

Article

The Fabrication and Mechanical Properties of Laminated ZrB₂-Mo₅SiB₂ Ceramics with an Mo-Mo₅SiB₂ Interlayer

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Abstract: The excellent physical and chemical properties of ultra-high temperature ceramics make them suitable for many high-temperature structural components, while their poor toughness and high sintering temperature become key limitations to their application. Laminated toughening has long been considered an effective toughening method to improve the mechanical properties of ceramics. In this study, laminated ZrB₂-Mo₅SiB₂ ceramics with an Mo-Mo₅SiB₂ interlayer were fabricated by tape casting and hot press sintering at 1900 °C for 2 h. Different layer thickness ratios between the matrix layer and the interlayer were designed to illustrate the toughening mechanism. Both the fracture toughness and flexural strength of the laminated ceramics showed a trend of first increasing and then decreasing with the increase of the layer thickness ratio. High fracture toughness ($9.89 \pm 0.26 \text{ MPa}\cdot\text{m}^{1/2}$) and flexural strength ($431.6 \pm 15.1 \text{ MPa}$) were obtained when the layer thickness ratio was 13. The improvement in fracture toughness of the laminated ceramics could be attributed to the generation of the residual stress, the deflection and the bifurcation of the cracks. Residual stress that developed in the laminated ceramics was also evaluated.

Keywords: laminated ceramics; ZrB₂; Mo₅SiB₂; toughening



Citation: Wang, Y.; Li, M.; Wang, H.; Shao, G.; Zhu, J.; Liu, W.; Wang, H.; Fan, B.; Xu, H.; Lu, H.; et al. The Fabrication and Mechanical Properties of Laminated ZrB₂-Mo₅SiB₂ Ceramics with an Mo-Mo₅SiB₂ Interlayer. *Metals* **2021**, *11*, 2018. <https://doi.org/10.3390/met11122018>

Academic Editor: Frank Czerwinski

Received: 7 November 2021

Accepted: 7 December 2021

Published: 14 December 2021

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1. Introduction

Zirconium diboride (ZrB₂), one of the most promising ultra-high temperature ceramic (UHTC) materials, presents an excellent combination of physicochemical properties, such as an ultra-high melting point (3245 °C), high hardness and stiffness, a relatively low thermal expansion coefficient and density, and excellent chemical and physical stability at high temperatures [1–5]. These properties make the material capable of many high-temperature structural applications, including use for hypersonic vehicles, high-temperature shielding, and thermal protection systems [6–9]. However, the strong covalent bond between Zr and B results in ZrB₂ ceramics with low fracture toughness and a high sintering temperature, factors which become a key limitation to their applications [10–12]. Once the material has microcracks due to internal stress or external force, catastrophic failure may occur, which greatly reduces the reliability of the material. Therefore, the low toughness of ZrB₂ ceramics needs to be improved in order to increase reliability and safety, especially for elevated temperature conditions.

A number of studies have attempted to enhance the toughness of the composites [13–22], one of the most effective toughening methods is laminated toughening [23–27]. This method has achieved great effects in materials, such as SiC/TiB₂, Al₂O₃/ZrO₂, Si₃N₄/BN laminated composites, etc. For laminated materials, residual internal stress is generated due to the difference of the thermal expansion coefficient of the critical layer, the change

of layer number and layer thickness ratio, or phase transformation. The stress acts at the crack tip to arrest crack propagation, achieving the strengthening and toughening of the materials.

In the past decade, ZrB₂ based laminated ceramics have been extensively studied. Zhang et al. [28] have prepared ZrB₂/SiC composites with a strong laminated structure by taking advantage of the differences in physical properties of the different layers. As such, the fracture toughness of the sample reached 7.03 MPa·m^{1/2}. Wang et al. [29] have fabricated laminated ZrB₂/Mo composites with a relatively high fracture toughness (7.52 ± 0.12 MPa·m^{1/2}) and high bending strength (451 ± 20 MPa) by introducing 6% Si and 4% B in the interlayers.

Mo₅SiB₂ can be synthesized by adding an appropriate amount of Si and B to Mo, which has a high melting point (2160 °C), good capability of anti-oxidation, and excellent high-temperature strength [30–34]. So far, a few studies have explored the performance of ZrB₂-MoSi₂ ceramics [35], but there are almost no papers devoted to the research of ZrB₂-Mo₅SiB₂ ceramics. Mo has a high melting point (2620 °C), low thermal expansion coefficient (5.2 × 10⁻⁶ K⁻¹), and excellent thermal shock resistance. These properties of Mo are close to ZrB₂, which prevent ceramic cracking due to the mismatch of thermal expansion coefficient and other parameters [36]. In addition, it was reported that better performance could be obtained after dispersing Mo₅SiB₂ into Mo [37]. K.Yoshimi et al. [38] produced Mo-Mo₅SiB₂ in situ composites, and the results showed that the composites had excellent high-temperature mechanical properties and oxidation resistance when the Mo volume fraction was 34%. Therefore, 70 wt.% Mo-30 wt.% Mo₅SiB₂ (that is 66 vol.% Mo-34 vol.% Mo₅SiB₂) was selected as the interlayer in this work.

The main objective of the present research was to fabricate laminated ZrB₂-Mo₅SiB₂ ceramics with high toughness. Herein, the laminated ceramics with ZrB₂-Mo₅SiB₂ as the matrix layer and Mo-Mo₅SiB₂ as the interlayer were prepared by tape casting and hot press sintering. The microstructure and mechanical properties of the laminated ceramics, such as fracture toughness and flexure strength, were characterized. The effect of the residual stress between the matrix layer and the interlayer was also discussed and analyzed in detail.

2. Materials and Methods

Commercial ZrB₂ powder (1–5 μm, 99.5%), Mo powder (0.5 μm, 99.0%), Si powder (0.5 μm, 99.0%), and B powder (1–5 μm, 99.5%) supplied by Zhongnuo New Material Technology Co., Ltd. were used as raw materials. Polyethylene glycol (PEG), Polyvinyl butyral (PVB), and Tricresyl phosphate (TCP) provided by Shanghai Macklin Biochemical Co. were chosen as the dispersant, binder, and plasticizer, respectively. In order to improve the fracture toughness of ZrB₂-based ceramics, laminated ZrB₂-Mo₅SiB₂ ceramics with Mo-Mo₅SiB₂ interlayer were designed based on the “brick-and-mortar” of the nacre. ZrB₂-Mo₅SiB₂ was used as “brick” material and Mo-Mo₅SiB₂ was used as “mortar” material, respectively. There were two composites as matrix layers: 80 wt.% ZrB₂-20 wt.% Mo₅SiB₂ (termed ZM20) and 70 wt.% ZrB₂-30 wt.% Mo₅SiB₂ (termed ZM30). There was one composite as an interlayer: 70 wt.% Mo-30 wt.% Mo₅SiB₂ (termed IL). The layer thickness ratio (λ) was the thickness of the matrix layer to the thickness of the interlayer. There were five different ratios (λ = 7, 10, 13, 16, 19) in this work. The schematic illustration of the designed laminated ceramics is shown in Figure 1.

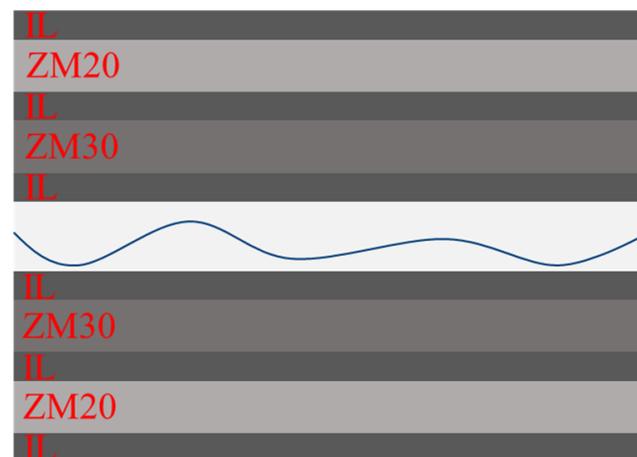


Figure 1. Schematic illustration of the designed laminated ceramics: ZM20 (80 wt.%ZrB₂-20 wt.%Mo₅SiB₂), IL (70 wt.% Mo-30 wt.%Mo₅SiB₂), ZM30 (70 wt.%ZrB₂-30 wt.%Mo₅SiB₂).

The fabrication process of the laminated ceramics is schematically illustrated in Figure 2. The preparation procedure could be summarized in the following steps: (a) synthesis of Mo₅SiB₂, (b) preparation of the slurries of the matrix layer and the interlayer, (c) tape casting the slurry of the matrix layer and cut into wafers, (d) brushing the interlayer slurry onto the matrix layer wafer according to the design of the layer thickness ratio, (e) packing the stacked layers into a graphite die in an alternated sequence, (f) burning out the organic matter of materials and hot pressing.

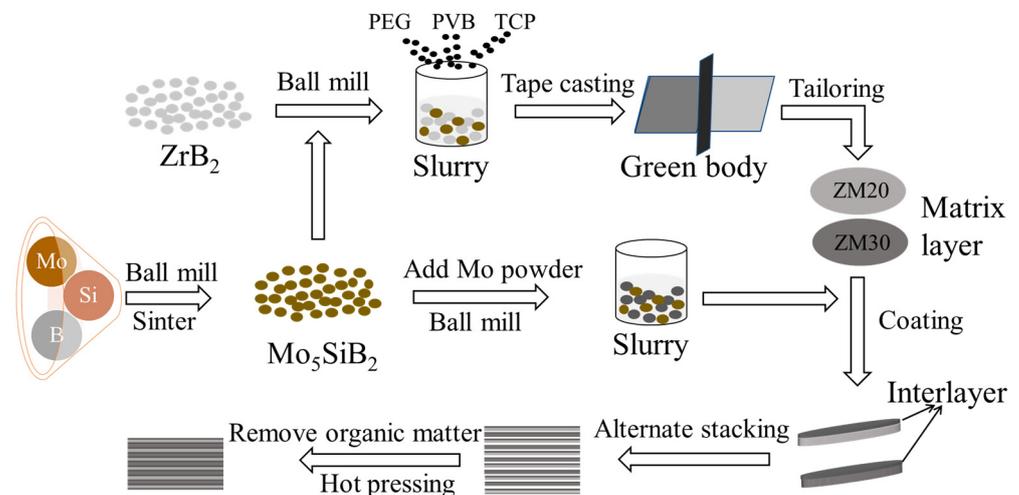


Figure 2. Flow chart for fabrication of the laminated ceramics.

For the synthesis of Mo₅SiB₂, the molar ratio of Mo: Si: B powder was 5:1:2. After ball milling, the mixed powder was sintered at 10 °C/min to 1000 °C and at 5 °C/min to 1450 °C for 2 h in an argon atmosphere. Afterwards, the two matrix compounds (80 wt.% ZrB₂-20 wt.% Mo₅SiB₂ and 70 wt.% ZrB₂-30 wt.% Mo₅SiB₂) and the interlayer compounds (70 wt.%Mo-30 wt.%Mo₅SiB₂) were separately ball-milled in a planetary ball mill at 300 r/min for 6 h. The ethanol and zirconia balls were used as the solvent and the grinding medium, respectively. The weight ratio of the powder to the grinding balls was 1 to 3. The uniformly dispersed powder was obtained after rotating evaporation and sieving. In order to obtain the uniform slurry, the obtained powders were firstly dissolved in alcohol, then mechanically stirred for 3 h with the addition of 0.2 wt.% PEG, and finally, mechanically stirred for 4 h after adding 4 wt.% PVB and 4 wt.% TCP in sequence. The amount of alcohol was approximately 22 wt.%. After tape casting, the green tape of the matrix layer

was obtained and cut into pieces with $\Phi 30$ mm. According to the design of the layer thickness ratio ($\lambda = 7, 10, 13, 16, 19$), the slurry of the interlayer was brushed on the pieces of the matrix layer. Then the matrix layer and the interlayer were stacked alternately into graphite die, as shown in Figure 1. In order to remove the organic matter in the material, the temperature was increased to 400 °C at a rate of 0.5 °C/min in air and kept for 8 h. The hot pressing was performed at 1900 °C and 30 MPa for 2 h in an Ar atmosphere.

The laminated ceramics were cut into bars and ground for the subsequent tests. Each specimen was polished with a diamond grinding disc. Flexural strength was tested via three-point bending on 2 mm \times 3 mm \times 16 mm bars, using a span of 10 mm and a crosshead speed of 0.5 mm/min by a Universal mechanical testing machine (WD-P4504, Jinan Test Machine Co., Ltd., Shenzhen, China). Fracture toughness was evaluated by a single edge notched beam (SENB) test with a span of 16 mm and a crosshead speed of 0.05 mm/min, using 2 mm \times 4 mm \times 22 mm bars with pre-fabricated cracks of 0.2 mm wide and 2 mm deep by a Universal mechanical testing machine. Microstructures of the laminated ceramics were observed by scanning electron microscopy (JSM-7500F, Japan Electron Optics Laboratory Co., Ltd., Tokyo, Japan). Phase constituents of the synthesized powder and the laminated ceramics were identified by X-ray diffraction analysis (XRD, PANalytical EMPYREAN, Amsterdam, the Netherlands).

3. Results

3.1. Synthesis of Mo_5SiB_2 Powders

The XRD pattern of the obtained Mo_5SiB_2 powder is shown in Figure 3. According to Figure 3, pure Mo_5SiB_2 was successfully synthesized after sintering at 1450 °C for 2 h in an argon atmosphere. Mo_5SiB_2 is the most thermodynamically stable intermetallic compound in the entire Mo-Si-B system. The phase region of Mo_5SiB_2 is extremely narrow, which also causes the slow diffusion of atoms in the Mo_5SiB_2 [39]. The impurities, such as Mo_3Si and Mo_2C , are easily formed at a low sintering rate. Therefore, the synthesis temperature of Mo_5SiB_2 should be reached as soon as possible during the heating process to avoid the formation of some impurities at lower temperatures.

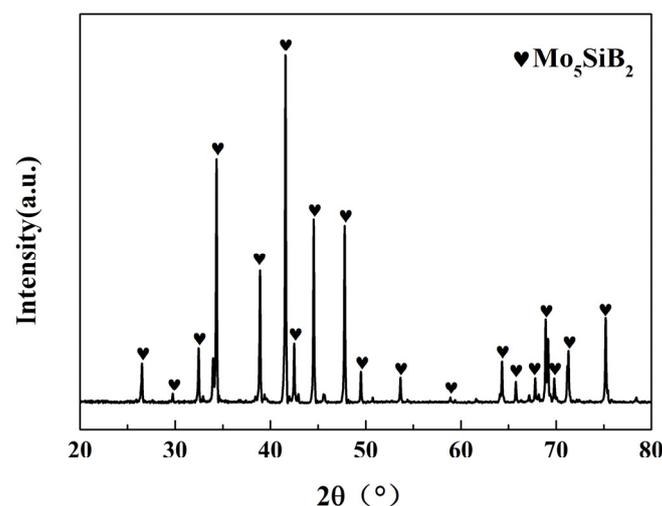


Figure 3. XRD pattern of the obtained Mo_5SiB_2 powder.

3.2. Phase and Microstructure

Figure 4 show the XRD patterns of the planes parallel to the interface (P_{\parallel}) and perpendicular to the interface (P_{\perp}) in the laminated ceramics. According to the results, the main phases of both samples were ZrB_2 , Mo_5SiB_2 , and Mo. Moreover, MoB was also detected. The presence of MoB may be caused by the reaction between Mo in the interlayer and ZrB_2 in the matrix layer, which was beneficial to improve the bonding strength between the different layers.

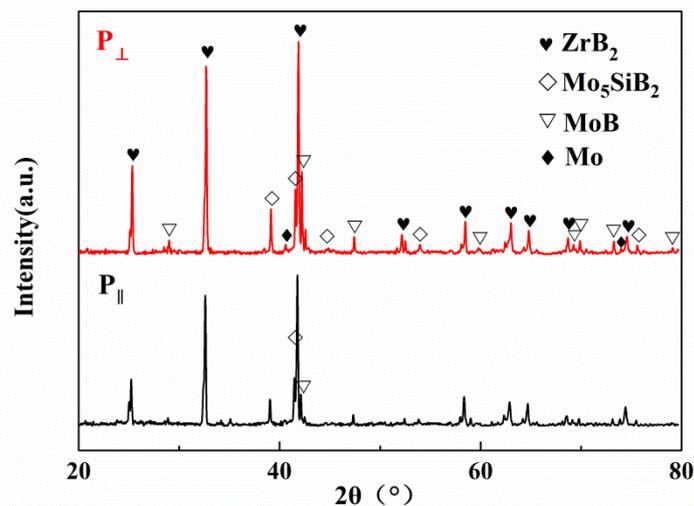


Figure 4. XRD patterns of the planes parallel to the interface ($P_{||}$) and perpendicular to the interface (P_{\perp}) in the laminated ceramics.

SEM micrographs of the polished cross-section of the laminated ceramics with different layer thickness ratios are provided in Figure 5. From the scanned images, it can be clearly seen that the matrix layers and the interlayers were alternately arranged and the thickness of the layers was uniformly distributed. The interface between the different layers was neat and distinct. The ZrB_2 - Mo_5SiB_2 matrix layers were dark and thick, and the Mo - Mo_5SiB_2 interlayers were light and thin. After debinding and sintering, the thicknesses of the matrix layer were reduced from 300 μm to 150 μm . With different thickness ratios between the matrix layer and the interlayer, the thicknesses of the interlayers were also reduced to 21.0 μm , 15.0 μm , 11.5 μm , 9.5 μm , and 8.0 μm , respectively.

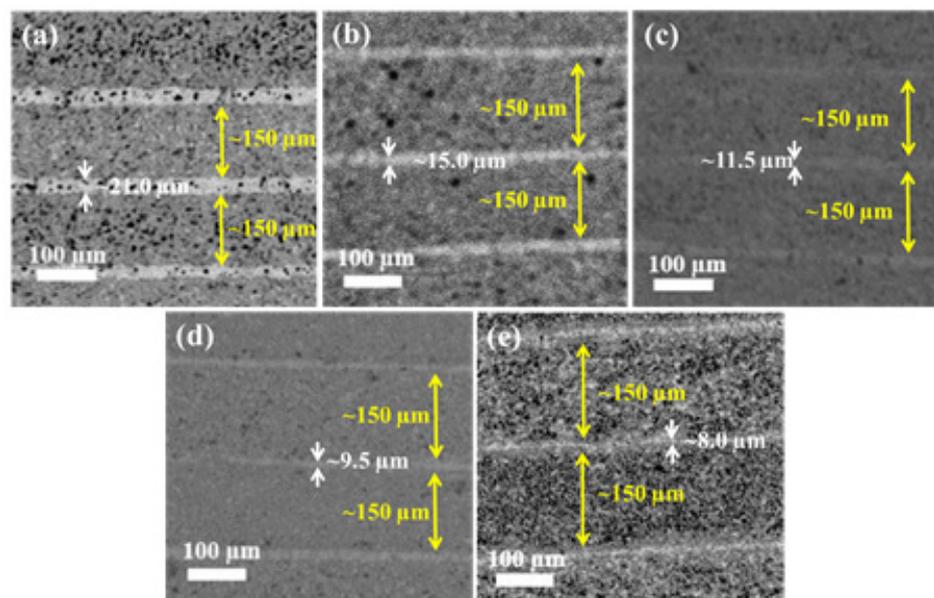


Figure 5. SEM images of the laminated ceramics with different layer thickness ratios: (a) $\lambda = 7$, (b) $\lambda = 10$, (c) $\lambda = 13$, (d) $\lambda = 16$, (e) $\lambda = 19$.

In order to illustrate the toughening mechanism, the crack propagation paths in the laminated ceramics with different thickness ratios were observed in Figure 6. The compositions of the two matrix layers adjacent to the Mo - Mo_5SiB_2 interlayers (IL) were 80wt.% ZrB_2 -20wt.% Mo_5SiB_2 (ZM20) and 70wt.% ZrB_2 -30wt.% Mo_5SiB_2 (ZM30), respec-

tively. The darker layer in the matrix layer was the ZM30 layer and the brighter layer was the ZM20 layer. As shown in Figure 6a,e, the crack propagation paths in the laminated ceramics with the layer thickness ratios of 7 and 19 presented a typically brittle fracture process, where the curves were flat without obvious deflection. As shown in Figure 6b,d, the cracks in the laminated ceramics with the layer thickness ratios of 10 and 16 exhibited multiple inflections inside the matrix layer and at the interface layer, resulting in an elongated crack propagation path. As shown in Figure 6c, the crack in the laminated ceramic with the layer thickness ratios of 13 deflected at the interface, extended laterally for a certain distance, and then extended to the next layer, forming a stepped crack appearance. The step-like cracks microstructure effectively increased the length of the crack, absorbed more fracture energy, and achieved the effect of toughening. The delamination crack at the interface was probably caused by the shear stress due to the difference of residual stress between the matrix layer and the interlayer. The phenomenon of crack deflection also existed in the matrix layer, which was also probably caused by the residual stress due to the thermal mismatch between ZrB_2 and Mo_5SiB_2 .

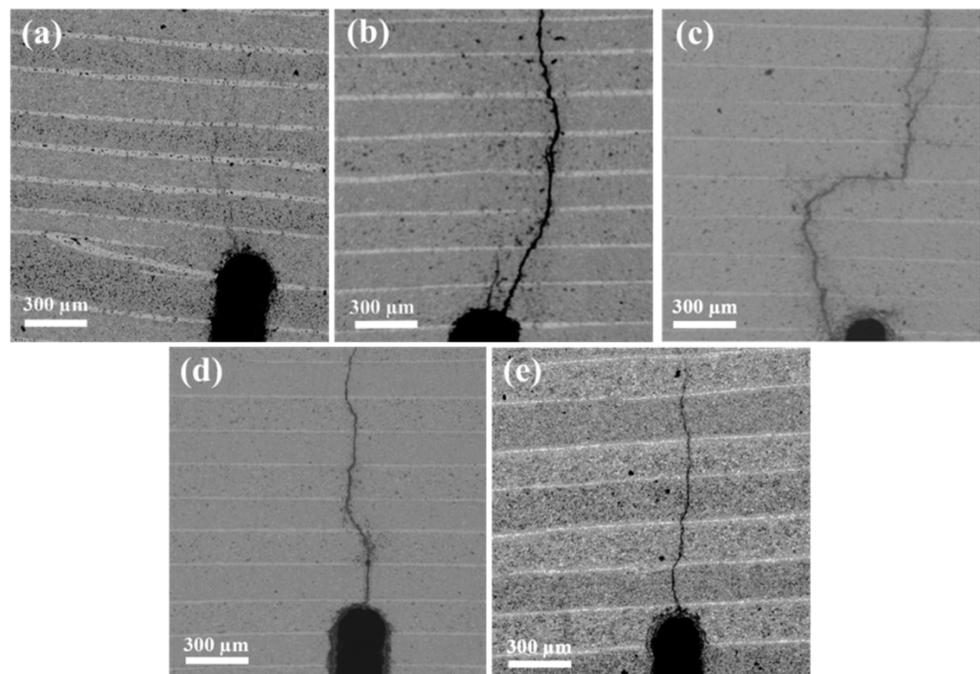


Figure 6. SEM images of the crack propagation paths in the laminated ceramics with different layer thickness ratios: (a) $\lambda = 7$, (b) $\lambda = 10$, (c) $\lambda = 13$, (d) $\lambda = 16$, (e) $\lambda = 19$.

According to the crack propagation paths in the laminated ceramics shown in Figure 6, it could be found that the laminated ceramic with a layer thickness ratio of 13 had the best performance. Figure 7 shows SEM images of the overall cross-section (a), the selected area including the matrix layer and the interlayer (b) and higher magnification of the selected area (c) of the laminated ceramic with the layer thickness ratio of 13. The laminated structure was observed in Figure 7a. As shown in Figure 7b, there were the ZM30 layer, the interlayer, and the ZM20 layer from top to bottom in the enlarged view of the laminated structure. According to the higher magnification image shown in Figure 7c, the fracture mode of the laminated ceramic was a mixture of transgranular fracture and intergranular fracture. Moreover, the matrix layer and the interlayer were tightly bonded because of the reaction of Mo and ZrB_2 at the interface.

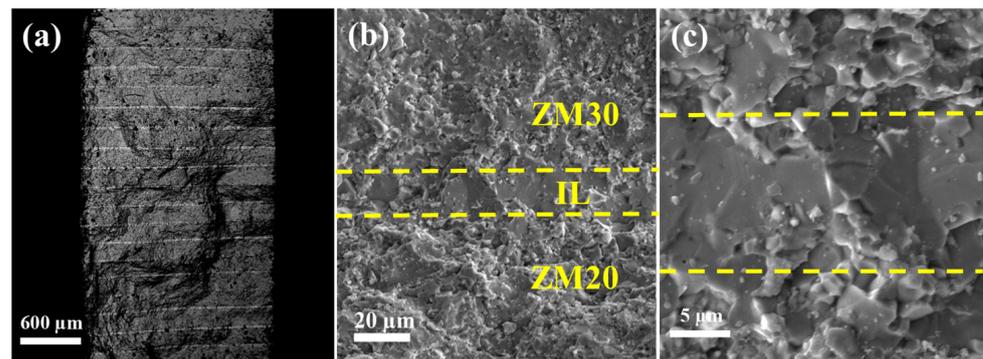


Figure 7. SEM images of the overall cross-section (a) the selected area including the matrix layer and the interlayer (b) and higher magnification of the selected area (c) of the laminated ceramic with the layer thickness ratio of 13.

3.3. Mechanical Properties and Residual Stress

The mechanical properties of the laminated ceramics with different layer thickness ratios are shown in Table 1. As shown, the flexural strength of the laminated ceramics with layer thickness ratios of 7, 10, 13, 16, and 19 were 360.4 ± 35.3 MPa, 375.2 ± 16.1 MPa, 431.6 ± 15.1 MPa, 366.7 ± 28.0 MPa, and 351.7 ± 26.9 MPa, as well as the fracture toughness were 5.54 ± 0.40 MPa·m^{1/2}, 6.94 ± 0.30 MPa·m^{1/2}, 9.89 ± 0.26 MPa·m^{1/2}, 7.06 ± 0.38 MPa·m^{1/2}, and 6.03 ± 0.51 MPa·m^{1/2}, respectively. Both flexural strength and fracture toughness exhibited a tendency of first increasing and then decreasing with the increase of the layer thickness ratio. Among them, the laminated ceramic with a layer thickness ratio of 13 had the highest flexural strength (431.6 ± 15.1 MPa) and fracture toughness (9.89 ± 0.26 MPa·m^{1/2}), which was consistent with the characteristics of the crack propagation paths shown in Figure 6e.

Table 1. Flexural strength and fracture toughness of the laminated ceramics with different layer thickness ratios.

Sample	Thickness Ratio (Matrix Layer/Interlayer)	Flexural Strength (MPa)	Fracture Toughness (MPa·m ^{1/2})
1	7	360.4 ± 35.3	5.54 ± 0.40
2	10	375.2 ± 16.1	6.94 ± 0.30
3	13	431.6 ± 15.1	9.89 ± 0.26
4	16	366.7 ± 28.0	7.06 ± 0.38
5	19	351.7 ± 26.9	6.03 ± 0.51

The crack propagation path in the laminated ceramic was strong evidence for analyzing the failure mechanism of the materials. Furthermore, the crack propagation resistance was impacted by the generation of residual stresses due to the difference in thermal expansion coefficients (CTE) between the matrix layer and the interlayer. Therefore, the residual stress affected the crack propagation direction in the laminated ceramics. In order to prevent excessive internal residual stress from causing block cracking, the design of the laminated structure should take into account the difference in CTEs of the matrix layer and the interlayer. The CTEs, elastic modulus, and Poisson's ratio of the composites can be calculated by the following Equations (1)–(3) [40]:

$$E_C = \frac{1}{2} \left(E_1 V_1 + E_2 V_2 + \left(\frac{1}{\frac{V_1}{E_1} + \frac{V_2}{E_2}} \right) \right) \quad (1)$$

$$\alpha_C = \alpha_1 V_1 + \alpha_2 V_2 \quad (2)$$

$$\nu = \nu_1 V_1 + \nu_2 V_2 \quad (3)$$

where E , α , ν , V are the elastic modulus, thermal expansion coefficient, Poisson's ratio and volume fraction, respectively. Table 2 show the elastic modulus, Poisson's ratio, and CTE of ZrB_2 , Mo_5SiB_2 , Mo , two kinds of matrix layers, and the interlayers. It can be found that the CTE of the two matrix layers was larger than the interlayer. Therefore, the matrix layer was subjected to residual tensile stress, and the interlayer was subjected to residual compressive stress, as shown in Figure 8. The residual stress can be calculated by the following Equations (4) and (5) [41]:

$$\sigma_{res1} = -\frac{nE_1E_2h_2(\alpha_2 - \alpha_1)\Delta T}{n(1 - \nu_1)E_2h_2 + (n + 1)(1 - \nu_2)E_1h_1} \quad (4)$$

$$\sigma_{res2} = \frac{(n + 1)E_1E_2h_1(\alpha_2 - \alpha_1)\Delta T}{n(1 - \nu_1)E_2h_2 + (n + 1)(1 - \nu_2)E_1h_1} \quad (5)$$

where σ , ν , n , h , and ΔT are the residual stress, Poisson's ratio, number of layers, layer thickness, and temperature change, respectively. The subscripts 1 and 2 represent the interlayer and matrix layer, respectively. According to the experiment process, the value of the number of layers was set as 19. From Figure 5, the layer thickness of the interlayer was 21.0 μm , 15.0 μm , 11.5 μm , 9.5 μm , and 8.0 μm , respectively. The value of ΔT was calculated from sintering temperature to room temperature, namely, $1900\text{ }^\circ C - 25\text{ }^\circ C = 1575\text{ }^\circ C = 2148.15\text{ K}$. h_2 was 150 μm , and h_1 was the thickness of the interlayer. The calculated value of the tensile stress and compressive stress of the laminated ceramics with different layer thickness ratios are shown in Table 3.

Table 2. Elastic modulus, Poisson's ratio, and CTEs of raw materials and the laminated ceramics.

Materials	Elastic Modulus (GPa)	Poisson's Ratio	CTE ($\times 10^{-6}\text{ K}^{-1}$)
ZrB_2	489	0.17	6.88
Mo_5SiB_2	383	0.26	8.50
Mo	324	0.31	6.00
80wt.% ZrB_2 - 20wt.% Mo_5SiB_2	471	0.18	7.12
70wt.% ZrB_2 - 30wt.% Mo_5SiB_2	462	0.19	7.25
70wt.% Mo -30 wt.% Mo_5SiB_2	345	0.29	6.96

IL	Compressive (−)
ZM20	Tensile (+)
IL	Compressive (−)
ZM30	Tensile (+)
IL	Compressive (−)

Figure 8. Diagram of the residual stresses applied to various layers in laminated ceramics.

Table 3. Residual stress in laminated ceramics with different layer thickness ratios.

Sample	Thickness Ratio (Matrix Layer/Interlayer)	Tensile Stress in the Matrix Layer (MPa)	Compressive Stress in the interlayer (MPa)
1	7	35.11	−203.15
2	10	22.20	−214.12
3	13	17.41	−218.19
4	16	14.31	−220.82
5	19	11.57	−223.15

As indicated in Table 3, as the layer thickness ratio increased, the residual compressive stress of the interlayer gradually increased from 203.15 MPa to 223.15 MPa, the residual tensile stress of the matrix layer decreased from 35.11 MPa to 11.57 MPa. According to Table 1, the laminated ceramic with a layer thickness ratio of 13 showed the highest flexure strength and fracture toughness. However, its residual compressive stress and the tensile stresses were 218.19 MPa and 17.41 MPa, respectively.

For laminated materials, residual compressive stress was beneficial for improving their mechanical properties and could keep them stable by shielding the applied stress at the crack tip [42,43]. Based on this theory, the surface of the laminated ceramics was designed as a compressive stress layer, which was useful for the improvement of flexural strength. However, the theoretical calculation and analysis of comparative performance revealed that the sample with the maximum residual compressive stress did not have the highest toughness. Since the maximum shielding effect was obtained at a distance equal to the thickness of the compressive layer from the notch tip, the thickness also played an important role. Therefore, the value of fracture toughness was influenced not only by the internal residual stress but also by the position of the notch tip and its distance from the interface. Zhang et al. [28] have also demonstrated this statement. In the case where the notch tip was located in the compressive layer, the crack would deflect at the tip. However, the crack would propagate straight through the layer when the notch tip lay in the tensile layer. Furthermore, the crack would continue to deflect as soon as it propagated to the next compressive layer. The notch depth in this experiment was uniformly set to half the thickness of the samples, and the notches were all located in the tensile layer because the compressive interface layer was so thin.

Within a certain range, such changes in internal residual stress were advantageous for the improvement of the mechanical properties. A possible explanation was that the mutual restraint of the laminated ceramics was primarily achieved through the shear stress between the matrix layer and the interlayer. When the layer thickness ratio increased, the reduction in the thickness of the interlayer caused the restraining force on the adjacent layers to decrease. Then the binding force between the matrix layer and the interlayer became weaker, which resulted in a decrease in the properties of the material. Therefore, the flexure strength and fracture toughness of the laminated ceramics decreased when the layer thickness ratio exceeded 13.

4. Conclusions

Laminated ZrB_2 - Mo_5SiB_2 ceramics with an Mo - Mo_5SiB_2 interlayer were fabricated by tape casting and hot press sintering at 1900 °C for 2h. Pure Mo_5SiB_2 powder was successfully fabricated at 1450 °C. Different layer thickness ratios between the matrix layer and the interlayer were designed. The optimum layer thickness ratio was 13. The maximum fracture toughness and flexural strength of the laminated ceramics were $9.89 \pm 0.26 \text{ MPa}\cdot\text{m}^{1/2}$ and $431.6 \pm 15.1 \text{ MPa}$, respectively. Both fracture toughness and flexural strength showed a trend of first increasing and then decreasing with the increase of the layer thickness ratio. The theoretical calculations of residual stress indicated that the compressive stress in the interlayer increased with the increase of the layer thickness ratio, and the tensile stress in the matrix layer was opposite. The existence of residual stress

made the crack propagate along the interlayer. The deflection and bifurcation of the cracks were the main reason for the toughening of the laminated ceramics.

Author Contributions: Conceptualization, H.W. (Hailong Wang); Data curation, W.L.; Formal analysis, J.Z.; Investigation, G.S.; Methodology, B.F.; Resources, H.X. and H.L.; Supervision, H.W. (Hailong Wang) and R.Z.; Validation, H.W. (Hailiang Wang); Writing—original draft, Y.W.; Writing—review & editing, M.L. All authors have read and agreed to the published version of the manuscript.

Funding: This research was funded by the National Natural Science Foundation of China, grant number 51772275.

Data Availability Statement: This study did not report any data.

Conflicts of Interest: The authors declare no conflict of interest.

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