIL805L, TA Instruments, Eschborn, Germany), and the experimental process was as follows: (1) heated to 1193 K at a heating rate of 1 K/s and kept for 180 s; (2) quenched to room temperature at a cooling rate of 20 K/s.

3. Results and Discussion

3.1. Al-Si Coating

The thickness of the Al-Si coating on the as-received boron steel sheet was approximately 30 μ m on each side (Figure 2a). The steel substrate showed refined ferrite/pearlite [36]. A reaction between the substrate and the Al-Si coating (Layer I) produced a transition layer (Layer II). After hot stamping at 1193 K, the microstructure of Layer I indicated the formation of a new intermetallic compound with a few voids (Figure 2b). Layer II became thicker (~7 μ m), and the steel substrate transformed into lath-like martensite.



Figure 2. Cross-sectional optical microstructures of Al-Si coated specimens, as-received state (**a**) and after hot-stamped at 1193 K (**b**).

EPMA quantitative analyses showed that the dominant phases of Layer I in the as-received steel sheet were α (Al) matrix and eutectic Al-Si, while the dominant phase of Layer II was Fe₂Al₇Si (τ_5) ternary intermetallic compound (Figure 3a). After hot stamping at 1193 K, three new phases formed: Fe(Al,Si) (β_2), Fe(Al,Si)₂ and α (Fe) [17,28,37–39] (Figure 3b). It should be noticed that although these colors in Figure 3 are false colors assigned by the instrument program, they provide more differential

contrast for human eye than the grey scale. The values corresponding to different color represent the number of photons detected by x-ray. The larger value indicates the higher content of the target element. The average chemical compositions of different phases at different locations have been listed in Table 2. All measurements were calibrated using pure elements, and five spots were performed for each phase. It is obvious that high-temperature hot stamping rapidly activated interdiffusion between the coating and the steel substrate, and because of the Kirkendall effect (that is, the movement between two metals that occurs with different diffusion rates), a few voids appeared inside the coating layer (Figures 2b and 3b) [17,28,40]. Cracks were also observed, which were closely linked to the differences between the thermal expansion coefficients of the newly-formed Fe-Al-Si phases in the coating [28].



Figure 3. Backscattered electron images and the corresponding elemental distributions in Al-Si coated specimens, as-received state (**a**) and after hot-stamped at 1193 K (**b**). The average chemistry in location 1 of Figure 3a indicated the existence of α (Al) matrix and eutectic Al-Si phase. The average chemistry in location 2 of Figure 3a indicated that the dominant phase in Layer II was Fe₂Al₇Si (τ_5) ternary intermetallic compound. The average chemistries in locations 3, 4 and 5 in Figure 3b indicated formation of Fe(Al,Si) (β_2), Fe(Al,Si)₂ and α (Fe) phases, respectively.

Elt	1	2	2	4		
Element	1	2	3	4	5	
Fe	0.2	21.0	51.3	33.5	85.0	
Al	94.6	69.3	35.5	63.6	10.9	

13.2

2.9

4.1

Table 2. The average chemical compositions of different phases at locations 1–5 as marked in Figure 3 (at.%).

9.7

Si

5.2

The DSC chart showed three phase transitions of the coated steel sheet within the measured temperature range from 700 K and 1150 K (Figure 4a). The first endothermic peak at 845 K upon heating was attributed to the melting of eutectic Al-Si phase, and the second peak at 882 K indicated the melting of α (Al) phase according to the Al-Si binary phase diagram [41]. The third peak at 1014 K corresponded to the austenitization of boron steel [42], which was consistent with the dilatometric data (Figure 4b). A relative length contraction took place between 1010 K (Ac_1) and 1099 K (Ac_3) upon heating, corresponding to the start and finish of ferrite \rightarrow austenite transformation, respectively [43]. The inflation started at 667 K upon cooling resulted from the martensitic transformation [44].



Figure 4. DSC (a) and dilatometric (b) curves of as-received Al-Si coated specimen.

The microstructures of steel substrate and the corresponding carbon (C) and aluminum (Al) elemental distributions were investigated in order to understand high-cycle fatigue behavior. In the hot-stamped coated specimen, lath-like martensite was observed (Figure 5a). Carbon distributed homogeneously in the substrate (Figure 5b). In the uncoated specimen, a decarburized layer with a width of approximately 80 µm was observed, where the carbon content was significantly lower relative to the interior region (Figure 5d). The number in the color bar of Figure 5 represents the area fraction of C/Al element within the measured area. It has been generally thought that decarburization is a surface degradation process involving the loss of near-surface carbon from steel during exposure in air at elevated temperatures, which would seriously deteriorate the mechanical properties of steels [45]. For example, Ren et al. [46] reported that the presence of decarburization reduced the microhardness from 425 HV to 260 HV in a 50CrMnMoVNb spring steel. Zhao et al. [47] found that the crack growth rate on the decarburized rail roller was over 4 times that of the non-decarburized rail roller under dry-wet conditions. Furthermore, the loss of the carbon element may have largely reduced the hardenability of the near-surface steel sheet [48] and thus the martensite transformation did not occur, and formed some ferrite/pearlite.

3.2. High-Cycle Fatigue Behavior

The statistic stress-fatigue life (*S*-*N*) curves obtained from two types of hot-stamped specimens showed that Al-Si coating enhanced fatigue endurance (Figure 6). At σ = 370 MPa stress level (stress amplitude), no Al-Si coated specimens failed at 10⁷ cycles. Nonlinear logistic fit was used (OriginPro 8.5.0 SR1, Originlab Corporation, Northampton, MA, USA) and two equations, σ = 391.326 + 112.691/ $\left[1 + \left(\frac{N}{100300.632}\right)^{5.329}\right]$ for Al-Si coated specimen and σ = 305.305 + 299.709/ $\left[1 + \left(\frac{N}{79217.244}\right)^{0.995}\right]$ for the uncoated specimen, were obtained. At σ = 305 MPa, two uncoated specimens reached 10⁷ cycles without failure. In other words, under 10⁷ cycles of fully reverse tension-compression loading, the Al-Si coated sheet. Low magnification images of fatigue fracture surface showed three distinctive fatigue regimes—crack initiation, propagation and final failure—for both coated and uncoated specimens (Figure 7a,c). Both types of specimens indicated surface-initiated failure, similar to the 22MnB5 steel sheets under a tension-tension loading mode [32]. The location of initiation, however, was different. Most of the cracks in the coated specimens were initiated on the side surface

(which was created by using wire-cutting) (circled in Figure 7a). All fractures for the uncoated specimens generated inside the decarburization layer (Figure 7c). That is, the decarburization layer acted as weaker location compared to the wire-cutting surface in the uncoated specimens. Enlarged facture surface images showed typical transgranular fracture feature [49] in both types of specimens, as pointed by the arrows in Figure 7b,d.



Figure 5. Cross-sectional SEM images and the corresponding C and Al elemental distributions after hot stamping at 1193 K for the coated (**a**,**b**) and uncoated (**c**,**d**) specimens.



Figure 6. Stress-fatigue life (*S*-*N*) curves of Al-Si coated (**a**) and uncoated (**b**) specimens after hot stamping at 1193 K.



Figure 7. Fracture macrographs of coated (**a**) and uncoated (**c**) specimens under 390 MPa after hot stamping at 1193 K. (**b**,**d**) are the enlarged images showing the crack initiation and early propagation (marked by arrows).

The prediction of the crack initiation location was conducted to validate the microstructural observation. The numerical model was established (SOLIDWORKS®Premium 2016, SP 1.0, Dassault Systèmes, Waltham, MA, USA) and as described above, the decarburized layer on both sides of the uncoated specimen was preset with a thickness of $80 \,\mu\text{m}$, and it was presumed that the UTS value linearly decreased from 1500 MPa to 900 MPa from the interface (green dashed line marked in Figure 5c) to the outer edge of the decarburization layer. The coated specimen was taken to be a uniform sheet for clarity since the narrow α (Fe) diffusion layer is less than 7 μ m. Figure 8 demonstrates the cloud images of damage for the fatigue specimens obtained by the finite element method (ANSYS®WorkbenchTM 2.0 Framework, 15.0.0, ANSYS, Inc., Canonsburg, PA, USA) under a sinusoidal tension-compression cyclic loading of 390 MPa. Miner's law was used based on the theory of linear cumulative damage. The fatigue damage was defined as the design life divided by the available life, and design life was preset at 10^9 in the present work. That is, the fatigue value is inversely proportional to the fatigue life cycles (N) and in the present work, it equaled $10^9/N$. The larger damage value indicated the shorter fatigue life where it was easier to initiate the crack. The weakest locations have been circled in Figure 8a,b, respectively. It can be seen that the simulation results agree well with the experimental observation of the wire-cutting surface for the coated specimen during cyclic loading. For the uncoated

specimen, the fatigue cracks would be easier to generate inside the decarburization layer. According to this conclusion, it seems necessary to further address the effect of the cutting surface quality (integrity) for the application of Al-Si coated high-strength boron steel sheet as fatigue resistant components.



Figure 8. Cloud images of damage for the coated (**a**) and uncoated (**b**) fatigue specimens under 390 MPa obtained by the finite element method with a sinusoidal cyclic loading mode.

4. Conclusions

Two types of 1500 MPa grade hot stamping boron steel sheets were compared, one with Al-Si coating and the other without. The following conclusions were obtained.

- (1) The phases of the as-received Al-Si coating were α (Al), eutectic Al-Si, and τ_5 . After hot stamping at 1193 K, Fe(Al,Si)₂, β_2 and α (Fe), phases formed and some voids/cracks appeared inside the coating. The Al-Si coating effectively prevented the occurrence of near-surface decarburization during high-temperature hot stamping.
- (2) The endurance limit of the hot-stamped coated steel sheet was 370 MPa compared to 305 MPa for the uncoated sheet under 10⁷ fully reversed tension-compression loading cycles. Both specimens were classified as the surface-induced transgranular fatigue fracture.
- (3) The high-cycle fatigue cracks were generated from the decarburization layer in the uncoated sheet, whereas for the coated sheet, the cracks were initiated at wire-cutting surfaces.

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