

Optimal Roving Winding on Toroidal Parts of Composite Frames

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Abstract: Frames made of polymer composites are increasingly used in the aerospace, automotive, and agricultural industries. A frequently used technology in the production line of composite frames is winding rovings onto a non-load-bearing frame to form the structure using an industrial robot and a winding head, which is solidified through a subsequent heat-treatment pressure process. In this technology, the most difficult procedure is the winding of the curved parts of a composite frame. The primary concern is to ensure the proper winding angles, minimize the gaps and overlaps, and ensure the homogeneity of the wound layers. In practice, the curved frame parts very often geometrically form sections of a torus. In this work, the difficulty of achieving a uniform winding of toroidal parts is described and quantified. It is shown that attaining the required winding quality depends significantly on the geometrical parameters of the torus in question. A mathematical model with a detailed procedure describing how to determine the number of rovings of a given width on toroidal parts is presented. The results of this work are illustrated with practical examples of today's industrial problems.

Keywords: curved composite frame; roving winding; optimized winding procedure; winding angle; torus; straight helix; toroidal helix

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1. Introduction

The progress in the development of high-performance materials and structures has been a challenge that has required the development of novel materials with superior and selective mechanical properties and physical features to overcome the standard demand for quality and reliability at different scales [1–3]. The need to increase conventional material efficiency continues to focus materials development on the exploration of materials as composites [4,5]. In this regard, new polymer composites are developed to have special microstructures with unique features to replace conventional materials (e.g., steel, glass, wood) that are frequently used in the design of advanced structures. They are characterized above all by mechanical properties such as tensile, compressive, and torsional strength, lightness, long lifespan, and weather resistance. These exceptional mechanical properties make possible composite structures that can endure extreme loads and boundary conditions [4,6,7]. The important role of such composites in various applications that require the development of structures with complex geometries, such as profiles with open or closed cross-section forms, curved frames with rectangle to circular geometries, and antisymmetric planer shapes [8,9], forced designers to innovate many fabrication methods, such as the vacuum-infusion process, pultrusion process, and robot winding, enabling the possibility of such fabrications [10]. Examples for application of such composite structures are reinforcements for the fuselages, wings, and doors of aircraft, or the

attachment part of windows to helicopter cockpits [11], or chassis reinforcements, car cabins, and door reinforcements in the automotive industry [12]. Composite frames (narrow curves, hollow structure) are utilized in many applications, such as orthopedic devices [13], the manufacture of sports equipment and bicycles [14], ship construction and fishery (hull reinforcements and masts) [5], internal parts of aircraft bodies [15], or to play the role of structural reinforcement [16]. As one of the important applications, such frame structures have been used in oil and petroleum industries as complex pipe or tank structures branching off different cross-sectional configurations of circular to elliptic shapes, utilized for transporting or storing oil and other petroleum liquid materials [17,18]. This is also due to the high potential of composite materials to bear severe loading under harsh environmental conditions. The composite frame structures with long wound fibers are normally fabricated using the filament-winding method by robot or machine to wind continuous strands of tow [19,20]. This winding process is highly adapted to arrange the fiber orientation in such a way that an ideal custom creation with lightweight structures is engineered to meet the desired strength characteristics as dictated by the application [21,22].

In robot filament winding, the winding of rovings on a non-load-bearing frame is performed by a winding head and an industrial robot (see Figure 1a). The frame is generally 3D; it can also have a geometrically complicated shape (see Figure 1b). The frame is attached to the end of the robot's working arm (robot end effector; see Figure 1a). The winding head contains usually three rotating rings (see Figure 2a). Several coils with rovings are placed evenly around the circumference of each ring (see Figure 2c).

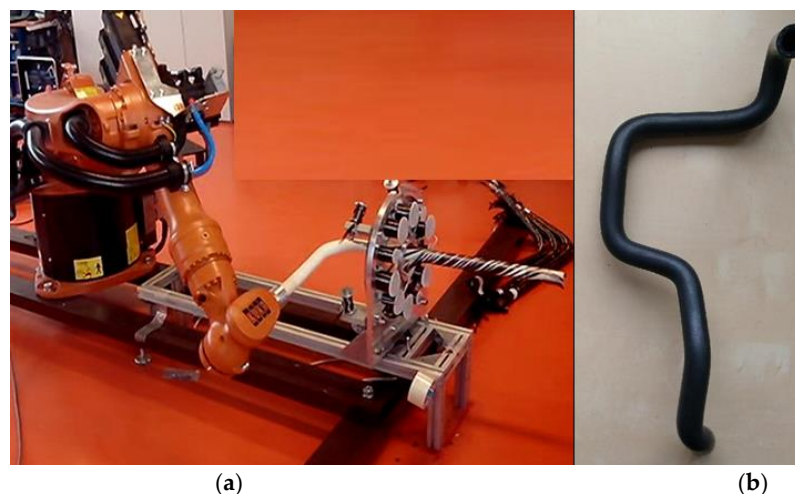


Figure 1. (a) The frame attached to the robot end effector passes through the winding head with a single rotating ring. One layer of winding is formed. (b) An example of a 3D frame with a complicated shape.

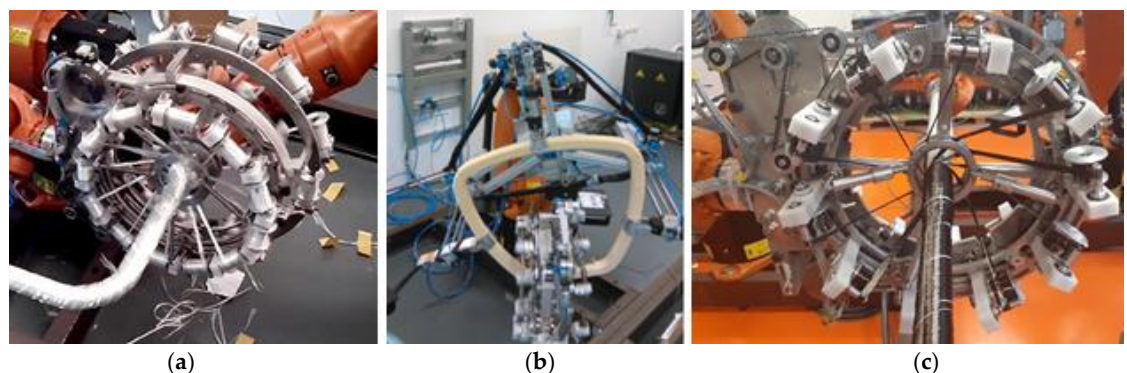


Figure 2. (a) Simultaneous winding of three layers of glass rovings on the open frame. (b) Fixing the closed frame to the robot end effector. (c) Rotating ring of winding head with coils with wound rovings.

Based on the determination of a suitable robot trajectory, the frame passes through the winding head at a constant speed. Each of the three rings performs the winding of a layer of rovings; based on the determination of the necessary angular speed of each ring (control provided by the robot's external axis), each layer is wound at the specified angle. Three layers of windings at different angles are thus created in one pass of the frame through the winding head. A detailed description of the calculation of the optimized robot trajectory is given in [23].

Note 1.

Roving is a fiber system that enables single filaments to be arranged in one parallel collection without twists. Fiber rovings (from carbon, glass, basalt, or aramid fibers) are used to produce 3D composite reinforcement.

Both open and closed frames can be wound using this winding procedure (see Figure 2a,b).

The quality of the composite frame significantly depends on maintaining the required winding angles, and ensuring the homogeneity of the windings (i.e., roving windings without overlaps and gaps). This article focuses on the quality of winding of the composite frame from a geometric perspective. At the same time, the quality of the composite also depends on the material properties of the rovings (e.g., rovings from carbon, glass, aramid, and rovings from recycled materials). However, studying the properties and quality of the fibers used in rovings is not included in the article.

Ideal roving winding can be formed on a frame with a circular cross-section if it forms a straight segment. In this case, a smooth, high-quality winding of the roving onto the frame can be realized. However, winding the curved parts of the frame is more difficult when high-quality winding is required. Simultaneous testing and ensuring the collision-free passage of the frame through the winding head is essential for 3D frames [23,24].

A constant speed of the frame through the winding head is assumed during the winding process. The winding angle is regulated by changing the angular speed of the winding rotated ring of the head. This issue is discussed in detail in [25]. It is also possible to calculate the distance of the roving winding on the frame from the rotating ring (this distance depends on the specified winding angle, the radius of the ring, and the radius of the wound frame; for detail see [25]). This enables smooth and continuous change from a given winding angle to another. The optimization of the number and width of the rovings used for specific winding is discussed in [26]. Based on this optimization, the formation of gaps and overlaps is minimized during the winding process.

As already mentioned, winding the curved parts of the frames is the most difficult process of winding technology. The curved sections of the frame often form parts of a *torus* (see next figures). Based on the literature review and to the best of the authors' knowledge, such a study has not been undertaken before. Therefore, this study focuses on the procedures for the optimal winding of the curved sections of frames shaped like parts of a torus. It is highlighted that achieving acceptable winding quality depends on the torus geometry. A mathematical model of the winding procedure and a detailed analysis of the possibility of achieving an acceptable and optimized winding of the frame with toroidal parts is described in the next sections. In addition, practical examples of the application of the various torus geometries are provided.

2. Materials and Methods

Winding the roving onto a straight frame of circular cross-section using a winding head creates a helix on the surface of the frame (see Figure 3). A standard helix wound on a straight frame is called a *straight helix*.

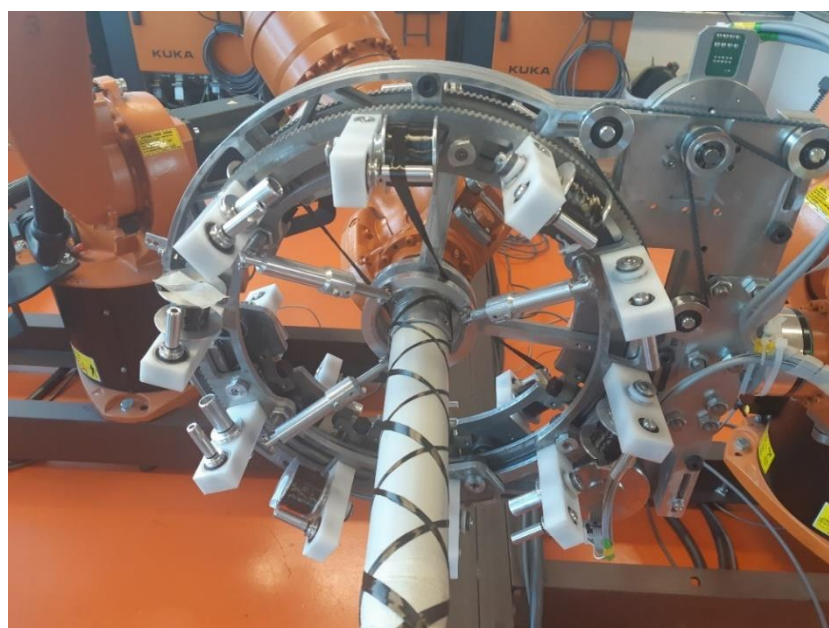


Figure 3. The first rotating ring of the winding head winds one roving at an angle of 45° and the following second rotating ring winds the roving at an angle of -45° .

If the wound roving forms a right-handed straight helix on the surface of the frame, it is said to have a positive winding angle (see Figure 4a). If a left-hand straight helix is formed, this is interpreted as a negative winding angle (see Figure 4b). One turn of the straight helix is shown in both cases Figure 4a,b. The following sections focus only on the winding in a positive direction and the creation of a right-handed straight helix. Winding at a negative angle is completely analogous.

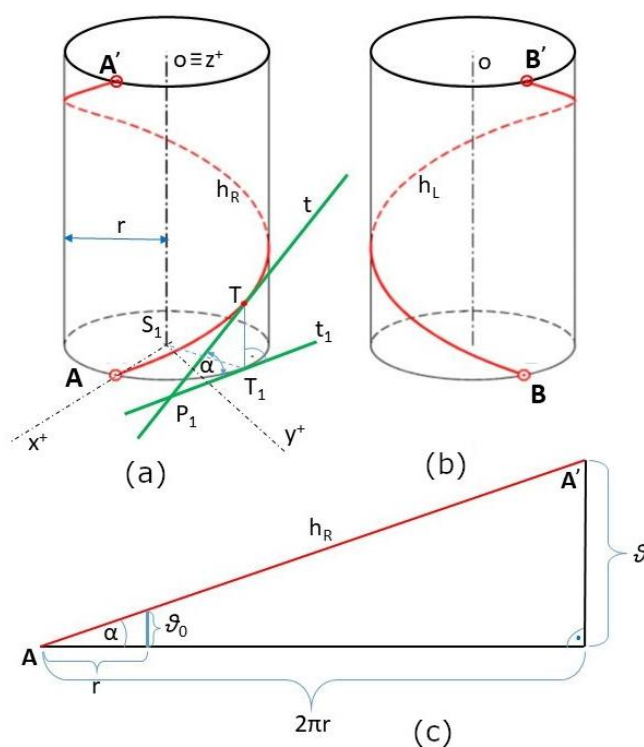


Figure 4. (a) One turn of a right-hand straight helix h_R . (b) One turn of a left-hand straight helix h_L . (c) Characteristics triangle of a straight helix.

One turn of right-handed straight helix h_R (initial point A and endpoint A') is shown in Figure 4a. This straight helix is defined by its axis o (longitudinal axis of the frame), radius r (radius of the frame), and pitch ϑ (height of one helix turn measured parallel to axis o of the helix), which is the Euclidean distance between points A and A' in Figure 4a; for detail see [27]. A characteristic triangle (see Figure 4c) defines the straight helix angle α , where

$$\operatorname{tg} \alpha = \frac{\vartheta}{2\pi r} = \frac{\vartheta_0}{r}. \quad (1)$$

Parameter ϑ is a pitch of straight helix per 2π , and parameter ϑ_0 is a pitch per one radian. Angle α is defined as an angle between tangent t to h_R at point T of the straight helix and its orthogonal projection t_1 into the ground plane (see Figure 4a). In the following, we will call the angle α defined by Relation (1); the winding angle. It is true that $\alpha \in (0, \pi/2)$. In case of $\alpha = \pi/2$, the roving is laid parallel to the axis o and longitudinal to the frame surface.

Specialists in the field of composite materials often call the β angle; the winding angle, as defined by the relationship

$$\beta = \frac{\pi}{2} - \alpha. \quad (2)$$

In this article, the winding angle will mean the angle α defined by Relation (1).

2.1. Torus-Shaped Part of the Frame

As already mentioned in the introduction of the article, winding the curved sections of frames belongs to the most difficult part of the winding technology when using a winding head and an industrial robot. These frame sections often form parts of a torus (see Figure 5).

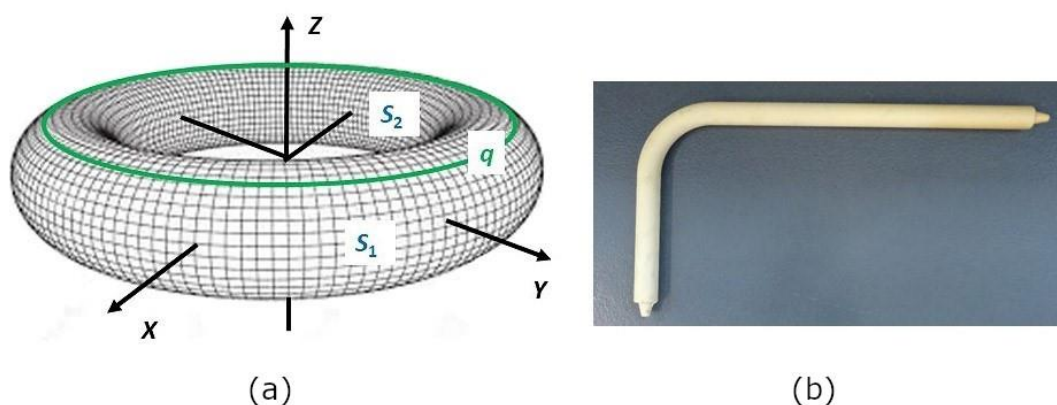


Figure 5. (a) Model of the torus. (b) Non-bearing polyurethane frame for winding rovings with a middle section forming part of the torus.

Instead of describing the winding of the curved part of the frame in the shape of a torus, for simplicity, we provide a description of the procedure for the case of winding rovings on the whole torus.

From a geometric point of view, a torus is a 3D body created by rotating a circle of radius r around a line lying in the plane of this circle and not intersecting this circle (see Figure 6a). The center of the circle of radius r is placed on the y -axis and its distance from origin S of the coordinate system is R , where $0 < r < R$. Rotation of this circle around axis z creates a torus (see Figure 6a). The value of R is called the *major radius* and r the *minor radius* of the torus.

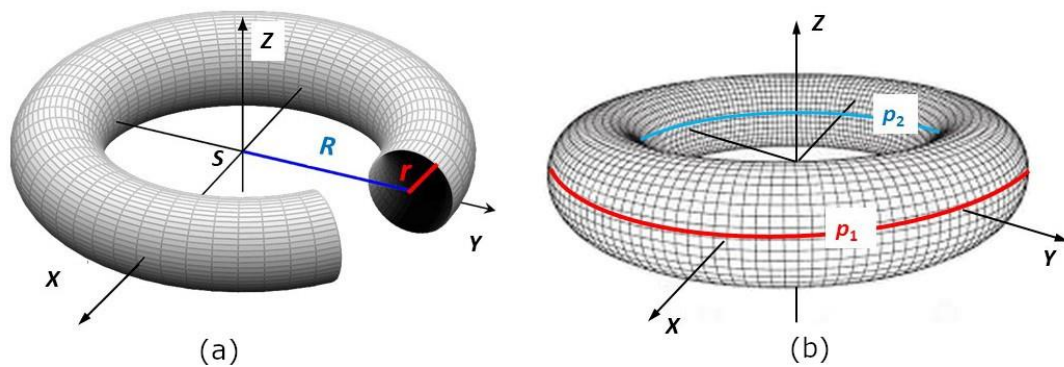


Figure 6. (a) An example of a torus. (b) Torus with outer circumferential circle p_1 and inner circumferential circle p_2 .

Similar to winding a roving on a straight frame at the α angle, it is necessary that the tangent t at point P of the intersection of frame axis o and plane ρ of winding of the roving on the frame is orthogonal to plane ρ (see Figure 7).

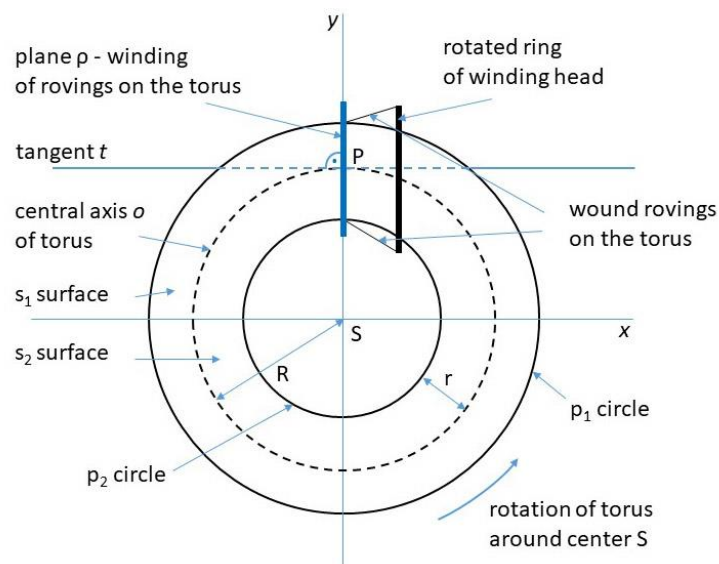


Figure 7. Schematic ground plan of the torus, its parameters, and rotated ring of winding head.

However, the roving is not wound at the specified constant α angle. The winding angle of the roving changes continuously during one turn (it is described in more detail in Sections 2.3 and 2.4; see also [28]).

2.2. Level of Difficulty of Roving Winding

In this paragraph, the focus is on determining the difficulty of winding the roving onto a torus.

The *aspect ratio* a of the torus is defined as

$$a = r/R. \quad (3)$$

The aspect ratio a significantly affects the difficulty of winding the torus. The smaller the value of a , the easier the torus can be wound. In the case of $a \rightarrow 0$, the torus transforms into a straight cylinder.

A vertical cut through the circle q (see Figure 5b) divides the surface s_{total} of the whole torus into outer part s_1 and inner part s_2 (see Figures 5a and 7). The winding difficulty is caused by the different surface sizes of part s_1 and part s_2 of the torus. Integral calculus is used to determine s_1 and s_2 (see Figure 8a). Radius $R + r$ of circle p_1 (see Figures 6a and 7) is called the *outer radius* of the torus and radius $R - r$ of circle p_2 the *inner radius* of the torus.

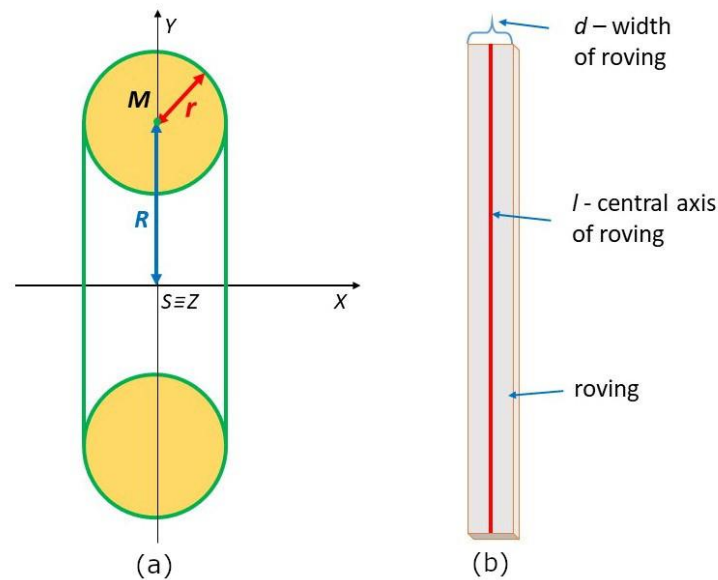


Figure 8. (a) Torus centered at the origin S , xy -plane cut, rotation of circle $k \equiv (M, r)$ around the x -axis. (b) Roving of width d with the central axis l .

Surface s_{total} of the whole torus is composed of partial surfaces s_1 and s_2 , i.e., $s_{total} = s_1 + s_2$, and (see [29], p. 26)

$$s_{total} = 4\pi^2 r R. \quad (4)$$

Surface s_{total} of the whole torus is thus equal to the contents of a rectangle with the lengths of the sides $2\pi R$ and $2\pi r$ (see Figure 9).

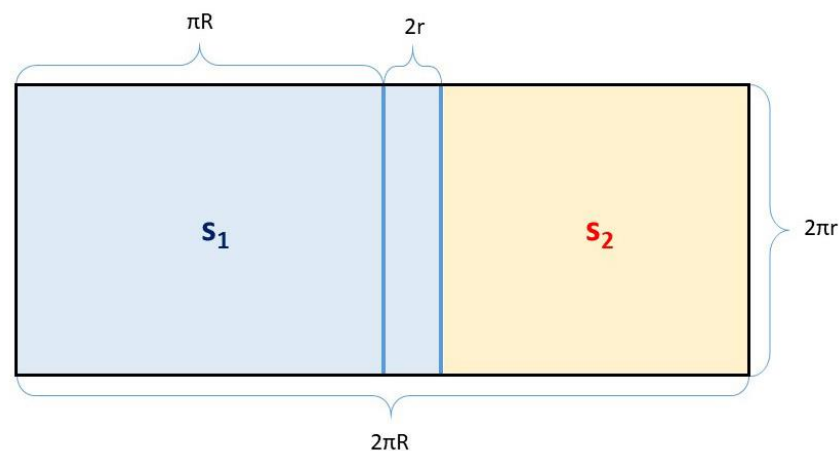


Figure 9. Graphical representation of the ratio of the size of partial surfaces s_1 and s_2 , and $s_{total} = s_1 + s_2$.

In addition, the focus is concentrated on the calculation of values s_1 and s_2 . Surface s_1 is created by rotating the curve $f(x) = R + \sqrt{r^2 - x^2}$ around the x -axis, where

$x \in (-r, r)$ (see [30], p. 107; Figure 8a). Therefore, the size of surface s_1 can be calculated by the following procedure:

$$\begin{aligned}
 s_1 &= 2\pi \int_{-r}^r f(x) \sqrt{1 + (f'(x))^2} dx = 2\pi \int_{-r}^r (R + \sqrt{r^2 - x^2}) \cdot \sqrt{1 + \left(\frac{-x}{\sqrt{r^2 - x^2}}\right)^2} dx = \\
 &= 2\pi \int_{-r}^r (R + \sqrt{r^2 - x^2}) \sqrt{1 + \frac{x^2}{r^2 - x^2}} dx = 2\pi \int_{-r}^r (R + \sqrt{r^2 - x^2}) \cdot \sqrt{\frac{r^2 - x^2 + x^2}{r^2 - x^2}} dx = \\
 &= 2\pi \int_{-r}^r (R + \sqrt{r^2 - x^2}) \cdot \frac{r}{\sqrt{r^2 - x^2}} dx = 2\pi Rr \int_{-r}^r \frac{1}{\sqrt{r^2 - x^2}} dx + 2\pi r \int_{-r}^r 1 dx = \\
 &= 2\pi \left[Rr \arcsin \frac{x}{r} + rx \right]_{-r}^r = 2\pi [Rr \arcsin 1 + r^2 - (Rr \arcsin(-1) - r^2)] = \\
 &= 2\pi \left[Rr \frac{\pi}{2} + r^2 - \left(Rr \left(-\frac{\pi}{2} \right) - r^2 \right) \right] = 2\pi^2 Rr + 4\pi r^2.
 \end{aligned}$$

In the previous derivation, the relation $\int \frac{dx}{\sqrt{r^2 - x^2}} = \arcsin \frac{x}{r}$ (see [30], p. 150) is used. This relationship also follows from the derivative of composite function $\arcsin \frac{x}{r}$:

$$\left(\arcsin \frac{x}{r} \right)' = \frac{1}{\sqrt{1 - \frac{x^2}{r^2}}} \cdot \frac{1}{r} = \frac{1}{\sqrt{r^2} \sqrt{1 - \frac{x^2}{r^2}}} = \frac{1}{\sqrt{r^2 - x^2}}.$$

The size of the surface s_1 is therefore given by the relation

$$s_1 = 2\pi^2 Rr + 4\pi r^2. \quad (5)$$

Since the following holds: $s_2 = s_{total} - s_1$, and from the Relations (4) and (5), this implies

$$s_2 = 2\pi^2 Rr - 4\pi r^2. \quad (6)$$

Thus, the value for the ratio s_2/s_1 and the use of Relations (5) and (6) is equal to

$$\frac{s_2}{s_1} = \frac{2\pi^2 Rr - 4\pi r^2}{2\pi^2 Rr + 4\pi r^2} = 1 - \frac{8\pi r^2}{2\pi^2 Rr + 4\pi r^2} = 1 - \frac{4r}{\pi R + 2r} < 1. \quad (7)$$

The more the ratio s_2/s_1 in Relation (7) approaches the value 1, the more acceptable the conditions for the winding of rovings are. It follows from Relation (7) that the larger the value of R with respect to r , the better the conditions for roving winding. The sizes of the areas corresponding to s_{total} , s_1 , and s_2 are shown graphically in Figure 9.

Figure 9 shows that the geometrical conditions for roving winding are better the smaller the blue marked area of size $2r \cdot 2\pi r$.

As stated in [26], circumference $o(p_1)$ of the *outer circumferential circle* p_1 (see Figures 6b and 7) is equal to $o(p_1) = 2\pi(R + r)$ and circumference $o(p_2)$ of inner circumferential circle p_2 is equal to $o(p_2) = 2\pi(R - r)$, while it is valid $R > r$ (see Figure 6). This then implies

$$\frac{o(p_2)}{o(p_1)} = \frac{2\pi(R - r)}{2\pi(R + r)} = \frac{R - r}{R + r} = \frac{R + r}{R + r} - \frac{2r}{R + r} = 1 - \frac{2r}{R + r} < 1. \quad (8)$$

It follows from Relation (8) that winding of the torus is easier the closer the $o(p_2)/o(p_1)$ ratio is to 1, i.e., the smaller the positive value of $2r/(R + r)$.

Relations (7) and (8) characterize the difficulty of winding rovings on the curved part of the frame.

Thus, it follows from Relations (7) and (8) that the smaller the value of aspect ratio a defined by Relation (3) (i.e., $R \gg r$), the more homogeneous the winding that can be achieved.

2.3. Mathematical Description of Roving Winding on the Torus

Our attention in this paragraph is focused on the procedure of winding rovings onto the surface of a torus. In the next mathematical model of the roving winding on the surface of the torus, only the central axis l (see Figure 8b) of the roving will be considered.

The torus can be parametrically defined in a 3D right-handed Euclidean space in the form (see [31], p. 65)

$$\begin{aligned}x(\theta, \varphi) &= (R + r \cos \theta) \cos \varphi, \\y(\theta, \varphi) &= (R + r \cos \theta) \sin \varphi, \\z(\theta, \varphi) &= r \sin \theta.\end{aligned}\quad (9)$$

Recall that major radius R denotes the radius of the central axis o of the torus (see Figure 7) and the minor radius r the radius of the tube (see Figures 6a–8a). Parameters θ and φ are the angles that make the whole torus, $\theta, \varphi \in [0, 2\pi)$. Angle θ represents rotation around the tube, whereas φ represents rotation around the torus's central axis o (see Figure 7).

The parametric expression of a right-handed helix wound on a torus can be expressed in the following form [28]

$$\begin{aligned}x(t) &= (R + r \cos(\omega t)) \cos t, \\y(t) &= -(R + r \cos(\omega t)) \sin t, \\z(t) &= r \sin(\omega t)\end{aligned}\quad (10)$$

for $t \in \mathbb{R}$, ω is a real positive constant; parameters R and r have the same meaning as in Relation (9). This winding defined by Relation (10) describes the helix wrapped around the torus and is called the right-handed *toroidal helix* (see Figure 10a).

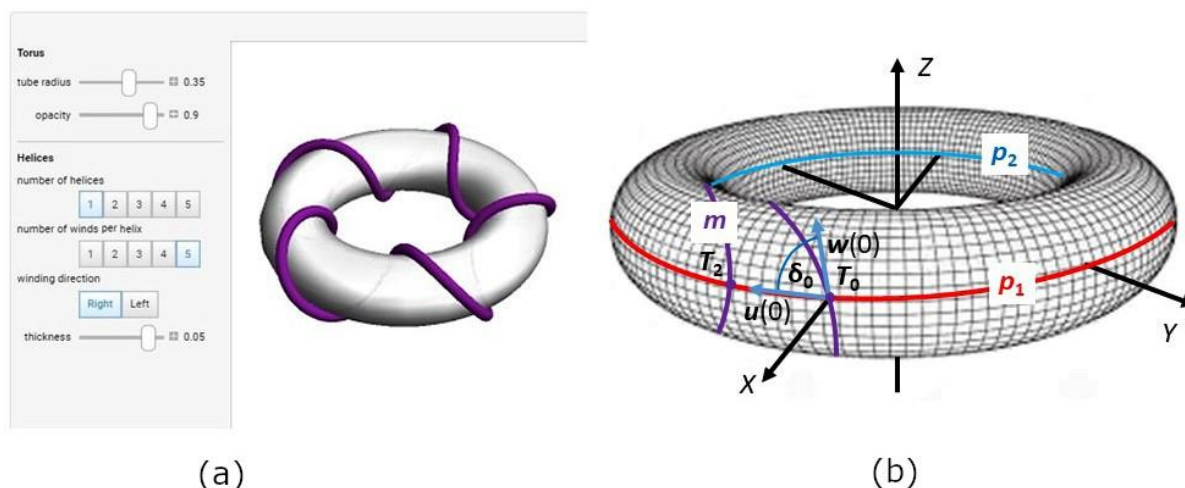


Figure 10. (a) Graph of right-handed toroidal helix for specified parameters $R = 100$, $r = 33$, $\omega = 5$ (number of winds per helix). (b) δ angle clamped by vectors $\mathbf{u}(0)$ and $\mathbf{w}(0)$ at point T_0 of toroidal helix δ . (Figures 10a and 12 are generated by „Toroidal Helices—Wolfram Demonstrations Project” graphics application freely available from <https://www.google.com/search?q=toroidal-helix&oeq=toroidal-helix&aqs=chrome..69i57j0i13i30.10920j0j15&sourceid=chrome&ie=UTF-8#imgsrc=HAW5MhPvHq4pfM>).

When ω is a natural number, the toroidal helix creates a closed loop and ω defines the number of times the toroidal helix coils around the torus (in more detail see [28]). The circumference $O(o)$ of the central axis o of the torus (see Figure 7) is equal to $O(o) =$

$2\pi R$. Following this, toroidal pitch H (specifies the length of repetition along the center axis o) and corresponding reduced toroidal pitch H_0 (pitch per one radian) are defined by the relations

$$H = \frac{2\pi R}{\omega}, \quad H_0 = \frac{H}{2\pi} = \frac{R}{\omega}. \quad (11)$$

The central axis o of the torus passes at the same speed through the winding head as in the case of a straight frame and this and Relation (1) imply that for toroidal pitch,

$$H = \vartheta = 2\pi r \operatorname{tg} \alpha. \quad (12)$$

Recall that α indicates the winding angle on the straight part of the frame.

2.4. Determination of Winding Angle on Torus

When the roving is wound onto a straight frame, a straight helix is formed with the same winding angle at all points of the resulting helix. When winding the toroidal helix, however, the winding angle continuously changes. This paragraph focuses on a more detailed description of the winding angle in the case of the toroidal helix.

A torus with major radius R and minor radius r in Figures 6a and 7 has its center S placed at the origin in the 3D right-handed Euclidean coordinate system. Circles p_1 , p_2 , and central axis o lie in the plane defined by the x and y axes. The points of the wound toroidal helix defined by Relation (10) and lying on the circles p_1 or p_2 can be determined by the following procedure. The z -coordinate of these points is zero. Therefore, it follows from Relation (10)

$$z(t) = r \sin(\omega t) = 0.$$

The relationship is valid when $\sin(\omega t) = 0$, which implies $\omega t = k \cdot \pi$, where k is an arbitrary integer number. From here it follows

$$t = \frac{k \cdot \pi}{\omega}. \quad (13)$$

Applying Relation (13) successively for $k = 0, 1, 2, 3$, it follows $t_0 = 0, t_1 = \pi/\omega, t_2 = 2\pi/\omega$, and $t_3 = 3\pi/\omega$. Points of toroidal helix $T_0 = [x(t_0), y(t_0), z(t_0)]$, $T_2 = [x(t_2), y(t_2), z(t_2)]$ lie on the outer circle p_1 of torus and points $T_1[x(t_1), y(t_1), z(t_1)]$, $T_3[x(t_3), y(t_3), z(t_3)]$ lie on the inner circle p_2 of the torus. The components of these points can be expressed using the relationship (10):

$$\begin{aligned} T_0 &= [x(t_0), y(t_0), z(t_0)] = [R + r, 0, 0], \\ T_1[x(t_1), y(t_1), z(t_1)] &= \left[(R - r) \cos \frac{\pi}{\omega}, -(R - r) \sin \frac{\pi}{\omega}, 0 \right], \\ T_2 &= [x(t_2), y(t_2), z(t_2)] = \left[(R + r) \cos \frac{2\pi}{\omega}, -(R + r) \sin \frac{2\pi}{\omega}, 0 \right], \\ T_3[x(t_3), y(t_3), z(t_3)] &= \left[(R - r) \cos \frac{3\pi}{\omega}, -(R - r) \sin \frac{3\pi}{\omega}, 0 \right]. \end{aligned} \quad (14)$$

Attention is focused on determining the winding angle on the outer circumference of the torus (circle p_1) and on the inner circumference of the torus (circle p_2). The tangent vector $w(t)$ at any point of the toroidal helix can be obtained by the following procedure. From Relation (10) it follows

$$\begin{aligned} \frac{\partial x}{\partial t} &= r \cdot (-\sin(\omega t)) \cdot \omega \cdot \cos t + (R + r \cdot \cos(\omega t)) \cdot (-\sin t) = \\ &= -r\omega \sin(\omega t) \cdot \cos t - (R + r \cos(\omega t)) \cdot \sin t, \\ \frac{\partial y}{\partial t} &= -[r \cdot (-\sin(\omega t)) \cdot \omega \cdot \sin t + (R + r \cdot \cos(\omega t)) \cdot \cos t] = \end{aligned} \quad (15)$$

$$= +r\omega(\sin(\omega t)) \cdot \sin t - (R + r \cdot \cos(\omega t)) \cdot \cos t,$$

$$\frac{\partial z}{\partial t} = r \cdot \cos(\omega t) \cdot \omega = r\omega \cdot \cos(\omega t).$$

The tangential direction vector $\mathbf{w}(t)$ at the point $[x(t), y(t), z(t)]$ has the expression

$$\mathbf{w}(t) = \left(\frac{\partial x}{\partial t}, \frac{\partial y}{\partial t}, \frac{\partial z}{\partial t} \right), \quad (16)$$

where $\frac{\partial x}{\partial t}$, $\frac{\partial y}{\partial t}$, and $\frac{\partial z}{\partial t}$ are defined by Relation (15).

Point T_0 lies on the circle p_1 and has coordinates $T_0 = [R + r, 0, 0]$ according to Relation (14).

Tangent vector $\mathbf{u}(t_0)$ to the circle p_1 at point T_0 lying in the plane of the x and y axes (ground plane) can then be expressed in the form $\mathbf{u}(t_0) = (0, R + r, 0)$; see Figure 10b.

Recall that in Euclidean space E_3 , the length of the vector \vec{u} is defined by the relation $\|\mathbf{u}\| = \sqrt{x_u^2 + y_u^2 + z_u^2}$. The scalar product $\mathbf{u} \cdot \mathbf{v}$ of vectors \mathbf{u} and \mathbf{v} is defined by $\mathbf{u} \cdot \mathbf{v} = x_u \cdot x_v + y_u \cdot y_v + z_u \cdot z_v$. The tangent vector $\mathbf{w}(t_0)$ to the toroidal helix at point T_0 is according to Relations (15) and (16) of the form $\mathbf{w}(t_0) = \mathbf{w}(0) = (0, -(R + r), r\omega)$. The angle δ_0 enclosed by the vectors $\mathbf{u}(0)$ and $\mathbf{w}(0)$ can be determined using the relation (see [32], p. 113)

$$\cos \delta_0 = \frac{\vec{u}(0) \cdot \vec{w}(0)}{\|\vec{u}(0)\| \cdot \|\vec{w}(0)\|} = \frac{(0, -(R+r), 0) \cdot (0, -(R+r), r\omega)}{(R+r) \cdot (\sqrt{(R+r)^2 + r^2 \omega^2})} = \frac{R+r}{(\sqrt{(R+r)^2 + r^2 \omega^2})},$$

thus

$$\delta_0 = \arccos \left(\frac{R+r}{(\sqrt{(R+r)^2 + r^2 \omega^2})} \right). \quad (17)$$

Similarly, tangent vector $\mathbf{w}(t_1)$ to the toroidal helix at point T_1 is $\mathbf{w}(t_1) = \mathbf{w}\left(\frac{\pi}{\omega}\right) = \left(-(R-r) \sin \frac{\pi}{\omega}, -(R-r) \cos \frac{\pi}{\omega}, -r\omega \right)$. Tangent vector $\mathbf{u}(t_1)$ to the circle p_2 at point T_1 lying in the plane of the x, y axes (ground plane) can be expressed in the form $\mathbf{u}(t_1) = (R-r) \cdot \left(-\sin \frac{\pi}{\omega}, -\cos \frac{\pi}{\omega}, 0 \right)$. Thus, it is true for the angle between vectors \mathbf{u}_1 and $\mathbf{w}(t_1)$

$$\cos \delta_1 = \frac{\mathbf{u}(t_1) \cdot \mathbf{w}(t_1)}{\|\mathbf{u}(t_1)\| \cdot \|\mathbf{w}(t_1)\|} = \frac{(R-r) \cdot \left(-\sin \frac{\pi}{\omega}, -\cos \frac{\pi}{\omega}, 0 \right) \cdot \left(-(R-r) \sin \frac{\pi}{\omega}, -(R-r) \cos \frac{\pi}{\omega}, -r\omega \right)}{\sqrt{(R-r)^2 \left(\sin^2 \frac{\pi}{\omega} + \cos^2 \frac{\pi}{\omega} \right)} \cdot \sqrt{(R-r)^2 \left(\sin^2 \frac{\pi}{\omega} + \cos^2 \frac{\pi}{\omega} \right) + r^2 \omega^2}} = \frac{(R-r)^2}{(R-r) \sqrt{(R-r)^2 + r^2 \omega^2}} = \frac{R-r}{\sqrt{(R-r)^2 + r^2 \omega^2}}.$$

Thus it is that

$$\delta_1 = \arccos \left(\frac{R-r}{(\sqrt{(R-r)^2 + r^2 \omega^2})} \right). \quad (18)$$

Note 2.

Let R, r, ω be real numbers and $R > r$. Thus,

$$\frac{R+r}{(\sqrt{(R+r)^2 + r^2 \omega^2})} > \frac{R-r}{(\sqrt{(R-r)^2 + r^2 \omega^2})}. \quad (19)$$

Proof. Assume the validity of Relation (19). After the removal of fractions, partial adjustments of the inequality are gradually made

$$\frac{R+r}{(\sqrt{(R+r)^2 + r^2 \omega^2})} > \frac{R-r}{(\sqrt{(R-r)^2 + r^2 \omega^2})}$$

$$(R+r) \cdot \sqrt{(R-r)^2 + r^2 \omega^2} > (R-r) \cdot \sqrt{(R+r)^2 + r^2 \omega^2}$$

$$(R + r)^2 \cdot ((R - r)^2 + r^2 \omega^2) > (R - r)^2 \cdot ((R + r)^2 + r^2 \omega^2)$$

$$(R + r)^2 \cdot (R - r)^2 + (R + r)^2 r^2 \omega^2 > (R - r)^2 \cdot (R + r)^2 + (R - r)^2 r^2 \omega^2$$

$$(R + r)^2 r^2 \omega^2 > (R - r)^2 r^2 \omega^2$$

$$(R + r)^2 > (R - r)^2$$

The last inequality holds for arbitrary real numbers R, r for $R > r$. From Relation (19) and the fact the \arccos function is decreasing in the interval $\langle 0, 1 \rangle$, it follows that $\delta_1 > \delta_0$. In accordance with Relation (2), it follows that $\tilde{\alpha}_0 = \frac{\pi}{2} - \delta_0$, $\tilde{\alpha}_1 = \frac{\pi}{2} - \delta_1$ and it is true $\tilde{\alpha}_0 > \tilde{\alpha}_1$. The transition from the point T_0 to point T_2 on the circle p_1 is made at one turn of the filament on the outer circumference of the torus. Analogously, transition from the point T_1 to T_3 on the circle p_2 is made also in one turn filament on internal circumference of the torus. This means that on the outer circumference of the torus, the filament is wound at an angle

$$\tilde{\alpha}_{ext} = \frac{\pi}{2} - \delta_0 \quad (20)$$

and in the internal circumference of the torus, the filament is wound at an angle

$$\tilde{\alpha}_{int} = \frac{\pi}{2} - \delta_1. \quad (21)$$

At the same time, $\tilde{\alpha}_{int} < \tilde{\alpha}_{ext}$ and the filament winding angle $\tilde{\alpha}$ varies continuously over the interval $\langle \tilde{\alpha}_{int}, \tilde{\alpha}_{ext} \rangle$. When winding the filament on a straight frame with a circular cross-section, the filament is wound at a constant α angle. However, if the filament is wound on a torus-shaped frame section, the wound $\tilde{\alpha}$ angle changes and is valid $\tilde{\alpha}_{int} < \alpha < \tilde{\alpha}_{ext}$.

2.5. Determination of Torodial Helix Parameter ω

One of the parameters defining the expression of the toroidal helix in Relation (10) is a real ω value. If ω is a natural number, it indicates the number of turns of the toroidal helix on the whole torus. According to Relations (11) and (12), $\frac{2\pi R}{\omega} = 2\pi r \cdot \tan \alpha$ holds. From here it follows

$$\omega = \frac{2\pi R}{2\pi r \cdot \tan \alpha} = \frac{R}{r \cdot \tan \alpha}. \quad (22)$$

The ω value determined by Relation (22) and used in the toroidal helix parametric Expression (10) ensures that the central axis o of the frame will pass through the winding head at the same speed when passing through both the straight and curved torus-shaped parts of the frame. In this case, the length ϑ on the o -axis at one turn of the straight frame part of the frame is equal to the length H on the o -axis at one turn of the toroidal helix on curved part of helix.

2.6. Optimal Number of Rovings Used during Winding

When winding the frame using rovings, it is desirable to ensure the following properties of the wound layer: the winding does not contain any gaps on the outer part of the torus, overlaps of adjacent rovings on the inner part of torus are minimized, and the approximate desired winding angle is maintained. The determination of the appropriate number of rovings when winding a curved torus-shaped frame section is the subject of this paragraph.

First, attention is paid to determining the length of the arc on the circle p_1 at one turn of the toroidal helix (i.e., the length of the arc with the starting point T_0 and the ending point T_2 on the circle p_1 ; see Figure 10b). Similarly, the length of the arc on the circle p_2 with starting point T_1 and ending point T_3 will be determined.

From the parametric expression of the coordinates of points T_0 , T_1 , T_2 and T_3 in Relation (14) it is clear that the vectors $\overrightarrow{ST_0}$ and $\overrightarrow{ST_2}$ (see Figure 11a) are at an angle $\gamma = \frac{2\pi}{\omega}$ and analogously vectors $\overrightarrow{ST_1}$ and $\overrightarrow{ST_3}$ are also at the same angle γ . At one turn of the toroidal helix, point $T_0 \in p_1$ corresponds to point $T_2 \in p_1$ and point $T_1 \in p_2$ corresponds to point $T_3 \in p_2$. The arc length l_{02} of circle p_1 with origin point T_0 and end point T_2 is given by relation (see [29], p. 11)

$$l_{02} = \gamma \cdot (R + r). \quad (23)$$

Angle γ is given in Relation (23) in arc measure. Analogously the arc length l_{13} of circle p_2 with origin point T_1 and end point T_3 is given by the relation

$$l_{13} = \gamma \cdot (R - r). \quad (24)$$

From Relations (23) and (24), it follows that the difference g of the arc lengths l_{02} and l_{13} is equal to $g = l_{02} - l_{13} = 2\gamma r$. As the g -value increases, it becomes more difficult to ensure quality winding of the rovings on the curved part of the frame.

Let d denote the width of the roving (see Figure 8b). The appropriate number of rovings when winding the curved part of the frame is determined by making one turn of the toroidal helix. Recall that δ_0 is the angle that the tangent vector $\mathbf{w}(0)$ of the toroidal helix makes with the tangent vector $\mathbf{u}(0)$ of the circle p_1 at the point T_0 (see Figure 10b, Relation (17)). It is valid (see Figure 11b) that $\sin \delta_0 = d/c_0$, where c_0 denotes the width of the wound roving on the circle p_1 . From here it follows

$$c_0 = d / \sin \delta_0. \quad (25)$$

Thus, the optimized number n of rovings used during torus winding is equal to

$$n = \left\lceil \frac{l_{02}}{c_0} \right\rceil. \quad (26)$$

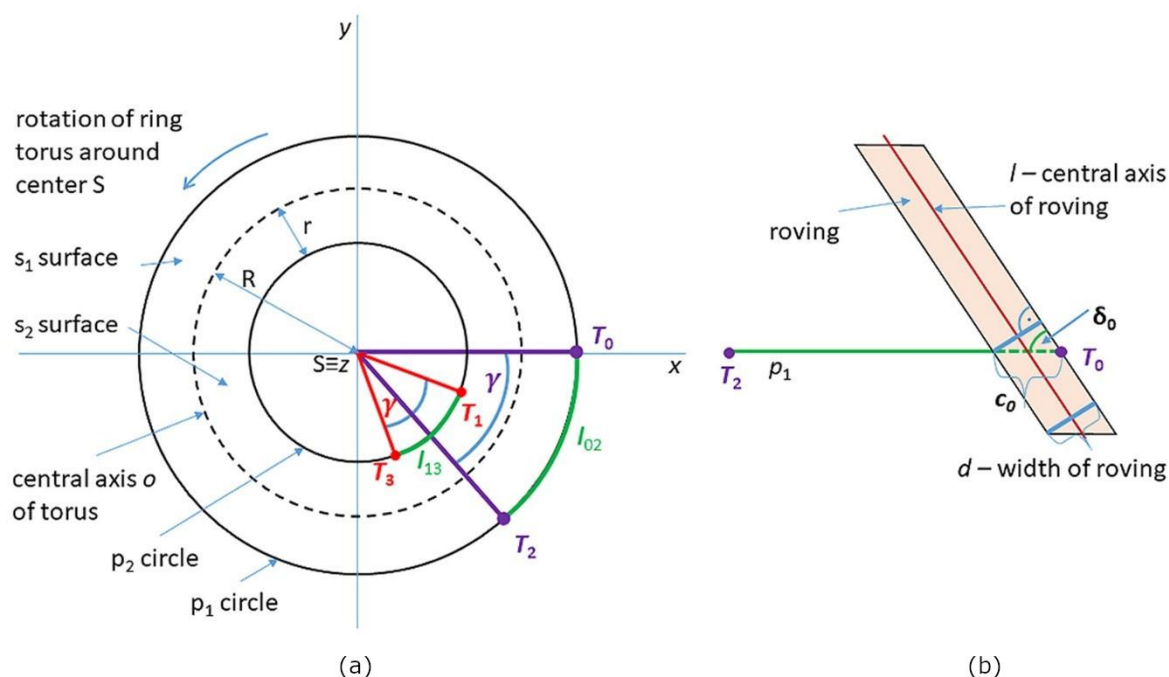


Figure 11. (a) Arc length part l_{02} of circle p_1 and arc length part l_{13} of circle p_2 . (b) Laying a roving of width d at an angle α_{ext} (relation 18a). Value c_0 indicates the length of the wound roving on connecting points T_0 and T_2 .

Note 3.

The ceiling $\lceil x \rceil$ of a real number, x , is defined as $\lceil x \rceil = \min\{p \in \mathbb{Z}; p \geq x\}$, where \mathbb{Z} denotes a set of integers.

Further the total sum ε_{02} of overlaps of adjacent rovings on the circle p_1 when winding a layer of rovings within one turn of the toroidal helix is equal to

$$\varepsilon_{02} = n \cdot c_0 - l_{02}. \quad (27)$$

The overlap $\tilde{\varepsilon}_{02}$ of two adjacent rovings on the circle p_1 is then equal to

$$\tilde{\varepsilon}_{02} = \frac{\varepsilon_{02}}{n} \quad (28)$$

Similarly, the angle δ_1 at the point T_1 lying on the circle p_1 is defined. Parameter c_1 denotes the width of the wound roving on the circle p_1 . Thus,

$$c_1 = d / \sin \delta_1 \quad (29)$$

and the total sum ε_{13} of overlaps of adjacent rovings on the circle p_2 within one turn of the toroidal helix is equal to

$$\varepsilon_{13} = n \cdot c_1 - l_{13}. \quad (30)$$

The overlap $\tilde{\varepsilon}_{13}$ of two adjacent rovings on the circle p_2 is then equal to

$$\tilde{\varepsilon}_{13} = \frac{\varepsilon_{13}}{n}. \quad (31)$$

Relations (26)–(30), (31) allow us to determine the optimized number of rovings when winding the torus. For a given roving width d , the minimum number of rovings used in the winding process can be determined. This prevents the formation of gaps between the rovings and at the same time ensures minimum overlaps between adjacent rovings on the outer circumference of the torus (circle p_1). At the same time, overlaps of adjacent rovings on the inner circumference of the torus (circle p_2) are minimized.

When using n rovings (n is defined by Relation (26)) of width d when winding the curved part of the torus-shaped frame, the n coils with wound rovings are distributed evenly around the circumference of the rotating ring of the winding head (see Figure 3). When winding the curved part of the frame, the relationships given in Section 2.6 apply. After the transition to the straight part of the frame, the rovings are wound at the desired α angle. The process of winding rovings onto a straight frame is discussed in detail in [26] and [25].

Note 4.

Relation (10) defines a toroidal helix wound on the torus. Consider hereafter only the central axes l of n rovings (see Figure 8b) wound on the torus. Following this, these axes form on the torus *regular toroidal n -helix* ([28]; see Figure 12).

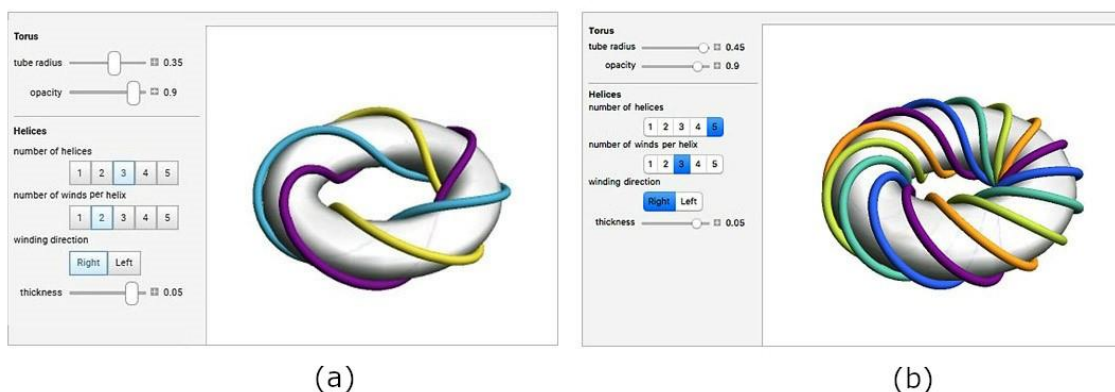


Figure 12. Examples of regular right-handed toroidal (a) 3-helix: major radius $R=10$, minor radius $r=2.5$, $\omega=2$ (number of winds per helix), (b) 5-helix: major radius $R=10$, minor radius $r=4.5$; $\omega=3$.

3. Results and Discussion

This chapter focuses on the practical applications of derived relationships presented in the previous paragraphs.

3.1. Determining the Difficulty of Torus Winding

As mentioned in the previous part of the article, winding a curved frame section with a circular cross-section is one of the most difficult parts of winding technology. Often the curved part of the frame is shaped in 2D and forms part of the torus (see Figure 5a). The three basic characteristics of the difficulty of performing a quality roving winding on a torus are applied in Table 1. The first column contains the values of major radius R and the second column minor radius r of the torus (see Figure 6a). The third column shows the values of aspect ratio a defined by Relation (3). The smaller the value of the parameter a is, the more suitable the conditions for winding (at $a \rightarrow 0$ the torus becomes a straight frame). The penultimate column contains the ratio of the surface area of the inner part s_2 and the outer part s_1 of the total torus surface (Figure 5a). The closer the ratio s_1/s_2 is to 1, the more suitable the torus is for winding. The last column shows the values of the ratios $o(p_2)/o(p_1)$. Here $o(p_1)$ denotes the circumference of the circle p_1 on the outer circumference of the torus (Figure 6b) and $o(p_2)$ the circumference of the circle p_2 on the inner circumference of the torus. Again, the closer the ratio $o(p_2)/o(p_1)$ is to 1, the better winding can be achieved.

Table 1. Characteristics indicating the level of difficulty of performing a quality roving winding.

Major Radius (R) [mm]	Minor Radius (r) [mm]	Aspect Ratio (a)	Ratio $\frac{s_2}{s_1}$	Ratio $\frac{o(p_2)}{o(p_1)}$
1000	20	0.02	0.9748	0.9607
	500	0.5	0.5171	0.3333
	800	0.8	0.2407	0.1111
500	50	0.1	0.8802	0.8181
	100	0.2	0.7741	0.6666
	400	0.8	0.3251	0.1111
100	20	0.2	0.7741	0.6666
	50	0.5	0.5171	0.3333
	90	0.9	0.2715	0.0526
50	10	0.2	0.7741	0.6666
	20	0.4	0.5941	0.4285
	30	0.6	0.4472	0.2500

Figure 13 shows three torus floor plans for the given pairs of R and r values from the fourth to sixth rows of Table 1. From Figure 13 and Table 1, it is clear that the best way to wind rovings onto the torus is in the a/ case and the worst way is in the c/ case.

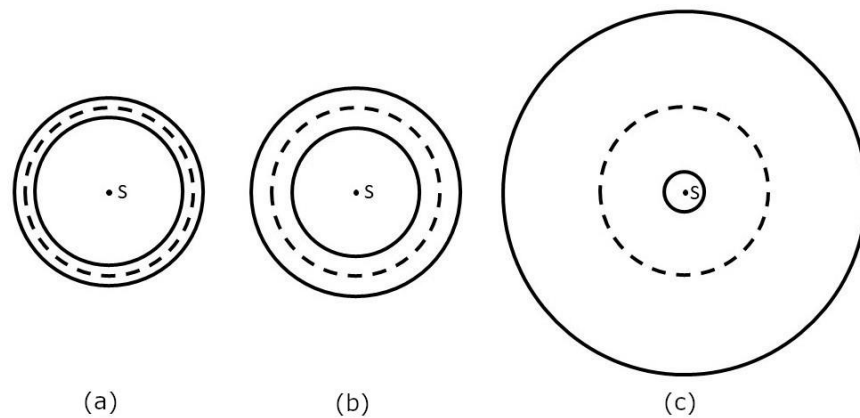


Figure 13. Floor plans of tori with parameters: $R = 500[\text{mm}]$; (a) $r = 50[\text{mm}]$, $a = 0.1$; (b) $r = 100[\text{mm}]$, $a = 0.2$; (c) $r = 400[\text{mm}]$, $a = 0.8$.

If the winding difficulty characteristics are unfavorable for the specified torus, then it is appropriate to consider either another production technology or the use of a differently shaped frame for the manufacture of the composite.

Figure 14 shows the values of the characteristics $a = r/R$ (Relation (3)), s_1/s_2 (Relation (6)) and $o(p_2)/o(p_1)$ (relation (8)). The best conditions for winding the roving on the torus occur in the case of $a \rightarrow 0$, $s_2/s_1 \rightarrow 1$ and $o(p_2)/o(p_1) \rightarrow 1$.

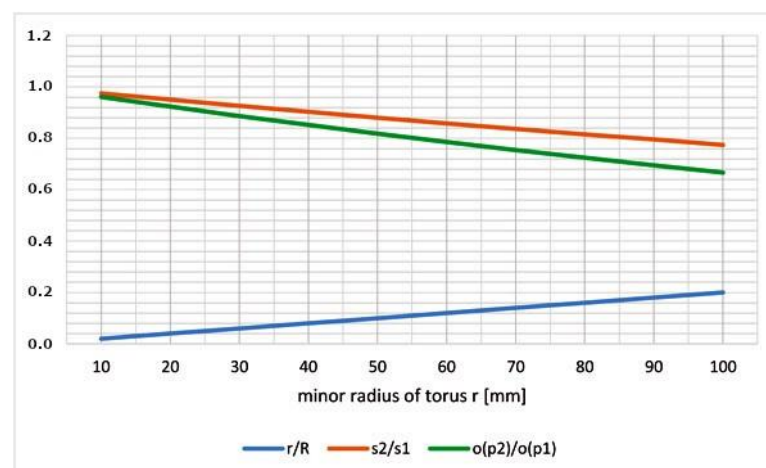


Figure 14. Graphical representation of $a = r/R$, s_2/s_1 , and $o(p_2)/o(p_1)$ values for constant major radius $R = 500[\text{mm}]$ and gradually increasing minor radius r .

It can be clearly seen from Figure 14 that as the value of r increases, the conditions for making a high-quality winding gradually deteriorate (aspect ratio of torus $a = r/R$ gradually increases and values of s_2/s_1 and $o(p_2)/o(p_1)$ gradually decrease). It is possible to use any quantity of $a = r/R$, s_2/s_1 , and $o(p_2)/o(p_1)$ as a measure of the difficulty of the winding, but we recommend the ratio $o(p_2)/o(p_1)$ as the most practically oriented measure.

3.2. Relations between Winding Parameters

The interrelationships of some parameters in winding the straight part of the frame and the curved part of the frame in the shape of a torus section are shown. An example of a frame composed of two straight parts and one curved part in the shape of a torus section is shown in Figure 5b. Table 2 gives examples of different parameter values for a frame with a circular cross-section, which includes a straight part and a curved part in the shape of a torus section. The first column contains the value of the major radius R and several

different values of the minor radius r of the torus in the second column. Aspect ratio a is defined by Relation (3). The following two columns contain the different values of the α angle under which the roving winding is required and the corresponding values of the tangent function. The penultimate column contains the values of ω parameter defined by Relation (22); this parameter is significant in the toroidal helix parametric Expression (10).

Table 2. Interrelation of parameters when winding the straight part of the frame and the curved part of the frame in the shape of the torus part.

Major Radius (R) [mm]	Minor Radius (r) [mm]	Aspect Ratio (a)	Winding Angle (α)		$\tan \alpha$	Parameter ω	Toroidal Pitch (H) [mm]
			[°]	[rad]			
500	25	0.05	5	0.0815	0.0874	228.8329	13.7287
			30	0.5235	0.5773	34.6440	90.68215
			45	0.7853	1.0000	20.0000	157.0796
	50	0.1	5	0.0815	0.0874	114.3118	27.4826
			30	0.5235	0.5773	17.32051	181.3799
			45	0.7853	1.0000	10.0000	314.1592
	100	0.2	5	0.0815	0.0874	61.3496	51.2080
			30	0.5235	0.5773	8.6610	362.7286
			45	0.7853	1.0000	5.0000	628.3185
	450	0.9	5	0.0815	0.0874	12.7128	247.1204
			30	0.5235	0.5773	2.1224	1480.2076
			45	0.7853	1.0000	1.1111	2827.4616

The last column of Table 2 contains values of toroidal pitch H calculated by the use of Relation (11).

Central axis o of the frame passes through the winding ring of the head at a constant speed. The required winding angle can be achieved by regulating the angular speed of the rotating ring of winding head when winding the straight part of the frame (angular speed is controlled by the robot's external axis; for details see [25]). When winding a part of the torus-shaped frame, the winding angle changes during one turn in the range of values $\tilde{\alpha}_{int}$ to $\tilde{\alpha}_{ext}$ defined by Relations (20) and (21). Maintaining the same angular speed of the rotating ring of head when winding the bent part of the frame corresponds to the determination of the ω parameter using Relation (22). Parameter ω is applied in the parametric expression of the toroidal helix in Expression (10). The ω parameter defines the toroidal pitch H (specifies the length of repetition along the centre axis o) by Relation (11).

3.3. Winding Angle of Rovings on the Torus

Based on the values of major radius R , minor radius r of the torus, and the desired winding angle α on the frame, the winding angle $\tilde{\alpha}_{int}$ of the roving on the inner circumference p_2 of the torus (see Figure 6b) and on the outer circumference p_1 of the torus $\tilde{\alpha}_{ext}$ can be determined. It always holds that $\tilde{\alpha}_{int} < \alpha < \tilde{\alpha}_{ext}$. Therefore, the winding angle $\tilde{\alpha}_{ext}$ on the outer circumference of the torus is larger than the winding angle $\tilde{\alpha}_{int}$ on the inner circumference of the torus. The angle of winding $\tilde{\alpha}$ roving on the torus surface changes continuously from $\tilde{\alpha}_{int}$ to $\tilde{\alpha}_{ext}$ and vice versa, $\tilde{\alpha}_{int} \leq \tilde{\alpha} \leq \tilde{\alpha}_{ext}$.

In Table 3, Relation (22) is used to calculate the parameter ω , Relations (18) and (21) are used to determine the angle $\tilde{\alpha}_{int}$, and Relations (17) and (20) are used to determine $\tilde{\alpha}_{ext}$.

Table 3. Determination of the winding angle $\tilde{\alpha}_{int}$ on the inner circumference of the torus (circle p_2) and the winding angle $\tilde{\alpha}_{ext}$ on the outer circumference of the torus (circle p_1) depending on the major radius R of the torus, the minor radius r of the torus and the desired winding angle α .

Major Radius (R) [mm]	Minor Radius (r) [mm]	Aspect Ratio (a)	Winding Angle (α)		$\text{tg } \alpha$	Parameter ω	Angle $\tilde{\alpha}_{int}$ [°]	Angle $\tilde{\alpha}_{ext}$ [°]
			[°]	[rad]				
100	20	0.2	5	0.0815	0.0874	57.2082	3.9968	5.9872
	50	0.5	30	0.5235	0.5773	3.4644	16.1007	40.8909
	90	0.9	45	0.7853	1.0000	1.4148	4.4904	56.1712
50	10	0.2	5	0.0815	0.0874	61.3496	3.73040	5.5857
	20	0.4	30	0.5235	0.5773	4.7755	17.4376	36.2379
	30	0.6	45	0.7853	1.0000	1.6666	21.8021	57.9956

Table 3 clearly shows that the deviations $\tilde{\alpha}_{int}$ and $\tilde{\alpha}_{ext}$ from the specified winding angle α increase with increasing aspect ratio a .

3.4. Determination of Optimal Number of Rovings

Based on the knowledge of the winding of the rovings on the frame from a geometrical point of view, the optimal number of rovings used in winding the new layer can be determined. Knowledge of the major radius R and minor radius r of the torus and the prescribed winding angle α is assumed. As shown in the previous Section 3.3, when the roving is wound onto the torus, the circumference $o(p_1)$ (see Figures 6b and 11a) is larger than the circumference $o(p_2)$. At the same time, it was shown that on the outer circle p_1 the roving is wound at a larger angle $\tilde{\alpha}_{ext}$ than on the inner circle p_2 with angle $\tilde{\alpha}_{int}$.

For the optimum number n of rovings to be used for roving width d , it is desired to create a winding without gaps and with zero or minimal overlap of two adjacent rovings on the outer circle p_1 . At the same time, the size of the overlap of two adjacent rovings can be determined on the inner circle p_2 . This overlap is always larger than on p_1 .

Table 4 shows the calculated values n of the optimal number of rovings used for a given width d and values of R , r , and winding angle α . From the knowledge of values R , r , α and Relation (22), the parameter ω can be determined. At the same time, the overlaps of two adjacent rovings on the outer circumference p_1 and the inner circumference p_2 are determined. By successively using Relations (17), (23), (25)–(28), the overlap $\tilde{\epsilon}_{02}$ of two adjacent rovings on the outer circumference p_1 of the torus can be determined. Similarly, by successively using Relations (18), (24), (29)–(31), the overlap $\tilde{\epsilon}_{13}$ on the inner circumference p_2 of torus can also be determined.

Table 4. Optimized number of rovings n used in winding and the size of overlaps $\tilde{\epsilon}_{02}$ on the outer and $\tilde{\epsilon}_{13}$ on the inner circumference of the torus for given values of R , r , d and α .

Outer Radius (R) [mm]	Inner Radius (r) [mm]	Param. a	Angle Winding (α) [°]	Param. ω	Roving Width (d) [mm]	Optimized Number of Rovings (n)	Outer Overlap ($\tilde{\epsilon}_{02}$) [mm]	Inner Overlap ($\tilde{\epsilon}_{13}$) [mm]
100	20	0.2	10	28.3607	9	3	0.3378	3.1811
	25	0.25	30	6.9282		11	0.7944	3.6238
	30	0.3	45	3.3333		17	0.3467	3.2236
200	10	0.05	10	113.4429	5	3	1.2083	1.5621
			30	34.6410		7	0.4055	0.7795
			45	20.0000		10	0.6526	0.9274

Note 5.

The carbon rovings are from Toho Tenax, a widespread manufacturer of winding rovings. Carbon roving 24K consists of twenty-four-thousand carbon filaments about a diameter $7[\mu\text{m}]$, creating a rectangular cross-section with a width of $9[\text{mm}]$. Carbon rovings marked 12K and 6K have a width of $12\text{K} = 5[\text{mm}]$ and $6\text{K} = 2[\text{mm}]$.

The curved part of a polymer-composite frame after the simultaneous successive winding of three layers of carbon rovings on a non-load-bearing frame under specified winding angles α equal to 45° , -45° , and 45° is shown in Figure 15. The curved part of the frame forms one-quarter of the torus. Subsequently, the wound frame is thermally treated.



Figure 15. Example of the curved part of polymer composite frame with the following parameters: major radius $R = 102.5[\text{mm}]$, minor radius $r = 17.5[\text{mm}]$, width of roving $d = 5[\text{mm}]$. The non-load-bearing polyurethane frame is visible in the vertical section (light colour of the cross-section).

3.5. Recommended Procedure before Starting Winding

Before starting the actual winding procedure on a frame with a curved section in the shape of a torus part, it is advisable to carry out the following preparatory steps.

1. Determine the suitability of winding the rovings on the non-load-bearing frame (Relations (3), (7), and (8) can be used, see Table 1). If the winding conditions are unfavorable, consider whether, for example, to use a differently shaped frame or to choose a different composite manufacturing technology.
2. Calculate parameter ω using Relation (22). Based on the knowledge of this parameter, an estimate of the number of roving revolutions on the whole toroidal helix can be obtained.
3. Determine the maximum winding angle of the roving $\tilde{\alpha}_{ext}$ on the torus at the outer circumference p_1 (see Figure 6b) using Relations (17) and (20). At the same time, determine the minimum winding angle $\tilde{\alpha}_{int}$ on the inner circumference of p_2 by applying Relations (18) and (21). For the required winding angle α for a given layer, the following relation holds: $\tilde{\alpha}_{int} < \alpha < \tilde{\alpha}_{ext}$. During the winding procedure, the winding angle $\tilde{\alpha}$ on the torus changes continuously and $\tilde{\alpha}_{int} \leq \tilde{\alpha} \leq \tilde{\alpha}_{ext}$. Due to the continuously changing winding angle $\tilde{\alpha}$, it is useful to determine whether the changing winding angle satisfies the winding requirements with respect to the planned loading of the polymer composite frame using a suitable modelling software tool (e.g., ABAQUS, ANSYS).
4. Determine the optimized number of rovings n for the winding of the layers at their specified width d . To the selected value of n , calculate the overlap $\tilde{\epsilon}_{02}$ of two adjacent rovings on the outer circumference p_1 and the overlap $\tilde{\epsilon}_{13}$ on the inner circumference p_2 . Following this, select the winding of the roving with the most suitable width d provided by the supplier of rovings.

Based on the above procedure, it is possible to define the optimized the winding procedure.

4. Conclusions

The article focuses on the problem of winding rovings on a non-load-bearing frame with a circular cross-section, and the problem of winding the curved part of the frame is solved. In particular, attention is paid to the case where the curved section of the frame forms part of the torus. The problem is solved from a geometric point of view. Based on the geometric parameters of the torus and the relations derived in the paper, the level of difficulty of the roving layer winding, including the real feasibility of homogeneous winding, can be determined. Based on the given torus and using the relations derived in Chapter 2, the parametric expression of the wound toroidal helix can be determined. As a result, the behavior of the roving when winding on a torus can be described analytically. The winding angle changes continuously within one turn of the roving.

In Section 2.2. we provide three alternative quantities, specifically $a = r/R$, s_2/s_1 , and $o(p_2)/o(p_1)$, that describe the level of difficulty of winding on given toroidal part of the frame. However, we recommend the quantity $o(p_2)/o(p_1)$ as the most practically oriented measure of the winding difficulty.

When winding the roving layer, it is necessary to avoid gaps in the winding on the outer circumference of the torus and at the same time it is necessary to minimize overlaps of adjacent rovings on the inner circumference of the torus. Using the relations from the previous section, the optimal number of rovings used to wind the layer onto the torus can be determined for a given roving width. At the same time, the overlap size of two adjacent wound rovings can be determined.

A greater number of rovings and their shorter length are required when the frame is wound at a greater angle (for a curved section of the frame in the shape of a torus section, a smaller ω parameter corresponds to a greater winding angle). When winding at a smaller winding angle, fewer rovings of greater length are required. The total amount of material required is practically the same, unless we consider the issue of overlapping adjacent rovings. Of course, different frame loads (tension, torsion, etc.) correspond to different suitable winding angles. Practical tests show that it is not advisable to wind the roving on the torus at an angle greater than 45° , as this usually causes the roving to “slide” on the surface of the frame and degrade the entire winding.

Meeting the necessary geometric conditions of winding is a prerequisite for a quality winding of the roving layer. As the frame passes through the winding head, three layers of roving are wound simultaneously at different angles (the winding head contains three rotating rings with coils of wound roving). If more layers of windings are required on the frame, the frame can be passed through the winding head repeatedly.

The problems of winding straight frames with circular cross-sections (especially the smooth transition to another winding angle, the distance of winding roving from the rotating ring of the winding head, and the determination of the optimal number of rovings when winding a layer of rovings) are analyzed mainly in previous published works [25,26]. These articles, together with this paper, comprehensively describe the problem of winding composite frames using rovings. The fulfillment of the required geometrical conditions of winding is a prerequisite for ensuring the production of high-quality polymer composite frames. A detailed procedure for calculating the optimal trajectory of the industrial robot during the winding process even for curved frames is described in [23,33].

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References

- Ghandvar, H.; Jabbar, K.A.; Idris, M.H.; Ahmad, N.; Jahare, M.H.; Koloor, S.S.R.; Petrú, M. Influence of barium addition on the formation of primary Mg₂Si crystals from Al–Mg–Si melts. *J. Mater. Res. Technol.* **2021**, *11*, 448–465.
- Joshani, M.; Koloor, S.; Abdullah, R. Damage mechanics model for fracture process of steel-concrete composite slabs. *Appl. Mech. Mater.* **2012**, *165*, 339–345.
- Shokravi, H.; Mohammadyan-Yasouj, S.E.; Koloor, S.S.R.; Petrú, M.; Heidarrezaei, M. Effect of alumina additives on mechanical and fresh properties of self-compacting concrete: A review. *Processes* **2021**, *9*, 554.
- Farokhi Nejad, A.; Bin Salim, M.Y.; Rahimian Koloor, S.S.; Petrik, S.; Yahya, M.Y.; Abu Hassan, S.; Mohd Shah, M.K. Hybrid and synthetic FRP composites under different strain rates: A review. *Polymers* **2021**, *13*, 3400.
- Rubino, F.; Nisticò, A.; Tucci, F.; Carlone, P. Marine application of fiber reinforced composites: A review. *J. Mar. Sci. Eng.* **2020**, *8*, 26.
- Koloor, S.; Ayatollahi, M.; Tamin, M. Elastic-damage deformation response of fiber-reinforced polymer composite laminates with lamina interfaces. *J. Reinf. Plast. Compos.* **2017**, *36*, 832–849.
- Khan, M.S.; Abdul-Latif, A.; Koloor, S.S.R.; Petrú, M.; Tamin, M.N. Representative cell analysis for damage-based failure model of polymer hexagonal honeycomb structure under the out-of-plane loadings. *Polymers* **2020**, *13*, 52.
- Hassan, M.H.; Othman, A.R.; Kamaruddin, S. A review on the manufacturing defects of complex-shaped laminate in aircraft composite structures. *Int. J. Adv. Manuf. Technol.* **2017**, *91*, 4081–4094.
- Mouritz, A.P.; Gellert, E.; Burchill, P.; Challis, K. Review of advanced composite structures for naval ships and submarines. *Compos. Struct.* **2001**, *53*, 21–42.
- Rajak, D.K.; Pagar, D.D.; Menezes, P.L.; Linul, E. Fiber-reinforced polymer composites: Manufacturing, properties, and applications. *Polymers* **2019**, *11*, 1667.
- Koloor, S.R.; Karimzadeh, A.; Abdullah, M.; Petrú, M.; Yidris, N.; Sapuan, S.; Tamin, M. Linear-nonlinear stiffness responses of carbon fiber-reinforced polymer composite materials and structures: A numerical study. *Polymers* **2021**, *13*, 344.
- Mlýnek, J.; Petrú, M.; Martinec, T.; Rahimian Koloor, S.S. Fabrication of high-quality polymer composite frame by a new method of fiber winding process. *Polymers* **2020**, *12*, 1037.
- Scholz, M.S.; Blanchfield, J.P.; Bloom, L.D.; Coburn, B.H.; Elkington, M.; Fuller, J.D.; Gilbert, M.E.; Muflahi, S.A.; Pernice, M.F.; Rae, S.I.; et al. The use of composite materials in modern orthopaedic medicine and prosthetic devices: A review. *Compos. Sci. Technol.* **2011**, *71*, 1791–1803. <https://doi.org/10.1016/j.compscitech.2011.08.017>.
- Jenkins, M. *Materials in Sports Equipment*; Woodhead Publishing: Sawston, UK, 2003; Volume 1.
- Kalanchiam, M.; Chinnasamy, M. Advantages of composite materials in aircraft structures. *Int. J. Aerosp. Mech. Eng.* **2012**, *6*, 2428–2432.
- Teshnizi, S.S.; Koloor, S.; Sharifishourabi, G.; Ayob, A.; Yazid, Y.M. Mechanical behavior of GFRP laminated composite pipe subjected to uniform radial patch load. *Adv. Mater. Res.* **2012**, *488*, 542–546.
- Okolie, O.; Latto, J.; Faisal, N.; Jamieson, H.; Mukherji, A.; Njuguna, J. Manufacturing defects in thermoplastic composite pipes and their effect on the in-situ performance of thermoplastic composite pipes in oil and gas applications. *Appl. Compos. Mater.* **2023**, *30*, 231–306.
- Khalid, H.U.; Ismail, M.C.; Nosbi, N. Permeation damage of polymer liner in oil and gas pipelines: A review. *Polymers* **2020**, *12*, 2307.
- Quanjian, M.; Rejab, M.; Idris, M.; Kumar, N.M.; Merzuki, M. Robotic filament winding technique (RFWT) in industrial application: A review of state of the art and future perspectives. *Int. Res. J. Eng. Technol.* **2018**, *5*, 1668–1676.
- Sorrentino, L.; Anamateros, E.; Bellini, C.; Carrino, L.; Corcione, G.; Leone, A.; Paris, G. Robotic filament winding: An innovative technology to manufacture complex shape structural parts. *Compos. Struct.* **2019**, *220*, 699–707.
- Sorrentino, L.; Marchetti, M.; Bellini, C.; Delfini, A.; Del Sette, F. Manufacture of high performance isogrid structure by Robotic Filament Winding. *Compos. Struct.* **2017**, *164*, 43–50.
- Kousta, I.; Papingiotis, T.; Vosniakos, G.-C.; Dine, A. On the development of a filament winding robotic system. *Procedia Manuf.* **2018**, *17*, 919–926.
- Martinec, T.; Mlýnek, J.; Petrú, M. Calculation of the robot trajectory for the optimum directional orientation of fibre placement in the manufacture of composite profile frames. *Robot. Comput.-Integr. Manuf.* **2015**, *35*, 42–54.
- Petrú, M.; Mlýnek, J.; Martinec, T. Numerical modelling for optimization of fibres winding process of manufacturing technology for the non-circular aerospace frames. *Manuf. Technol.* **2018**, *18*, 90–98.

25. Mlýnek, J.; Rahimian Koloor, S.S.; Martinec, T.; Petrů, M. Fabrication of high-quality straight-line polymer composite frame with different radius parts using fiber winding process. *Polymers* **2021**, *13*, 497.
26. Mlýnek, J.; Petrů, M.; Ryvolová, M.; Rahimian Koloor, S.S. Winding optimization of composite frame by dry fiber rovings. *J. Ind. Text.* **2022**, *52*, 15280837221114639.
27. Shifrin, T. *Differential Geometry: A First Course in Curves and Surfaces*; University of Georgia: Athens, GA, USA, 2015; p. 24.
28. Olsen, K.; Bohr, J. Geometry of the toroidal N-helix: Optimal-packing and zero-twist. *N. J. Phys.* **2012**, *14*, 023063.
29. Benenson, W.; Harris, J.W.; Stöcker, H.; Lutz, H. *Handbook of Physics*; Springer Science & Business Media: Berlin/Heidelberg, Germany, 2006.
30. Jeffrey, A.; Dai, H.H. *Handbook of Mathematical Formulas and Integrals*; Elsevier: Amsterdam, The Netherlands, 2008.
31. Do Carmo, M. *Differential Geometry of Curves and Surfaces*; Prentice-Hall, Inc.: Englewood Cliffs, NJ, USA, 1976.
32. Kiselev, A.P. *Kiselev's Geometry: Stereometry*; Sumizdat: El Cerrito, CA, USA, 2008.
33. Petru, M.; Mlynek, J.; Martinec, T.; Broncek, J. Mathematical modelling of fibre winding process for composite frames. *Commun. Sci. Lett. Univ. Zilina* **2016**, *18*, 103–111.

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