

Article



Prediction of Deflection of Reinforced Concrete Beams Strengthened with Fiber Reinforced Polymer

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Abstract: The article analyses the calculation of the deflection of reinforced concrete beams strengthened with fiber reinforced polymer. This paper specifically focuses on estimating deflection when the yielding of reinforcement is reached. The article proposes a simple method for calculating deflection that was compared with the experimentally predicted deflection. The carried out comparison has showed that the proposed method is suitable not only for the strengthened beams but also for the reinforced concrete beams with a varying reinforcement ratio. The suggested calculation method is based on the effective moment of inertia, such as the one introduced in the ACI Committee 318 Building Code Requirement for Structural Concrete (ACI318). The development of deflection was divided into three stages, and equations for the effective moment of inertia were proposed considering separate stages. In addition, the put forward equations were modified attaching additional relative coefficients evaluating a change in the depth of the neutral axis.

Keywords: strengthening; FRP; deflection; yielding; effective moment of inertia

1. Introduction

One of the greatest advantages that can provide strengthening with carbon fiber reinforced polymer (CFRP) is an increase in the flexibility of the beam. Failure in the reinforced concrete beam is related to steel yielding, concrete crashing, or shear failure. Short-term and long-term experiments have showed that strengthening RC beams with CFRP can delay steel yielding [1–6]. Evenly, if steel yielding is reached or steel is rusted, the strengthened beams can serve until the rupture, delamination of the CFRP layer, steel fatigue fracture, or concrete crashing are achieved [7–11]. Due to high strength and high elasticity, the tensioned layer of CFRP can intercept tensile forces (stresses) when the yielding of reinforcement is reached. That is why the deflection of the beam can develop, thus reaching the yielding of reinforcement at a later stage. However, there is a danger for premature debonding of CFRP layer. In order to prevent this, proper additional anchoring can delay this phenomenon [12]. As well near surface mounted CFRP due to a larger perimeter-to-sectional-area-ratio can ensure better bond performance [13].

Various researches demonstrate that deflection development and reached yielding depend on the reinforcement (steel) ratio [14,15]. This may be related to the exploitation of the compressed concrete. If the reinforcement ratio is low, the exploitation of the compressed concrete is also greatly reduced until the yielding of reinforcement is reached. Therefore, the deflection (when the yielding of reinforcement is reached) of the strengthened beams with a low reinforcement ratio is the biggest. This is due to the unexploited deformability of the compressed concrete.

The existing methods for calculating deflection can perform estimation until the yielding of reinforcement is reached. The most common and simplest methods are based on design guidelines ACI318 [16] and the Eurocode 2 [17]. In addition, the multi-layer method can be used for calculating *Materials* **2019**, *12*, 1367; doi:10.3390/ma12091367 www.mdpi.com/journal/materials

the deflection of the strengthened beams; however, this method is not that convenient for engineers, and therefore will not be discussed in this article. The calculation method based on ACI318 [16] evaluates the effective moment of inertia, and the method based on Eurocode 2 [17], usually evaluates the average curvature of the bending element. Both methods evaluate the moment of the inertia of the full cross-section and the moment of the inertia of the cross-section where the crack is opened. However, these methods evaluate stress strain state in the cross-section before yield stresses in reinforcement are reached. There are several methods [18–21] that can evaluate stress-strain state in the cross-section after yield stresses are reached but these methods are difficult to be applied by the designer. Several contributions based on the moment-curvature modeling are available [22,23]. The accuracy of the proposed model [22,23] is impressive, however certain parameters like moment of inertia, depth of the neutral axis remains unknown.

The load carrying capacity of the strengthened beams can significantly increase such that the increased service load can locate in the range of the load-deflection curve where steel yielding is reached. The main objective of this article is to calculate the deflection of the strengthened beam when steel yielding is reached and when only the layer of CFRP intercepts tensile forces.

2. Analyzed Beams

RC strengthened beams with various reinforcement ratios were chosen to perform the calculation of deflection. The data about beams were collected from various research. The references and titles of the analyzed beams with a short description are presented in Table 1. The chosen beams are suitable for deflection analysis, because deflection develops when the yielding of reinforcement is reached. As mentioned above, a lower reinforcement ratio allows a higher increment in deflection when the yielding of reinforcement is reached.

The mechanical parameters of the material such as the modulus of elasticity and tensile strength are required in order to calculate the deflection of the beam. This and other mechanical parameters are presented in Table 2.

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Table 1.	Characteristics	of investigated	experimental beams.	

Author	Beam Name	l, m	Load Positions, m	b, m	h, m	As1	As2	d 1, m	d2, m	Af	
	V1	_			0.178	2Ø6	_			-	
	V1R1	_			0.17	2Ø6	_			1 × 1.45 × 9.59	
	V2	_			0.173	3Ø6	_	0.024		_	
Berries et al. 2005 [24]	V2R2	- 1 5		0.1	0.177	3Ø6	- 100	0.024	0.025	$2 \times 1.45 \times 9.59$	
barros et al., 2005 [24]	V3	1.5	0.5 + 0.5 + 0.5	0.1	0.175	2Ø6	- 208		0.025	_	
	V3R2				0.175	2Ø6 + Ø8				$2 \times 1.45 \times 9.59$	
	V4	_			0.175	3Ø8		0.025		_	
	V4R3				0.18	3Ø8		0.025		3 × 1.45 × 9.59	
	Ref_c_no_1		0.925 + 0.25 + 0.925	_						_	
	Ref_d_no_1		Distributed load	_						_	
	$\underline{\text{EBR}}_{c}1.4 \times 40_{1}$	_	$0.925 \pm 0.25 \pm 0.925$			5 2010				56 mm ²	
Bilotta et al., 2015 [25]	$EBR_c_{1.4} \times 40_2$		0.925 + 0.25 + 0.925	_				0.05		56 mm ²	
	$EBR_d_{1.4} \times 40_1$	- 21	Distributed load	0.12	0.16		2Ø10		0.025	56 mm ²	
	$EBR_d_{1.4} \times 40_2$	2.1		0.12	0.10	2010			0.000 -	56 mm ²	
	$NSM_c_2_{1.4} \times 10_1$		0.925 + 0.25 + 0.925	_						28 mm ²	
	NSM_d_2_1.4 × 10_1		Distributed load							28 mm ²	
	$NSM_c_3_{1.4} \times 10_1$		_	0.925 + 0.25 + 0.925	_						42 mm ²
	$NSM_d_3_{1.4} \times 10_1$		Distributed load							42 mm ²	
	P1	_								_	
David et al., 2003 [26]	P2	2.8	0.9 + 1.0 + 0.9	0.15	0.3	2Ø14	2Ø8	0.027	0.024	1.2 (cm ²)	
	P5									2.4 (cm ²)	
	REF	_								-	
	CN1	_								71.26 (mm ²)	
	CN2	_								2 × 71.26 (mm ²)	
	GN1	_								71.3 (mm ²)	
EL-Gamal et al., 2016 [27]	GN2	2.36	0.93 + 0.5 + 0.93	0.2	0.3	2Ø12	2Ø8	0.04	0.032	2 × 71.3 (mm ²)	
	CHVB									71.26 + 25.8	
		_								(mm ²)	
	CHVB									71.3 + 25.8	
	GIIID									(mm ²)	

	REF-II									_
	CN1-II	_				4Ø12				71.26 (mm ²)
	CN2-II	_								2 × 71.26 (mm ²)
E : (1 2002 [20]	A1	2.0		0.15	0.05	0014	200	0.005	0.005	_
Ferrier et al., 2003 [28]	A2	- 2.0	0.7 + 0.6 + 0.7	0.15	0.25	2014	208	0.025	0.025	120 (mm ²)
	CON1									_
	A0	_								0.22 × 75
	A10	_			0.2					0.22 × 75
Gao et al., 2004 [29]	A20	1.5	0.5	0.15		2Ø10	2Ø8	0.038	0.027	0.22 × 75
	B0	_								0.44×75
	B10	_								0.44×75
	B20	_								0.44×75
	2O									_
	2N6	_					2018	0.038		
	2T625-1	_				2010				$(\times 0.11 \times 150$
	2T650-1	1 5	0.5	0.15	0.2				0.027	6 × 0.11 × 150
Gao et al., 2006 [30]	2T675-1	- -		0.15	0.2	2010	208		0.027	
	2N4									
	2T450-1									$4\times0.11\times150$
	2T4100-1	_								
Hofferman 1007 [21]	Conventional	_ 10	16 + 16 + 16	0.2	0 5720	2Ø25 +	20210	0.074	0.067	-
nelleman 1997 [31]	CFRP strengthened	- 4.0	1.0 + 1.0 + 1.0	0.5	0.3739	Ø20	2010	0.074	0.067	65.5 (mm ²)
Heffernan and Erki 2004	Conventional	- 285	$11 \pm 0.65 \pm 1.1$	0.15	03	2Ø20 +	20210	0.041	0.027	_
[32]	CFRP strengthened	2.85	1.1 + 0.05 + 1.1	0.15	0.5	Ø10	2010	0.041	0.037	89.4 (mm ²)
	SREF	_								_
Hossoini et al. 2014 [22]	S2L-0	- 24	00+06+00	0.6	0.12	1018	306	0.024	0.023	
110sseini et al., 2014 [55]	S2L-20	2.4	$0.9 \pm 0.0 \pm 0.9$	0.0	0.12	400	500	0.024	0.023	$2 \times 1.4 \times 20$
	S2L-40									
	B-C	_								_
Khalifa et al 2016 [24]	B-S-2	- 22	$0.95 \pm 0.3 \pm 0.95$	0.15	0.26	0.26 2Ø12	2Ø12	0.041	0.021	60 (mm ²)
Khalifa et al., 2016 [34] -	B-S-4		$0.90 \pm 0.0 \pm 0.90$	0.15	0.20			0.041	0.031	120 (mm²)
	B-N-1-2									60 (mm ²)

	B-N-2-2									60 (mm ²)
	B-N-2-4								_	120 (mm²)
Kataria at al. 2008 [25]	B-08S	4.0	1 4 . 1 4 . 1 4	0.15	0.2	2012	2010	0.02 *	0.02 **	60 (mm²)
Kotynia et al., 2008 [35]	B-083m	- 4.2	1.4 + 1.4 + 1.4	0.15	0.3	3012	2010	0.03 *	0.03	58.5 (mm²)
	G1									_
Kalania at al. 2011 [26]	G2		1.2 + 1.2 + 1.2 + 1.2 +	1.0	0.22	7010		0.03143	0.024	120 (mm²)
Kotynia et al., 2011 [36]	G3	6.0	1.2		0.22	7012	708	*	**	120 (mm²)
	G4								-	120 (mm ²)
	B12-a		10.10.10.10.							1.2×100
Kotynia et al., 2014 [37]	B12-asp	6.0	1.2 + 1.2 + 1.2 + 1.2 +	0.5	0.22	4Ø12	4Ø8	0.031	0.029	1.2×100
-	B16-asp		1.2						-	1.2×100
	BO									_
	B1-NP						2Ø10	0.057	-	
Omran et al., 2012 [38]	B1-P1	5.0	2 + 1 + 2	0.2	0.4	3Ø15			0.036	0
-	B1-P2									2 × 2 × 16
	B1-P3									
	Control									_
	Non prestressed		0.9 + 0.4 + 0.9					0.035	-	
Rezazadeh et al., 2014 [39]	20% prestressed	2.2		0.15	0.3	2Ø10	ð10 2Ø10		0.025	1.4 00
	30% prestressed									1.4×20
	40% prestressed									
	СВ									_
	LB1C1								-	1Ø8
	LB1G1								-	1Ø8
	LB2C1			0.16	0.00	0010	200	0.00(0.024	2Ø8
Sharaky et al., 2014 [40]	LB2G1	- 2.4	0.8 + 0.8 + 0.8	0.16	0.28	2012	2608	0.036	0.034 -	2Ø8
	LA2C1								-	2Ø8
	LA2G1								-	2Ø8
-	LB1G2								-	1Ø12
	C-0									_
Soudki et al., 2007 [41]	T-0	2.25	0.75	0.15	0.25	2Ø10	2Ø6	0.025	0.023	4×0.11
	S-0								-	50 × 1.2

	B0									_
-	B500								-	
Teng et al., 2006 [42]	B1200	3.0	1.2 + 0.6 + 1.2	0.15	0.3	2Ø12	2Ø8	0.036	0.034	2 ~ 1(
_	B1800									2 × 16
	B2900									
	B6.1C									$0.1(7.(cm^2))$
_	B6.2C					2Ø6			_	0.167 (CIII ²)
	B6.5				200		$- 2\alpha$		_	_
_	B8.1C			100	200		200			0.167 (cm ²)
Valivonis et al., 2010 [14] — — —	B8.2C	— 1 2	$0.4 \pm 0.4 \pm 0.4$	100		2Ø8		0.025	0.025 -	0.107 (CIII-)
	B8.3	1.2	$0.4 \pm 0.4 \pm 0.4$					0.023	0.025	_
	B12.1C			-	203				_	$0.167 (cm^2)$
	B12.2C				200	2012	2018			0.107 (CIII-)
_	B12.5			104	198	2012	200			
	B12.6			105	201					_
_	Control								_	_
_	B11								_	Ø7.9
_	B21									2017 9
W_{11} of al. 2014 [43] -	B22	<u> </u>	$0.6 \pm 0.6 \pm 0.6$	0.15	03	30/11	2016	0.037	0.033 -	201.)
	BP11	1.0	$0.0 \pm 0.0 \pm 0.0$	0.15	0.5	5014	200	0.057	0.035	
_	BP12									Ø7 9
_	BP13									Q7.9
	BP14									
Xiong et al., 2007 [44]	Ра					2×10		0.03	_	_
	2C	2.1	0.7	0.125	0.2	0.2	2×8	0.05	0.024	0.22×100
	Pb					2 × 12		0.031		_

* $a_5 = h-A_{s1}/\dot{n}_{s1}$ ·b; ** evaluated individually; 1-span length; b-total width of the beam; h-height of the beam; A_{s1} -cross-section of the tensioned steel bars; A_{s2} -cross-section of the compressed steel bars; d_1 -position of the tensioned steel bars; d_2 -position of the compressed steel bars; A_f -cross-section of the tensioned fibers or FRP; $\dot{n}s1$ -reinforcement ratio by A_{s1} .

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Table 2. Mechanical characteristics of investigated experimental beams materials.

Author	Beam Name	fc, MPa	fct, MPa	Ec, GPa	f _{y1} , MPa	f _{y2} , MPa	Es1, GPa	Es2, GPa	ff,fe, MPa	Ef,fe, GPa
	V1	46.1	2.27	22.25	720				_	_
	V1R1	40.1	5.57	33.33	730	_			2740	158.8
	V2	46.1	2 50	26 F	720	-			_	_
Berries et al. 2005 [24]	V2R2	46.1	3.38	36.5	730	EE4 22	200	200	2740	158.8
Barros et al., 2005 [24]	V3	46.1	2 21	24.90	730	554.52	200	200	_	_
	V3R2	40.1	3.21	34.89	730; 554.32	-			2740	158.8
	V4	16 1	2 42	25.96	EE4 22					-
	V4R3	40.1	5.45	35.66	554.52				2740	158.8
	Ref_c_no_1								-	-
	Ref_d_no_1								_	-
	$EBR_c_{1.4} \times 40_1$							200		
	$EBR_c_{1.4} \times 40_2$						200			
Bilotto et al. 2015 [25]	$EBR_d_{1.4} \times 40_1$	174	1 34	25.08	540	540				
Dilotta et al., 2015 [25]	$EBR_d_{1.4} \times 40_2$	17.4	1.34	25.98	540	340	200	200	2052	171
	$NSM_c_2_{1.4} \times 10_1$								2052	1/1
	$NSM_d_2_{1.4} \times 10_1$									
	NSM_c_3_1.4 × 10_1									
	NSM_d_3_1.4 × 10_1									
	P1	38.7	2.94 ¹	33.02 ²					_	-
David et al., 2003 [26]	P2	39.2	2.97 ¹	33.14 ²	500	500	205 ³	205 ³	2400	150
	P5	40.1	3.03 1	33.37 ²					2400	130
	REF								_	_
	CN1								1588	110 /
	CN2								1300	119.4
	GN1								1105	ED 24
EL-Gamal et al., 2016 [27]	GN2	49.62	2.99	35.57 ²	480	455	205 ³	205 ³	1165	32.34
	CHYB								2096 *	147.47 *
	GHYB								1800 *	98.22 *
_	REF-II								_	_
	CN1-II								1588	119.4

	CN2-II									
	A1								_	_
Ferrier et al., 2003 [28]	A2	39	2.96 ¹	31	550	550 ³	210	210 ³	650	80
	CON1								_	_
	A0									
	A10									
Gao et al., 2004 [29]	A20	35.7	2.75^{1}	25	531	400	200	200	1000	
	B0								4200	235
	B10									
	B20									
	2O								_	_
	2N6									
	2T625-1									
Gao et al., 2006 [30]	2T650-1	62.1	4 20 1	07.1	460	160	200	205		
	2T675-1		4.29	37.1	460	460	200	205	4200	235
	2N4									
	2T450-1	•								
	2T4100-1									
Lighterman 1007 [21]	Conventional	22.0	2.56 ¹	01 4E 2			200	200	-	_
Helleman 1997 [31]	CFRP strengthened	32.9		31.45 -2	-	-	200	200		325
Hoffermen and Erki 2004 [22]	Conventional	27	2 02 1	22 E7 2	E11 0- 111	111	210	210	-	_
Hellerhan and Erki 2004 [52]	CFRP strengthened	37	2.03 1	32.37 -	511 & 411	411	210	210		233
	SREF								_	_
Hossoini et al. 2014 [22]	S2L-0	167	2 / 2 1	20.7	186	161	200	200		
110ssenii et al., 2014 [33]	S2L-20	40.7	5.45	29.1	400	404	200	200	2483.9	153.2
	S2L-40									
	B-C									
	B-S-2									
——————————————————————————————————————	B-S-4	35	271	28	400	400	400 200	200	2800	165
	B-N-1-2	55	2.1	20	400	400 400		200	2000	
	B-N-2-2									
	B-N-2-4									

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	P 096	22.2	0 E 0 1	21 27 2	400	E24	105	200	201E	170
Kotynia et al., 2008 [35]	D-085	32.3	2.52 1	31.27 2	490	524	195	209	2915	1/2
	B-083m	34.4	2.66 1	31.87 2	436	524	220	209	3500	230
	Gl	45	3.33 1	34.55 2						
Kotynia et al., 2011 [36]	G2	46.2	3.4 1	34.82 2	554	561	200	200	2800	165
	G3	45.9	3.39 1	34.75 ⁻²						
	G4	45.6	3.37 1	34.68 ²					2235	149
	В12-а	45.3	3.35	24.3	539.6	416.2	191.3	186.1	-	
Kotynia et al., 2014 [37]	B12-asp	32.2	2.51	23.7	511.4	583.2	191.4	200.7	2800	173.3
	B16-asp	49	3.57	25.4	595	555.8	198	196.4		
	BO	_							_	-
	B1-NP	_								
Omran et al., 2012 [38]	B1-P1	40	3.02^{1}	27.84	478	500	200	200	2 610	120 E
	B1-P2	-							2010	150.5
	B1-P3	-								
	Control								_	_
	Non prestressed	-								
Rezazadeh et al., 2014 [39]	20% prestressed	32.2	2.51 ¹	27.4	585	585	208	208	1000	174
	30% prestressed								1922	104
	40% prestressed	-								
	CB								_	-
	LB1C1	-							2350	170
	LB1G1	-							1350	64
	LB2C1	-	2.0	01 5		- 4-	005	0.05	2350	170
Sharaky et al., 2014 [40]	LB2G1	32.4	2.8	31.7	545	545	205	205	1350	64
	LA2C1	-							2350	170
	LA2G1	-							1350	64
	LB1G2	-							1350	64
	C-0								_	_
Soudki et al., 2007 [41]	T-0	35	2.7	32.04	460	460	205	205	3480	230
	S-0	-							2800	165
Teng et al., 2006	B0		0.07.	04.01.3	_	_	010	210	_	_
[42]	B500	- 44	3.27 1	34.31 2			210	210	2068	131

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	D10 00									
-	B1200	_								
	B1800	_								
	B2900									
	B6.1C								4900	001
	B6.2C	34.4	2.93	32.45	358	358	205	205	4000	231
	B6.5								_	-
	B8.1C								4800	001
	B8.2C	29.7	2.63	30.91	557	358	195	205	4000	231
	B8.3								_	_
	B12.1C	- 20.4	2.67	21.14					4800	001
	B12.2C	- 30.4	2.67	51.14	210	420	204.0	204.1	4000	231
	B12.5	- 207	2 56	20 55	318	420	204.9	204.1	_	_
_	B12.6	- 28.7	2.36	30.55					_	_
	Control								_	_
	B11					• 10	••••			
	B21									
Wu et al., 2014	B22	- 04.4	0 ((1	01 05 1	240			200		
[43]	BP11	- 34.4	2.66 1	31.87 -2	340	240	200	200	2629	170
-	BP12	_								
-	BP13	_								
-	BP14	_								
	Pa				411		200		_	_
	2C	30.71	2.41 ¹	30.8 2	411	233	200	210	3652	252
	Pb	_	2.11		606	_	210	-	_	_

¹ $f_{ctm} = 0.3(f_{cm}-8)^{2/3}$ equation from Eurocode 2 [17]; ² $E_{cm} = 22(f_{cm}/10)^{0}$. ³ equation from Eurocode 2 [17]; ³ evaluated individually; f_c —concrete compressive strength; f_{ct} —concrete tensile strength; E_c —modulus of elasticity of the concrete material; f_{y1} —yielding strength of the tensioned steel bars; f_{y2} —yielding strength of the compressed steel bars; E_{s1} —modulus of elasticity of the tensioned steel bars; E_{s2} —modulus of elasticity of the tensioned fibers or FRP; $E_{f,fe}$ —modulus of elasticity tensioned fibers or FRP; *—calculated by the law of the mixture.

3. Calculation of Deflection

The development of the deflection of the strengthened and unstrengthened beams is divided into stages. At the first stage, deflection develops until vertical cracks open in the tensioned part of the cross-section. At the second stage, deflection develops when the vertical crack is opened until the yielding strength of the tensioned reinforcement is reached. At the third stage, deflection develops when the yielding strength of reinforcement is reached and only a layer of CFRP intercepts tensile force. Therefore, two deflection development stages exist for the unstrengthened beams and three stages for the strengthened ones (Figure 1). Bending moments M_I and M_{LS} are shown in (Figure 1), which is the cracking moment of the unstrengthened and strengthened beam, respectively. Due to the CFRP layer, the contribution cracking moment of the strengthened beam is slightly bigger than that of the unstrengthened beam (M_{LS} > MI). Bending moments (M_{LS} and M_I) correspond to the end of the first stage. The maximal carrying bending moment of the strengthened beam (M_{LS} = M_I) is smaller than that of the bending moment of the strengthened beam (M_{LS}) when the yielding of reinforcement is reached. These bending moments correspond to the end of the second stage. The maximum carrying bending moment of the strengthened beam is designated as M_{R-S} = M_{III} and corresponds to the end of the third stage.



Figure 1. The development of the deflection of the strengthened and unstrengthened beam.

The deflection of the beams at a certain stage is influenced by different flexural stiffness. Generally, bending stiffness $E \cdot I$ (the product of the modulus of elasticity and the moment of inertia) is influenced by the moment of inertia. The current methods for calculating deflection usually evaluate the modulus of elasticity like for an elastic material. Then, the development of deflection undergoes all stages, cracks in the tensioned part of the cross-section develop, therefore, the moment of inertia is not constant. Thus, at a certain stage, the depth of the neutral axis and the moment of inertia are different. A change in the depth of the neutral axis of the strengthened and unstrengthened beams is presented in Figures 2 and 3. Thus, there are parts of the cross-section containing and having no cracks. Therefore, the effective moment of inertia should be evaluated. The prediction of the depth of the neutral axis at each stage confirms that the distribution of strains is linear. Stresses in the compressed part of the section are in the elastic range. In addition, a hypothesis about the plane section is valid. The strain of internal and external reinforcement is equal to the surrounded concrete strain (bond slip is not evaluated).



Figure 2. A change in the depth of the neutral axis of the RC strengthened beam: (**a**) Cross-section of the strengthened beam; (**b**) depth of the neutral axis before vertical cracks will open; (**c**) depth of the neutral axis when vertical cracks are opened; (**d**) depth of the neutral axis when steel yielding is reached.



Figure 3. A change in the depth of the neutral axis of the RC beam: (**a**) Cross-section of the beam (**b**) depth of the neutral axis before vertical cracks will open; (**c**) depth of the neutral axis when vertical cracks are opened.

The deflection of the strengthened beam at stage 1 up to the cracking of the tensioned part of the cross-section can be predicted by the equation:

$$\omega_{I,S}(M_I) = \frac{3 \cdot l^2 - 4 \cdot a^2}{24} \cdot \frac{M_I}{E_{cm} \cdot I_{I,red}}.$$
(1)

where l—the span length of the beam, a–distance from the support to the external load position, M_l —acting moment, E_{cm} —the modulus of elasticity of concrete, $I_{l.red}$ —the reduced moment of the inertia of the total cross-section according to the neutral axis of the cross-section.

At stage 1, the evaluated acting moment is $0 < M_l \le M_{l,s}$, and the ultimate bending moment of stage 1 is the cracking moment:

$$M_{crc} = M_{I.S} = f_{ct} \cdot \frac{I_{I.red}}{y_{c.I}}.$$
 (2)

where f_{ct} —the tensile strength of concrete, $y_{c.t}$ —the centre of the gravity of the cross-section at stage 1. The center of gravity can be predicted by the following equations:

$$A_{red} = b \cdot h + \alpha_f \cdot A_f + (\alpha_{s1} - 1) \cdot A_{s1} + (\alpha_{s2} - 1) \cdot A_{s2},$$
(3)

$$S_{red} = b \cdot h \cdot \left(\frac{h}{2} + t_f\right) + \alpha_f \cdot A_f \cdot \frac{t_f}{2} + (\alpha_{s1} - 1) \cdot A_{s1} \cdot (d_1 + t_f) + (\alpha_{s2} - 1) \cdot A_{s2} \cdot (h + t_f - d_2),$$
(4)

$$\alpha_f = \frac{E_f}{E_c},\tag{5}$$

$$\alpha_{s1} = \frac{E_{s1}}{E_c},\tag{6}$$

$$\alpha_{s2} = \frac{E_{s2}}{E_c},\tag{7}$$

$$y_{c.I} = \frac{S_{red}}{A_{red}}.$$
(8)

where A_{red} —the reduced cross-section of the strengthened beam, A_f —the cross section of carbon fibers, A_{s1} , A_{s2} —the cross-section of steel bars, S_{red} —the static moment of the reduced cross-section of the strengthened beam, α_{f} , α_{s1} , α_{s2} —coefficients of reduction, E_f —the modulus of elasticity of fibers, E_{s1} , E_{s2} —the modulus of elasticity of the steel bars.

The reduced moment of the inertia of the cross-section can be predicted by the following equation:

$$I_{I.red} = \frac{b \cdot h^3}{12} + b \cdot h \cdot \left(\frac{h}{2} + t_f - y_{c.I}\right)^2 + \alpha_f \cdot A_f \cdot \left(y_{c.I} - \frac{t_f}{2}\right)^2 + (\alpha_{s1} - 1) \cdot A_{s1} \cdot (y_{c.I} - t_f - d_1)^2 + (\alpha_{s2} - 1) \cdot A_{s2} \cdot (h + t_f - y_{c.I} - d_2)^2.$$
(9)

The deflection of the strengthened beam at stage 2, when the tensioned part of the cross-section is cracked and the yielding of the tensioned reinforcement is not reached, can be predicted by the equation:

$$\omega_{II}(M_{II}) = \frac{3 \cdot l^2 - 4 \cdot a^2}{24} \cdot \frac{M_{II}}{E_c \cdot I_{II}(M_{II})}.$$
(10)

The acting bending moment at stage 2 is M_{II} and the moment $M_{LS} < M_{II} \le M_{II.S}$. The moment when the yielding of reinforcement is reached is $M_{II.S}$. The effective moment of inertia is evaluated using the Branson [45] equation for parameter I_{II} :

$$I_{II}(M_{II}) = I_{I.red} \cdot \left(\frac{M_{I.u}}{M_{II}}\right)^3 + I_{II.red} - I_{II.red} \cdot \left(\frac{M_{I.u}}{M_{II}}\right)^3.$$
(11)

If change of the neutral axis is evaluated, then Equation (11) is modified like:

$$I_{II}(M_{II}) = I_{I.red} \cdot \left(\frac{M_{I.u}}{M_{II}}\right)^3 + I_{II.red} \cdot \gamma_{1.c} \cdot \gamma_{1.t} - I_{II.red} \cdot \left(\frac{M_{I.u}}{M_{II}}\right)^3 \cdot \gamma_{1.c} \cdot \gamma_{1.t}.$$
 (12)

Where *I*_{*U,red}—the reduced moment of the inertia of the cross section where the vertical crack is opened. This moment of inertia can be predicted by the equation:</sub>*

$$I_{II.red} = \frac{b \cdot x_{II}^3}{12} + b \cdot x_{II} \cdot \left(\frac{x_{II}}{2}\right)^2 + \alpha_f \cdot A_f \cdot \left(h + t_f - x_{II} - \frac{t_f}{2}\right)^2 + \alpha_{s1} \cdot A_{s1} \cdot (h - x_{II} - d_1)^2 + (\alpha_{s2} - 1) \cdot A_{s2} \cdot (x_{II} - d_2)^2.$$
(13)

Coefficients $\gamma_{1,t}$ and $\gamma_{1,t}$ evaluate a change in the neutral axis and can be predicted by equations:

$$\gamma_{1.c} = \frac{x_{II}}{x_I},\tag{14}$$

$$\gamma_{1,t} = \frac{h + t_f - x_{II}}{h + t_f - x_I}.$$
(15)

The depth of the neutral axis at stage 1 is predicted by the equation:

$$x_{I} = h + t_{f} - y_{c.I}.$$
 (16)

The prediction of the depth of the neutral axis in the section having an opened crack is based on the previously mentioned assumptions. The hypothesis of plain sections is valid. The distribution of strains through the height of the section is linear (Figure 4b). Then, by the similarity of triangles, strains at each layer, in proportion with the strain of the compressed concrete layer, can be expressed, and the depth of the neutral axis should be expressed from the square equation. The depth of the neutral axis at stage 2 can be predicted by the equation:

$$x_{II} = \frac{-B + \sqrt{B^2 + 4 \cdot A \cdot C}}{2 \cdot A}.$$
(17)

where coefficients *A*, *B*, and *C*:

$$A = b \cdot 0.5,\tag{18}$$

$$B = \alpha_f \cdot A_f + \alpha_{s1} \cdot A_{s1} + (\alpha_{s2} - 1) \cdot A_{s2}, \qquad (19)$$

$$C = \alpha_f \cdot A_f \cdot \left(h + \frac{t_f}{2}\right) + \alpha_{s1} \cdot A_{s1} \cdot d + (\alpha_{s2} - 1) \cdot A_{s2} \cdot d_2.$$
(20)



Figure 4. Stress-strain state in the strengthened RC beam until the yielding of reinforcement is reached: (a) Depth of the neutral axis; (b) distribution of strains; (c) distribution of stresses; (d) internal forces.

The deflection of the strengthened beam at stage 3, when the yielding strength of tensioned reinforcement is reached, can be predicted by the equation:

$$\omega_{III}(M_{III}) = \frac{3 \cdot l^2 - 4 \cdot a^2}{24} \cdot \frac{M_{III}}{E_c \cdot I_{III}(M_{III})}.$$
(21)

The acting bending moment at stage 3 is M_{III} and the moment $M_{II.u} < M_{III} \le M_{III.u}$. The ultimate bending moment at stage 3 is $M_{III.u}$. The new effective moment of inertia is evaluated in the equation for parameter I_{III} :

$$I_{III}(M_{III}) = I_{I.red} \cdot \left(\frac{M_{I.u}}{M_{III}}\right)^3 + I_{II.red} \cdot \left(\frac{M_{I.u}}{M_{III}}\right)^3 - I_{II.red} \cdot \left(\frac{M_{I.u}}{M_{III}}\right)^3 + I_{III.red} \cdot \left(\frac{M_{III}}{M_{III}}\right)^3 - I_{III.red} \cdot \left(\frac{M_{III}}{M_{III}}\right)^3.$$
(22)

If change of the neutral axis is evaluated, then Equation (22) is modified like:

$$I_{III}(M_{III}) = I_{I.red} \cdot \left(\frac{M_{I.u}}{M_{III}}\right)^3 + I_{II.red} \cdot \left(\frac{M_{II.u}}{M_{III}}\right)^3 \cdot \gamma_{1.c} \cdot \gamma_{1.t} - I_{II.red} \cdot \left(\frac{M_{I.u}}{M_{III}}\right)^3 \cdot \gamma_{1.c} \cdot \gamma_{1.t} + I_{III.red} \cdot \left(\frac{M_{III}}{M_{III}}\right)^3 \cdot \gamma_{2.c} \cdot \gamma_{2.t} - I_{III.red} \cdot \left(\frac{M_{I.u}}{M_{III}}\right)^3 \cdot \gamma_{2.c} \cdot \gamma_{2.t}.$$

$$(23)$$

where *I*_{III.red}—the reduced moment of the inertia of the cross section where the vertical crack is opened. This moment of inertia can be predicted by the equation:

$$I_{III.red} = \frac{b \cdot x_{III}^3}{12} + b \cdot x_{III} \cdot \left(\frac{x_{III}}{2}\right)^2 + \alpha_f \cdot A_f \cdot \left(h + t_f - x_{III} - \frac{t_f}{2}\right)^2 + (\alpha_{s2} - 1) \cdot A_{s2} \cdot (x_{III} - d_2)^2.$$
(24)

Coefficients $\gamma_{2.c}$ and $\gamma_{2.t}$:

$$\gamma_{2.c} = \frac{x_{III}}{x_{II}};$$
(25)

$$\gamma_{2.t} = \frac{h + t_f - x_{III}}{h + t_f - x_{II}}$$
(26)

The depth of the neutral axis at stage 3 is also predicted from the similarity of triangles (Figure 5b).



Figure 5. Stress-strain state in the strengthened RC beam when the yielding of reinforcement is reached: (a) Depth of the neutral axis; (b) distribution of strains; (c) distribution of stresses; (d) internal forces.

The depth of the neutral axis at stage 3 is predicted by the equation:

$$x_{III} = \frac{-B + \sqrt{B^2 + 4 \cdot A \cdot C}}{2 \cdot A}.$$
(27)

Were coefficients *A*, *B*, and *C*:

$$A = b \cdot 0.5; \tag{28}$$

$$B = \alpha_f \cdot A_f + (\alpha_{s2} - 1) \cdot A_{s2}; \tag{29}$$

$$C = \alpha_f \cdot A_f \cdot \left(h + \frac{t_f}{2}\right) + \left(\alpha_{s2} - 1\right) \cdot A_{s2} \cdot d_2.$$
(30)

The deflection of the unstrengthened beams can be predicted by the same Equations (1) and (10). However, the parameters of the FRP layer in other equations should be ignored. If the beams are strengthened with the prestressed FRP, in this case it is necessary to calculate the additional curvature and the deflection from prestress force. The total deflection is obtained by summing up all the deflections.

4. Results

A comparison of deflections (Figures 6–9) shows that the equation method is suitable for RC beams with various reinforcement ratios. Calculated deflections of all mentioned beams are presented in the Appendix A. In these figures, designation "Calc. I" is related to Equations (11) and (22). Designation "Calc. II" related with Equations (12) and (23). It is clear that the theoretical equation method gives brake points such as the cracking moment and steel yielding moment on the load deflection curve. The difference between the calculated and experimental deflection increases when the load level increases. This may happen because the theoretical method evaluates the elastic work of concrete and the constant depth of the neutral axis. Thus, the deflection curve curvature

depends just from ratio of the bending moments. In order to increase the accuracy of the theoretical method, nonlinear stress-strain distribution across the height of the cross-section should be evaluated. The proposed method evaluates linear stress-strain distribution. The evaluation of nonlinear stress-strain distribution can be complex for designers, and thus triangular distribution is easier to assess. Furthermore, a comparison of the position of the center of the parabolic and triangular form gives little difference. The difference in results is also influenced by the accuracy of the experiment. In certain experiments, deflection at the cracking moment to big. The main drawback of the suggested method is the prediction of the bending moment when steel yielding is reached. It is difficult to predict the moment when the FRP layer is incorporated, because strains are not known in the compressed concrete and tensioned CFRP layer. In such a case, the problem must be solved by the iteration approach until the balance of internal forces is reached. This is also a complex task for designers. For this research values of cracking, yielding and ultimate moment were predicted from the deflection evolution plots.

Experiments in which the deflection was measured from the frame mounted on a beam gives a more precise result. Calculated deflection (Calc. I) using the effective moment of inertia equation without any coefficients is suitable for this measurement system. Equation of the effective moment of inertia must be without coefficients—it is related with the neutral axis. Please note that the second stage does not have a horizontal straight line. The other experimental "deflection" results, which are more close to the "Calc. II" can be associated with the measured displacement.



Figure 6. Bending moment–deflection curves, (**a**) beam V1R1; (**b**) beam NSM_c_ $2 \times 1.4 \times 10_1$; (**c**) beam P1; (**d**) beam CN1.



Figure 7. Bending moment–deflection curves, (**a**) beam A2; (**b**) beam A0; (**c**) beam 2N4; (**d**) beam CFRP Strengthened; (**e**) beam CFRP strengthened; (**f**) beam S2L-0; (**g**) beam B-N-1-2; (**h**) beam B-08S.



Figure 8. Bending moment–deflection curves, (**a**) beam G2; (**b**) beam B12-a; (**c**) beam B1-NP; (**d**) beam Non prestressed, (**e**) beam LB1C1; (**f**) beam T-0; (**g**) B500; (**h**) B6.1C.



Figure 9. Bending moment–deflection curves, (a) beam B11; (b) beam 2C.

5. Conclusions

According to the proposed method for calculating the deflection of the strengthened RC beam, it is possible to predict deflection when steel yielding is reached. When the deflection is calculated using the usual expression of an effective moment of inertia (Equations (11) and (22)), in some cases smaller deflections are obtained. This discrepancy may be due to an incorrectly determined experimental deflection, since in some experiments it is not clear whether the deflection is determined by compensating the lift of the neutral axis at the supports. In most cases, the most accurate calculation using the normal expression of an effective inertia moment (Equations (11) and (22)). Estimating the change in the neutral axis (Equations (12) and (23)) results in bigger deflections but are more precise when the deflections are lower with normal expression (Equations (11) and (22)). Another important criterion related to the accuracy of deflections is the coefficient of estimating the nature of the external load, since after the strengthening the evolution of cracks changes, the curvature development change too. In order to verify the accuracy of the experimental and computational results, further finite element analysis is required.

Notation

А, В, С	the designation of the equation for the depth of the neutral axis;
A_c	the cross section of the compressed concrete layer;
A_f	the cross-section area of carbon fiber;
Ared	the transformed cross section of the beam;
A_{s1} and A_{s2}	the cross-section area of the tensioned and compressed reinforcement;
Ecm	the modulus of elasticity of concrete;
E_f	the modulus of elasticity of carbon fiber;
Es1, Es2	the modulus of elasticity of steel bars;
In, Im	the effective moment of inertia at stages 2 and 3;
II.red, III.red, IIII.red	the moment of the inertia of the transformed cross-section at stages 1, 2, and 3;
MI and MI.S	the cracking moment of the unstrengthened beam and strengthened beam respectively;
MII.s	the bending moment of the strengthened beam when the yielding of reinforcement is reached;
Mr, MII	the maximum carrying bending moment of the unstrengthened beam;
Mr.s, MIII	the maximum carrying bending moment of the strengthened beam;
Sred	the static moment of the transformed cross-section;
а	distance from the support to loading;
b	the width of the beam;
d_1 and d_2	distance from the beam edge to the center of the tensioned and compressed reinforcement;

f_c	the compressive strength of concrete cylinders;
f_{ct}	the tensile strength of concrete;
h	the height of the beam;
k_1	the coefficient evaluating the shape of stress distribution;
l	the span length of the beam;
t_f	the thickness of the carbon fiber layer;
x_1 , x_{11} and x_{111}	the depth of the neutral axis at stages 1, 2 and 3;
Ус.I, Ус.II, Ус.III,	the centre of the gravity of the beam cross-section at stages 1, 2 and 3;
α_{f} , α_{s1} , α_{s2}	relative coefficients;
γ1.c, γ1.t	relative coefficients evaluating a change in the depth of the neutral axis at stage 2;
Y2.c, Y2.t	relative coefficients evaluating a change in the depth of the neutral axis at stage 3;
Ec	the strain of the compressed concrete;
E c1	strain when the maximum strength of concrete material is reached;
Ef	the strain of the carbon fiber layer;
$\mathcal{E}s$	the strain of the tensioned reinforcement;
Es2	the strain of the compressed reinforcement;
σ_c	stresses in the layer of the compressed concrete;
σ_{f}	stresses in the layer of carbon fiber;
σ_s	stresses in the tensioned reinforcement;
σ_{s2}	stresses in the compressed reinforcement;
ωι, ωπ	the deflection of the control beam up to the end of stages I and II.
$\omega_{I.S}, \omega_{II.S},$	the deflection of the strengthened beam up to the end of stages L.U. and UL
ω III.s	the denection of the strengthened beam up to the end of stages 1, 11, and 111.

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Figure A1. Barros et al., 2005 [24] research beams displacement compared with calculated deflection, (a) beam V1; (b) beam V1R1; (c) beam V2; (d) beam V2R2; (e) beam V3; (f) beam V3R2; (g) beam V4; (h) beam V4R3; (i) scatter of the results at 60% and 80% of the ultimate load.



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Figure A2. Bilotta et al., 2015 [25] research beams deflection compared with calculated, (a) beam Ref_c_no_1; (b) beam Ref_d_no_1; (c) beam EBR_c_1.4 × 40_1; (d) beam EBR_c_1.4 × 40_2; (e) beam

 $EBR_d_{1.4} \times 40_1$; (f) beam $EBR_d_{1.4} \times 40_2$; (g) beam $NSM_c_2 \times 1.4 \times 10_1$; (h) beam $NSM_d_2 \times 1.4 \times 10_1$; (i) beam $NSM_c_3 \times 1.4 \times 10_1$; (j) beam $NSM_d_3 \times 1.4 \times 10_1$; (k) scatter of the results at 60% and 80% of the ultimate load.



Figure A3. David et al., 2003 [26] research beams deflection compared with calculated, (**a**) beam P1; (**b**) beam P2; (**c**) beam P5; (**d**) scatter of the results at 60% and 80% of the ultimate load.





Figure A4. El-Gamal et al., 2016 [27] research beams deflection compared with calculated, (**a**) beam REF; (**b**) beam CN1; (**c**) beam CN2; (**d**) beam GN1; (**e**) beam GN2; (**f**) beam CHYB; (**g**) beam GHYB; (**h**) beam REF-II; (**i**) beam CN1-II; (**j**) beam CN2-II; (**k**) scatter of the results at 60% and 80% of the ultimate load.



Figure A5. Ferrier et al., 2003 [28] research beams deflection compared with calculated, (**a**) beam A2; (**b**) beam A0; (**c**) scatter of the results at 60% and 80% of the ultimate load.





Figure A6. Gao et al., 2004 [29] research beams deflection compared with calculated, (**a**) beam CON1; (**b**) beam A0; (**c**) beam A10; (**d**) beam A20; (**e**) beam B0; (**f**) beam B10; (**g**) beam B20; (**h**) scatter of the results at 60% and 80% of the ultimate load.



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Figure A7. Gao et al., 2006 [30] research beams displacement compared with calculated deflection, (a) beam 2O; (b) beam 2N4; (c) beam 2N6; (d) beam 2T450-1; (e) beam 2T625-1; (f) beam 2T650-1; (g) beam 2T675-1; (h) beam 2T4100-1; (i) scatter of the results at 60 % and 80% of the ultimate load.





Figure A8. Heffernan 1997 [31] research beams displacement compared with calculated deflection, (a) beam Conventional; (b) beam CFRP Strengthened; (c) scatter of the results at 60% and 80% of the ultimate load.



Figure A9. Heffernan and Erki 2004 [32] research beams displacement compared with calculated deflection, (**a**) beam CFRP Strengthened; (**b**) beam Conventional; (**c**) scatter of the results at 60% and 80% of the ultimate load.





Figure A10. Hosseini et al., 2014 [33] research beams deflection compared with calculated, (**a**) beam SREF; (**b**) beam S2L-0; (**c**) beam S2L-20; (**d**) beam S2L-40; (**e**) scatter of the results at 60% and 80% of the ultimate load.





Figure A11. Khalifa et al., 2016 [34] research beams deflection compared with calculated, (**a**) beam B-C; (**b**) beam B-S-2; (**c**) beam B-S-4; (**d**) beam B-N-1-2; (**e**) beam B-N-2-2; (**f**) beam B-N-2-4; (**g**) scatter of the results at 60% and 80% of the ultimate load.



Figure A12. Kotynia et al., 2008 [35] research beams deflection compared with calculated, (**a**) beam B-08S; (**b**) beam B-083m; (**c**) scatter of the results at 60% and 80% of the ultimate load.



Figure A13. Kotynia et al., 2011 [36] research beams deflection compared with calculated, (**a**) beam G1; (**b**) beam G2; (**c**) beam G3; (**d**) beam G4; (**e**) scatter of the results at 60% and 80% of the ultimate load.





Figure A14. Kotynia et al., 2014 [37] research beams deflection compared with calculated, (**a**) beam B12; (**b**) beam b12-a; (**c**) beam B12-asp; (**d**) beam B12-asp-e; (**e**) beam B16; (**f**) beam B16-asp; (**g**) beam B16-asp-e; (**h**) scatter of the results at 60% and 80% of the ultimate load.



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Figure A15. Omran et al., 2012 [38] research beams deflection compared with calculated, (**a**) beam B0; (**b**) beam B1-NP; (**c**) beam B1-P1; (**d**) beam B1-P2; (**e**) beam B1-P3; (**f**) scatter of the results at 60% and 80% of the ultimate load.





(e)

(**f**)

Figure A16. Rezazadeh et al., 2014 [39] research beams deflection compared with calculated, (**a**) beam Control; (**b**) beam Non prestressed; (**c**) beam 20% prestressed; (**d**) beam 30% prestressed; (**e**) beam 40% prestressed; (**f**) scatter of the results at 60% and 80% of the ultimate load.







Figure A17. Sharaky et al., 2014 [40] research beams deflection compared with calculated, (**a**) beam CB; (**b**) beam LB1C1; (**c**) beam LB1G1; (**d**) beam LB2C1; (**e**) beam LB2G1; (**f**) beam LA2C1; (**g**) beam LA2C1; (**g**) beam LA2C1; (**h**) beam LB1G2; (**i**) scatter of the results at 60% and 80% of the ultimate load.



Figure A18. Soudki et al., 2007 [41] research beams deflection compared with calculated, (**a**) beam T-0; (**b**) beam S-0; (**c**) beam C-0; (**d**) scatter of the results at 60% and 80% of the ultimate load.





Figure A19. Teng et al., 2006 [42] research beams deflection compared with calculated, (a) beam B0; (b) beam B500; (c) beam B1200; (d) beam B1800; (e) beam B2900; (f) scatter of the results at 60% and 80% of the ultimate load.



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Figure A20. Valivonis et al., 2010 [14] research beams deflection compared with calculated, (**a**) beam B6.5; (**b**) beam B6.1C; (**c**) beam B6.2C; (**d**) beam B8.3; (**e**) beam B8.1C; (**f**) beam B8.2C; (**g**) beam B12.5; (**h**) beam B12.6; (**i**) beam B12.1C; (**j**) beam B12.2C; (**k**) scatter of the results at 60% and 80% of the ultimate load.



(i)

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Figure A22. Xiong et al., 2007 [44] research beams deflection compared with calculated, (**a**) beam Pa; (**b**) beam Pb; (**c**) beam 2C; (**d**) scatter of the results at 60% and 80% of the ultimate load.

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